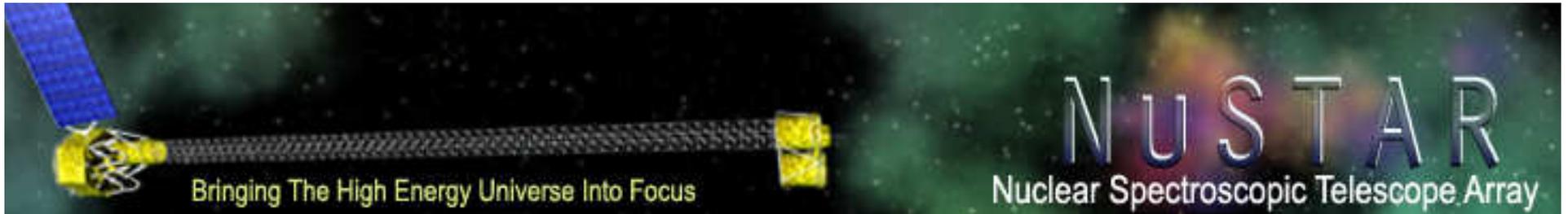


National Aeronautics and Space Administration



The Nuclear Spectroscopic Telescope Array (NuSTAR)

A New View of the High Energy Universe

Yunjin Kim

Jet Propulsion Laboratory

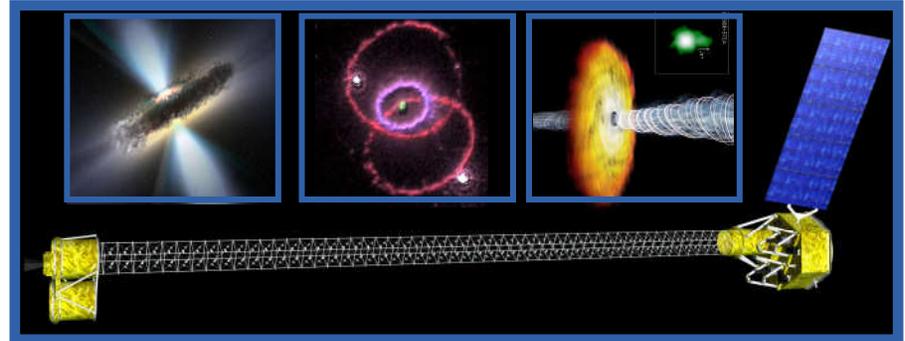
California Institute of Technology

and NuSTAR Team

Project Overview (1)

Salient Features

- PI-led (PI: Fiona Harrison, Caltech) SMEX mission
- *NuSTAR* will carry the first high-energy X-ray focusing telescope
- *NuSTAR* partners include Caltech, JPL, GSFC, Orbital, ATK, UCB, DNSC and Columbia University
- JPL managed project
- Launch readiness date: August 15, 2011
- Science operations: 2 years



