



Pre-launch Testing and Post-Launch Performance
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Preflight Characterization of the OCO-3 Imaging Spectrometer

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acknowledged.

- Orbiting Carbon Observatory Missions 2002-2022
- Atmospheric state retrievals from 3 narrow NIR spectral bands
- OCO-2 Science Highlight
- From OCO-2 Spare Spectrometer to OCO-3 Payload
- OCO-3 Test Program
 - Ground Support Equipment Overview
 - Radiometry and sphere calibration with NIST
 - Heliostat spectra & verification with TCCON
- Inflight Calibration Strategies
- Conclusion: Expected launch 2/19





OCO, OCO-2, and OCO-3



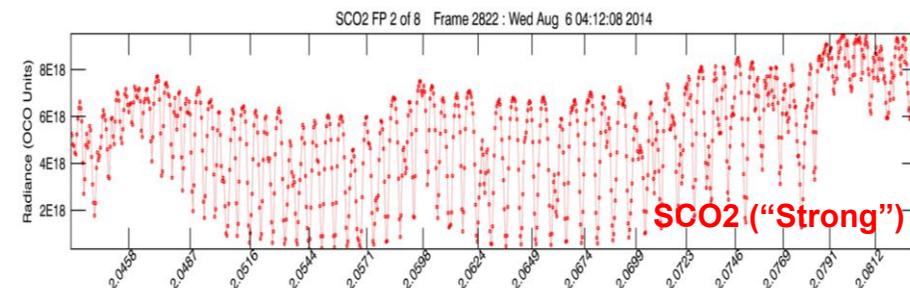
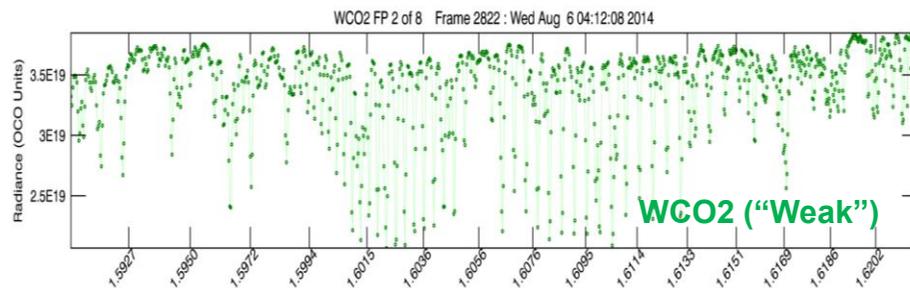
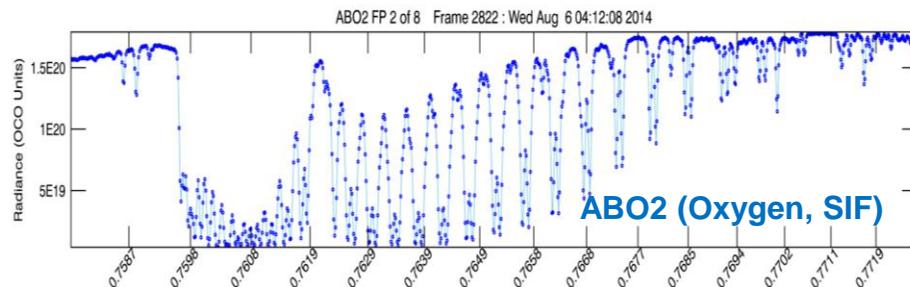
- Orbiting Carbon Observatory approved within the Earth System Science Pathfinder program in July 2002
- Launched on February 24, 2009 but did not achieve orbit due to launch vehicle failure
- Work on OCO-2 began in March 2010
- Launched into A-Train successfully from Vandenberg Air Force Base in California on July 2, 2014
- Following the successful OCO-2 launch, work began on converting the spare spectrometer into OCO-3
- Launch to ISS scheduled for February 2019, planned duration 3 years





Measurement Basics

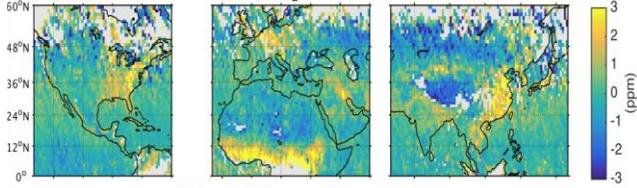
- Three-channel grating spectrometer with common entrance optics
 - 758-772, 1594-1619, 2045-2081 nm
- High spectral resolution
 - $\lambda\Delta\lambda = 17000-20000$
- Infer several atmospheric properties from the depths of the absorption lines
 - Surface pressure
 - Aerosols
 - Clouds
- 24 Soundings acquired per second
 - Onboard averaging in spatial dimension compresses 160 rows into 8 footprints (~2.5 km² on ground)



