

Next Generation Simulation Framework for Robotic and Human Space Missions

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The *Dartslab* team at NASA's Jet Propulsion Laboratory (JPL) has a long history of developing physics-based simulations based on the *Darts/Dshell* simulation framework that have been used to simulate many planetary robotic missions, such as the Cassini spacecraft and the rovers that are currently driving on Mars. Recent collaboration efforts between the Dartslab team at JPL and the Mission Operations Directorate (MOD) at NASA Johnson Space Center (JSC) have led to significant enhancements to the Dartslab DSEENDS software framework. The new version of DSEENDS is now being used for new planetary mission simulations at JPL. JSC is using DSEENDS as the foundation for a suite of software known as COMPASS that is the basis for their new human space mission simulations and analysis. In this paper, we will describe the collaborative process with the JPL Dartslab and the JSC MOD team that resulted in the redesign and enhancement of the DSEENDS software. We will outline the improvements in DSEENDS that simplify creation of new high-fidelity robotic/spacecraft simulations. We will illustrate how DSEENDS simulations are assembled and show results from several mission simulations.

I. Introduction and Background

A. Prior History the Dartslab DSEENDS simulation frameworks at JPL

DSEENDS is a physics-based engineering simulator for space missions developed by the Dartslab⁵ team at NASA's Jet Propulsion Laboratory [1]. DSEENDS models the spacecraft as a multi-body system where the spacecraft position, attitude, articulation and body flexibility states (and their rates) interact with gravity, atmospheres, terrain, and on-board spacecraft devices in response to ground commands and flight-software directed sensing and control actions. DSEENDS is a deployment of the Dshell multi-mission simulation framework [2]. It was originally designed to provide functionality for Entry, Descent and Landing (EDL) problems but has since been generalized to provide capabilities relating to spacecraft ascent, orbit, proximity operations, rendezvous descent and surface operations (e.g. roving).

The DSEENDS tool is in use at JPL for technology/concept development – all the way from Pre-Phase A analysis to flight operations. It is used by NASA/JPL missions for performance studies, cross-validation of other simulations and tools, and flight-critical EDL mission operations including lander targeting. It has been used by NASA/JPL Technology Programs, Program Offices, and Mission Analysis teams as a high-fidelity simulator to support proposal development, as an integration platform and test-bed for studies, and as tool for algorithm and software development.

As part of JPL's end-to-end Mission systems it interoperates with JPL's Interplanetary Mission design and navigation software *Monte* [2]. In flight-operations it is used to verify the actions of mission actions (e.g. determine the landing footprint), perform targeting operations (e.g. to design interplanetary trajectory correction maneuvers).

DSEENDS development started with the 1991 development of the DARTS dynamics engine. The Dshell framework was initially developed in 1992 for the Cassini mission, with development of the DSEENDS deployment commencing in 2000. In 2007 DSEENDS was used in the Mars Phoenix EDL Mission Operations and more recently it was used for the Mars Science Laboratory EDL operations. It is currently in use at JPL by the Low Density

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⁵ For more information about the Dartslab, please visit <http://dartslab.jpl.nasa.gov>.

Supersonic Decelerator (LSD) project for design of their integrated test campaign. It has also been selected by JSC-MOD as the basis for their next generation flight dynamics tool for all ascent, descent, orbit, proximity-operations, and rendezvous simulations.

To users, DSEENDS provides a modern scripting language (Python) to enable data-driven building of applications at run-time, easy configuration setup & initialization, and automated testing. The underlying Dshell framework heavily leverages various Open Source software resources. High computational performance is achieved using C++ implementations of module functions and fast algorithms, with auto-generation of the hooks from the C++ level to Python. Visualization and introspection tool configurations are dynamically generated from the simulation configuration.

B. Motivation for improvements to the DSEENDS framework

Due to the production nature of the first users of DSEENDS, the software package delivered to them was fairly stable and static. Changes were typically due to bug fixes and mission-specific enhancements or models. In the meantime, the Dartslab team had improved the simulation framework software, including enhancements to multibody modeling (Ndarts [3], [4]), 3D visualization (using Ogre [6]), very large scale terrain modeling (SimScape [6]). Having to maintain an older version of the software while active development is going on the latest version is inefficient and error-prone. A path forward that would allow the DSEENDS user base to migrate to a current version of the software was needed.