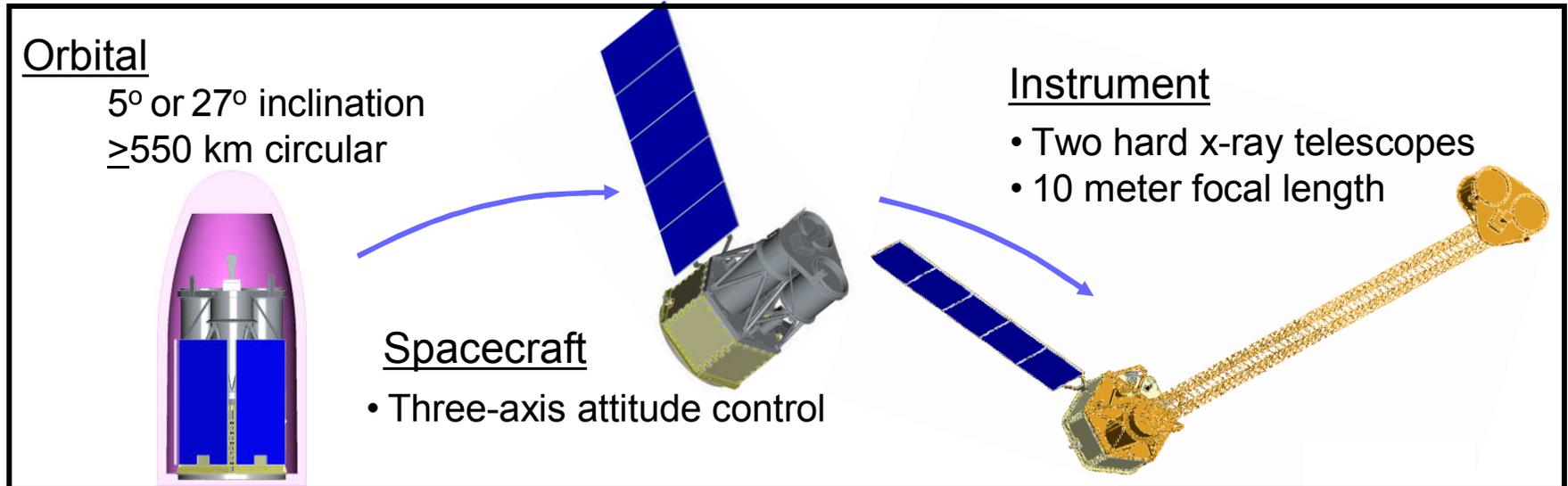
Partners:

Caltech, JPL, Orbital Sciences Corporation, GSFC, ATK-Goleta, UC Berkeley, Danish Technical University, Columbia University

Science

- ***NuSTAR will open a new window on the Universe by making maps of the high-energy X-ray sky (6 keV to 79 keV) that are more than 100 times deeper than from any previous mission***
- ***Objective 1: Determine how massive black holes are distributed through the cosmos, and how they influence the formation of galaxies like our own***
- ***Objective 2: Understand how stars explode and forge the elements that compose the Earth***
- ***Objective 3: Determine what powers the most extreme active black holes***

Mission Overview (2)



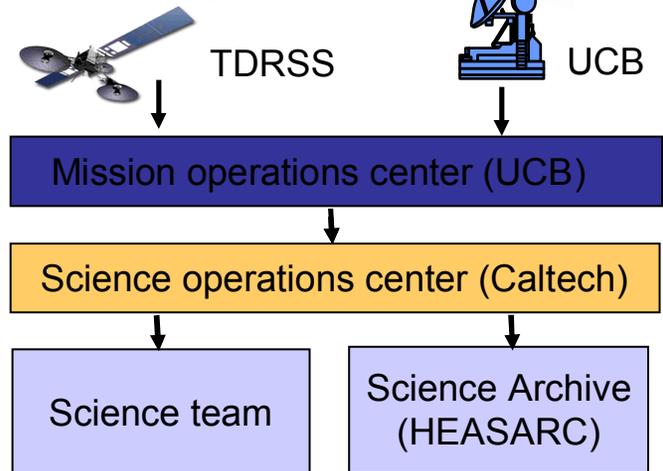
Mission Profile:

Long pointed observations of survey fields, specific science targets, ToOs.

