



Jet Propulsion Laboratory
California Institute of Technology

SysML and Beyond: Applications of MBSE at the NASA Jet Propulsion Laboratory

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Jet Propulsion Laboratory, California Institute of Technology

12 April 2018 – Cornell University, Ithaca, NY, USA

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The decision to implement Mars Sample Return will not be finalized until NASA's completion of the NEPA process. This document is being made available for information purposes only.

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About the Presenter

- Software Systems Engineer at NASA JPL
- Education:
 - Ph.D., Mechanical Engineering (Georgia Tech)
 - B.Sc. + M.Sc., Aerospace Engineering and Computer Science (Technical University Munich)
- MBSE lead for two efforts:
 - Thirty Meter Telescope APS Subsystem
 - Planning for a Mars Sample Return effort
- MBSE infrastructure development software engineer
- Research on model-based design space exploration, and applications of Artificial Intelligence for advancing MBSE



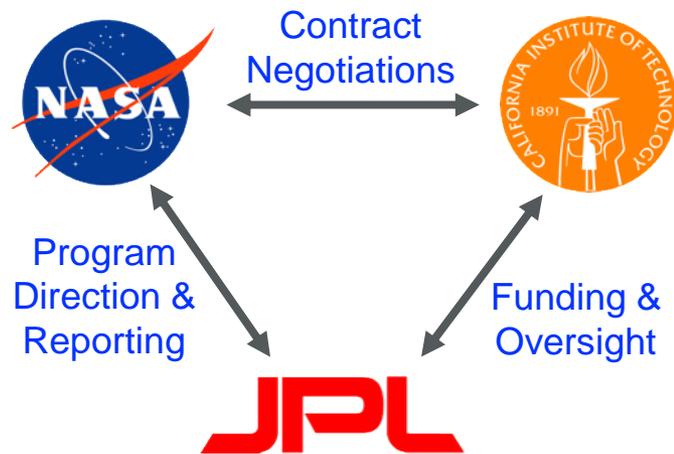
Fun fact: built my first UML model when I was 14!

Agenda

- Introduction
- Model-based Systems Engineering at JPL
- Applications of MBSE with SysML at JPL
 - Thirty Meter Telescope Project (TMT)
 - Planning for a Mars Sample Return (MSR) Effort
- Infusion & Adoption Challenges
- Vision Forward
- Summary & Conclusions

NASA Jet Propulsion Laboratory (JPL)

- Located in Pasadena, CA
- NASA-owned *"Federally-Funded Research and Development Center"*
- University-operated
- ~5,000 employees



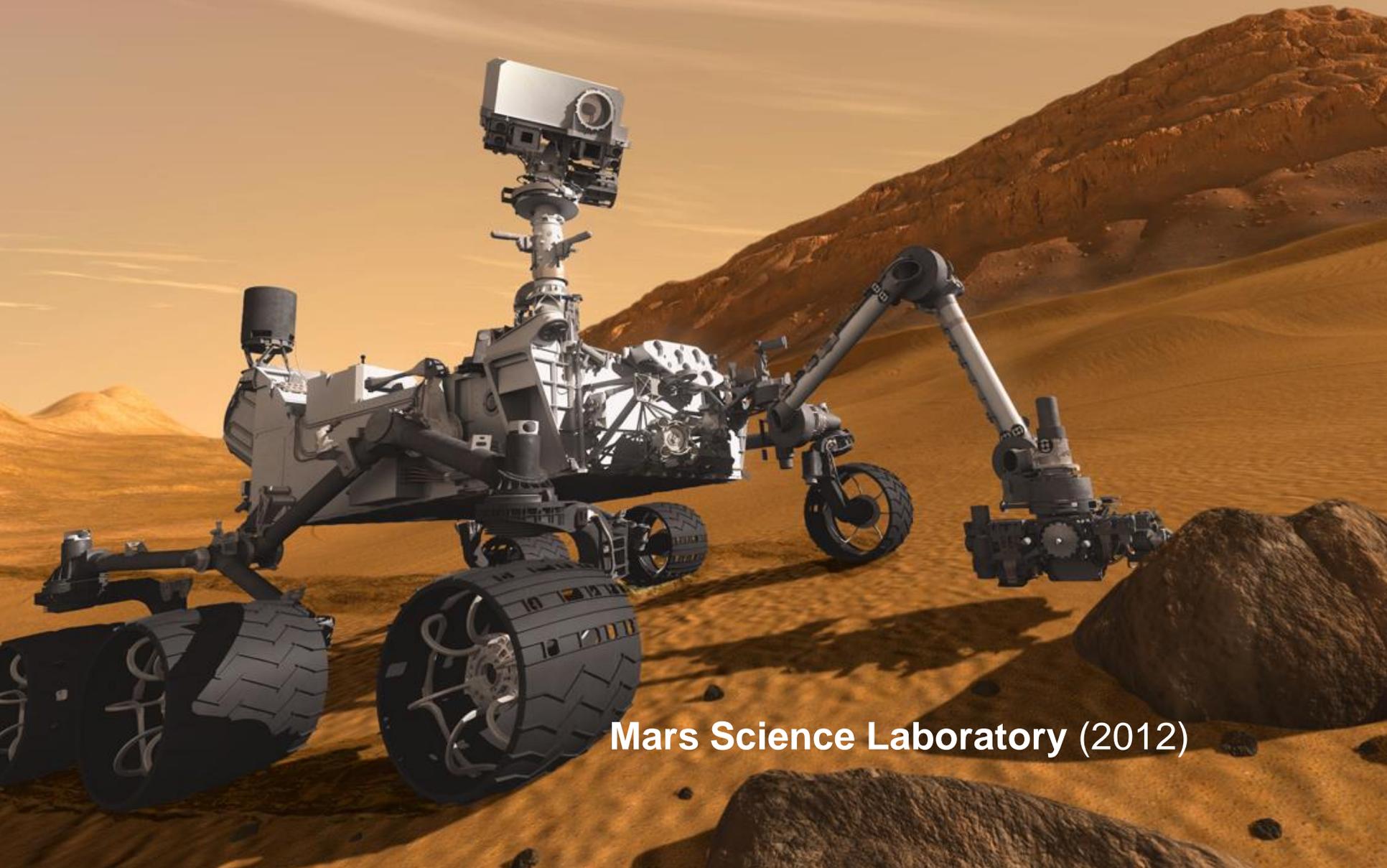
Source: Lin et al., 2011

**You May Know Some
of Our Missions...**



Voyager 1 & 2 (1977)

JPL's Mission is Robotic Exploration



Mars Science Laboratory (2012)

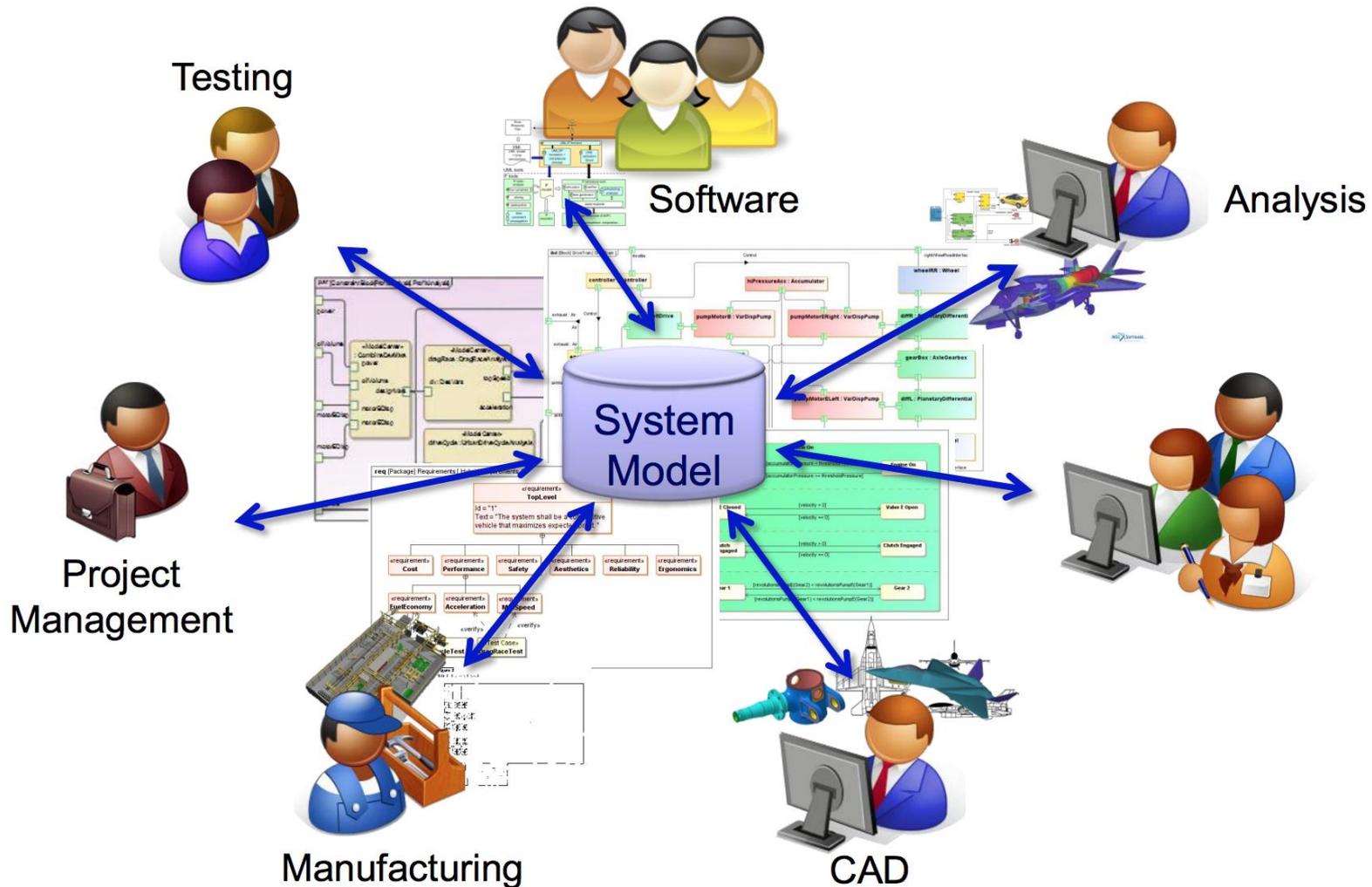
Our Motivation for Adopting MBSE

- Strengthen quality of formulation products by allowing for [exploration of a more comprehensive option space](#)
- More, and [integrated engineering analysis](#) and less paper management
- [Validation](#) of systems [early and often](#)
- Improve quality of [communication and understanding](#) among system and subsystem engineers
- Achieve greater [design re-use](#)
- [Reduce number of product and mission defects](#) in the face of growing complexity, and [increase productivity, reduce cost](#)

Source: Nichols & Lin, 2014

MBSE is Much More Than Just Applied SysML

Information Management Across All Disciplines and the Life Cycle



Source: Paredis, 2012

Agenda

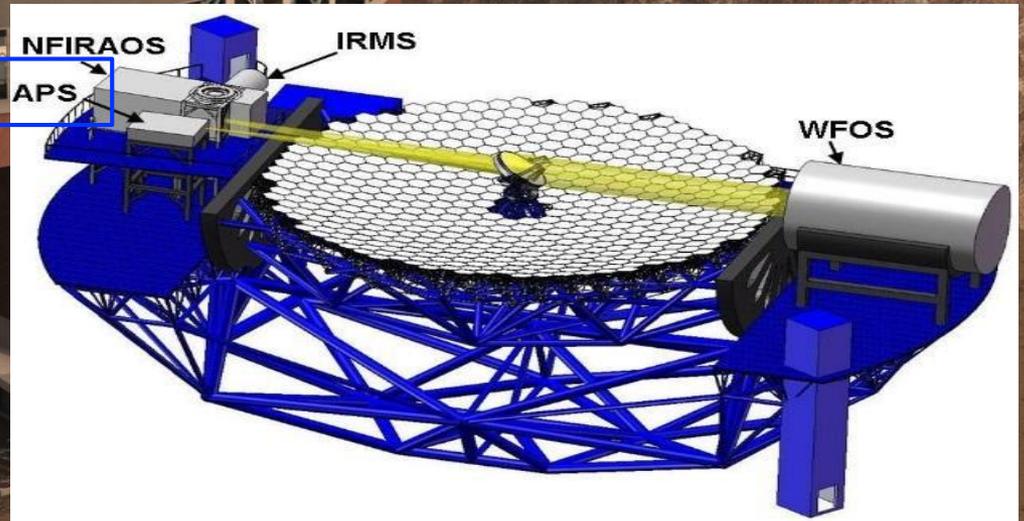
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- ➔ Example Applications of MBSE with SysML at JPL
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Applications of MBSE with SysML at JPL

- JPL is already applying MBSE practice in a wide variety of projects, across a number of lifecycle phases
 - Planned mission to Europa / Europa Clipper
 - Mars 2020 (next Mars Rover)
 - InSight
 - Thirty Meter Telescope
 - Ground system development
 - Planning for a Mars Sample Return Effort
 - Cyber security
 - ...
- MELs, PELs, resource allocation and analysis, system decomposition, libraries / capturing reusable models, etc.
 - ➔ **We use MBSE to do SE**

Not just spacecraft missions! Not just early phases of design!

The Thirty Meter Telescope (TMT) Project



TMT MBSE Objectives

- Define an **executable SysML model**
- Use the model to **analyze the system design and verify requirements** on power consumption, mass, duration, pointing errors, ...
- Produce **engineering documents**
 - Requirement Flow Down Document
 - Operational Scenario Document
 - Design Description Document
 - Interface Control Documents
- Use **standard languages and techniques, and COTS tools where practical** to avoid custom software development

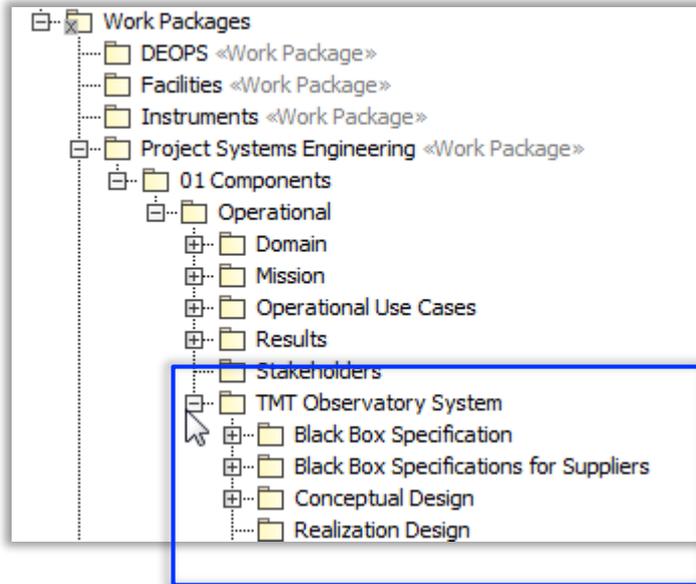
Modeling Approach

- **OOSEM**, but with additional activities focusing on building an executable model
- Use case driven model development
- Challenges:
 - JPL is a **supplier** for a number of subsystems of TMT (the **customer**)
 - Model is used by a number of teams, including TMT directly

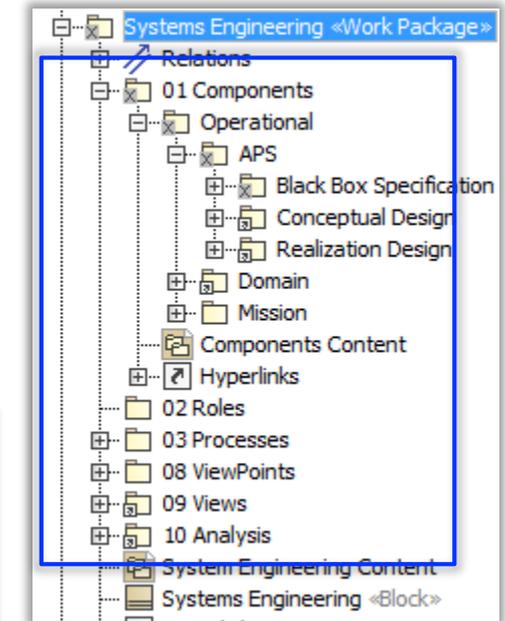
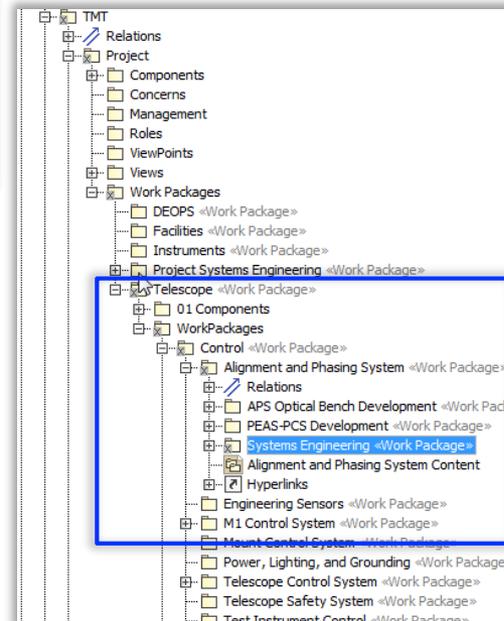
Package Organization