When the JSC Missions Operations Directorate team approached us, they needed a new simulation framework. With the retirement of the Space Shuttle, the MOD team was transitioning into a dynamic role of supporting various vehicles with rapidly changing missions. This required agile new simulation tools that are adaptable to a wide range of human space flight missions. The JSC MOD Flight Dynamics team evaluated a variety of options and completed a 6-DOF simulation benchmarking effort, and determined that DSEENDS “offers the most capable and flexible simulation path for the division”.

In the process of training the JSC MOD team in the use of latest version of DSEENDS, it became apparent that there were some parts of DSEENDS that needed improvements for their use cases. For instance, DSEENDS could only support one spacecraft at a time. This led to a collaborative effort to refactor and improve DSEENDS.

C. Collaborative development

In order to collaboratively develop a new version of DSEENDS, in 2011 we formed a joint JPL-JSC development partnership led by JPL (although most of the development team was from the JSC MOD). Our goal was to refactor much of the simulation software underlying DSEENDS so that it would meet the needs of both the JPL and the JSC MOD teams. The primary focus was on creating a new module "DshellCommon" that constitutes a complete re-write of the corresponding functionality of the previous version of DSEENDS. There were also extensive changes to the underlying software to incorporate the framework enhancements outlined in the previous section, as well as changes to increase the usefulness, usability, and robustness of the software.

Our development team used Agile methodologies [7] including Scrum principles [8] to define short-term incremental releases on a two-to-three week development cycle. Each cycle typically involved a review of the prior DSEENDS functionality and code, planning of how to adapt or rewrite the code, implementing the desired re-coding, and writing relating regression tests.

Due to inter-center complications of sharing a software repository, the team in each center maintained a separate software repository (using subversion). The JPL team took responsibility for merging changes from both sides based on JPL-JSC team discussions and consensus building, and committing the changes to both software repositories. Most of the software evolution took place in the DshellCommon module. The JPL team made sure that all changes worked in the JSC environment and the JPL environment (which was evolving independently of the JSC environment). Every several months, the JPL team put together a consolidated package of all the modules and provided it to JSC as a new baseline package. This prevented divergence of the code base over time.

We also conducted a face-to-face Technical Interchange Meeting (TIM) once every 6 months to discuss progress in the software development, talk about technical issues, and plan development strategy for the coming months.

This collaborative process has worked extremely well and has produced a new very useful and high-quality version of DSEENDS. The new version of DSEENDS was derived from earlier versions of DSEENDS but is essentially a new implementation that supports a wide range of spacecraft and mission related scenarios. Not only is the JSC MOD team using the new DSEENDS, but the JPL Dartslab team has transitioned to the new DSEENDS for all current and new work.

D. The end product: A new DSEENDS

The new version of DSEENDS is significant step up from earlier versions of DSEENDS in terms of flexibility, usability, robustness, and general software quality. DSEENDS is made of several parts as shown in this figure on the right. From the bottom up, these components are:

- **Dartslab core SW, libraries** - Underlying software libraries such as vector libraries, terrain modeling libraries, visualization libraries, *etc.*
- **Darts multibody code** – Darts multibody dynamics engine.
- **Dshell** - The Dshell framework of classes for models, data flow management, parameter handling, integration, data logging, *etc.*
- **Dartslab models** - Library of reusable models for system hardware and subsystems.
- **DSEENDS models** - Models for simulations involve aerodynamics and other EDL-related capabilities.
- **DshellCommon** - Simulation executives, run-script handling code, and classes for "assemblies" (a higher level model construct which construct and connect sets of related primitive models).
- **User code and run scripts** – Python libraries and run scripts written by the user to make use of all the lower levels to model and simulate complete systems.

The JSC MOD team uses the DSEENDS software as the core of their system called COMPASS (Core Operations, Mission Planning, and Analysis Spacecraft Simulation). This block diagram illustrates how COMPASS uses the underlying DSEENDS framework.

This is the same as the previous layout except that JSC has implemented a variety of models and assemblies in the item labeled 'COMPASS Components' related to specific vehicle hardware and simulation scenarios they plan to support as part of their mission.