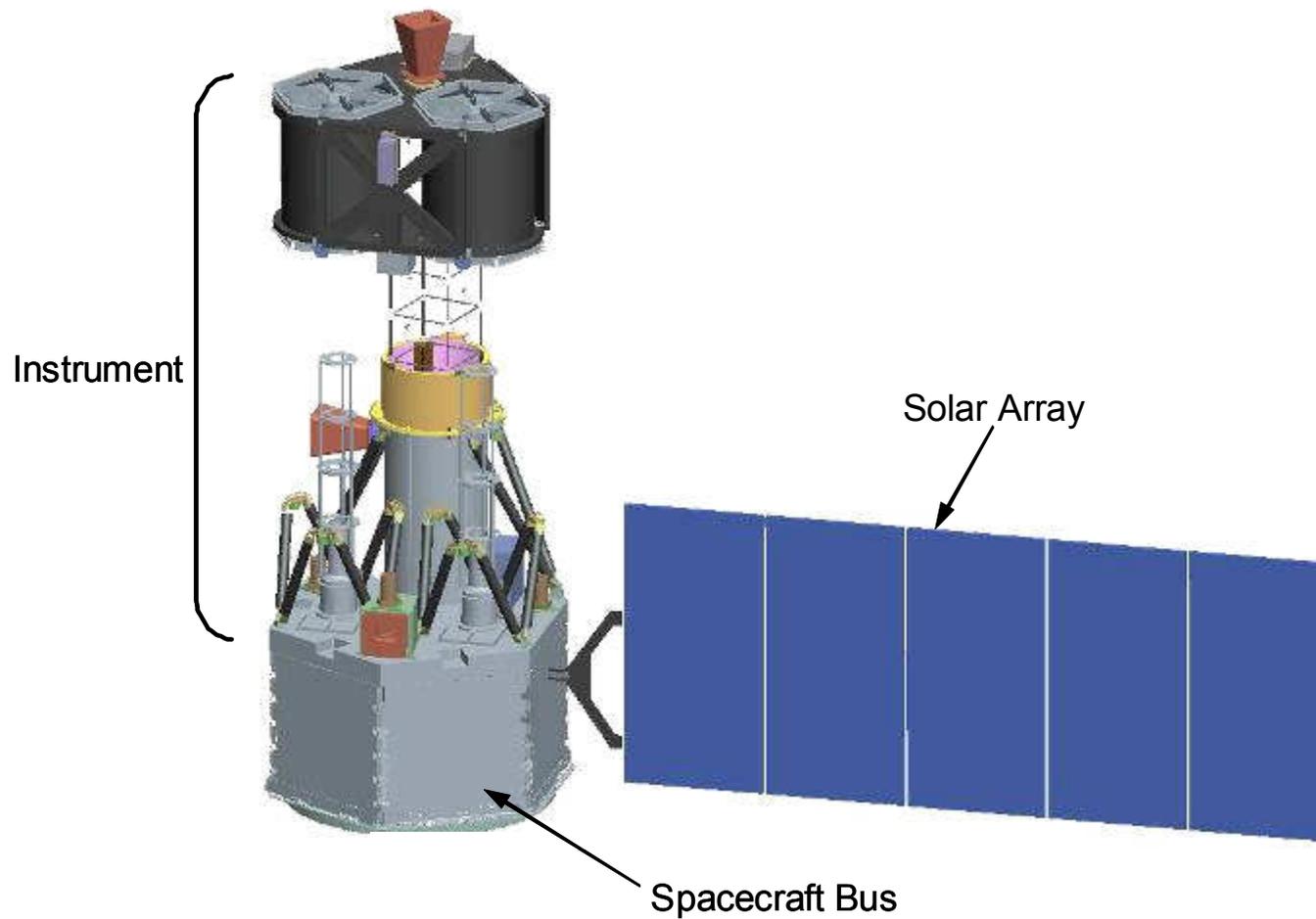
Daily data downlinks, infrequent uplinks.