Model Organization Principles

**OOSEM
abstraction
layers**



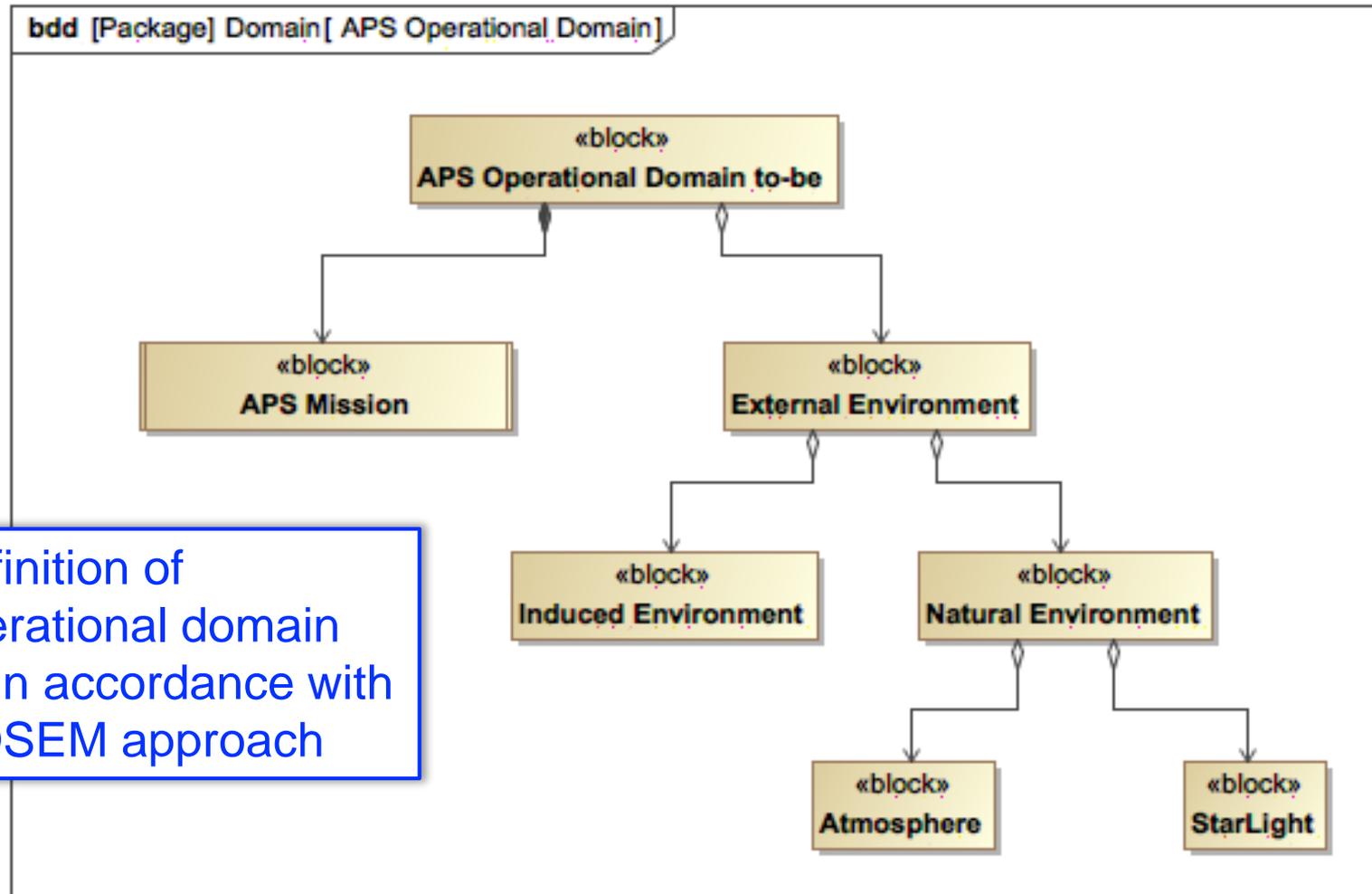
**Customer / supplier
relationship**



**Work
breakdown
structure**

Operational Domain

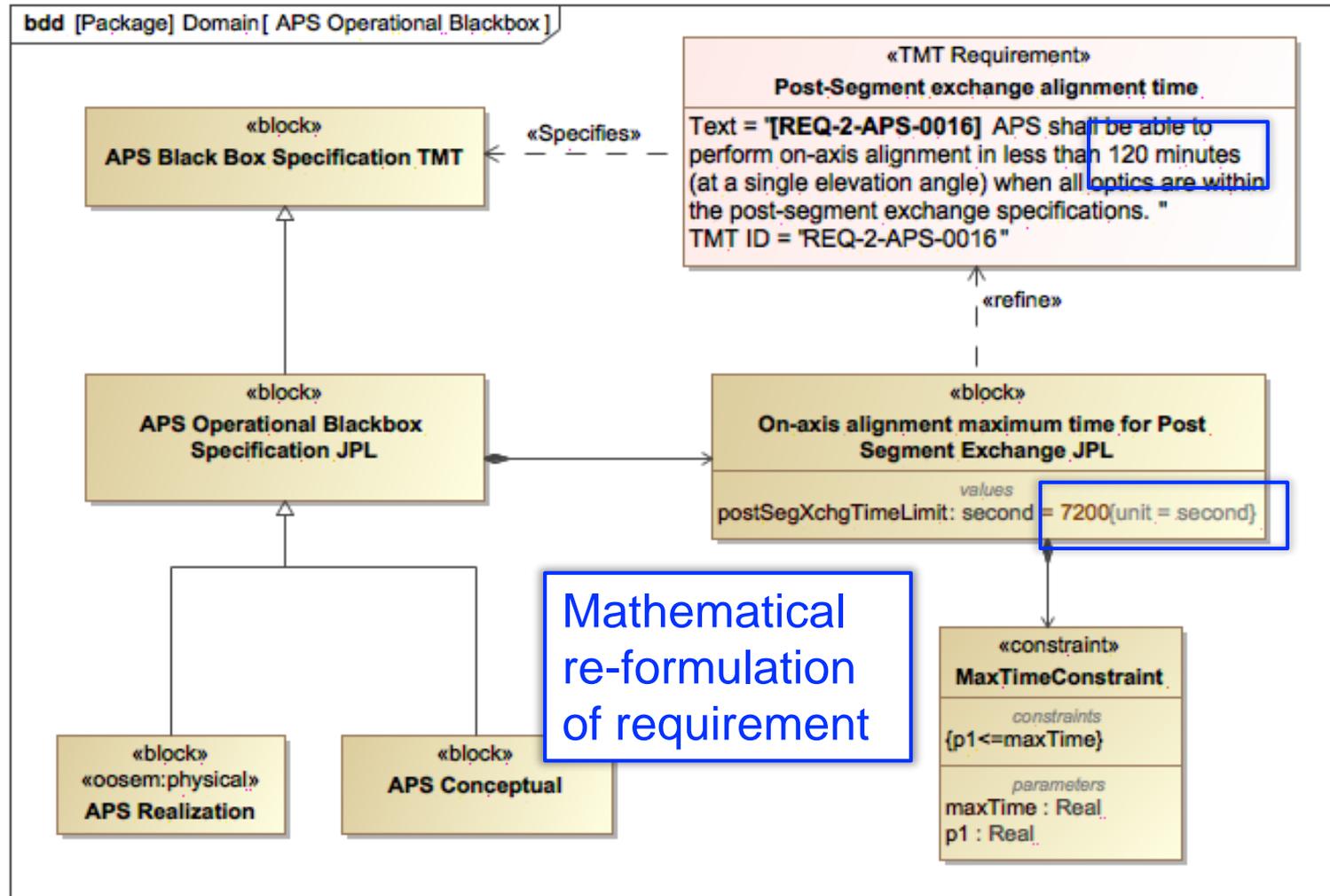
Thirty Meter Telescope



Definition of operational domain as in accordance with OOSEM approach

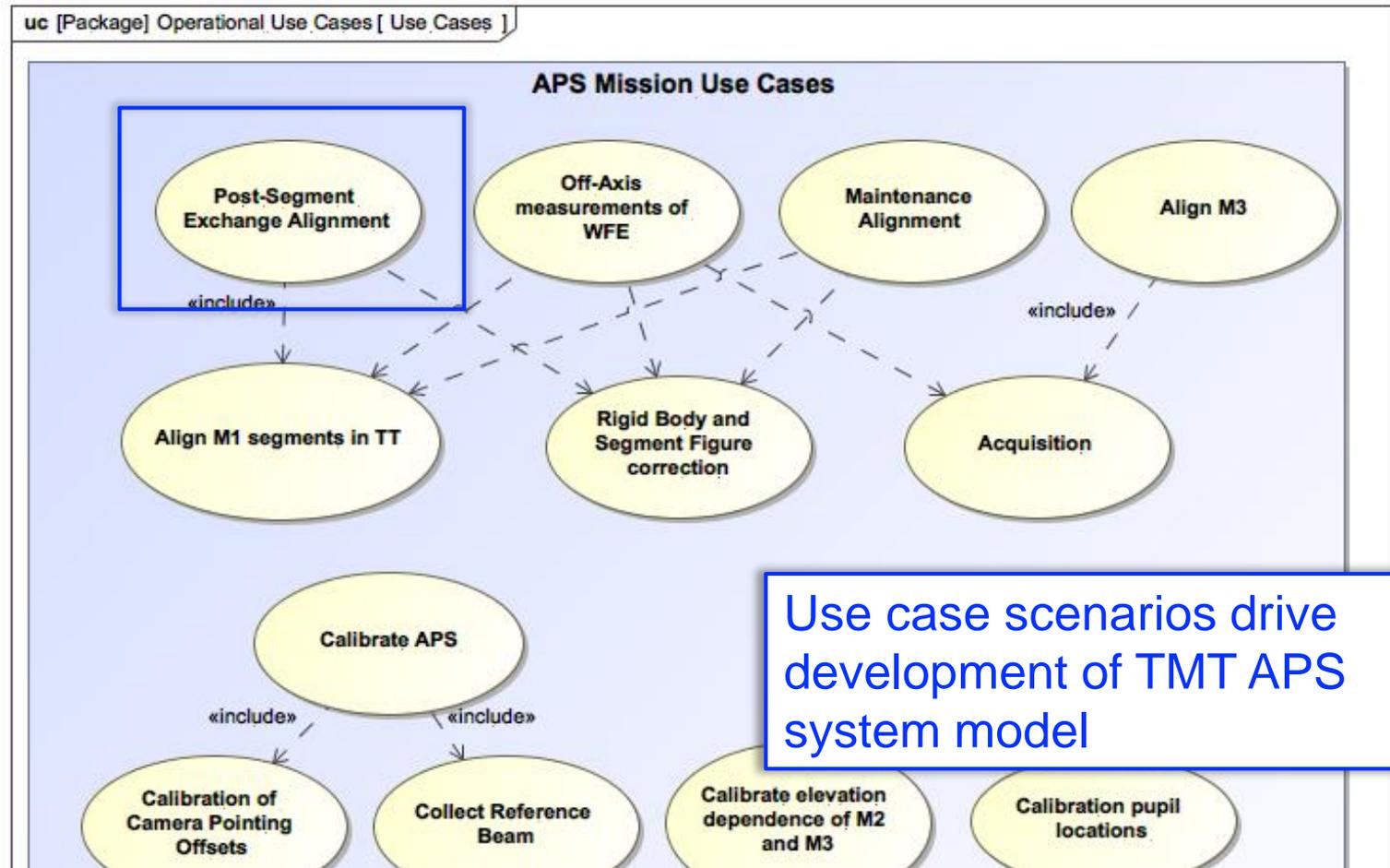
Formalizing Requirements

Thirty Meter Telescope



Use Cases

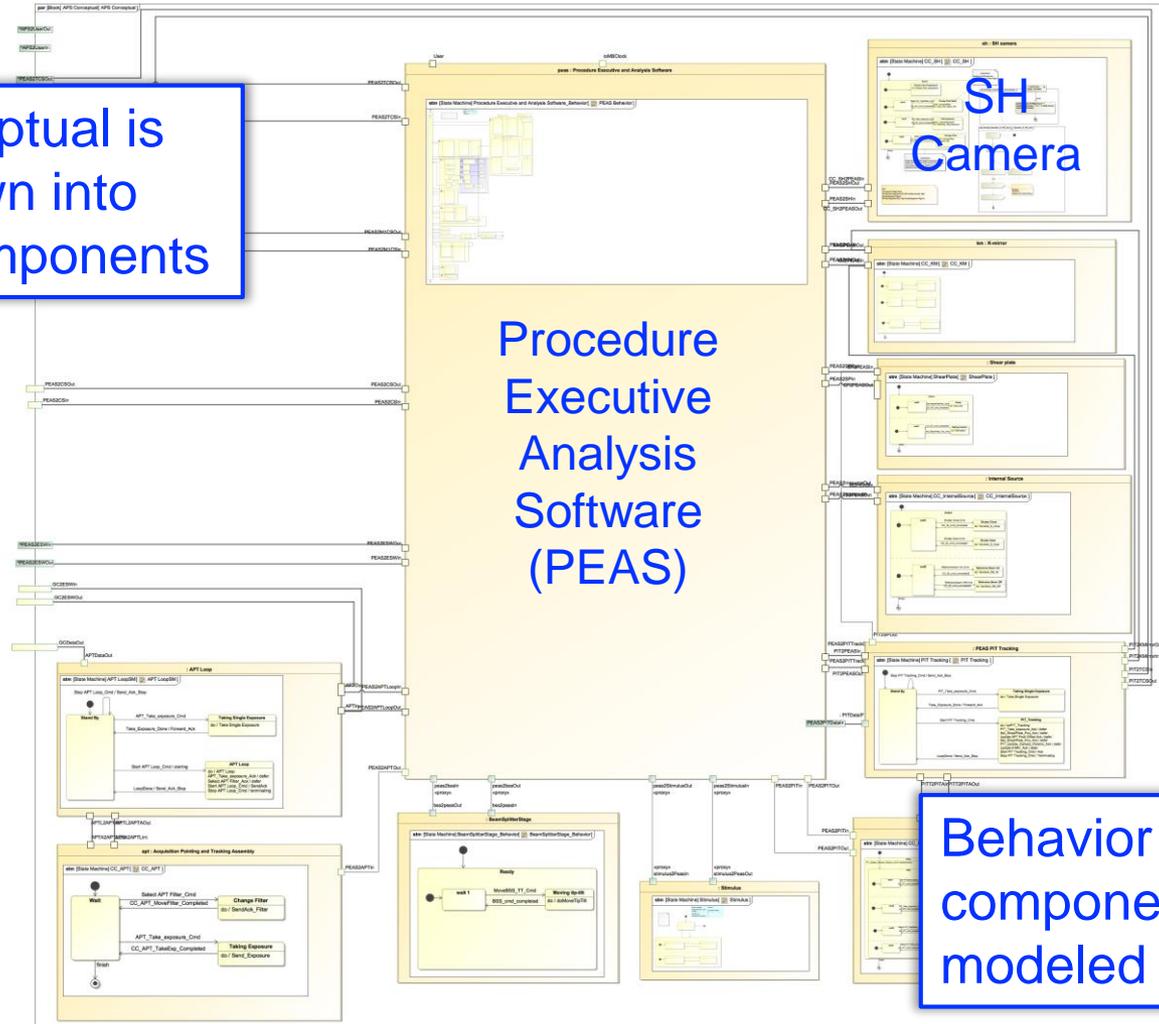
Thirty Meter Telescope



Conceptual Architecture

Thirty Meter Telescope

APS conceptual is broken down into several components



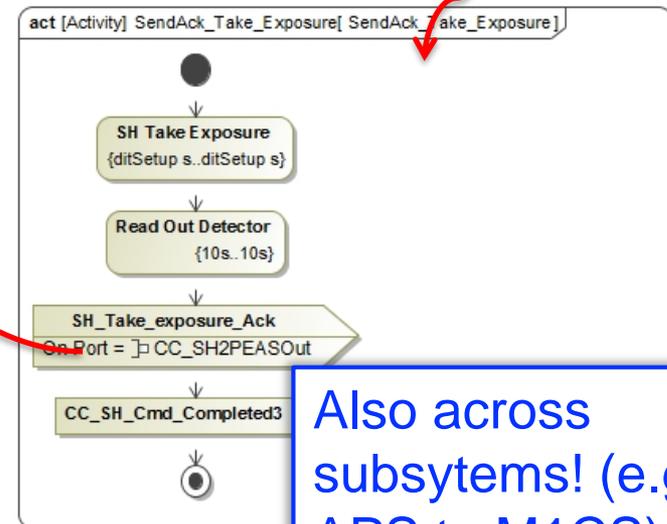
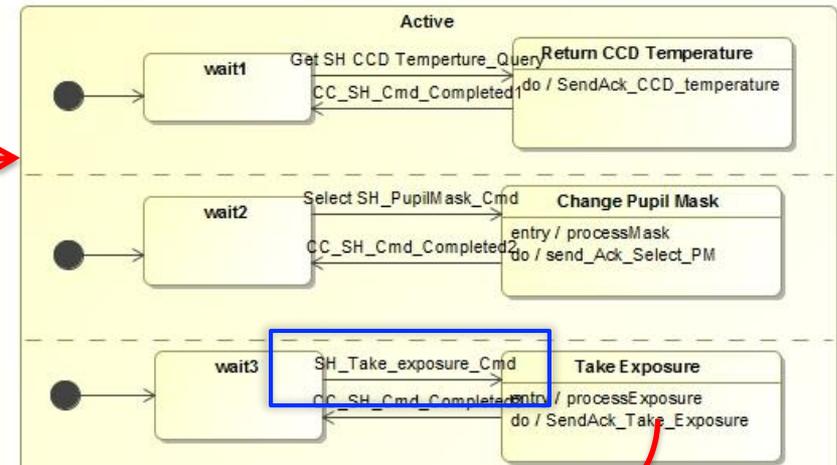
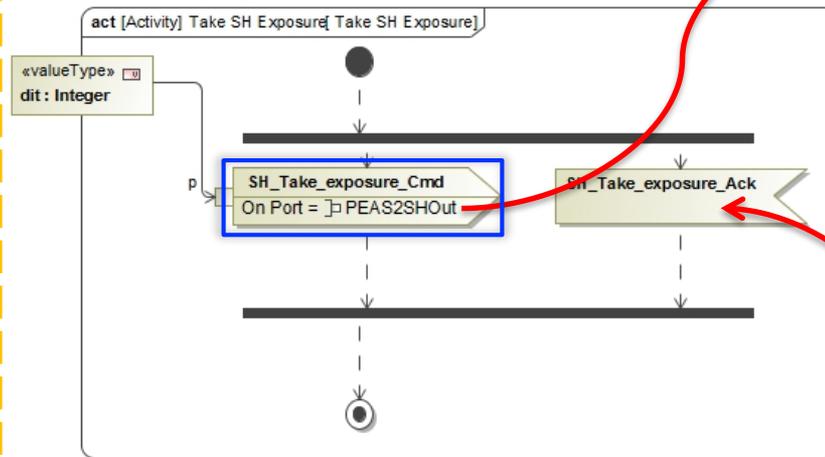
Behavior of all components modeled

Interactions Between Components

Thirty Meter Telescope

SH Camera Context

PEAS Context

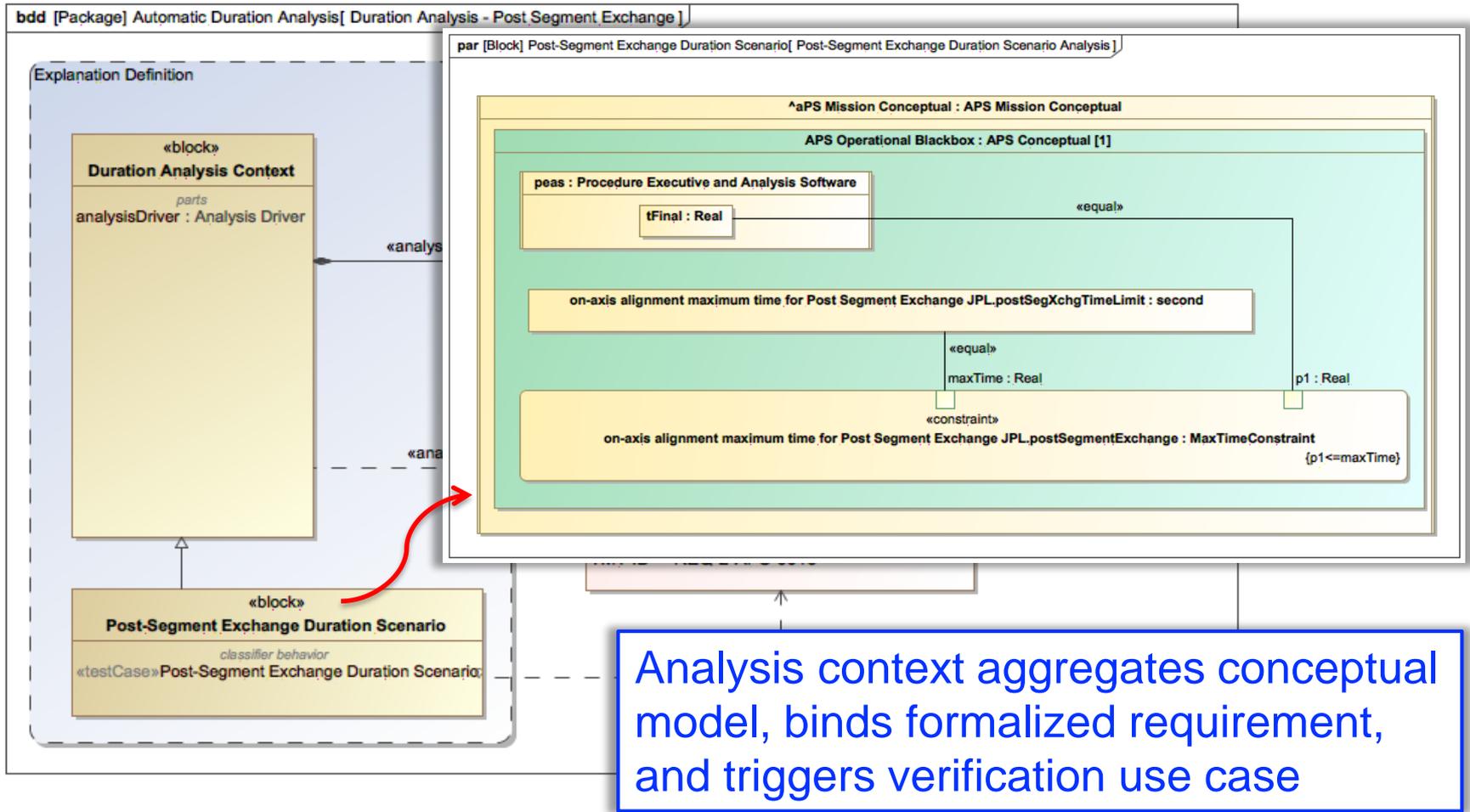


Use of signals sent over ports to simulate a message passing mechanism between components

Also across subsystems! (e.g., APS to M1CS)