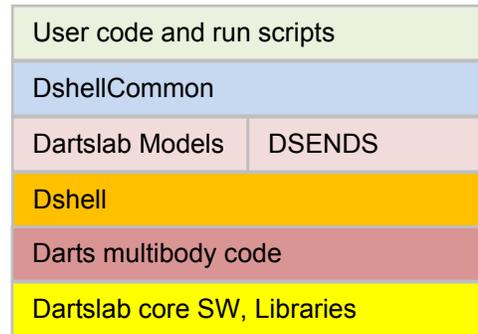


Figure 1 - DSEENDS Components

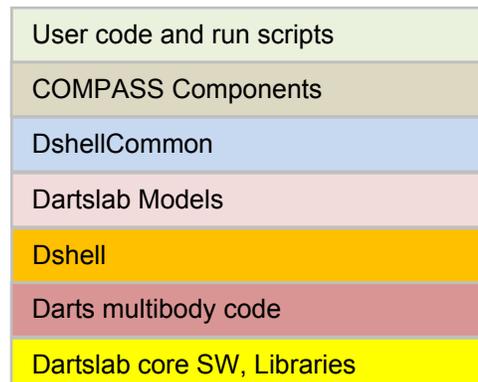


Figure 2 - JSC MOD COMPASS Components

II. Simulations with DSEENDS (How does it work?)

E. What kind of systems can DSEENDS model?

In short, the DSEENDS can model *any* type of space, aerial, marine surface, or underwater vehicle. The fidelity of the model is only limited by how detailed the implementers are willing model out the multibody system and the various hardware and software components.

DSEENDS can handle arbitrary multibody systems including chain, tree, and closed-chain topologies. The Darts multibody dynamics code, winner of 1997 NASA Software of the Year Award, can handle closed loops with constraint embedding so that a minimum number of coordinates are used [4], [4]. Darts can also model bodies that are flexible using modal descriptions of the flexible-body dynamics.

DSEENDS can model actuators that inject forces into the multibody system (such as a thruster) as well as sensors that can read information out of the multibody system (such as joint encoder). DSEENDS also supports many different models for specific types of hardware (IMUs, *etc.*). If the user has special subsystems they need to model, they can create new Dshell models (in C++) and use them in simulations just like any existing model.

DSEENDS includes higher-level models called *Assemblies* that set up sets of primitive models, connect their data flows appropriately, and initialize the primitive model parameters as needed. DshellCommon and other modules include a library of Assemblies that model typical spacecraft subsystems (such as a robot arm that includes bodies, joints, joint motors, encoders, *etc.*)

DSEENDS can model aerodynamic forces acting on arbitrary bodies in a simulation. This has been used to perform launch vehicle breakup analysis to determine vehicle dispersion on vehicle failure.

DSEENDS can handle simulations involving multiple spacecraft and multiple planetary bodies. The spacecraft can be independent or attached to each other. Spacecraft can detach from their parent body and attach to any other body at run time.

DSEENDS provides a wide range of planetary bodies (Earth, Moon, Mars, *etc.*) as well as sophisticated atmospheric models based on the latest NASA planetary atmospheric codes. DSEENDS can also model small bodies such as asteroids with a polygonal model that captures the variable gravity of oddly shaped bodies very accurately.

DSEENDS provides tools for finite state machines that can control the sequencing and behavior of the simulation to construct and execute a very sophisticated mission simulation sequence.

DSEENDS can run parameter sweeps or Monte Carlo simulations involving thousands of separate runs to determine dispersions. This capability was used extensively to plan entry trajectories and Entry, Descent, and Landing strategy for the Mars Phoenix and Mars Science Laboratory missions.

DSEENDS provides tools for 3D visualization of all parts of a simulation (spacecraft, planets, trajectories, *etc.*) The user can control the view point of the visualization “camera” to better watch and understand the simulation. Once a simulation is developed, the 3D visualization can be captured in a movie.

F. What is "under the hood" of DSEENDS?

Underneath the surface of DSEENDS, there are many excellent features and capabilities. Simulations based on DSEENDS are physics-based. The figure on the right shows the basic functionality of the DSEENDS simulation framework.

The high degree of physics fidelity starts with the underlying multibody code, 'Darts'. Darts is a multibody dynamics engine based on spatial operator algebra [9] with $O(n)$ performance that scales well to hundreds or thousands of degrees of freedom.

Since it is an inherently efficient algorithm and is implemented in C++, in most situations Darts offers faster than real-time performance. Another key feature is the ability to support a wide range of 'Dshell' models for various hardware devices or subsystems. The models can insert forces or torques into the multibody system (actuators) or read information out of the multibody system (encoders, sensors). The fidelity of the simulation is primarily limited by the fidelity of the models. Dshell models have the advantage of being very efficient since they are written in C++.