Mission Operations



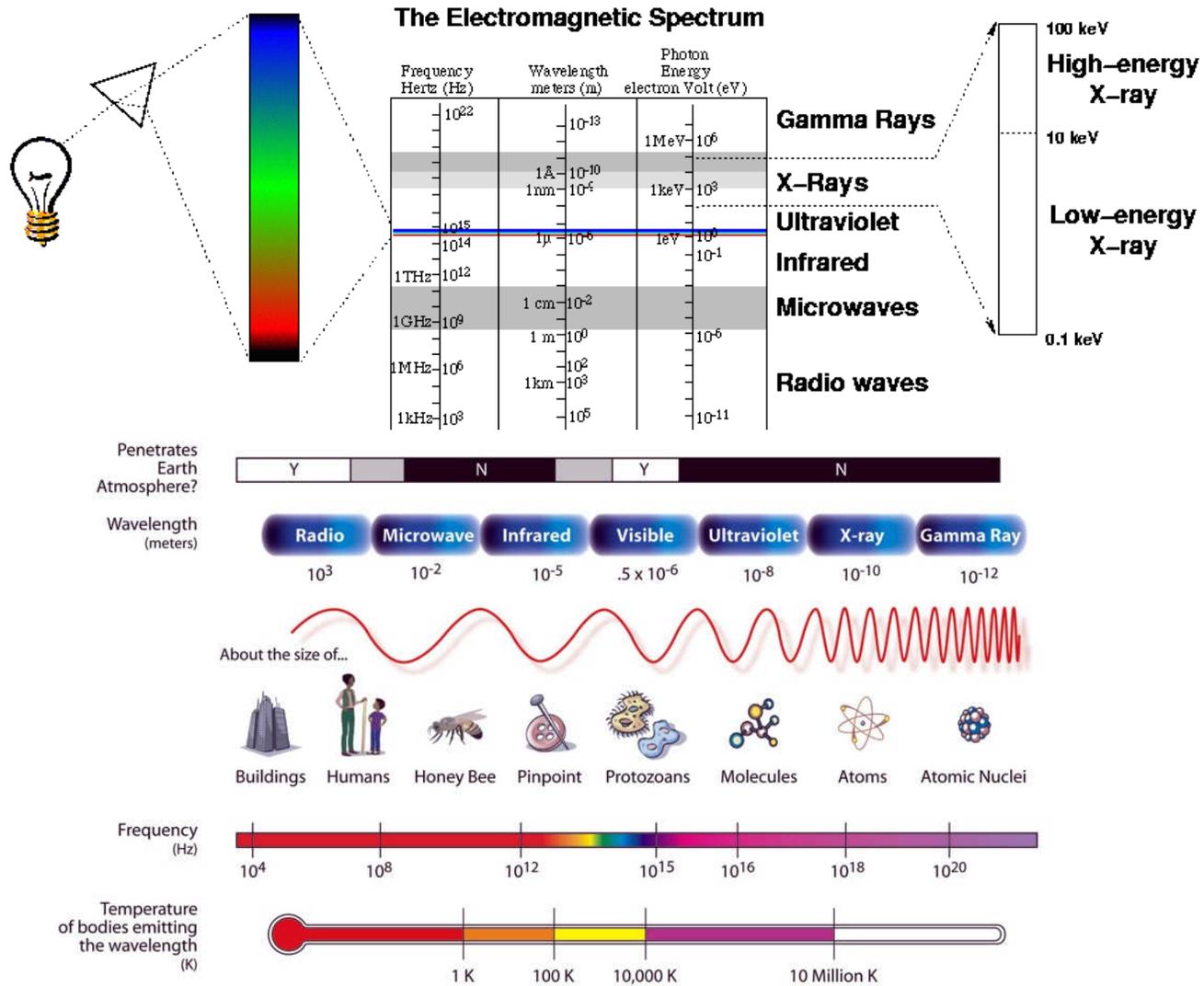
NuSTAR Observatory



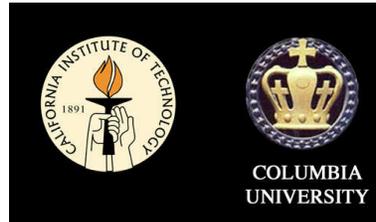
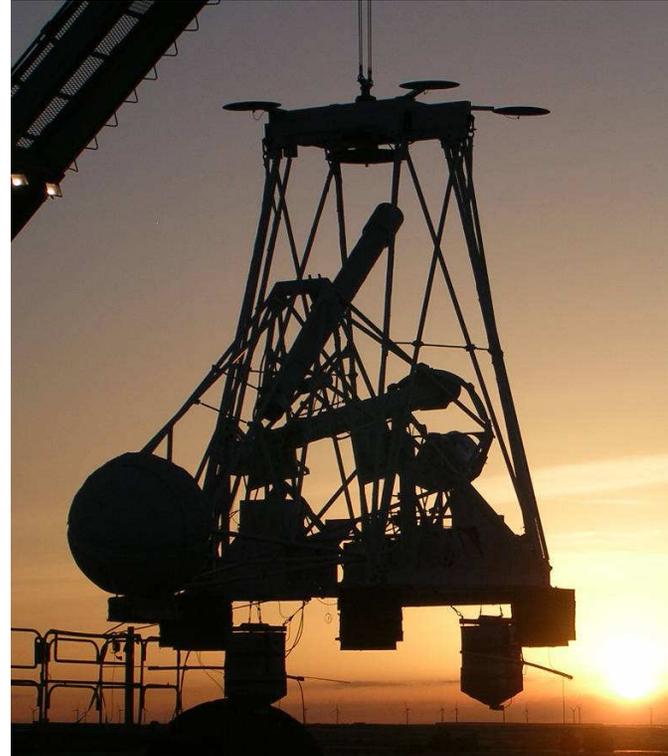
September 10, 2008

