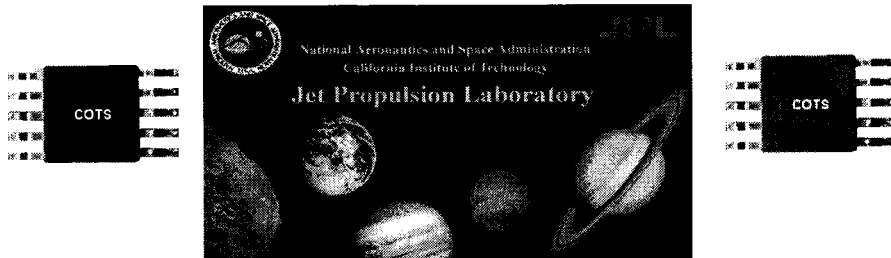


Military & Aerospace /Avionics COTS Conference

August 22-25, 2009

Commercial Off-The-Shelf (COTS) Program Methodology and Results of Upscreening Electronic Parts - An Update



Mike Sandor, Shri Agarwal
4800 Oak Grove Drive
Pasadena, CA 91109

JPL 8-24-00

Phone: (818) 354-0681 FAX: (818) 393-4559

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AGENDA:

- ADVOCACY FOR COTS
- DRAWBACKS WHEN IMPLEMENTING COTS
- JPL COTS⁺⁺ CRITICAL SCREENING FLOW
- JPL COTS⁺⁺ CRITICAL QUALIFICATION
- COST & SCHEDULE TRADEOFFS
- COTS⁺⁺ Upscreening Results
- C-SAM Update and Ongoing Work
- COTS DPA Failures
- SUMMARY

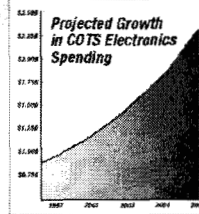
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Advocacy for Using COTS(plastic packages):

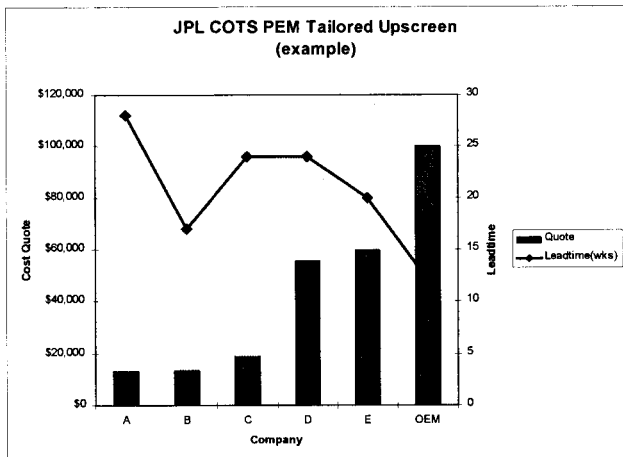
1. State of-the-art parts are mostly available as COTS
2. COTS plastic parts performance capabilities continue to increase (e.g. processing power & high density memories)
3. COTS plastic parts enable reduction of hardware weight and volume
4. COTS plastic parts initial acquisition cost is less than ceramic
5. COTS plastic parts have been reported to demonstrate good to excellent reliability in commercial and aerospace applications
6. Often they are the only option when Grade 1 is not offered or available



Drawback to COTS Implementation (plastic packages):

1. Upscreening cost is coupled to the following influences and therefore cannot be tightly controlled (no standard exists)
 - Finding suitable test expertise
 - Minimum quantities often dictate cost
 - Manufactures unwillingness to upscreen
 - Costs of ownership depends on risk accepted
2. Upscreening schedules can jeopardize project schedules unless
 - Flows and processes are in writing & approved
 - Engineering/QA help is available daily
 - Vendor commits to screening schedule
 - Material in-process status is monitored weekly
3. Risk is not totally eliminated with upscreening