Connecting models are data flows called 'signals'. Signals not only allow models to communicate with each other, they also define a partial order that allows model sorting (to control order of model execution during simulations).

In order to set up low-level models, data flows, and initialize parameters, we use Python. Each software module is compiled separately and set up so it can be loaded using the Python module import system. This means that simulations can be set up without compiling a large application executable (although this is possible, if desired). The users run script can load the necessary modules for their simulation. Therefore, we can shift away from the “application binaries” paradigm and use Python as the bedrock to glue together appropriate mix libraries at run-time.

Another very useful feature of DSEENDS is that it has an underlying data representation called DVar (similar to a URL addressing scheme) that allows the user to access all multibody bodies and joints, model parameters, states, and signals in a simple consistent way. This makes operations such as changing simulation parameters and logging much simpler and more flexible.

Dshell also includes the typical simulation capabilities such as selecting integrators, controlling simulation step sizes, processing events, and executing the simulations.

Finally, DSEENDS includes a wide range of data analysis tools for run time (such as dials, gauges, and strip charts) as well as post-run analysis (such as plotting and data mining).

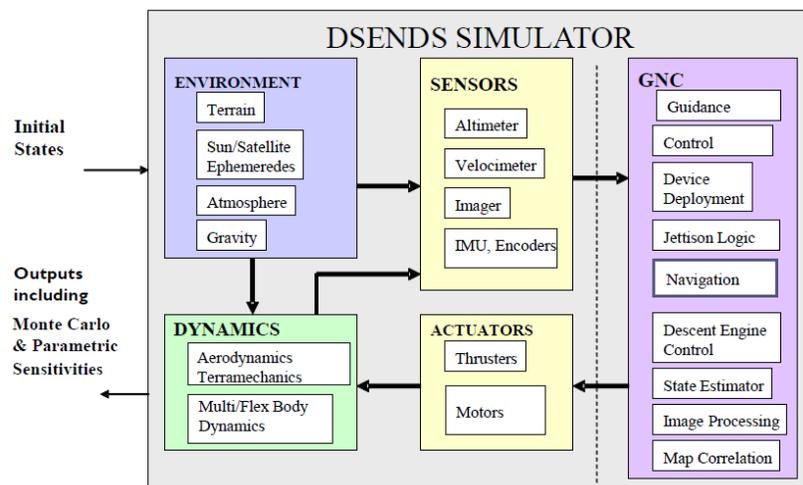


Figure 3 - DSEENDS Functional Block Diagram

G. What does a run script look like?

The simulation 'run' scripts are Python files that have the following parts:

1. **Setup** - importing classes for the simulation executive and parameter definition files (not shown).

2. **Create the simulation executive:**

```
sim = SimulationExecutive()
```

This creates the simulation executive that will process and execute the rest of the simulation specification in the rest of this user run script.

3. **Configure the simulation configuration and parameterization:**

```
config = {
    'Mars' : {
        'class' : 'TargetAssembly',
        'params' : { 'Target' : targets['Mars'],
                    'Bodies' : bodies['Target']['Bodies'],
                  }
    },
    'SC1' : {
        'class' : 'VehicleAssembly',
        'basename' : 'CapsuleBase',
        'params' : { 'Bodies' : bodies['SC1']['Bodies']
                  },
        'assemblies' : {
            'grav' : {
                'class': 'GeneralGravityActuatorAssembly',
                'context' : { 'body' : 'CapsuleBase' }
            }
        }
    }
}
```

Definition
for target
planet
(Mars)

Definition for
spacecraft
'SC1',
including
contained
gravity model

This section of the user run script extracts parameters from the 'targets' and 'bodies' objects. These are defined by the user in separate Python files (targets.py and bodies.py) which are not shown here.

4. **Create the simulation:**

```
sim.createAssemblies(config)
sim.bindState()
sim.resetState(0.0)
```

These statements create the simulation objects (such as the spacecraft 'SC1') as well as set up its initial configuration properly.