AIAA Space 2008

X-ray Observation



The High Energy Focusing Telescope (HEFT)

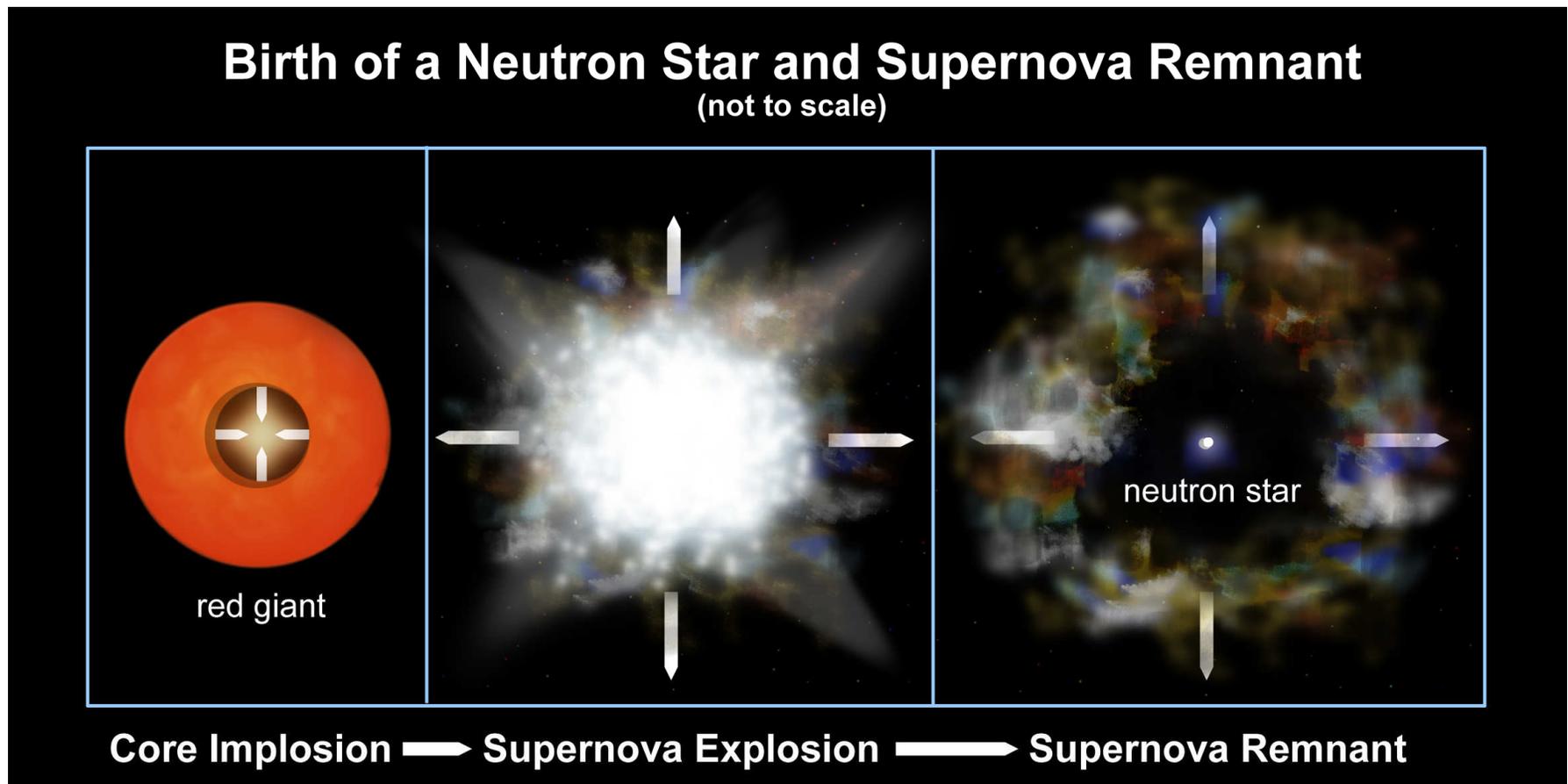


September 10, 2008

AIAA Space 2008

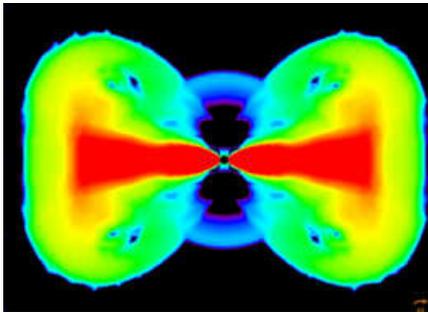
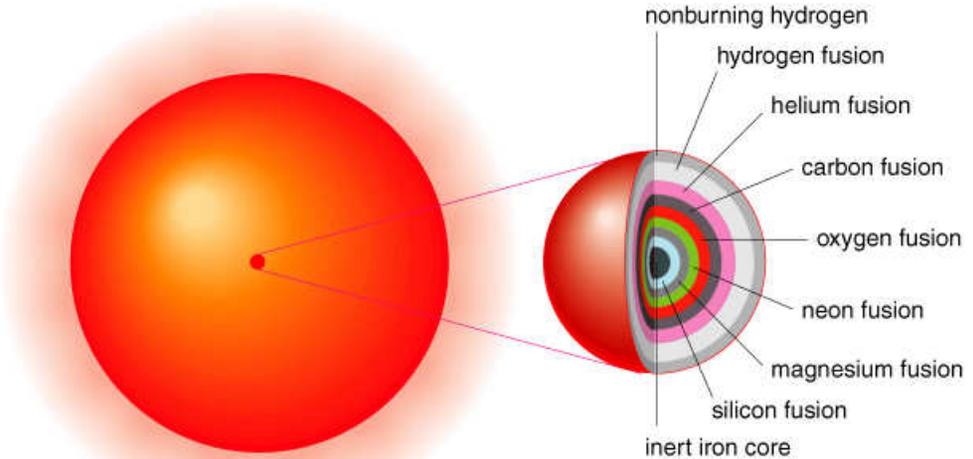
6

Birth of a Neutron Star & Supernova Remnant



To map young remnants to understand how stars explode and distribute the elements

NuSTAR will be a new tool for looking inside of a supernova; probing down to the very core.



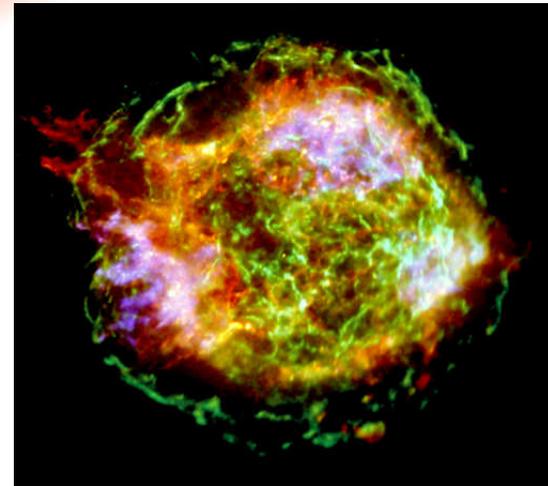
The explosion
(theory)

© Addison-Wesley Longman

?