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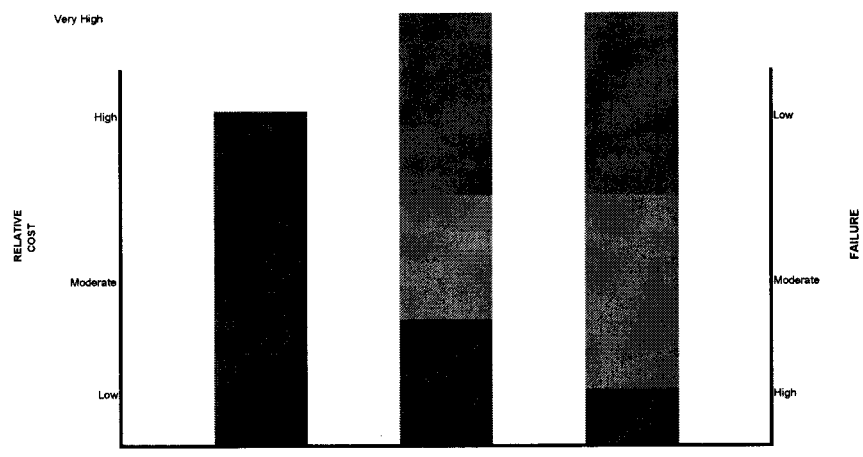
Competitive bidding demonstrates cost & schedule selection tradeoffs

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JPL **Likelihood of Part Failure Vs Cost for Space Flight Applications**



Cost Legend:

Procurement	Risk Mitigation	Replacement Before Launch
Variables are 1. OEM 2. Distributor 3. Leadtime 4. Substitutes 5. Volume	Variables are 1. Application 2. Requirements 3. Radiation 4. Reliability 5. Design	Variables are 1. Spacecraft 2. System 3. Sub-assembly 4. Board 5. Component

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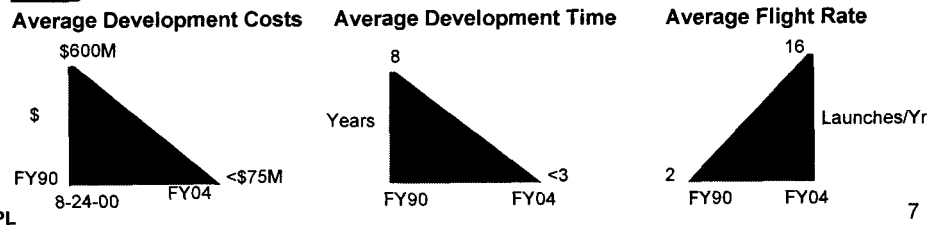
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More Risk Management is Needed:

JPL/NASA Project Drivers:

- Must infuse the latest technology
- Must significantly reduce development costs
- Must significantly reduce development time
- Per NASA, Better, Faster, Cheaper is here to stay



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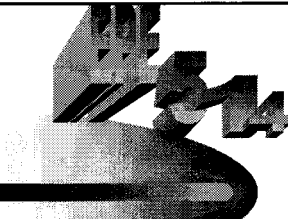
COTS PEM Risk Mitigation Addresses the Following Concerns:

- Narrow Temperature Range for Commercial Grade
- Plastic Assembly Quality
- Lot Non-Uniformity & Traceability (including radiation)
- Adequacy of Vendors Testing
- Infant Mortality
- Die Construction and Quality

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Radiation Issues of Using COTS for Space Applications:

Rad Hard Assurance Varies from the same processing lot

Radiation Assurance has little statistical confidence

TID response depends on process-

"Positive" process changes can reduce radiation tolerance

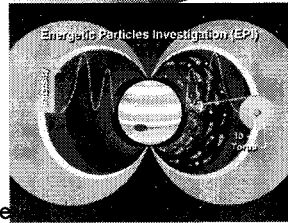
SEE depends on circuit design and dimensions-