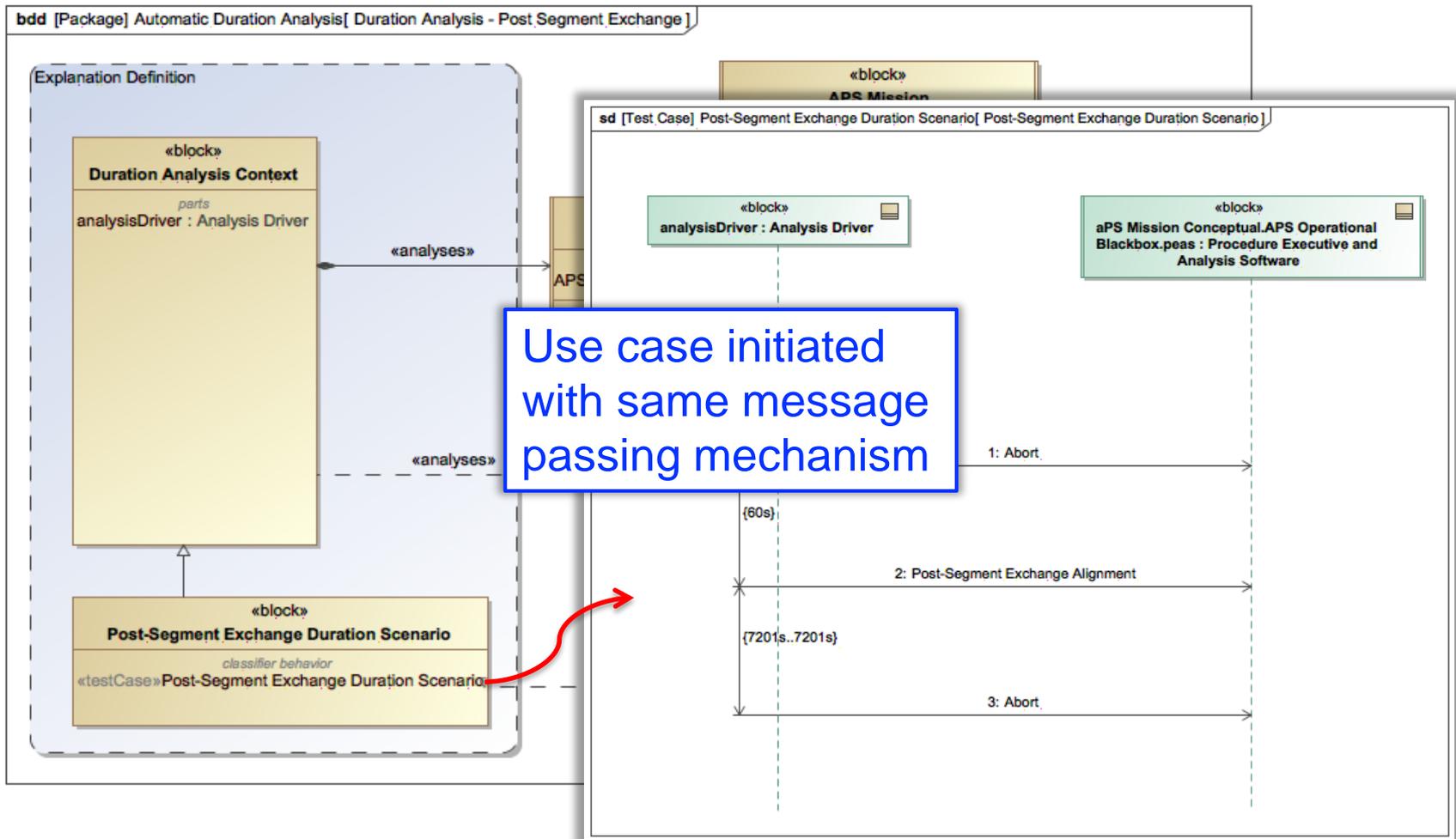
Verifying Timing Requirements by Simulation

Thirty Meter Telescope



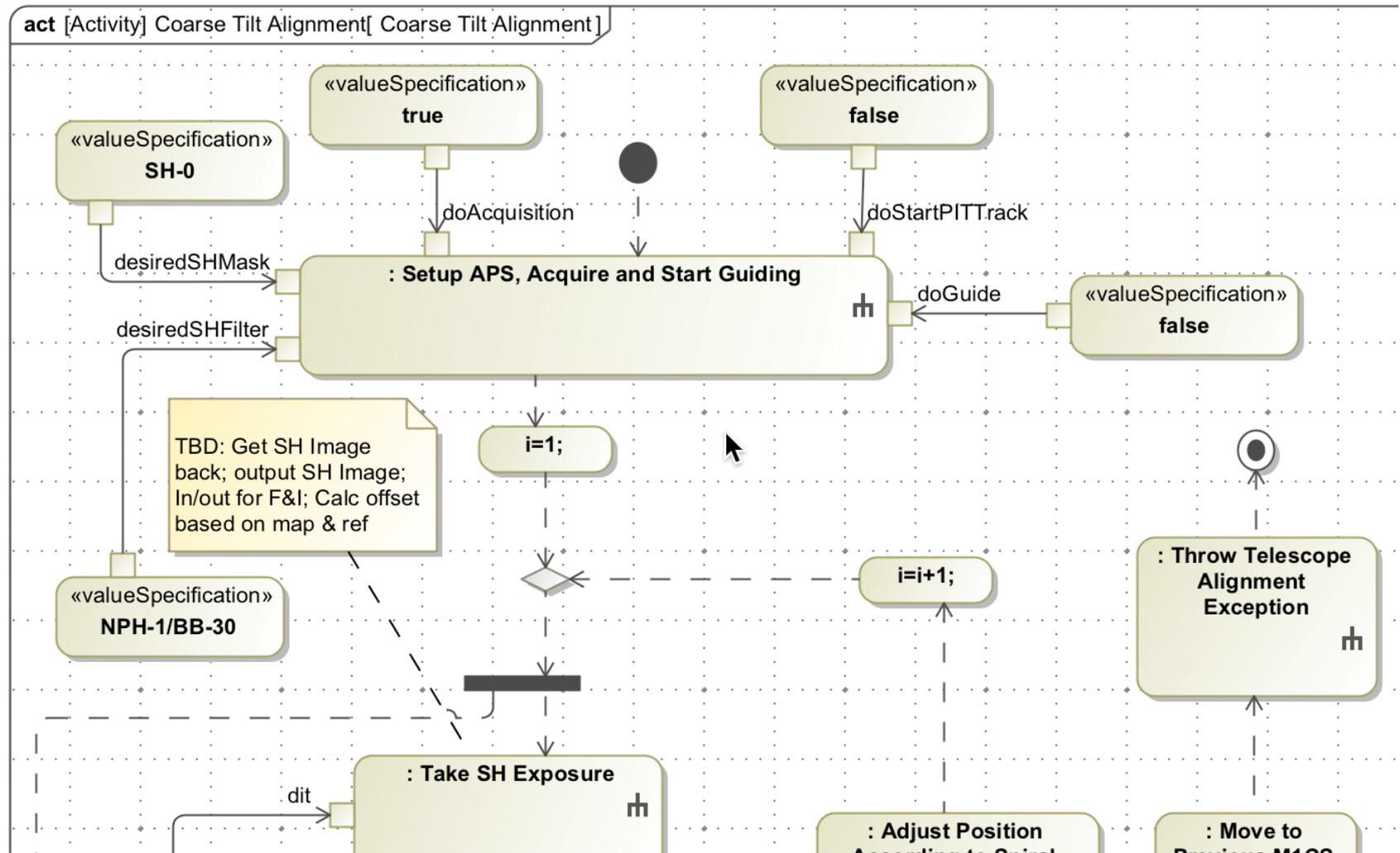
Verifying Timing Requirements by Simulation

Thirty Meter Telescope



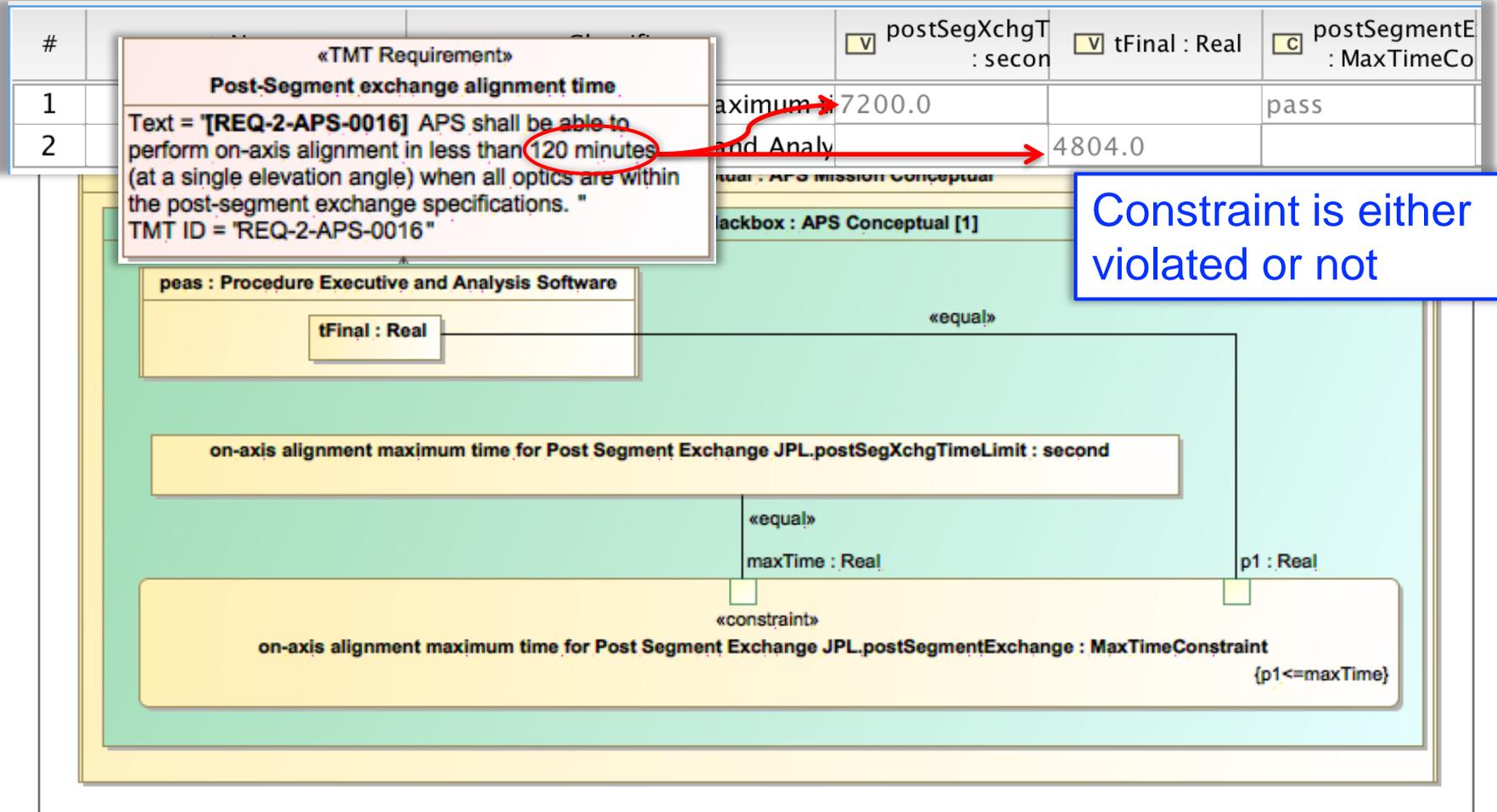
Verifying Timing Requirements by Simulation

Thirty Meter Telescope



Verifying Timing Requirements by Simulation

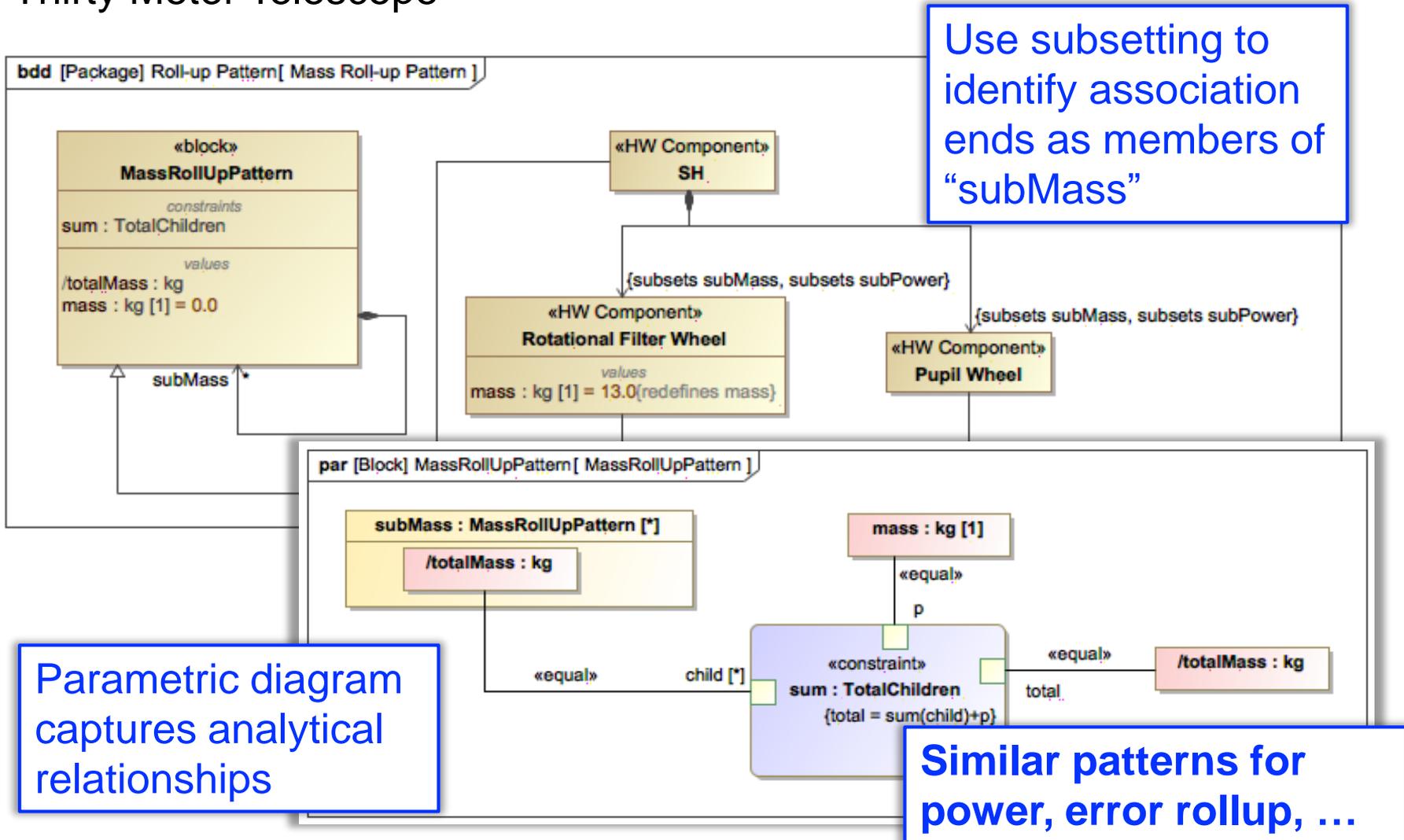
Thirty Meter Telescope



Constraint is either violated or not

“Static” Rollup Analyses – Example: Mass

Thirty Meter Telescope



Lessons Learned

Thirty Meter Telescope

- Possible to use only “vanilla” SysML and COTS tools, but requires **use of advanced UML features**
 - Deep knowledge of UML / SysML required
 - Makes model less accessible, and sometimes cluttered and brittle
 - Sometimes no visual representation of advanced concepts
- Approach can require **human interpretation of model**, e.g., based on names of elements (“everything is a block”)
 - Some typical SE concepts not natively supported: e.g., function
 - Some relationships semantically weak (sometimes on purpose): e.g., allocate, dependency
- Some open questions: modeling variants, relating conceptual and physical architectures, ...

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MBSE with SysML for Potential Mars Sample Return



MSR MBSE Objectives

Mars Sample Return Effort

- Build a model of the **concept of operations** of a Mars Sample Return effort
 - Goals and objectives
 - Policies, and constraints affecting the system
 - Organizations, activities, and interactions among participants and stakeholders
 - Clear statement of responsibilities and authorities delegated
 - Specific operational processes for fielding the system
 - Processes for initiating, developing, maintaining, and retiring the system
- Have a **verifiably consistent** model
- Focus is on **analyzability** of systems engineering information
- Automated generation of **reports & engineering documents**

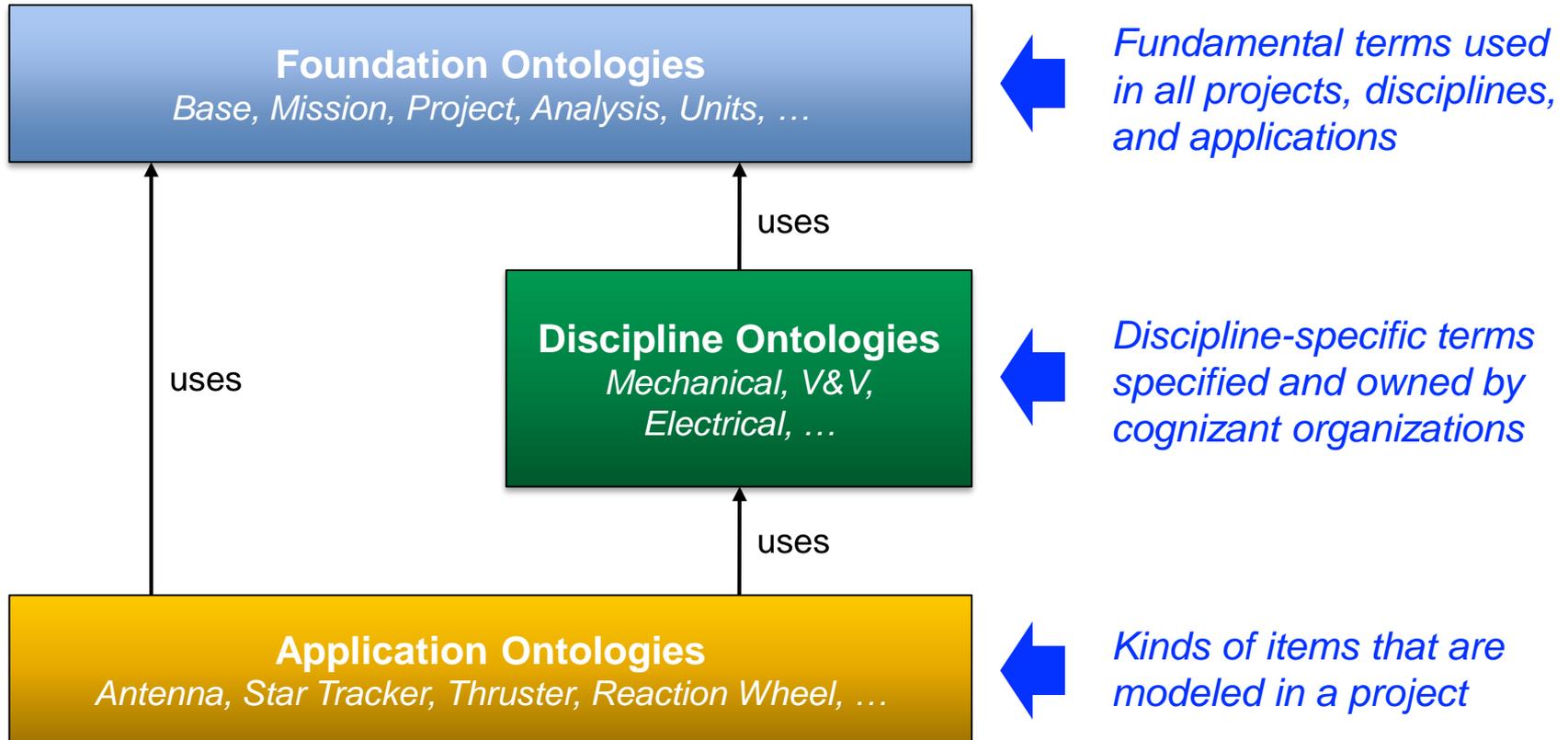
MSR MBSE Approach

Mars Sample Return Effort

- Leverage **OOSEM** where meaningful
- **Apply best practices & lessons learned** from past projects
 - Make use of UML / SysML standards to exploit benefits of different visual syntaxes, executability
 - Use an extended systems engineering vocabulary produced by the Integrated Model-Centric Engineering (IMCE) effort over past decade
- Build an **integrated model** of the concept of operations:
 - Functional decomposition
 - Operational scenarios
 - Structural decomposition
 - Requirements and traces to other model elements
 - Authority delegation