5. **Configure finite state machines to sequence the simulation (not shown).**

6. **Initialize any simulation-specific initial values (not shown).**

7. **Execute the simulation:**

```
sim.step()
```

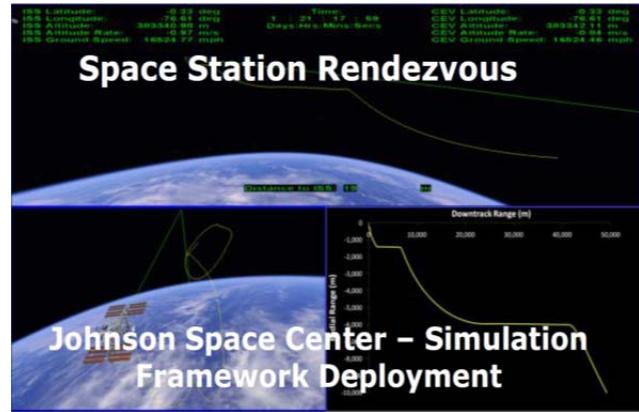
Users are free to include loops to execute as many time steps as desired. They may also want to interleave commands (although this can usually be better handled via finite state machines).

III. Applications of DSENDS (What is it good for?)

H. JSC MOD applications

The JSC MOD team has applied the DSENDS/COMPASS framework to variety of mission scenarios including:

- Ascent scenarios that start fixed on the surface, launch, and reach a goal orbit.
- Rendezvous scenarios that involve two spacecraft in orbit. One spacecraft maneuvers to rendezvous with the other (passive) spacecraft (such as a spacecraft rendezvousing with the International Space Station). See the figure to the right.
- Descent scenarios that start with a spacecraft in orbit, perform a de-orbit thruster firing and track the vehicle downwards through entry, parachute deploy, and surface landing.
- Other scenarios such as orbital proximity operations and vehicle breakup analysis.



I. MSL Entry Descent and Landing Visualization

DSENDS has been used in a variety of ways in the Mars Science Laboratory mission (Striep, et al. 2006, vol.43 no.2). Entry, Descent, and Landing was planned out using a prior version of DSENDS [10]. During the landing on August 5, 2012, the MSL Mission Control team used an MSL EDL visualization tool developed by the Dartslab team using the new version of DSENDS [12]. This was visible on of the main screens and on screens of several controller workstations. A screenshot of this visualization tool from the actual landing is shown here. Here the DSENDS models are driven by the real-time telemetry data from the MSL relay satellite (Mars Odyssey) instead of by solving of the underlying dynamics. Note that because of the availability of the dynamics capability, a physics-based extrapolation of the system trajectory and behavior is possible in case of telemetry data drop outs or slow data rates. However in the control room, only the last known status and position (from telemetry) were shown on the screen. In addition DSENDS also uses all the geometrical models (terrain, coordinate frames, etc) to display the telemetry data in the most appropriate mode for a given mission phase (e.g. inertial display versus surface fixed frame relative display). Also, the current positions of the Mars communications satellites (Mars Odyssey and Mars Reconnaissance Orbiter) were shown at all times using the standard DSENDS ephemerides updates. The 3-d views of the spacecraft are overlaid with event data as well as visual displays in the form of dials and text.

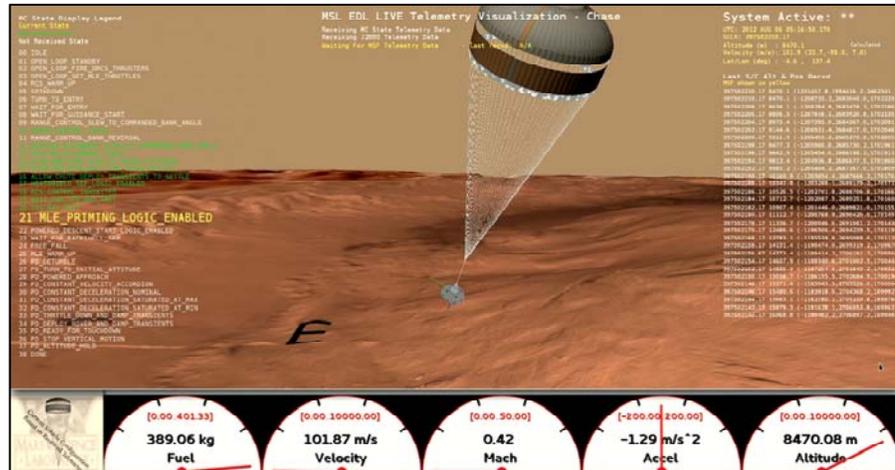


Figure 4 - MSL EDL Visualization Tool based on the new DSENDS used in MSL Mission Control during Mars Entry, Descent, and Landing

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