Commercial vendor can change these without notice

No good way of predicting radiation response without extensive testing-

Exception is a controlled Rad Hard process line

Radiation risk mitigation techniques are often required- \$\$\$

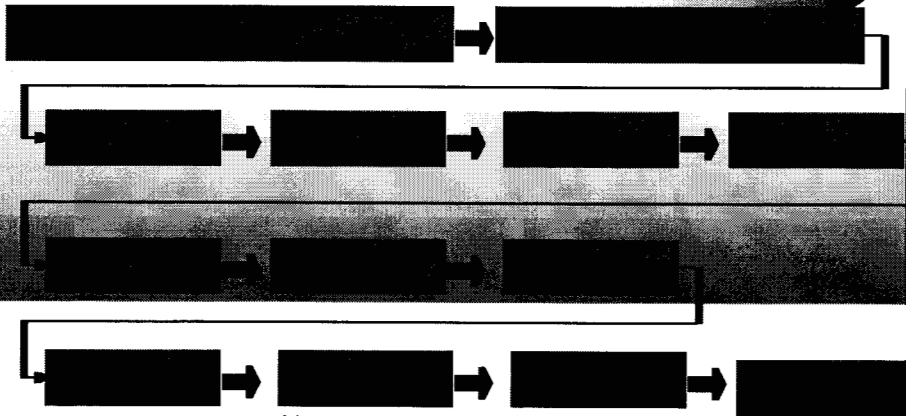


Methods to Insure Low Risk COTS in Critical Space Applications

Target Guidelines

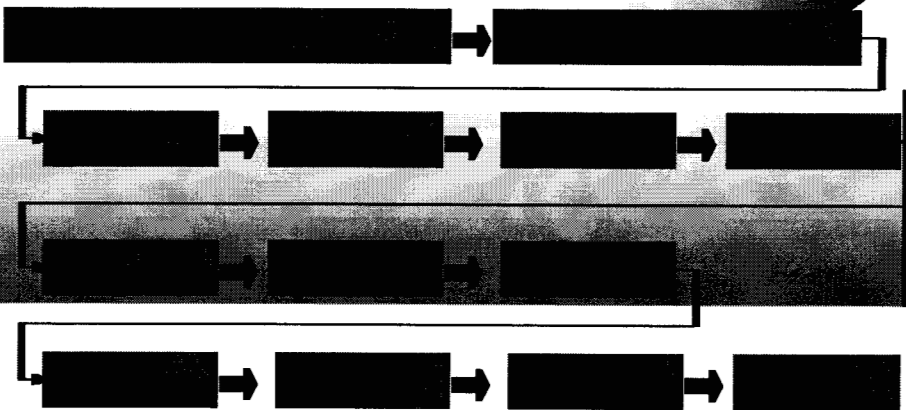
15 yr mission:	JPL Upscreen/ Qual	+	Derate/WLA/ RLAT/DPA/QML
10 yr mission:	JPL Upscreen/ Qual	+	Derate/WLA/ RLAT/DPA/QML
5 yr mission:	JPL Upscreen/ Qual	+	Derate/ RLAT/DPA/QML
≤1 yr mission:	JPL Upscreen	+	DPA/Generic Data

**Part Level
Screening**



COTS⁺⁺ Plastic Infusion Critical Screening Flow
(Tailored for Project application/mission requirements)

**Part/ Lot
Qualification**



COTS⁺⁺ Plastic Infusion Critical Qualification
(Tailored for Project application/mission requirements)



COTS++ PEM Upscreen Impact on Risk Mitigation

	<u>Amplifier</u>	<u>ADC</u>	<u>DC-DC Conv.</u>	<u>Reg.</u>
• Narrow Temp. Range for Commercial Grade	1	1	3	9
• Plastic Assembly Quality	3	9	9	1
• Lot Non- Uniformity & Traceability	1	9	3	3
• Adequacy of Vendors Testing	1	9	3	9
• Infant Mortality	1	9	1	9
• Die Construction and Quality	1	1	1	1
Total Score	8	38	20	31
COTS++ Impact on Lowering Risk	Low	High	High	High
Fallout	4%	65%	26%	25%

8-24-00 Risk mitigation weighting factors used: Minimum = 1, Moderate = 3, Significant = 9



COTS++ Upscreening Rejects by Part Type & Vendor

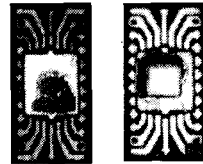
	<u>Amplifier- A</u>	<u>ADC- B</u>	<u>ADC2-B</u>	<u>DC-DC Con.-C</u>	<u>Voltage C-A</u>	<u>S.Regulator-B</u>
DPA:	0/4	1/8	TBD	0/4	0/4	0/4
Incoming:	0/78	n/a	4/79	1/78	0/80	8/80
C-SAM:	3/78	38/78	9/75	16/77	5/80	0/80
Temp Cycle:	0/78	10/78	0/75	3/77	0/80	3/72
Burn-In:	0/78	3/68	0/75	0/74	0/80	9/69
QCI:	0/10	0/10	TBD	0/10	0/10	0/10
Total:	3/78	51/78	TBD	20/78	5/80	20/80