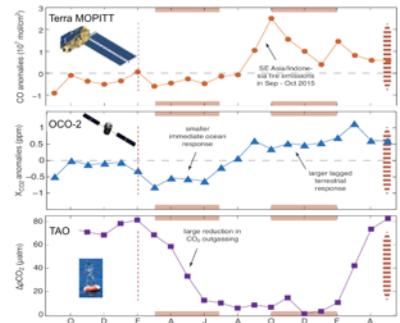
Notable OCO-2 Science



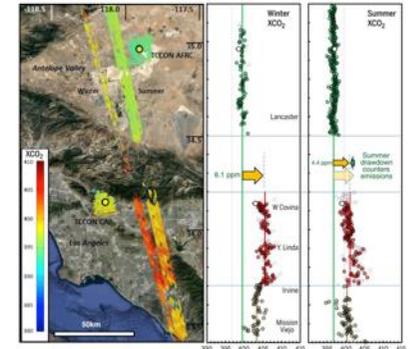
OCO-2 mean XCO₂ anomalies, 2014-2016



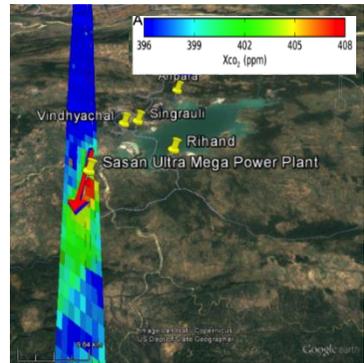
Large-Scale Anthropogenic Emissions
 (Hakkarainen et al, GRL, 2016)



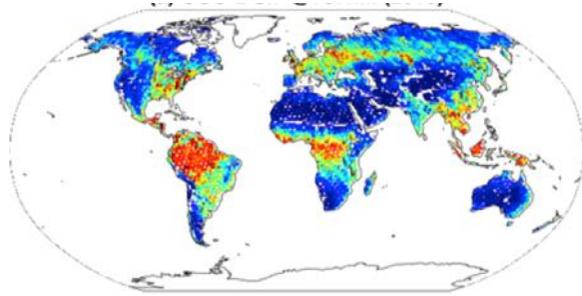
Ocean Response to 2015-16 El Nino
 (Chatterjee et al, Science, 2017)



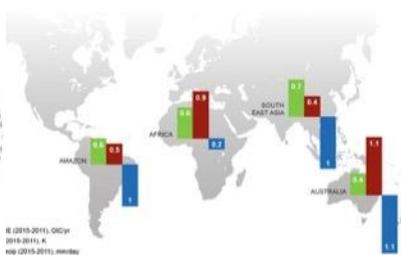
Detection of Urban & Volcanic Emissions
 (Schwandner et al, Science, 2017)



Quantifying Power Plant Emissions
 (Nassar et al, GRL, 2017)



Global SIF Measurements
 (Sun et al, Science, 2017)



Tropical Response to 2015-16 El Niño
 (Liu et al, Science, 2017)





OCO-3 Thermal Vacuum Testing

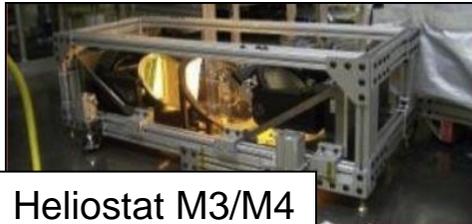
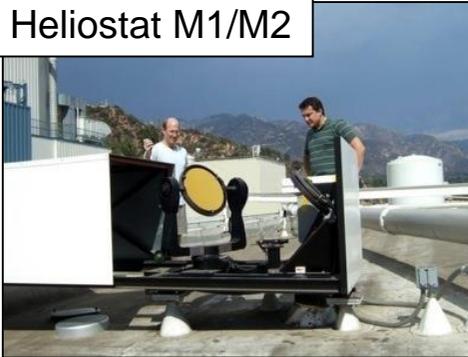