Leveraging the IMCE SE Ontologies

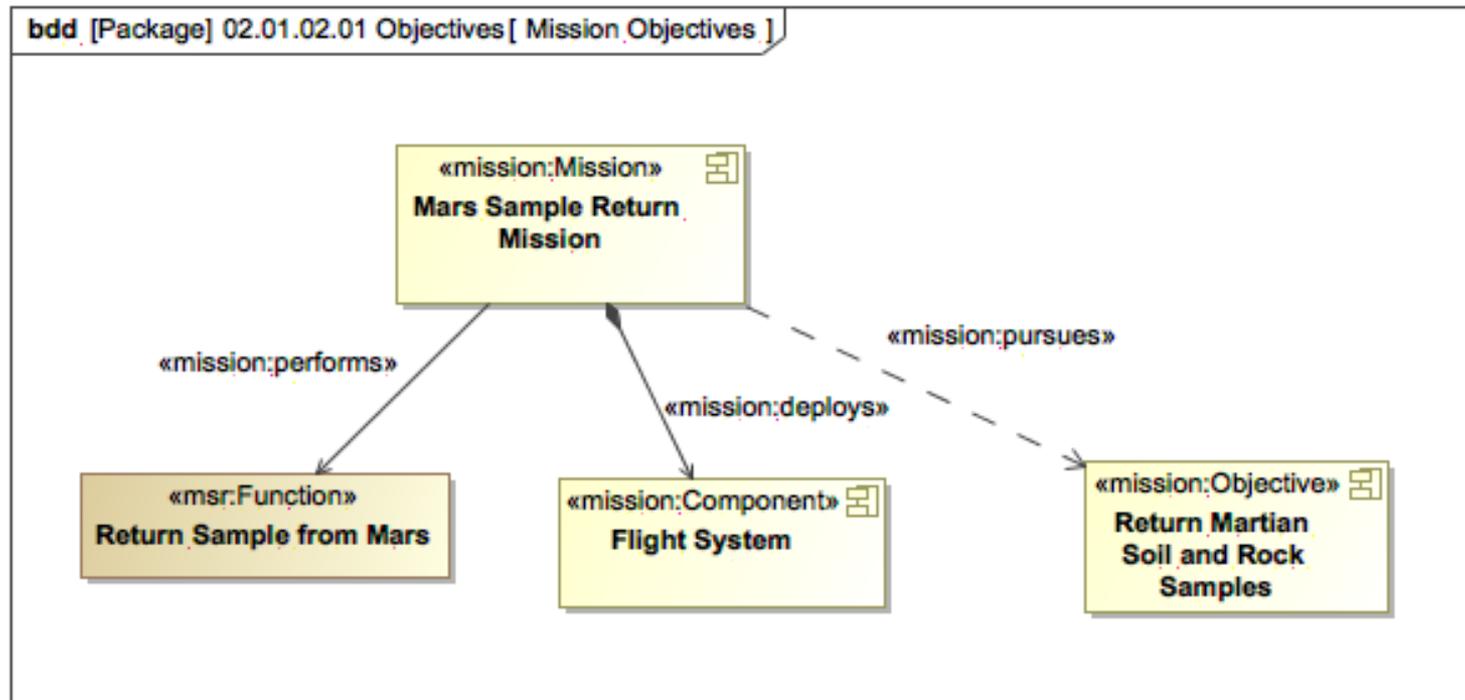
Mars Sample Return Effort



***An ontology describes concepts and relationships
for a domain of interest***

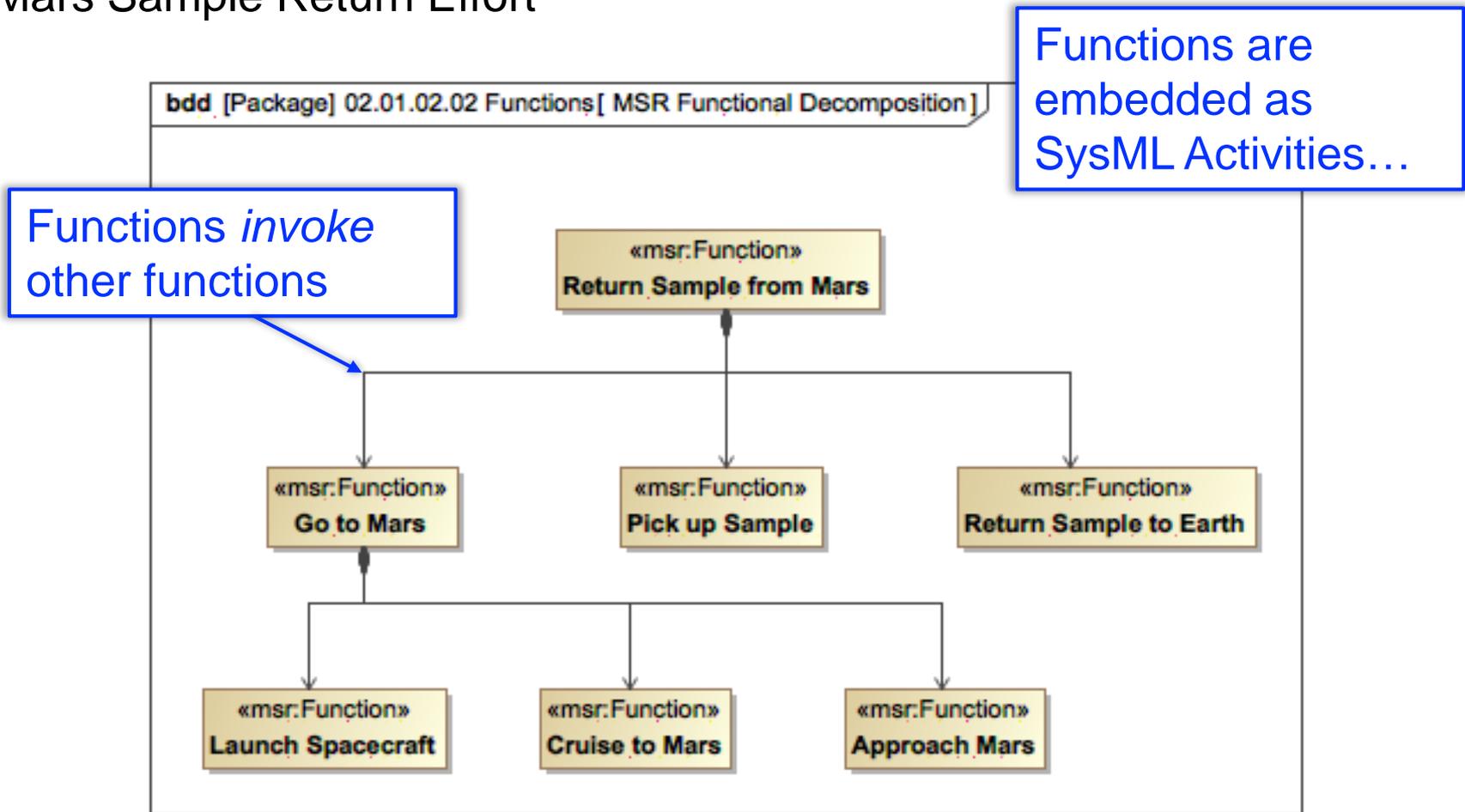
Capturing Goals & Objectives

Mars Sample Return Effort



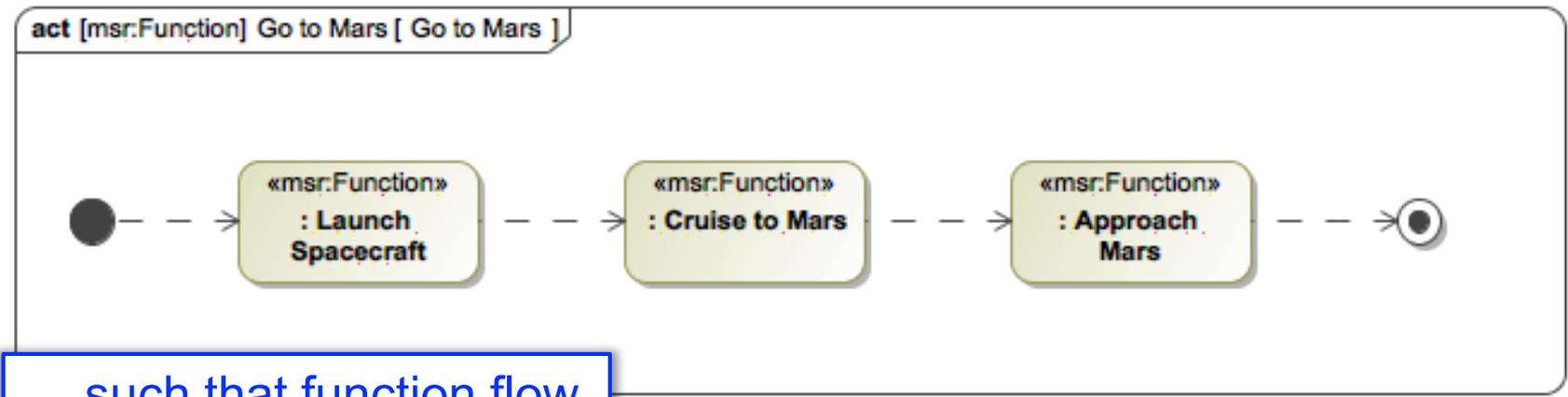
Functional Decomposition

Mars Sample Return Effort



Function Flow & Operational Scenarios

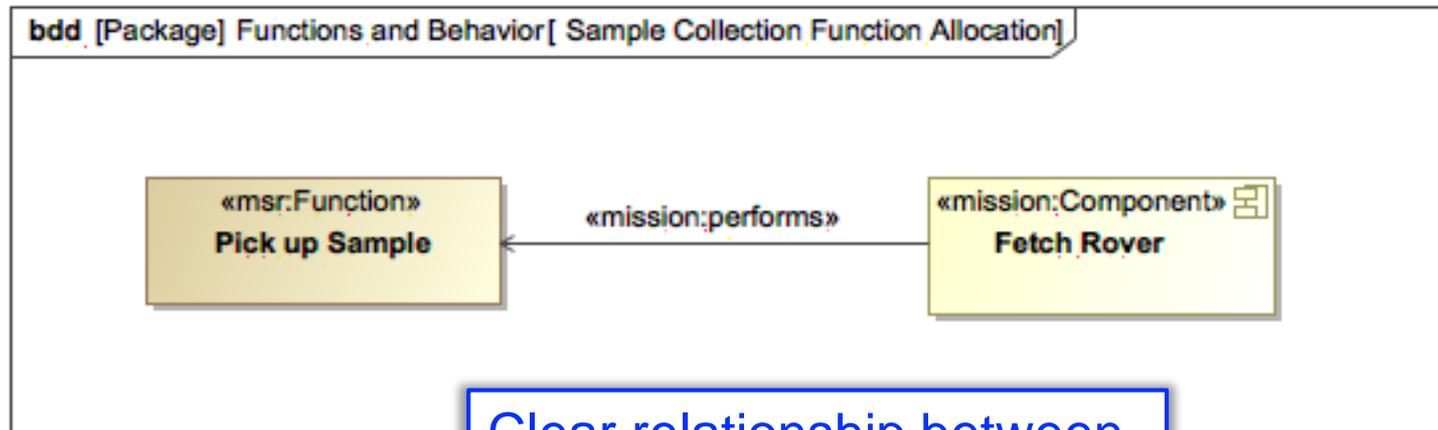
Mars Sample Return Effort



... such that function flow can be captured cleanly

Function Allocation

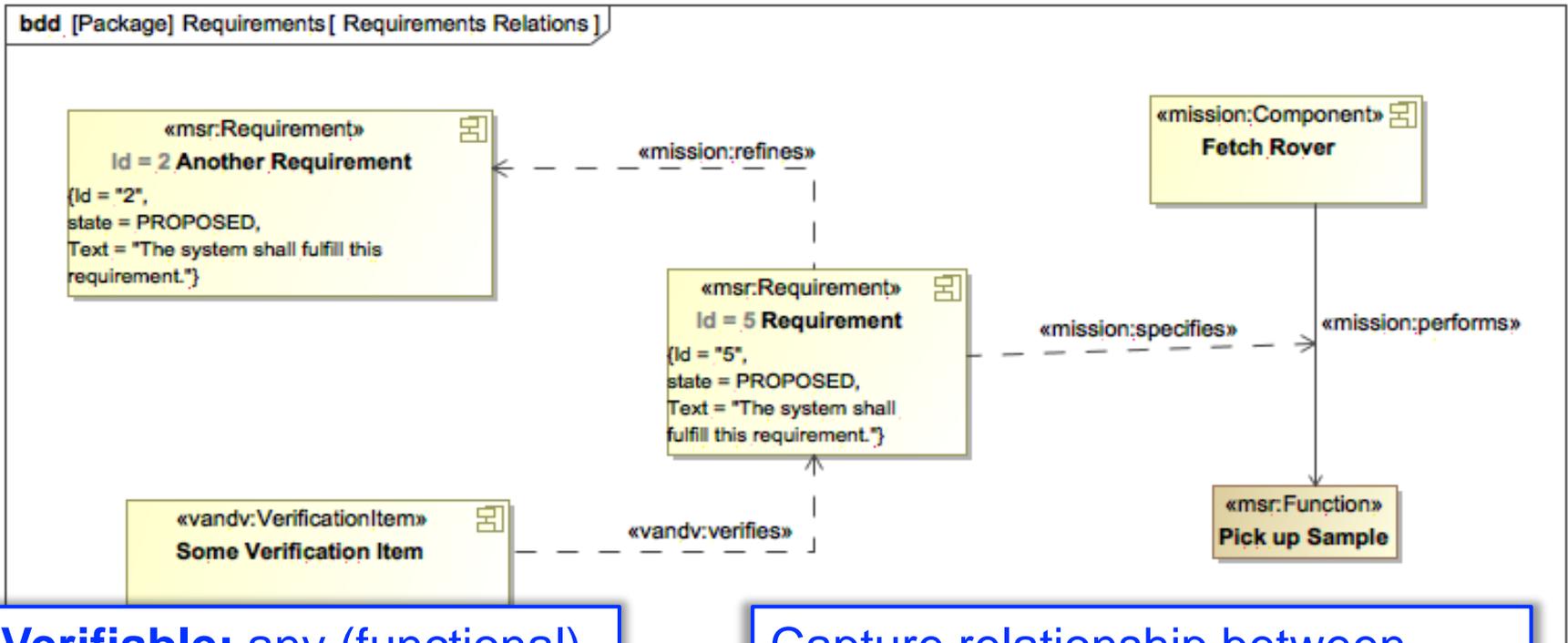
Mars Sample Return Effort



Clear relationship between functions and components
→ **Result of design synthesis activity**

Requirements Tracing

Mars Sample Return Effort

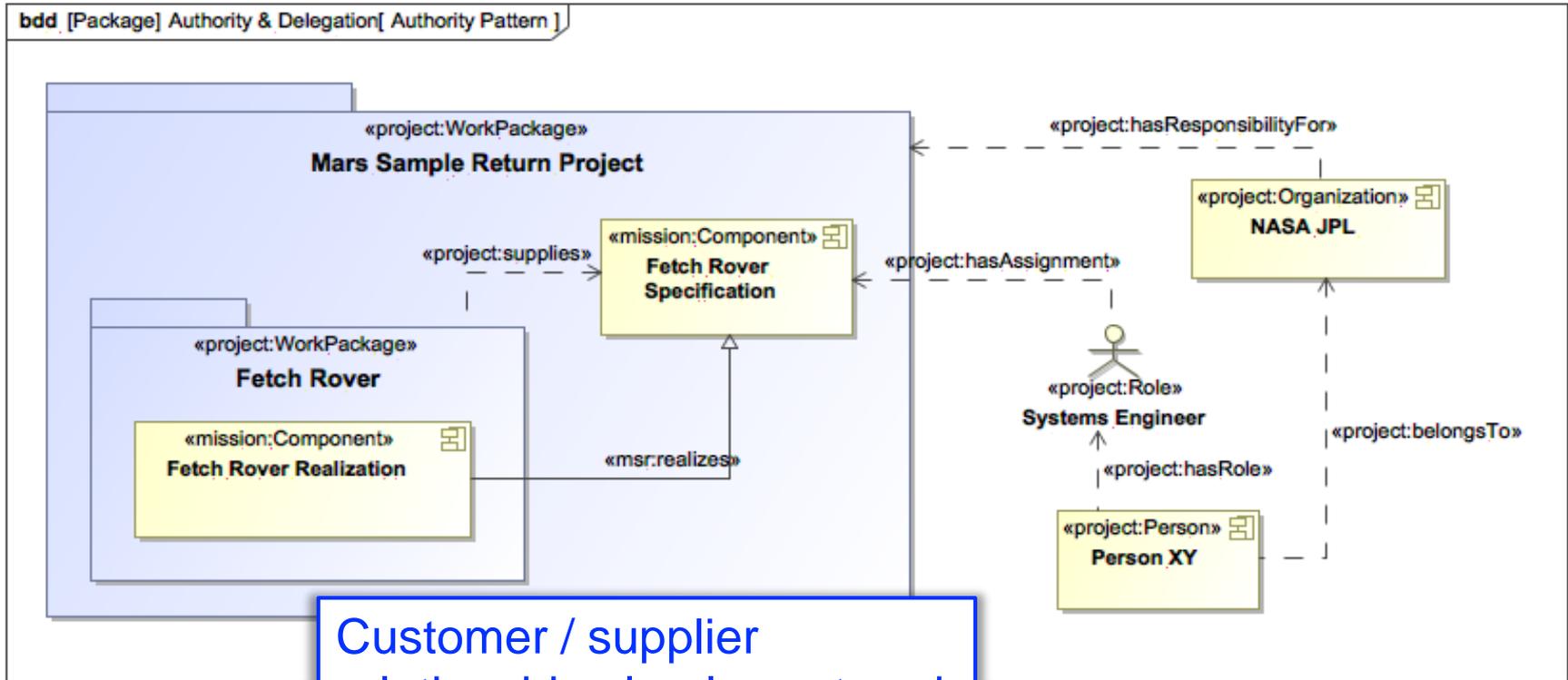


Verifiable: any (functional) requirements that do not specify a function? Any requirements that have no verification method?

Capture relationship between requirements and other system elements, and methods for how requirements are verified

Responsibilities & Authority Delegation

Mars Sample Return Effort



What do we do with the Modeled Information?