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Failure Mechanisms from PEM Delamination:

- Stress-induced passivation damage over the die surface
- Wire bond degradation due to shear displacement
- Accelerated metal corrosion
- Die attach adhesion
- Intermittent electricals at high temperature
- Popcorn cracking
- Die cracking



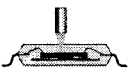
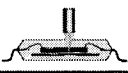
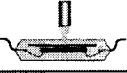
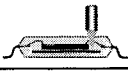
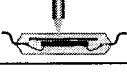
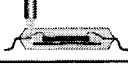




CSAM Yields 06/12/2000

Part Type	Manufacturer	Yield
NPN Transistor 1	A	83%
Switching Diode	A	0%
NPN Transistor 2	A	100%
Zener Diode	A	50%
NPN Transistor 3	A	100%
Op-Amp 1	B	87%
Op-Amp 2	C	0%
Op-Amp 3	C	7%
Phase Detector	D	100%
Mini Circuit	E	40%

**Results are
package/ vendor
assembly dependent**

Lot sizes range
from 15-30 parts each.

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	Circuit Side Scan	Non-Circuit Side Scan
Type I Delamination: Encapsulant/Die Surface		
Type II Delamination: Die Attach Region		
Type III Delamination: Encapsulant/Substrate (Die Side)		
Type IV Delamination: Substrate/Encapsulant (Backside)		
Type V Delamination: Encapsulant/Lead Interconnect		
Type VI Delamination: Intra-laminar (Laminated Substrates Only)		
Type VII Delamination: Heat Sink/Substrate		

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IC defect
descriptions are
now identified in
J-STD-035
(Acoustic Microscopy for
NonHermetic Encapsulated
Electronic Components)

Source: Sonoscan Inc.

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**A New Failure Characterization Study
is Underway Utilizing Plastic Part C-SAM Rejects**

Objectives:

- Identify C-SAM reject parts by criteria(s)
- Measure Material Properties including sonic test, IR, X-ray
- Apply extreme temperature cycle stresses
- Repeat Material Properties Measurements including C-SAM at different intervals
- Identify all failure mechanisms and risk rate C-SAM rejects

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A Failed Chip Scale Board Assembly is under investigation utilizing C-SAM inspection on components/board

Objectives:

- Identify component delaminations
- Identify board layer delaminations
- Make correlation to CSP package thermal cycle failures



- CTE Mismatch
- Package Proximity and Location on Board
- Ball Bond Size and Location

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Updated Examples of COTS Parts/Die Failing DPA



Op Amp

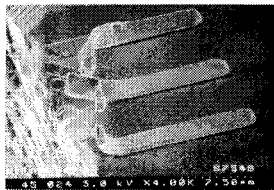
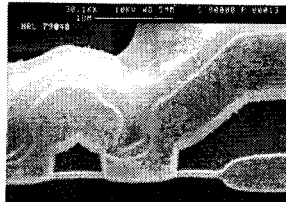


Figure 1. 4000X SEM micrograph of localized metallization traces along edge of die.



PROM

FIGURE 8-22
SAMPLE # 3
MAG. 20,000X
SUBJECT
SEM PHOTOGRAPHY OF IRREGULAR CONTACT
NOTE METAL REMAINS
ON WITHOUT BARRIER METAL
NOW WITH BARRIER METAL.

Metallization anomalies are
the predominant failures

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Summary/Conclusions:

- The concerns/risks anticipated with using COTS PEMS can be reduced to acceptable medium risk levels using JPL upsampling.
- A part qualification plan has been added to JPL's existing screening flows to further insure the reliability of parts used by Projects when application requirements are different.
- Further investigations/studies are being conducted on individual components and board assemblies using C-SAM analysis. This information will provide more understanding of the correlation between delamination and component/ board failure mechanisms.