- The OCO-2 spare spectrometer was stored after testing in May 2013
- Intermediate tests in 2016 and 2017 to confirm performance and evaluate new entrance optics
- The OCO-3 payload completed its final thermal vacuum test in May 2018
 - Two weeks of optical testing with additional thermal tests
 - Derived spectral and radiometric calibration coefficients for launch
- Additionally, verified dozens of requirements including:
 - Field of View
 - Slit Alignment
 - Focus
 - Saturation
 - Bad Pixels
 - Polarization extinction

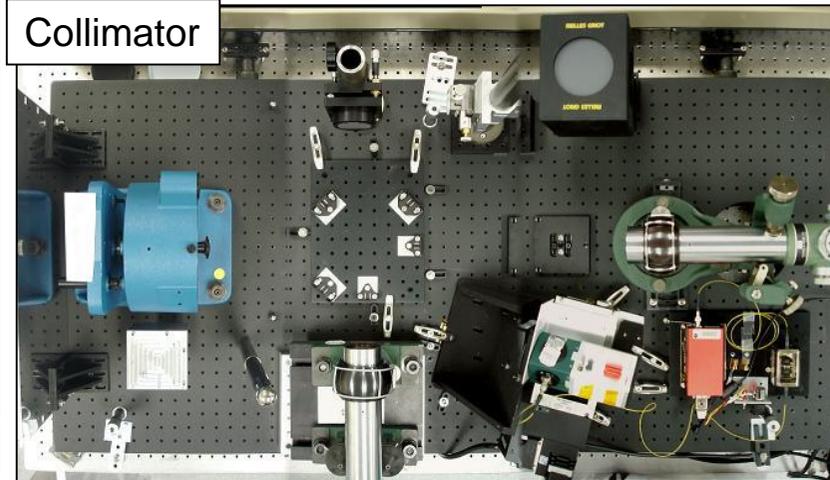
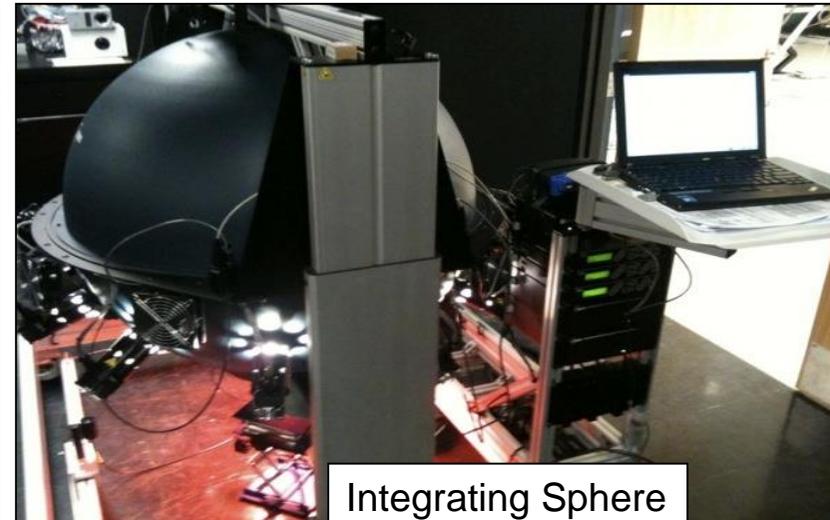
Optical GSE Configuration