Mars Sample Return Effort

- Produce **reports and documents** through the use of model transformations
- Formally **reason** about design
 - Completeness checks & model audits
 - Correctness checks / checking compliance with business rules
- Verify requirements through **simulation**
- Detecting **logical fallacies**
 - Use of a *formal ontology* that carries description logic semantics
 - Formally prove consistency and satisfiability

➔ **Applied formal methods for Systems Engineering!**

Early Lessons Learned

Mars Sample Return Effort

- Extended vocabulary leads to **significantly less ambiguity** in model, but **can be overwhelming** due to the large scope
 - Significantly easier to process (especially computationally), allowing for extensive automated verification
 - More understandable by a human, but potentially vast amount of information
- Effective use **requires tool customizations**
- Requires **training, evangelism** (“does not look like the SysML I’m used to from books”)
- Some **pushback**, since focus is not on ease of use and accessibility, but rather on completeness, precision and rigor
→ **Requires investment and commitment**

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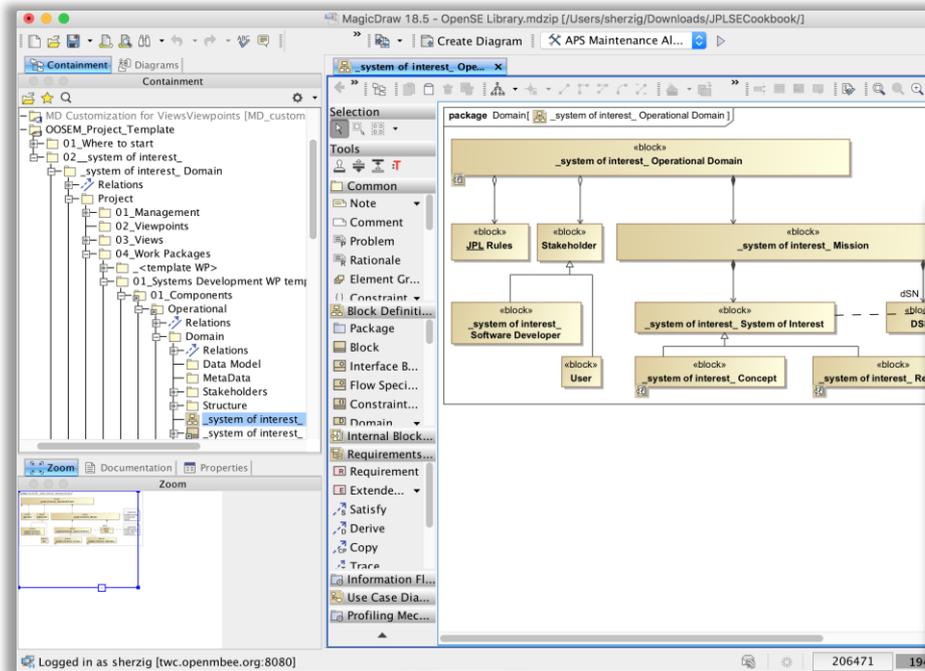
Infusion & Adoption Challenges

- MBSE with SysML still quite a jump for most SE's that are used to Visio, Excel, ...
 - Perceived complexity can be overwhelming, benefits may not be immediately clear (“it worked before”)
 - Mistrust in new technology, since not sufficiently proven
- Oftentimes little resources available for training
 - Is not intuitive, and methods / tools still maturing
 - Search for “easier solution” often leads to formality shortcuts
- Still early in its development
 - Tool support lacking
 - Lack of tried & tested methodologies, standards

How We Are Addressing These Challenges

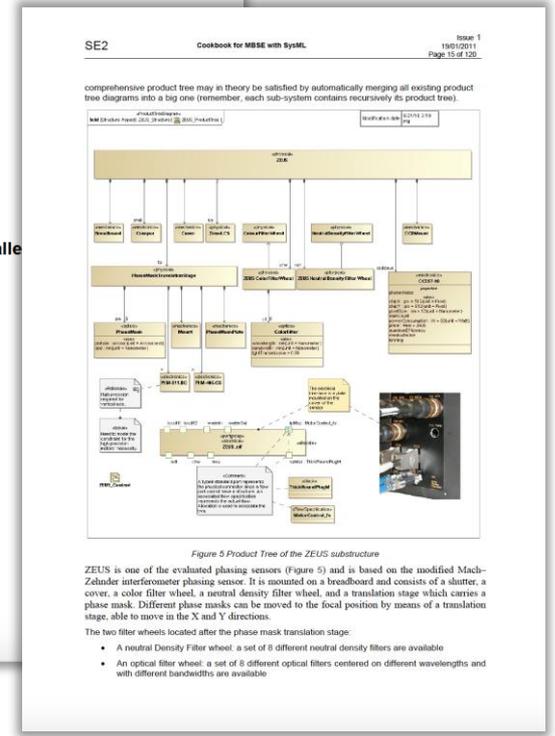
- Clear, consistent message from leadership / line organization
- Dedicated modeling teams composed of modeling experts
 - Either act as “scribes” for systems engineering team
 - Or are part of systems engineering team
- Generation of textual documents and tabular views that have familiar “look and feel”, but interface with model
 - Often required for deliverables anyway
- Provision of internal training: “MBSE bootcamp”, tutorials, “JPL MBSE Cookbook”, modeling patterns development, ...

JPL SE “Cookbook” and Template Model



“Cookbook” for modeling methodology & patterns

MBSE Initiative – SE2 Challenge
COOKBOOK FOR MBSE WITH SYSML
Issue 1
19/01/2011



Template models to be used by projects as a starting point, with recommended organization, model libraries, etc.

Document & Report Generation via View Editor

The screenshot displays the View Editor interface for a document titled "2.1.6 Time to execute". The interface includes a navigation pane on the left with a tree view of the document structure, a main content area with text and a diagram, and a right-hand sidebar with various tool icons. The document content is as follows:

2.1.6 Time to execute

The table below shows our current bottom-up time estimate for each of the activities that make up this use case. The total time estimate is ~75 (TBR) minutes, which is to be compared with our requirement of 120 min (as shown in the figure below).

At Keck, we routinely perform post-segment exchange alignment in 120 minutes or less. However, at Keck the segment shapes are measured in a separate test, with each segment measured separately, but adjustment of the segment warping harnesses is manual and occurs the next day. We will measure the TMT segment shapes in parallel as part of the rigid body and segment figure activity and immediately adjust the segment shapes during the night via the motorized warping harnesses and iterate the control at least once. In addition the CCD read out time for APS is significantly faster than at Keck, ~10 vs ~55 seconds, given the post-segment exchange alignment takes ~60 frames, this accounts for 45 minutes. Given our bottom up estimate and our Keck experience we have a high degree of confidence we can met the 120 minute requirement.

The diagram below illustrates the relationship between the Duration Analysis Context and the APS Mission Conceptual model. The Duration Analysis Context (left) is a block that contains a part named "analysisDriver" of type "Analysis Driver". The APS Mission Conceptual (middle) is a block that contains a part named "APS Operational Blackbox" of type "APS Conceptual [1]" which redefines the "aPS Operational Blackbox JPL". It also has a value named "maxPhasingTime" set to "s = 300.0" which redefines the "maxPhasingTime". The APS Mission Conceptual block satisfies the Duration Analysis Context block, as indicated by the "«satisfy»" relationship arrow.

```
graph LR
    subgraph "bdd [Package] Automatic Duration Analysis [ Duration Analysis - Post Segment Exchange]"
        subgraph "Explanation Definition"
            D[«block» Duration Analysis Context]
            D -- «analyses» --> A[«block» APS Mission Conceptual]
        end
        subgraph "Explanation Results Instance"
            R[Post-Segment Exchange Alignment Timing Analysis Results]
        end
    end
```

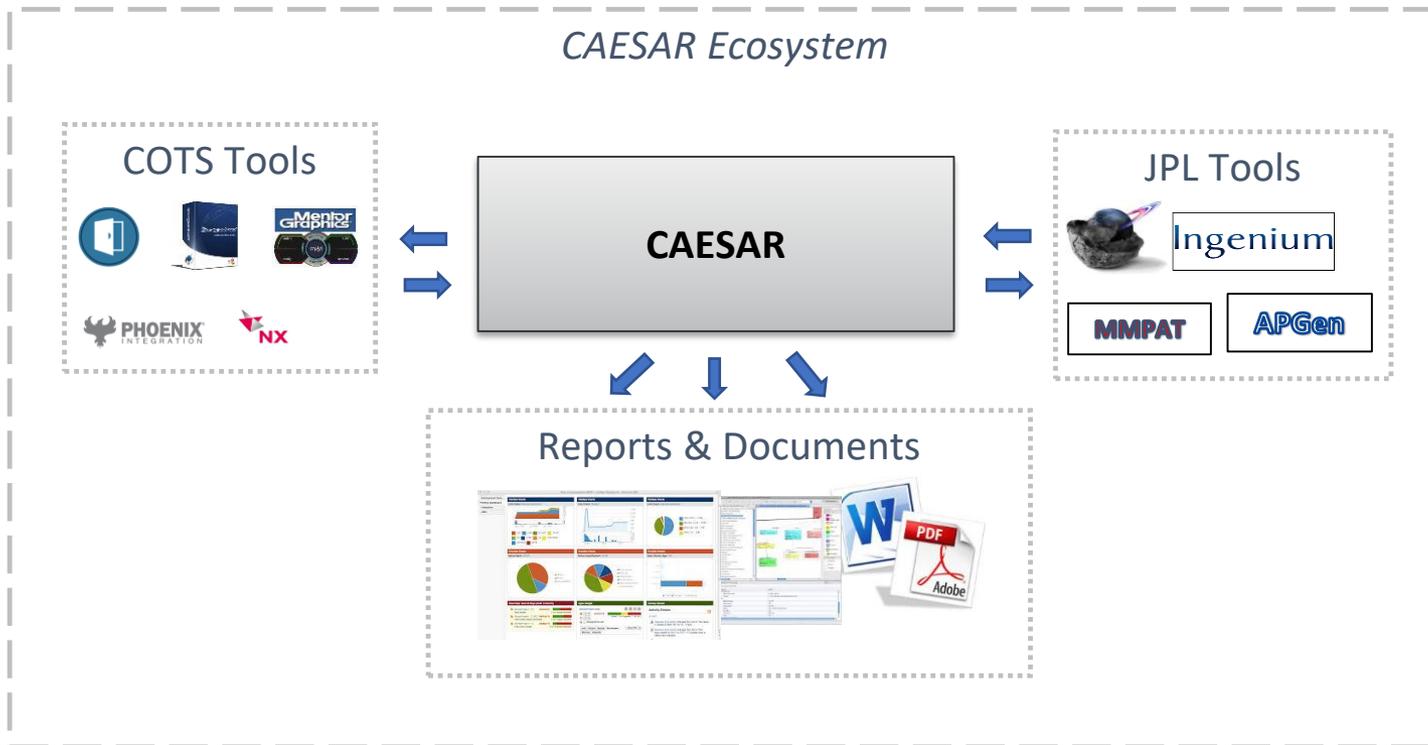
* OpenMBEE / ViewEditor is open source, and available at <http://github.com/Open-MBEE>

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Next-Generation Infrastructure Development

Ontology-based Systems Engineering Information Management

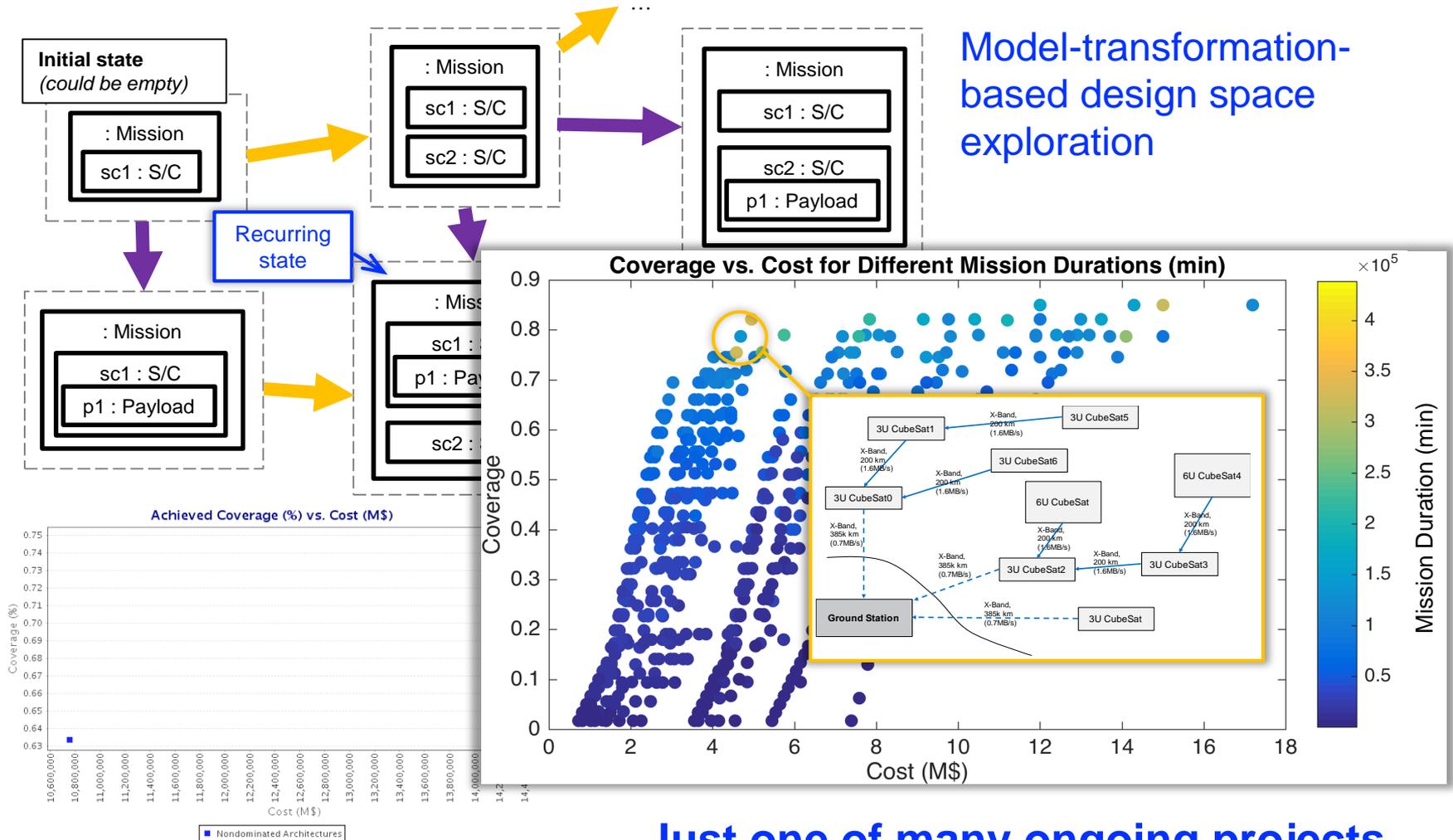


Focus is on integration of information rather than tool or model integration!

Model-based Design Space Exploration

Cutting Edge MBSE Research at JPL

Model-transformation-based design space exploration



Just one of many ongoing projects...

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Summary & Conclusions

- JPL is successfully applying MBSE with SysML to numerous projects - large and small – over different life cycle phases
- Clear benefits: early verification, consistency, less ambiguity
- There has been tremendous progress in tools and methods in the past decade – and we're only just starting
 - Many lessons learned, tools, techniques integrated from MBSE practice
 - Strong developments in methodology and theory
- Magnitude of paradigm shift still leads to skepticism and adoption challenges → transition inevitable, but slow

MBSE is the future of SE - and all of you will take part in shaping it!

Acknowledgements

This work was carried out at the Jet Propulsion Laboratory, California Institute of Technology, under a contract with the National Aeronautics and Space Administration.

The TMT Project gratefully acknowledges the support of the TMT collaborating institutions. They are the Association of Canadian Universities for Research in Astronomy (ACURA), the California Institute of Technology, the University of California, the National Astronomical Observatory of Japan, the National Astronomical Observatories of China and their consortium partners, and the Department of Science and Technology of India and their supported institutes. This work was supported as well by the Gordon and Betty Moore Foundation, the Canada Foundation for Innovation, the Ontario Ministry of Research and Innovation, the National Research Council of Canada, the Natural Sciences and Engineering Research Council of Canada, the British Columbia Knowledge Development Fund, the Association of Universities for Research in Astronomy (AURA) and the U.S. National Science Foundation.

References

- [1] C. Lin, D. Nichols, H. Stone, S. Jenkins, T. Bayer, D. Dvorak: *Experiences Deploying MBSE at NASA JPL*. Frontiers in Model-based Systems Engineering Workshop, Georgia Institute of Technology, Atlanta, Georgia, USA, April 2011.
- [2] Dave Nichols and Chi Lin: *The Application of MBSE at JPL Through the Life Cycle*. INCOSE International Workshop, January 2014.
- [3] S.J.I. Herzig, S. Mandutianu, H. Kim, S. Hernandez, T. Imken: *Model-Transformation-Based Computational Design Synthesis for Mission Architecture Optimization*. AIAA / IEEE Aerospace, March 2017.
- [4] Open-MBEE, available online at <http://github.com/Open-MBEE>.
- [5] TMT SysML Model, available online at <http://github.com/Open-MBEE/TMT-SysML-Model>.
- [6] C. Paredis: *Model Transformations in Model-based Systems Engineering*.

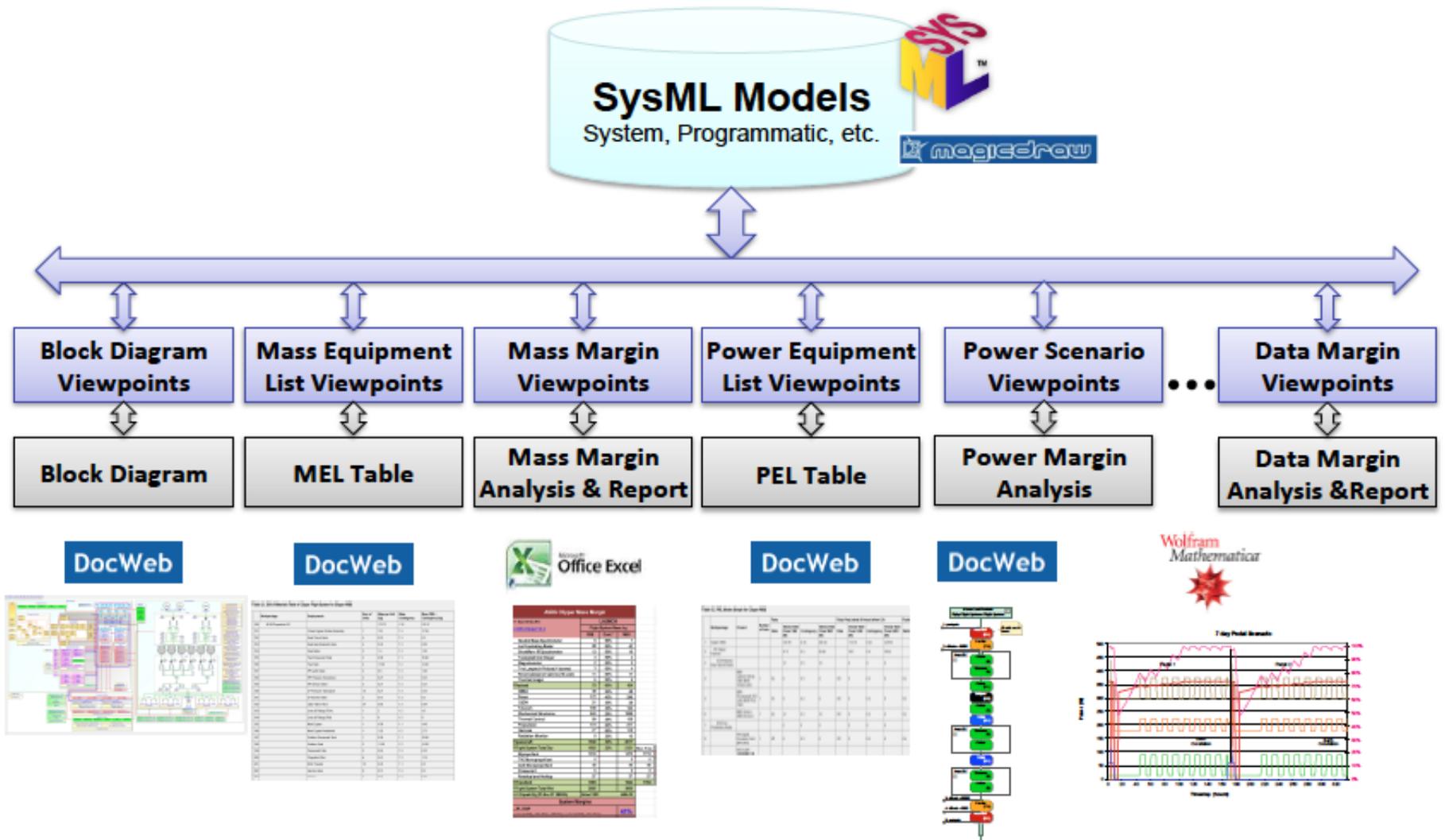


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BACKUP SLIDES

Europa System Model Framework



Pre-Decisional Information -- For Planning and Discussion Purposes Only

Source: Nichols & Lin, 2014

Integrated Power / Energy Analysis

System Model:

