

Natural Satellite Ephemerides at JPL

A Report to the IAU Commission 20 Working Group On Natural Satellites

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There are currently seventy-seven recognized and thirty-eight provisional natural satellites in the Solar System. ~~NASA's~~ Jet Propulsion Laboratory (JPL) maintains ephemerides for all and makes these ephemerides available electronically through JPL's On-Line Solar System Data Service known as Horizons (Giorgini *et al.* [1996]). Because of the intimate connection between the orbits of the Earth and Moon, the ephemeris of the latter is determined as part of the development of JPL's planetary ephemerides.

We use numerically integrated orbits as the basis for the ephemerides of most of the satellites. The exceptions are the Martian satellites, the five major Uranian satellites, Pluto's satellite, and majority of the minor inner satellites.

The orbits for the Martian satellites, Phobos and Deimos, are represented by the theory of Sinclair (1972,1989) as extended by Morley (1990) and Jacobson (1996a). Orbits from the ESAPHO and ESADE theories developed by Chaporn-Touzé (1988,1990a,1990b) are also available.

The orbits for the five major Uranian satellites are from the theory of Laskar (1986) which was fit to observations by Laskar and Jacobson (1987). The numerically integrated ephemerides used in Voyager operations (Jacobson *et al.* [1986]) are also available.

The orbit of Pluto's satellite, Charon, is modeled as a two body conic orbit with elements from Tholen(1990) and the revised semi-major axis of Null *et al.* (1993).

Precessing ellipses represent the minor satellites' orbits. The sources of the ellipse elements are:

- the four inner Jovian satellites (Jacobson [2001a]).
- four of the minor satellites of Saturn, Prometheus, Pandora, Atlas, and Pan (Jacobson [1995,1996b])
- the ten minor Uranian satellites (Jacobson [1998b])
- the six minor Neptunian satellites (Owen *et al.* [1991]).

To support the Galileo mission we developed numerically integrated orbits for the Galilean satellites (Jacobson [2001a,2002]). These orbits are still being refined using data acquired by the Galileo

spacecraft. We have also extended the integration to include the inner satellites Amalthea and Thebe in order to process the Galileo data from the close flyby of Amalthea.

The Cassini mission is relying on integrated ephemerides for the eight major Saturnian satellites, Phoebe, the inner satellites Janus and Epimetheus, and the Lagrangian satellites Helene, Telesto, and Calypso. The initial integrations were described in (Jacobson [1996b,1998a]). The final pre-encounter ephemeris development is under way; Jacobson (2003) contains a preliminary report on the work.

The ephemerides for Triton and Nereid are from the numerical integration by Jacobson *et al.* (1991). These were used for the Voyager mission to Neptune.

The orbits of the irregular satellites of Jupiter, Saturn, and Uranus are all numerically integrated. Discussions of some of the orbits may be found in Jacobson (1999,2000,2001b). All are being continually updated as additional observations become available.

We also provide, via our website "<http://ssd.jpl.nasa.gov>" (Chamberlin *et al.* [1997]), tables of satellite orbital elements. For all of the satellites with integrated orbits the elements represent a precessing ellipse fit to an extended integration. The elements are mean in the sense that they represent an orbit differing from the integrated orbit by only periodic effects over the fit interval.

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