OGSE was inherited from OCO, OCO-2 and was used for previous OCO-3 TVACs

Heliostat M1/M2



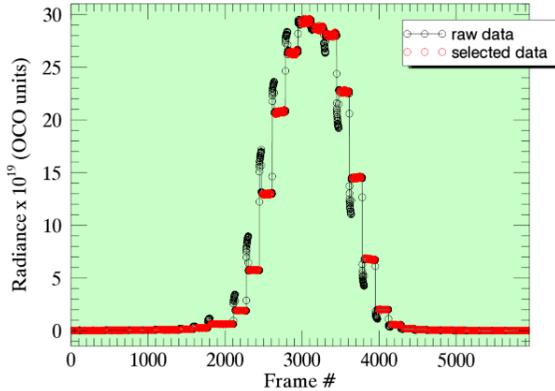
Heliostat M3/M4



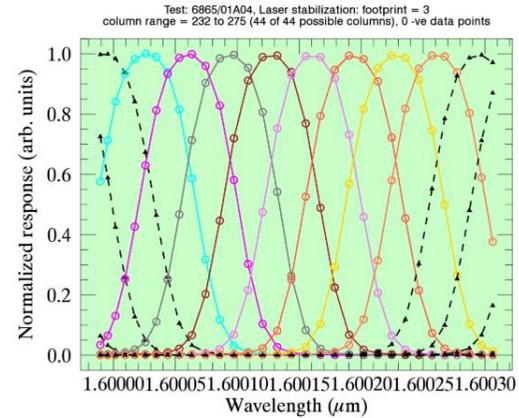
Measuring Instrument Line Shapes

Single laser scan – raw data

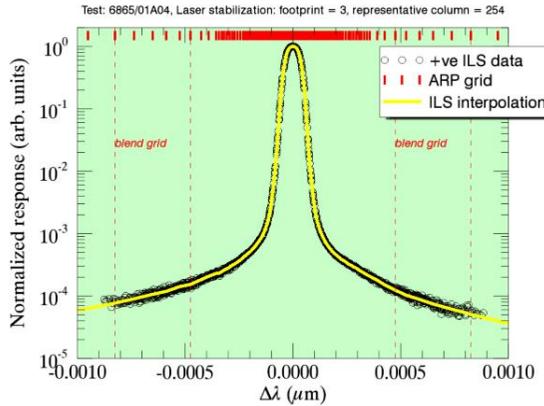
Test: 6865/01A04, Laser stabilization: column = 253, footprint = 3
file = oco3_L1alnSci01A04SC_96865a_180508143447s.h5



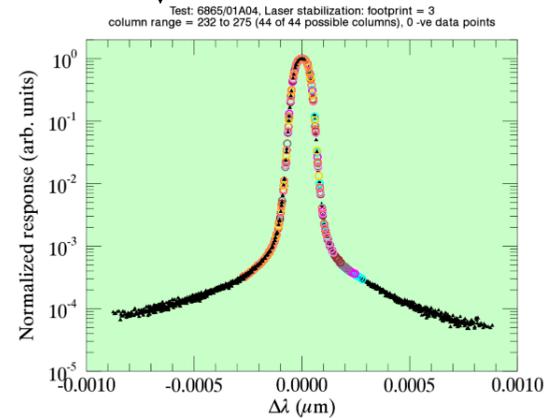
Illuminates ~8 spectral samples



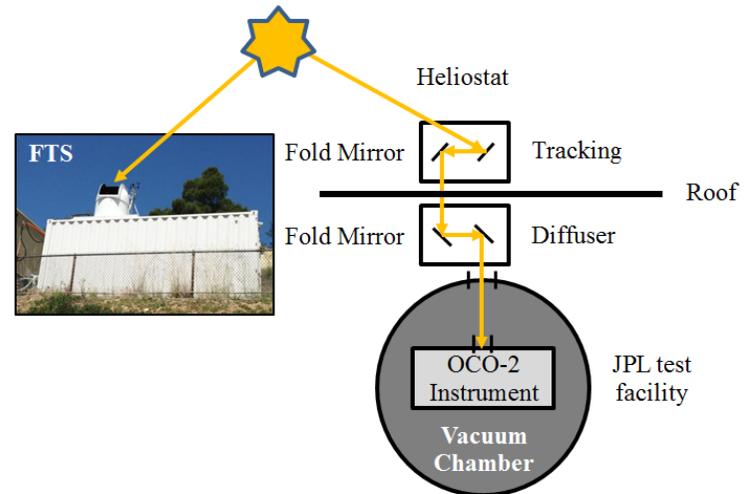
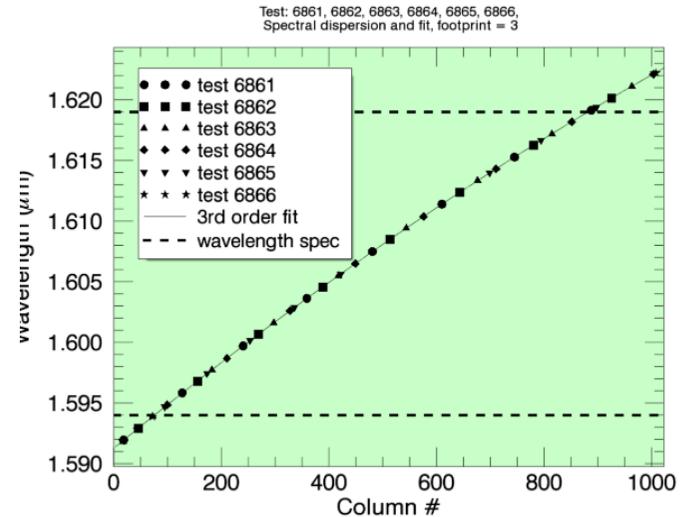
ILS fit