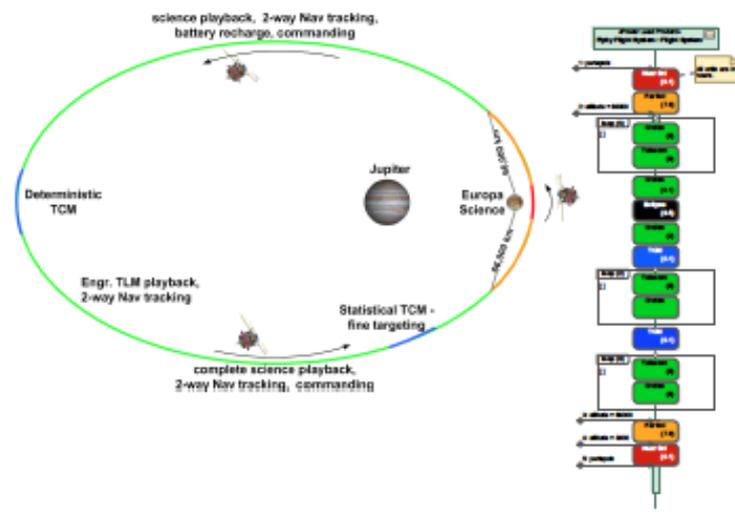
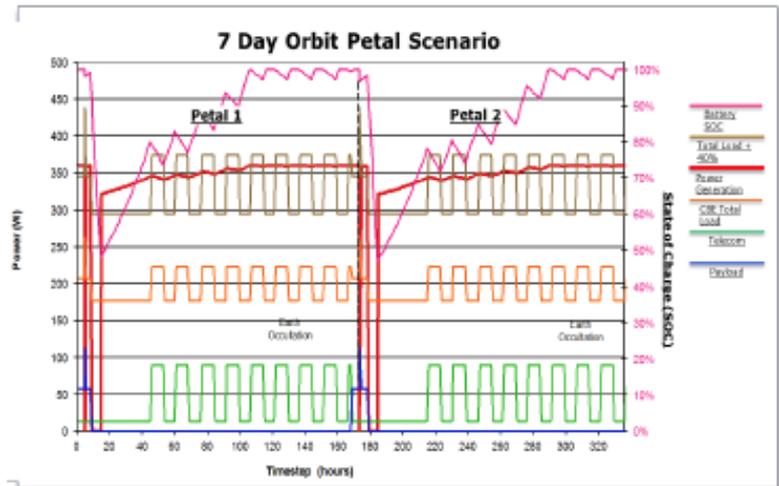
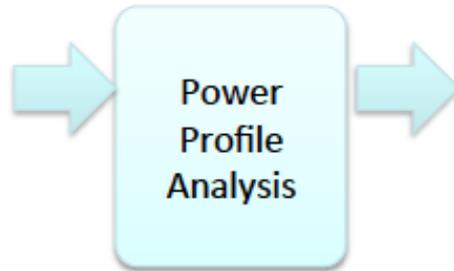
- Equipment List
- Demand vs Mode
- Scenario Definitions

Subsystem Power Models

- Power Source Models
- Battery Models
- Load Profile Simulation

Integrated Power/Energy Analysis

Wired Package	Product	Power (W)					
1	1	1	1	1	1	1	1
2	2	2	2	2	2	2	2
3	3	3	3	3	3	3	3
4	4	4	4	4	4	4	4
5	5	5	5	5	5	5	5
6	6	6	6	6	6	6	6
7	7	7	7	7	7	7	7
8	8	8	8	8	8	8	8
9	9	9	9	9	9	9	9
10	10	10	10	10	10	10	10
11	11	11	11	11	11	11	11
12	12	12	12	12	12	12	12
13	13	13	13	13	13	13	13
14	14	14	14	14	14	14	14
15	15	15	15	15	15	15	15
16	16	16	16	16	16	16	16
17	17	17	17	17	17	17	17
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19	19	19	19	19	19	19	19
20	20	20	20	20	20	20	20

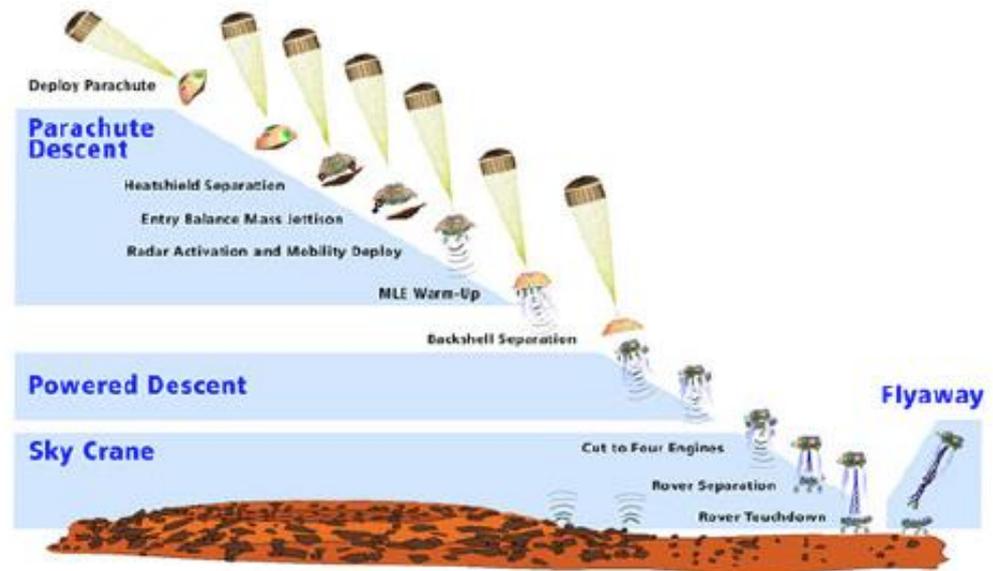
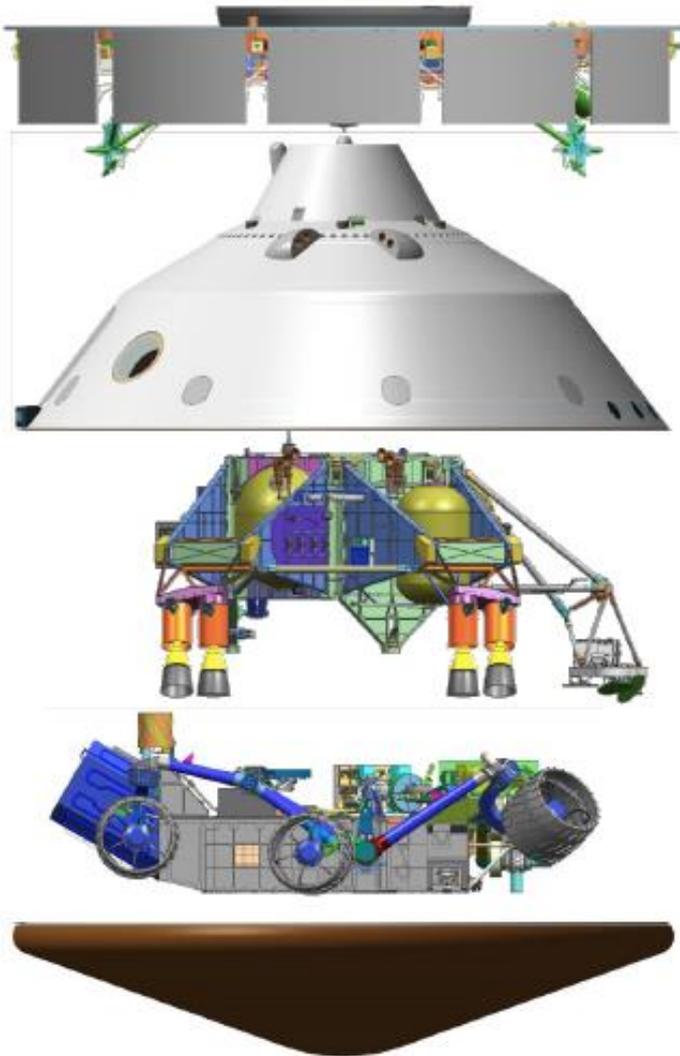


Pre-Decisional Information -- For Planning and Discussion Purposes Only

Source: Nichols & Lin, 2014

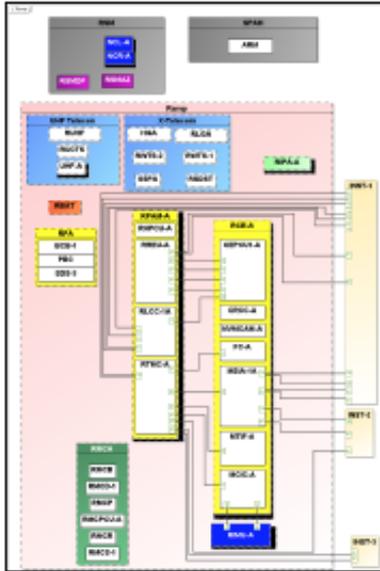
Mars 2020 - Coping with Complexity

- Mars 2020: [follow-on to MSL](#)
- Challenge: engineer inherently complex mission and system at [lower cost](#), and [changes to payload instruments](#)

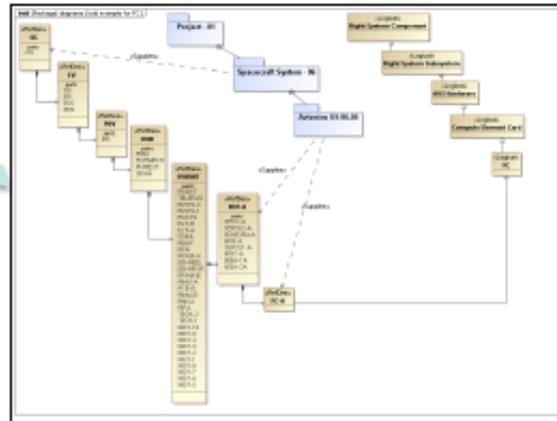


Example System Modeling (Derived) Products

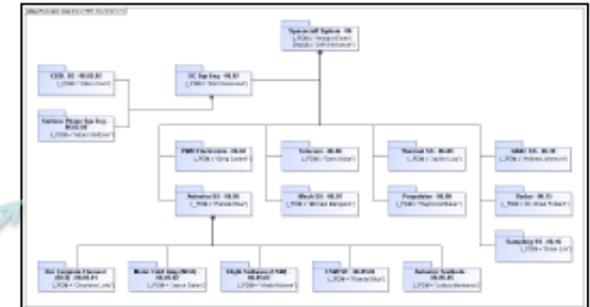
System Block Diagrams and Interfaces



Physical Decomposition, Logical Decomposition, and WBS

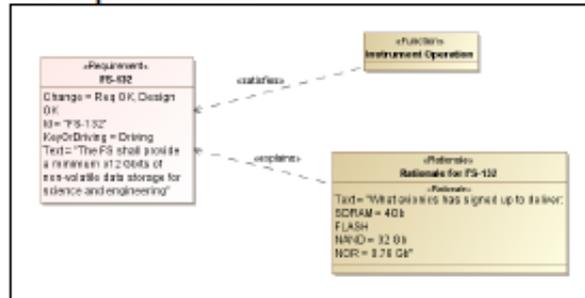


Org Chart



Linking information to core components (Reference Designators)

Assessment of Key & Driving Requirements



Resource Tracking (e.g., subset of web-accessible MEL)

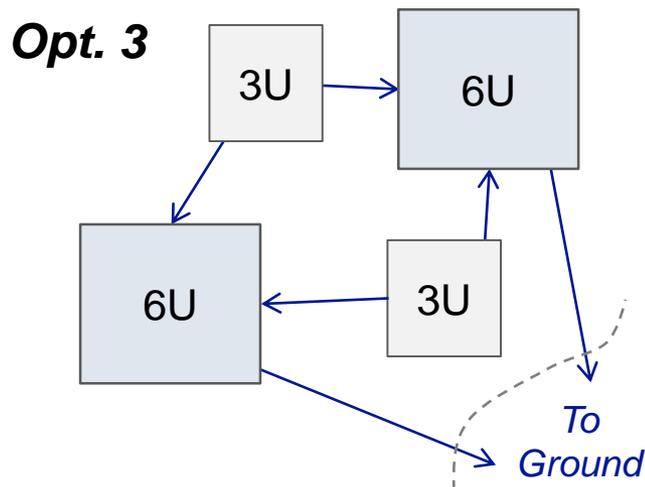
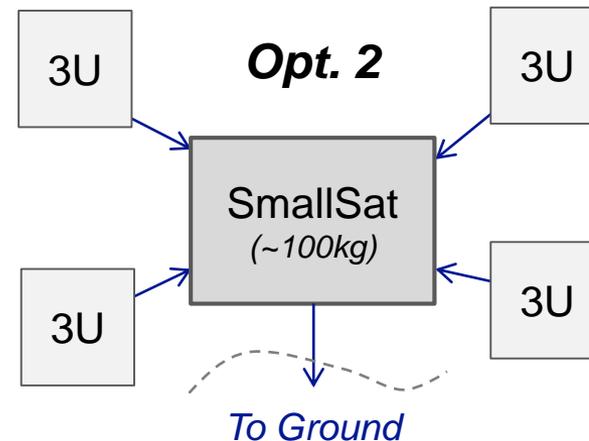
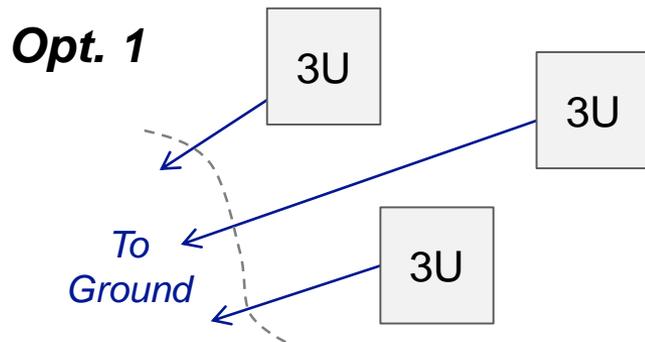
Flight System	Flight Quantity	CBE (kg)	MEY (kg)	Contingency (Percent)	Contingency Level	CBE All Count (kg)	All MEY All Count (kg)
Flight System	1	915.06	942.27	2.97	N/A	915.06	942.27
EPS	1	44.79	45.68	2.00	N/A	44.79	45.68
RTG	1	44.79	45.68	2.00	N/A	44.79	45.68
_PAYLOAD	1	72.23	73.73	2.04	N/A	72.23	73.73
Thermal	1	41.14	41.96	2.00	N/A	41.14	41.96
RVRESTAT	12	0.01	0.01	2.00	N/A	0.16	0.16
RIPA	1	14.58	14.87	2.00	N/A	14.58	14.87
RVRETRM	1	17.28	17.63	2.00	N/A	17.28	17.63
CHESFL	1	0.90	0.91	2.00	N/A	0.90	0.91
RVPRPT	182	0.00	0.00	2.00	N/A	0.28	0.28

Subset of patterns are extended from institutionally-and Europa derived patterns

System model provides integrated, consistent, and broadly-accessible design information and change assessment

Source: Nichols & Lin, 2014

Which Architecture is Optimal?



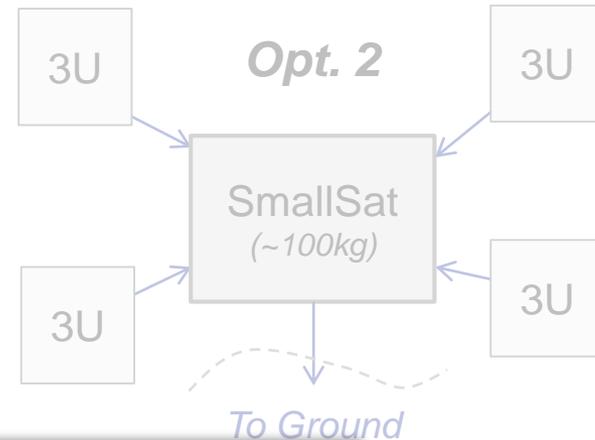
Challenge: transmit very large data volume from LLO to Earth

- How many spacecraft?
- Are all equipped with interferometry payload? Are some just relays?
- Who communicates with Earth?
- What frequency bands? Multi-hop?
- ...
- Optimal w.r.t. cost? Science value?

Which Architecture is Optimal?

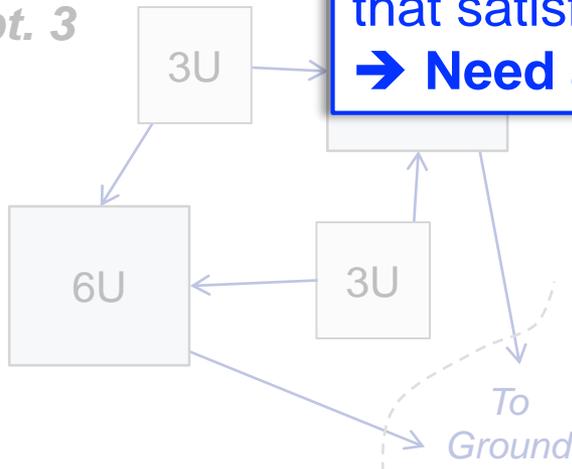
Opt. 1

Same functionality, different qualities / performance
→ **Examine trade-offs**



Opt. 3

Very large number of architectures that satisfy mission objectives
→ **Need automation**

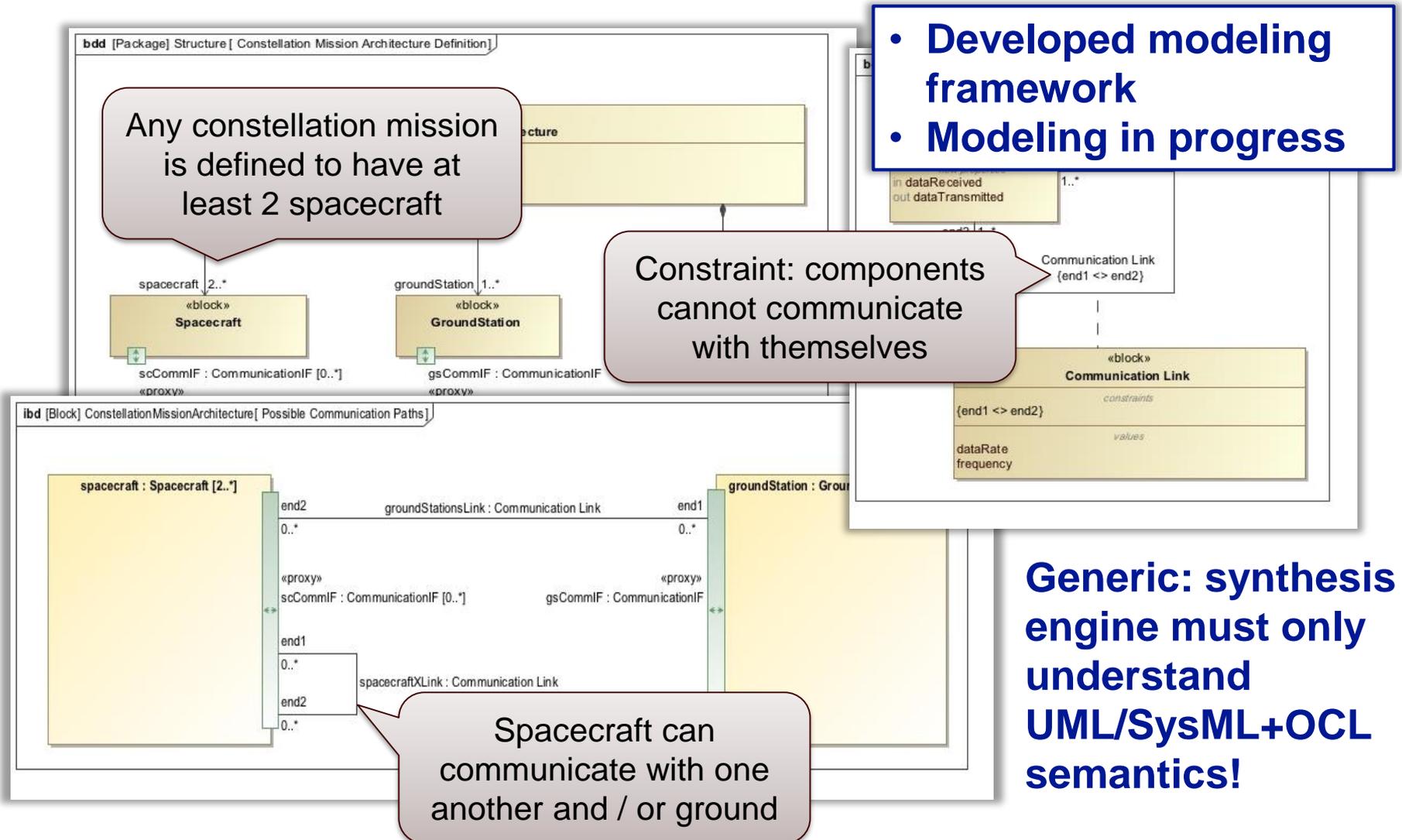


- Very large data Earth
- How many spacecraft?
- Are all equipped with interferometry payload?
 - Who owns the data?
 - What frequency bands? Multi-hop?
 - ...
 - Optimal w.r.t. cost? Science value?

Functional allocation is key
→ **Synthesis problem**

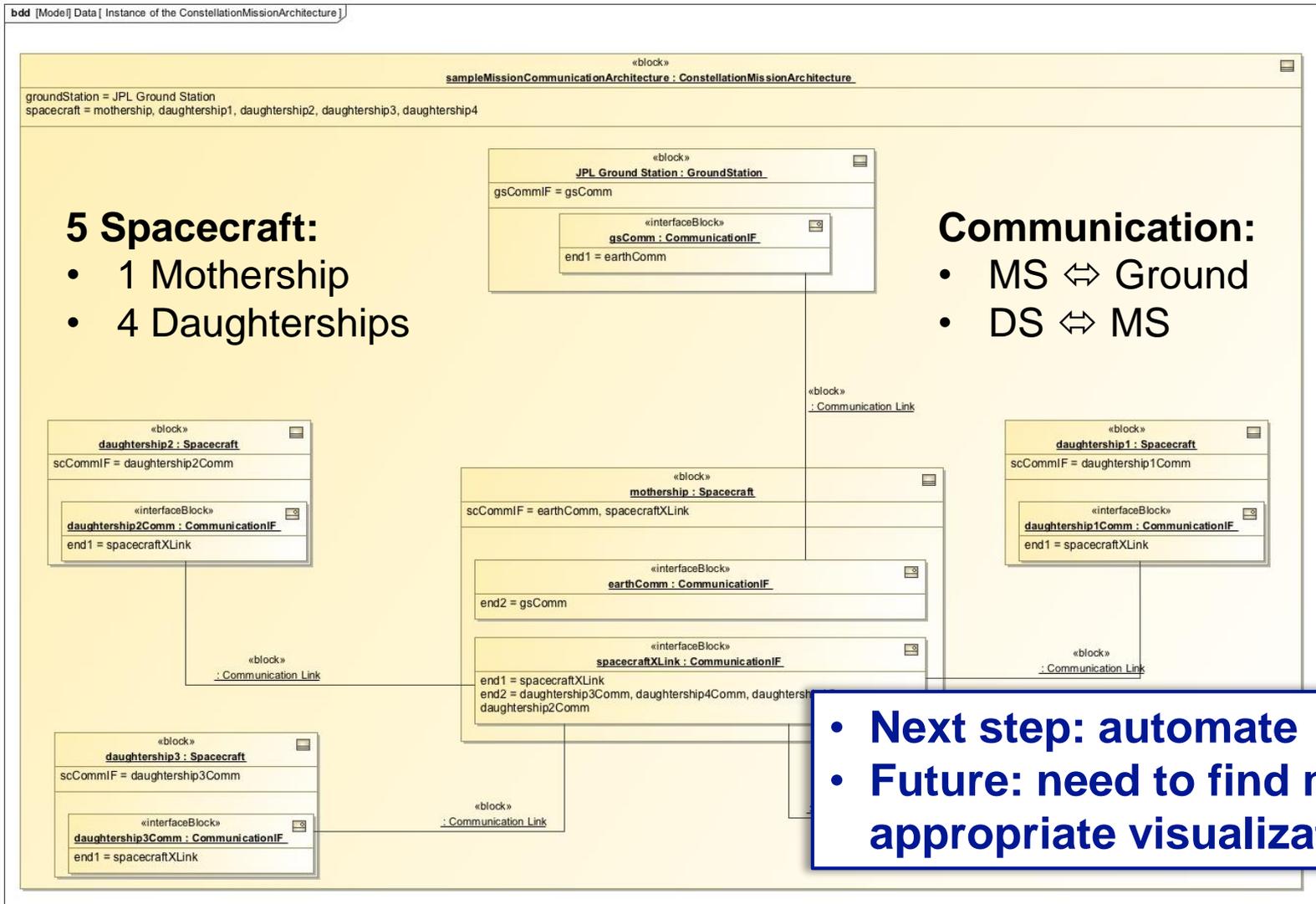
Model Library Development

Capturing Domain Elements



Instance Creation: Example Partial Instance

Optimized Integrated Network Constellation Design



• Next step: automate

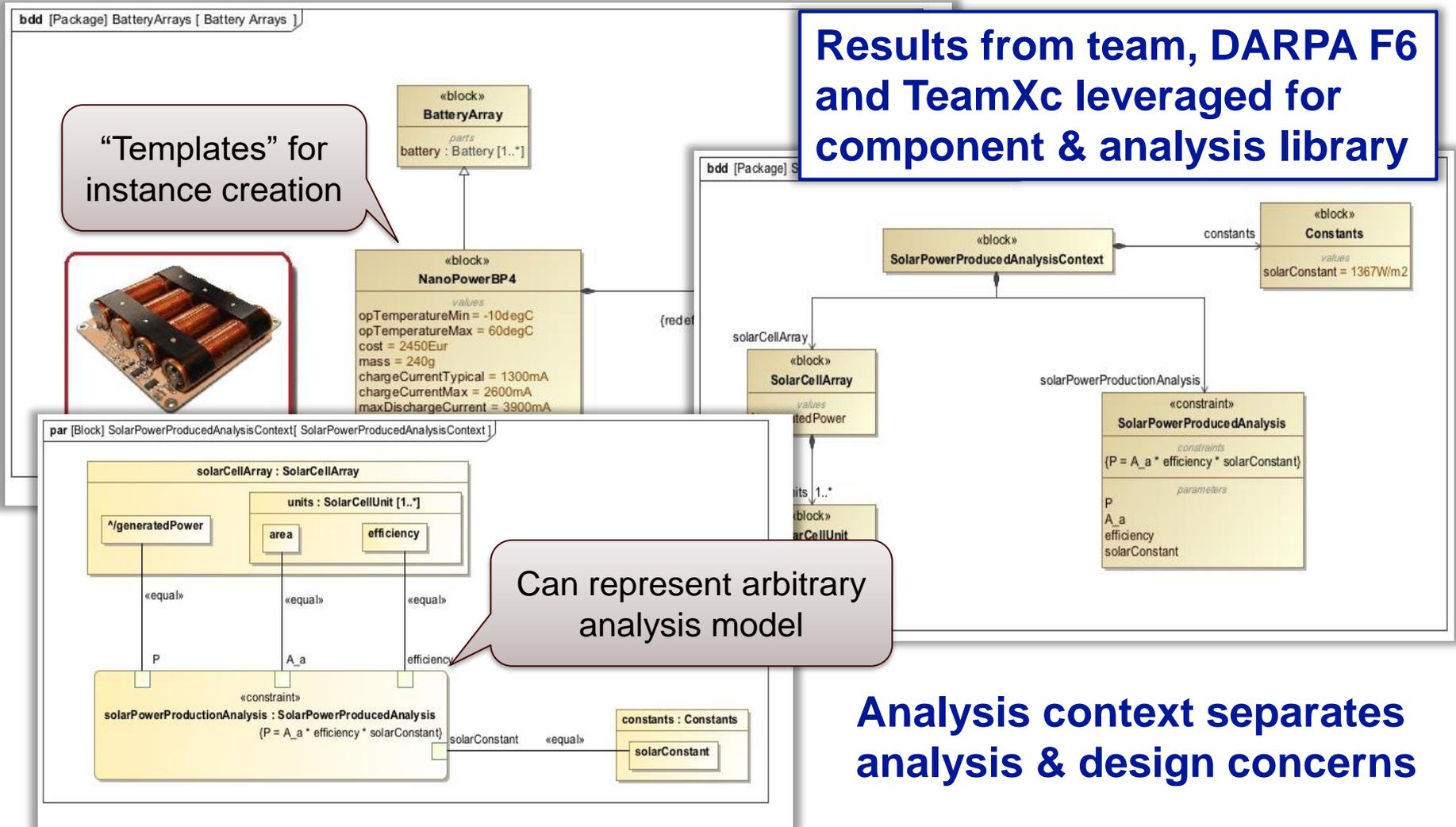
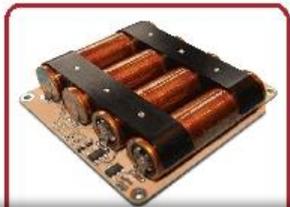
• Future: need to find more appropriate visualization

Model Library Development

Capturing Physical Components & Parametric Relationships

Results from team, DARPA F6 and TeamXc leveraged for component & analysis library

“Templates” for instance creation

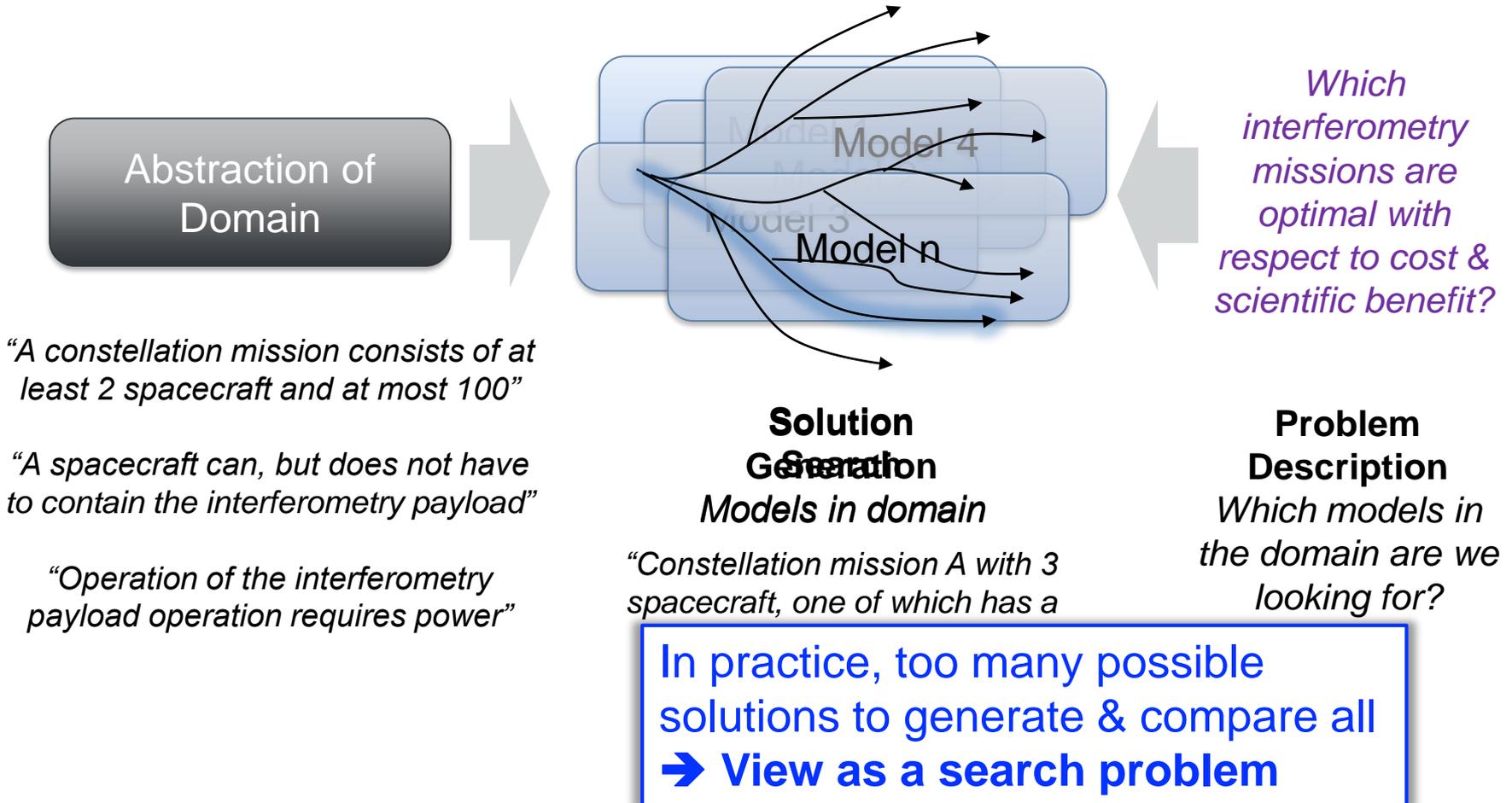


Can represent arbitrary analysis model

Analysis context separates analysis & design concerns

Mission Architecture Trade Space Exploration

Mechanized Exploration

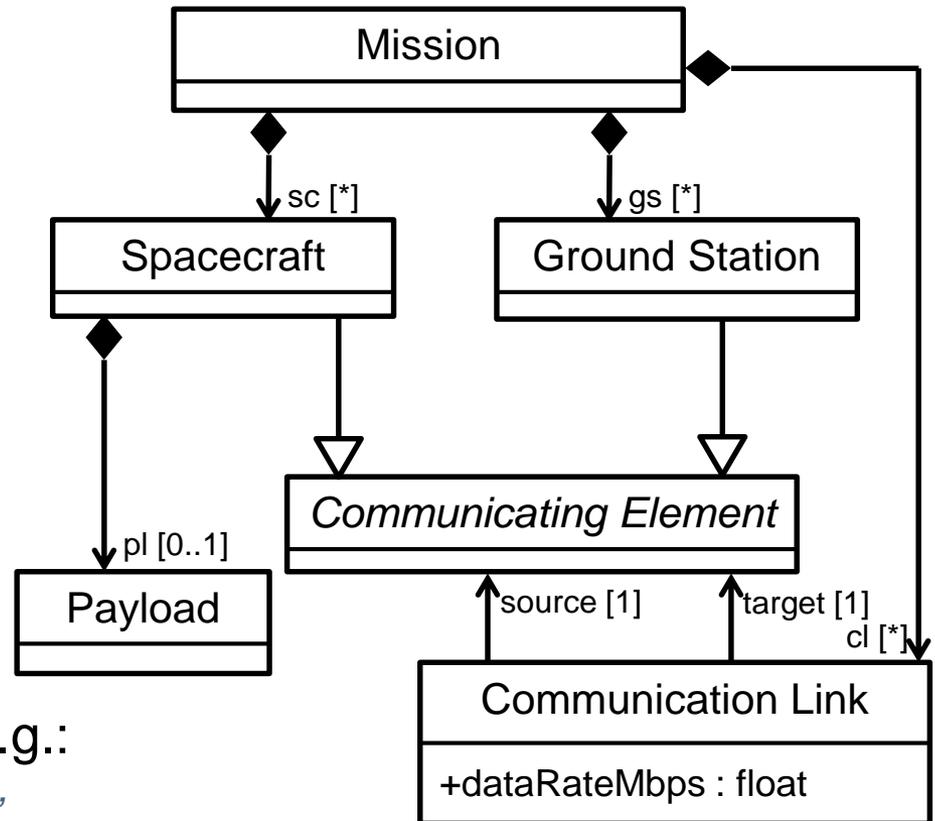


Domain Model & Well-Formedness Constraints

- Domain model
 - Concepts
 - Associations / relations
 - Attributes
 - Describes a **universe of discourse**: many models in domain
 - Describes structural part of the problem
- Typically annotated with addl. well-formedness constraints, e.g.:

“No communication loops may exist”

“All spacecraft must (transitively) be connected to at least one ground station through a communication link”

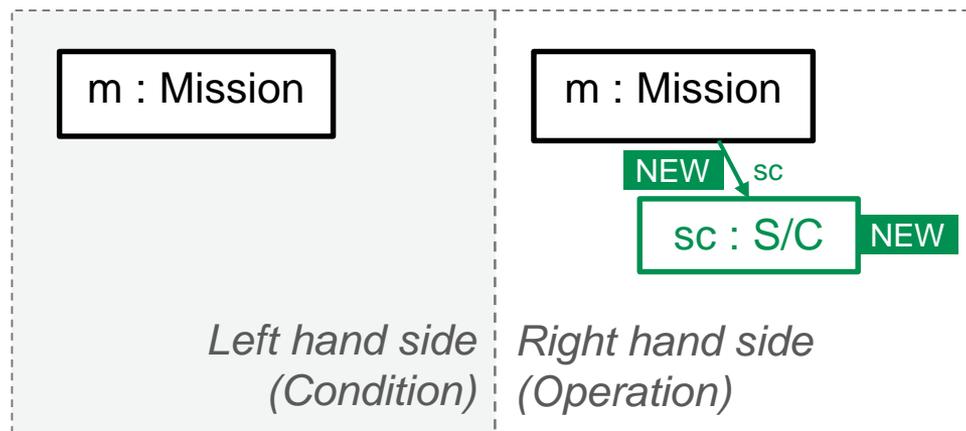


Any model in the domain
is a (structurally) valid
solution

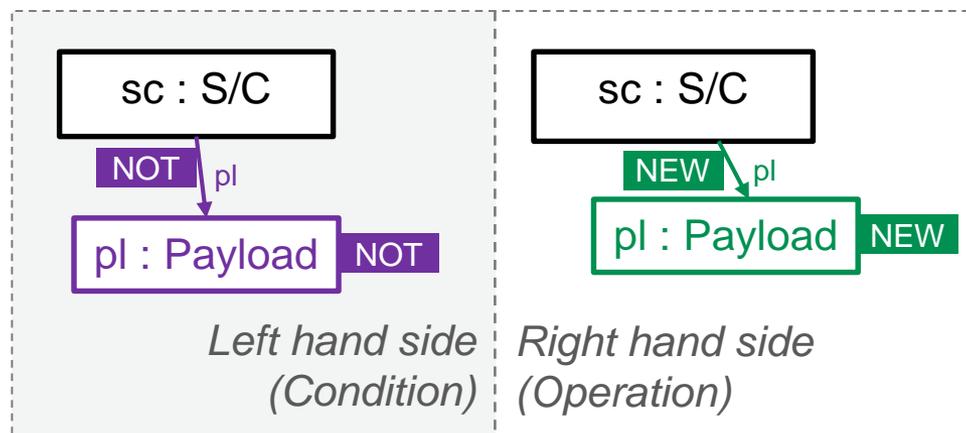
Model-Transformation-Based Exploration

Model Transformation Rules as Enablers for Evolving Solutions

- Transformation Rules
 - **LHS:** **Condition** for match in input model (e.g., “*find an element of type Mission*”)
 - **RHS:** **Operation** to be performed (e.g., “*create a new element of type S/C (Spacecraft) and attach it to the matched mission*”)
- Here: *endogenous* transformations
 - Source and target meta-models are the same
- Used for generating **models in domain** (~design rules)



Rule “createSpacecraft”



Rule “addPayload”

Driving Exploration Towards Optima

Using Evolutionary Algorithms to find Pareto-Optimal Solutions

Crossover

Individual x:
(Selection from population)

Add Spacecraft	Add X-Band Comm	Add Spacecraft	Add Comm Link
----------------	-----------------	----------------	---------------

fitness=0.6
(Obj. Fct. Values)

Individual y:

Add Spacecraft	Add Ka-Band Comm	Add Payload	Add Spacecraft
----------------	------------------	-------------	----------------

fitness=0.5

Here, individuals are **sequences of transformation rule activations**

→ Each genome in population is a variable with set of trafo rules as range

New:

(Recombined individual in next generation)