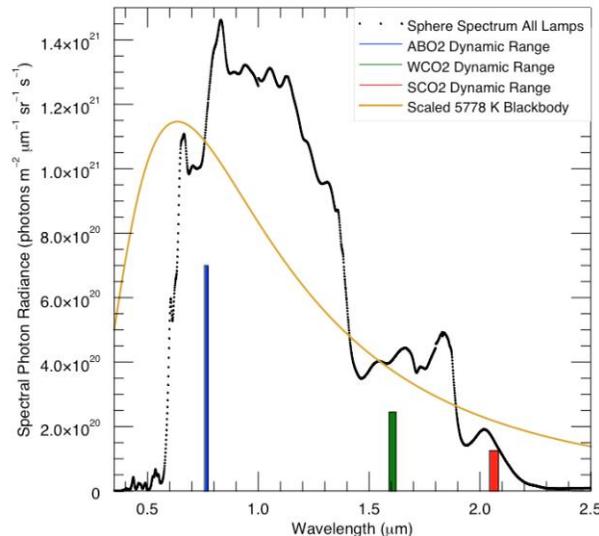
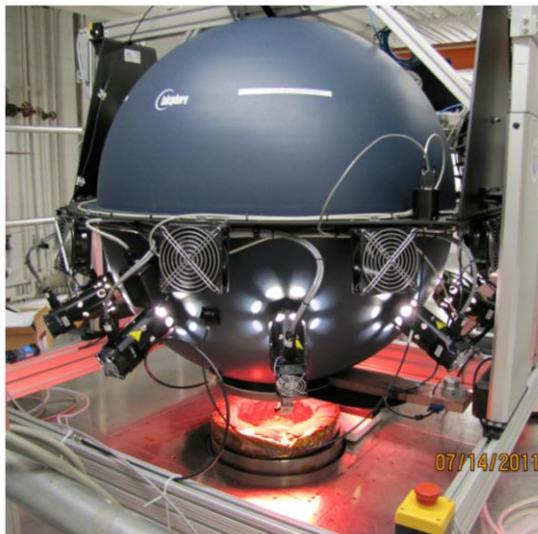
spectral samples combined



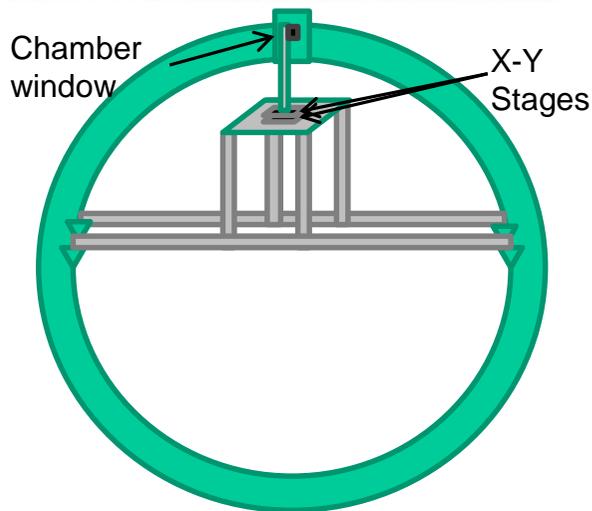
- ~ 40 laser scans allows ILS determination/interpolation for 1016 spectral channels, eight footprints, and three bands, yielding 24,384 individual ILS functions.
- Initial laser based dispersion also determined from these ~40 laser scans
- The laser based ILS & dispersion further optimized by comparing solar spectra recorded simultaneously on the ground by the OCO-3 flight instrument and a collocated high-resolution Fourier transform spectrometer (FTS).