Add Ka-Band Comm

fitness=0.8

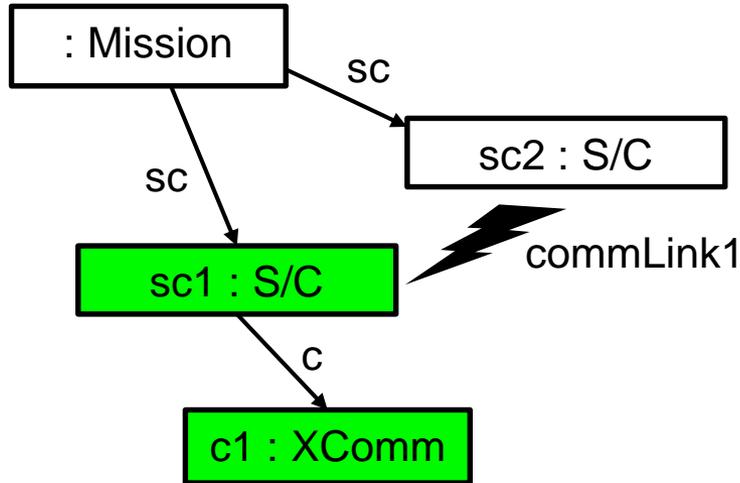
Mutation

Could also be a “placeholder” transformation (= rule “do nothing”)

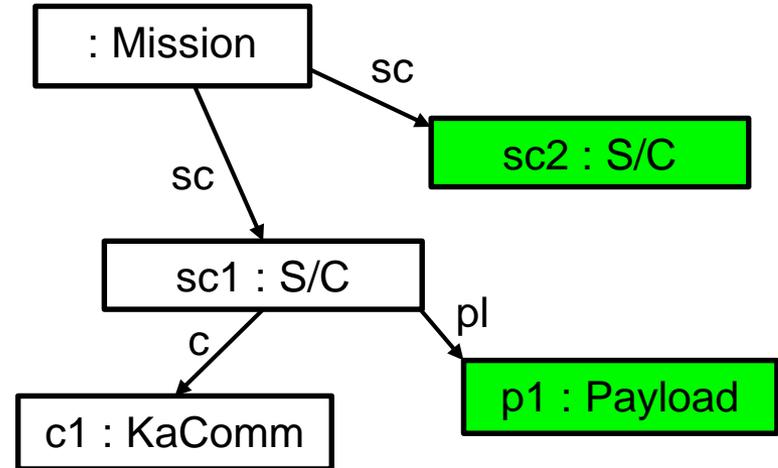
Driving Exploration Towards Optima

Models Resulting from Executing Transformations

Individual x:

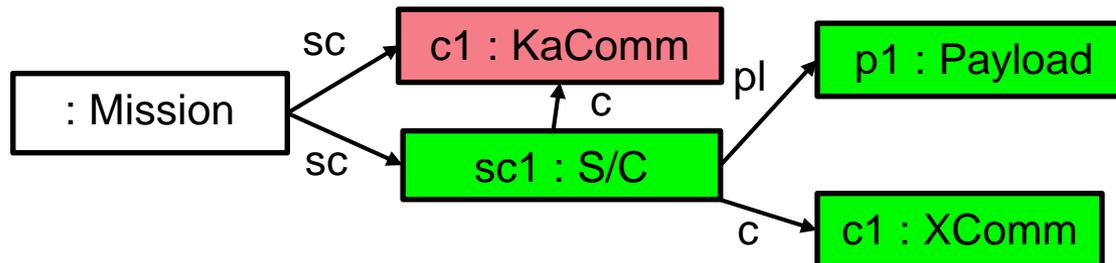


Individual y:



recombined to
Mutation

New:



Implementation

Open Source Technologies Used in Implementation

- Representation of Domain
→ **Ecore / Eclipse EMF + OCL**
- Exploration Rules
→ **Henshin (or Viatra)**
- Analyses / Fitness Functions
→ **Java**
- Optimization Using Genetic Algorithms
→ **MOMoT, MOEA (or Viatra DSE)**

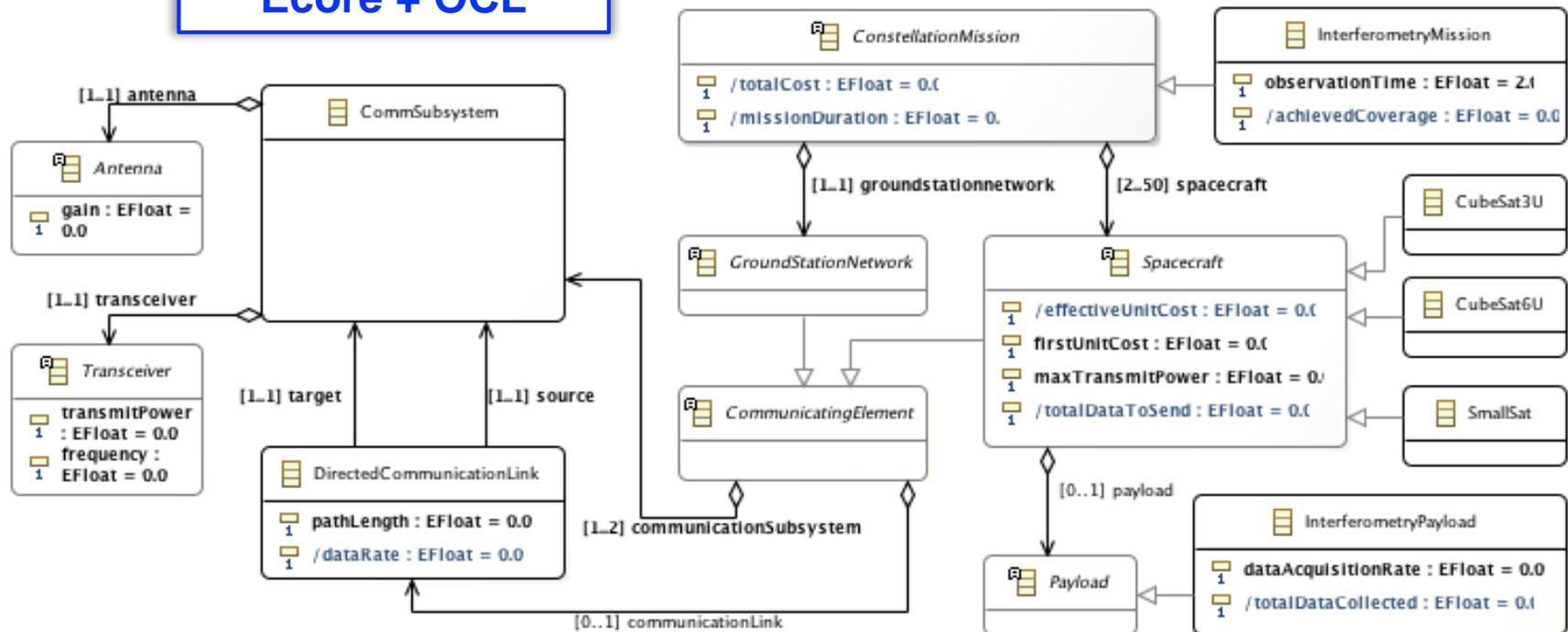


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Application to Case Study

Representation of Domain (Excerpt)

Domain model in
Ecore + OCL



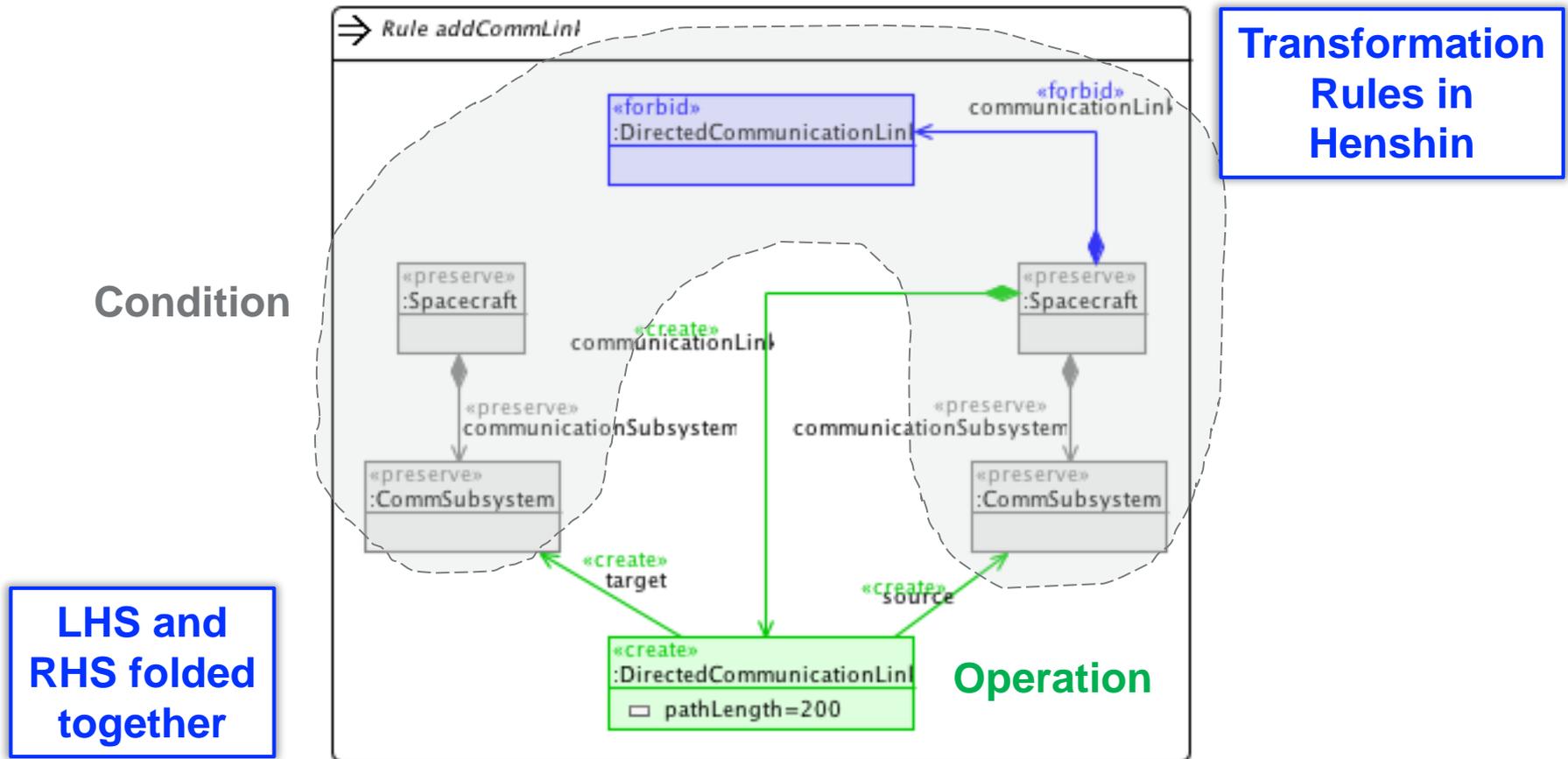
20 concepts, 9 associations, 15 attributes / parameters

> 48¹⁰ possible models

Too many for
exhaustive search

Application to Case Study

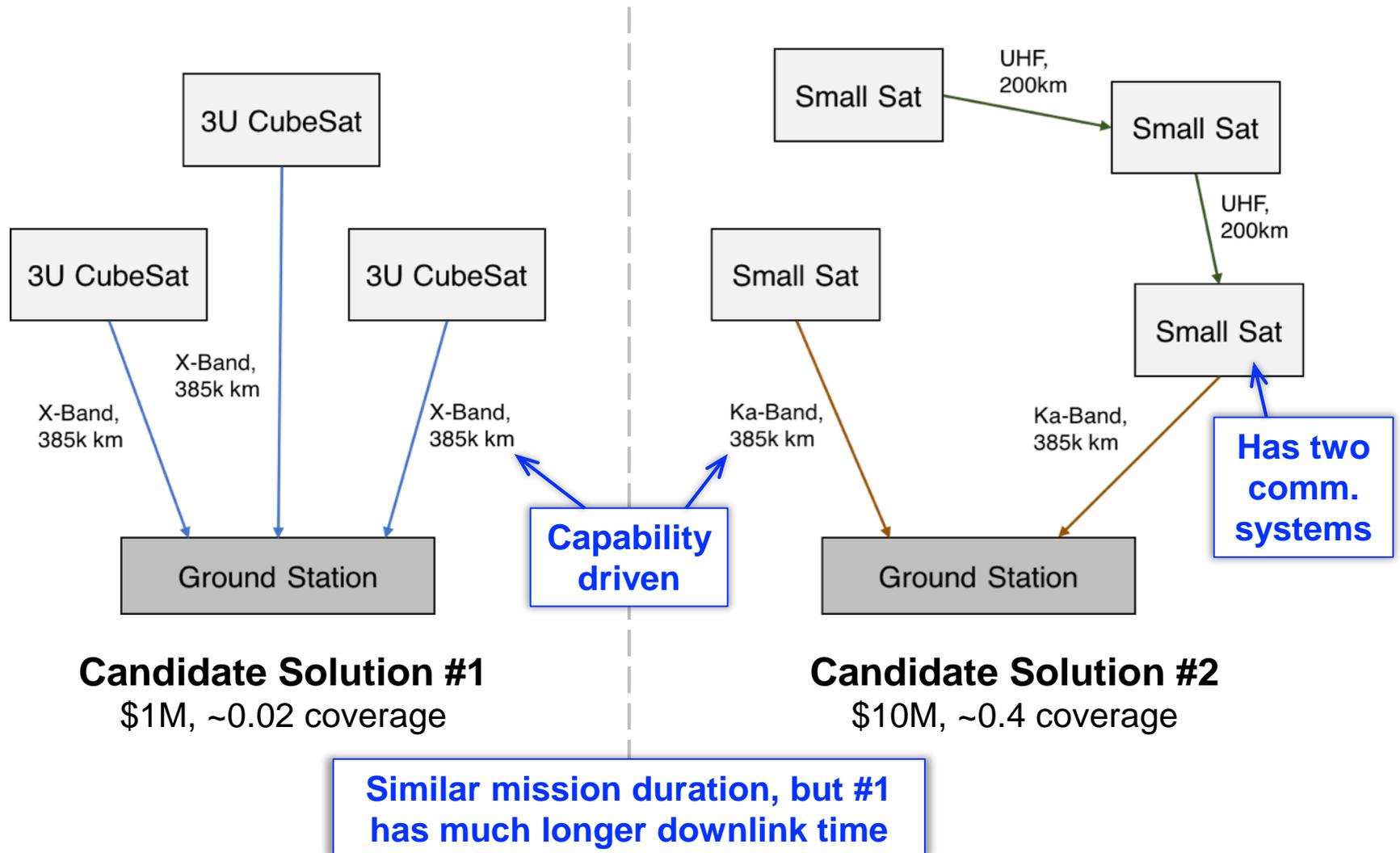
Transformation Rule Example (Henshin Syntax): Add Comm. Link



In Prose: "Find 2 distinct spacecraft instances, and add a communication link between them"

Results from Application to Case Study

Examples of Pareto-Optimal (Nondominated) Solutions



Tool & Language for Integrated Optimization

Development of the "Kigen" Optimization Environment

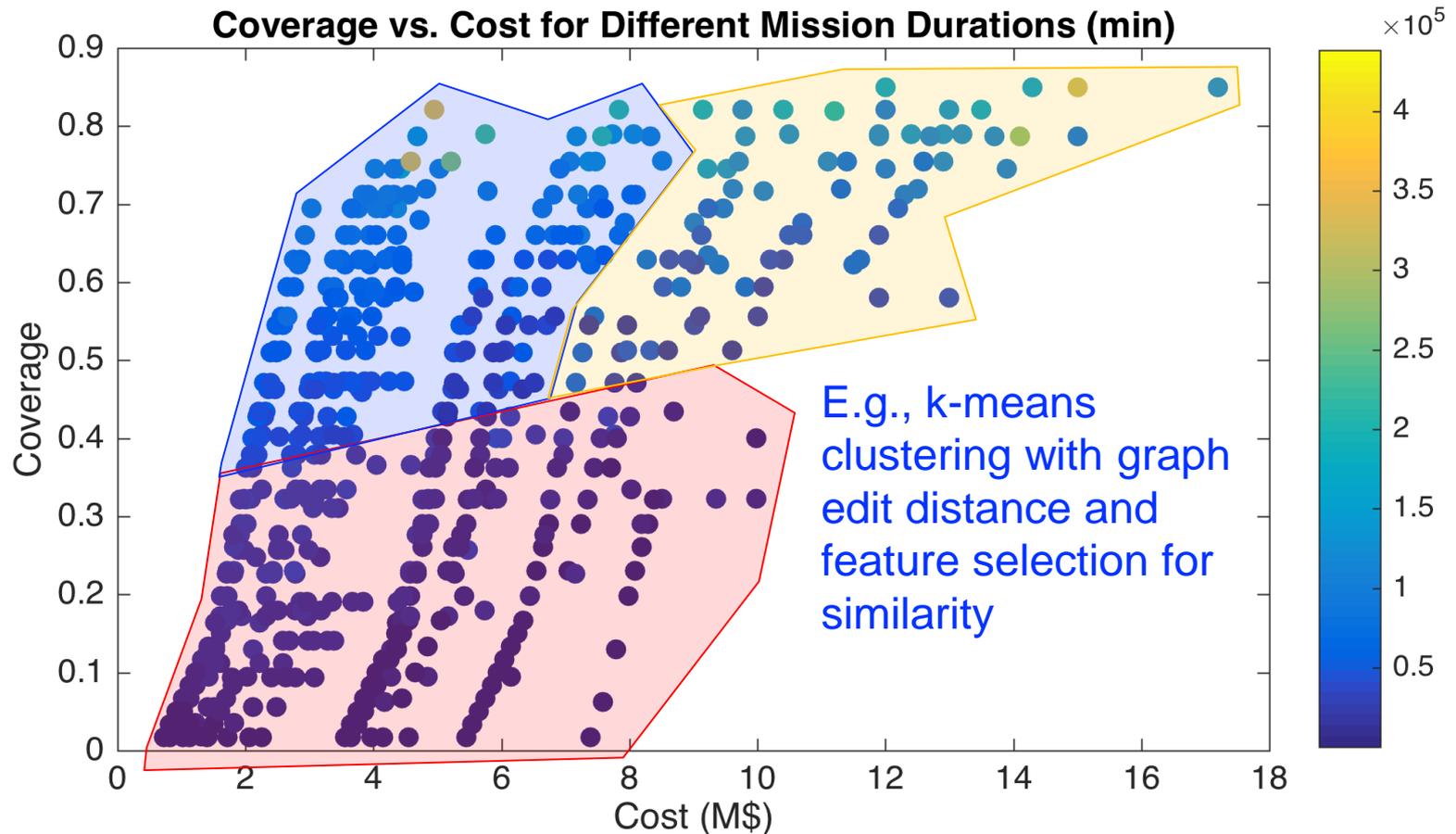
- Able to concisely define an optimization problem
- Similar to LP, QP languages
- Supports:
 - Variables, constants
 - Custom types **← Unique!**
 - Constraints, helper functions
 - Multiple objectives
- Can directly call Java functions
- Able to generate a set of Pareto-optimal mission concepts
- **Since Q1 also in use by 6x R&TD task "MOSAIC" for trade studies**

```
Example.kml
1 type FlightSystem {
2   var spacecraft : Spacecraft [ 1 .. 100 ]
3
4   fun cost = {
7
8     constraint mustBeInBudget: cost < 100 //M$
9   }
10
11 type Spacecraft {
12   var mass = 1 .. 50 //kg
13
14   var payload : Instrument [ 0 .. 2 ]
15 }
16
17 abstract type Instrument
18
19 type Camera extends Instrument
20 type Radar extends Instrument
21
22 minimize FlightSystem.cost
23 maximize FlightSystem.spacecraft.size
24
25 solve optimize using 'nsgaii' 1 times with
26   populationSize 25,
27   maxGenerations 10,
28   maxSolutionLength 20
--
```

Notional Example of a KigenML program

What's Next?

Clustering of Similar Architectures



The cost information contained in this document is of a budgetary and planning nature and is intended for informational purposes only. It does not constitute a commitment on the part of JPL and/or Caltech.



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