Preflight Radiometric Calibration with NIST



- 5% absolute performance requirement
- Sphere has dedicated ASD spectroradiometer
- NIST ASD in chamber before and after testing transfers calibration from standard sources and helps to correct artifacts



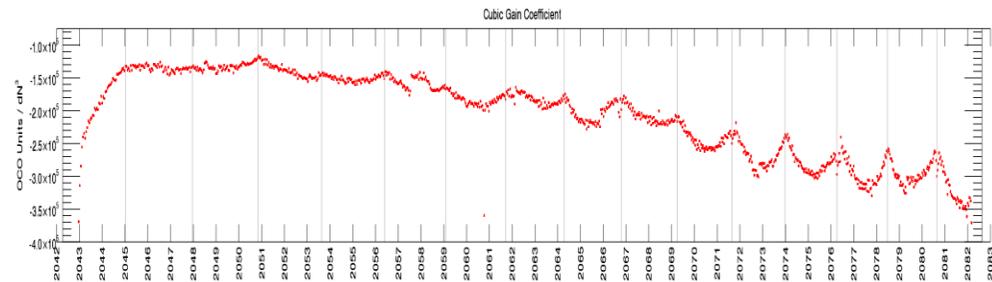
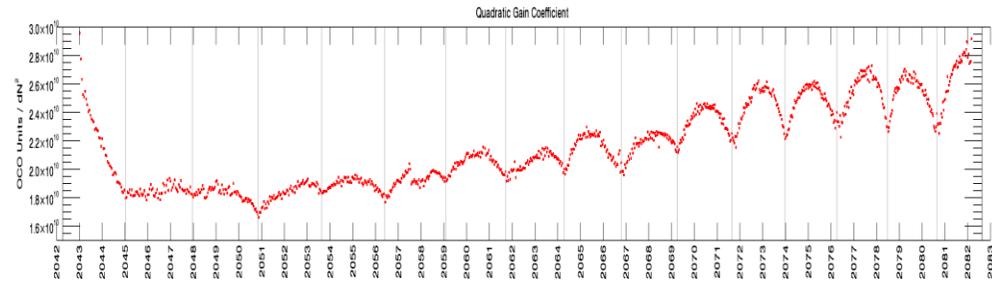
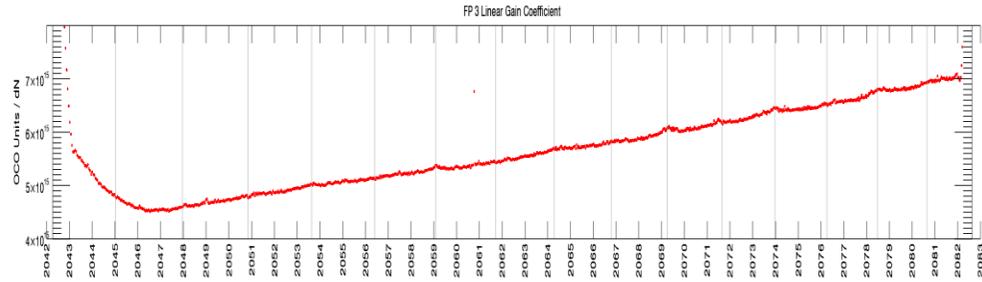
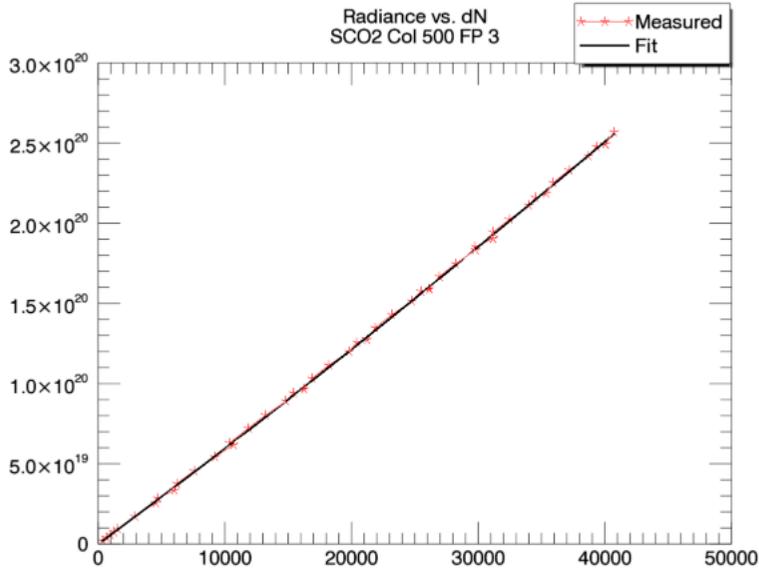
Integrating sphere has 10 external halogen lamps with filters, one has a variable attenuator



Example Gain Fits: SCO2 FP 3



- Cubic gain polynomial for every spectral sample with constant term set to zero because dark correction is performed separately





Inflight Calibration Chain