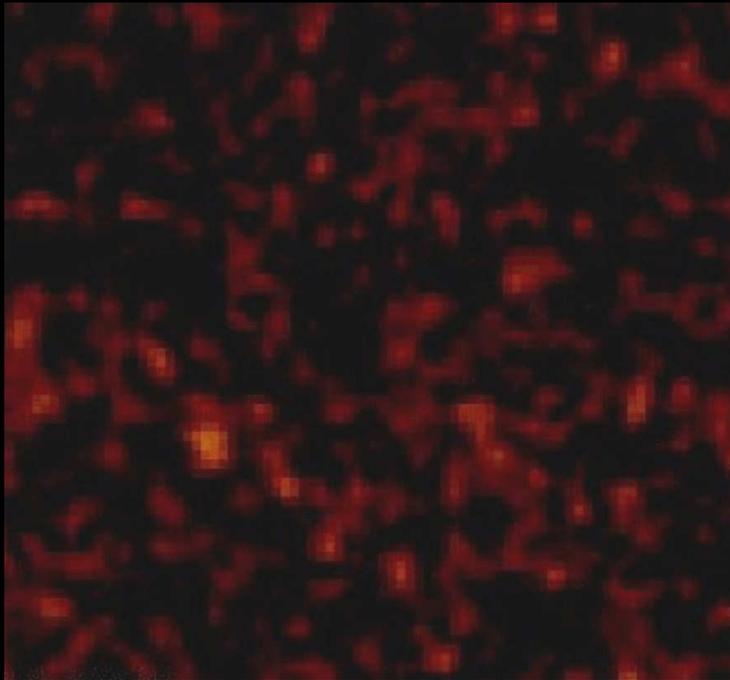


The remnant



Steve Boggs, UC Berkeley, for the *NuSTAR* Science Team

NuSTAR's map of the hard X-ray sky

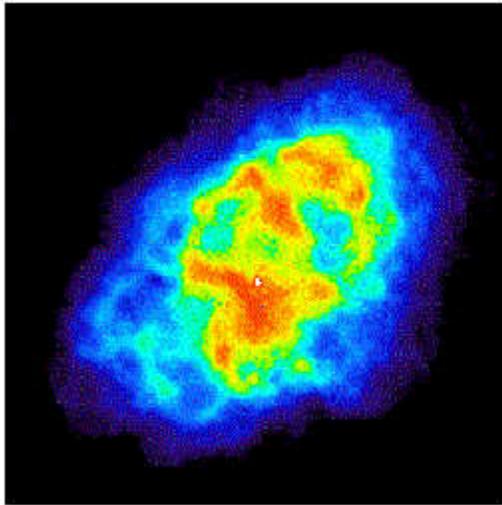


INTEGRAL/IBIS - 2 x 2
degrees.



NuSTAR simulated image -
2 x 2 degrees.

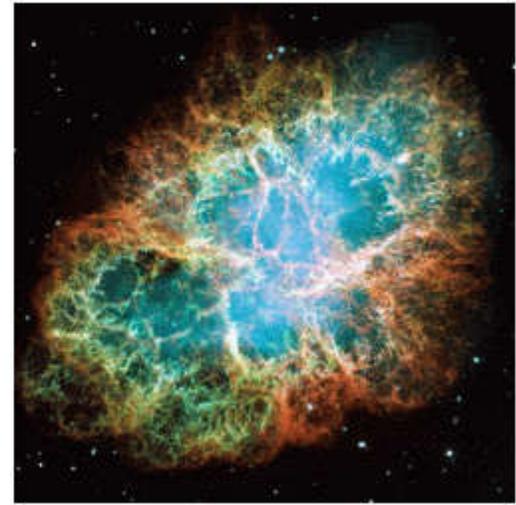
Crab Nebula: Remnant of an Exploded Star (Supernova)



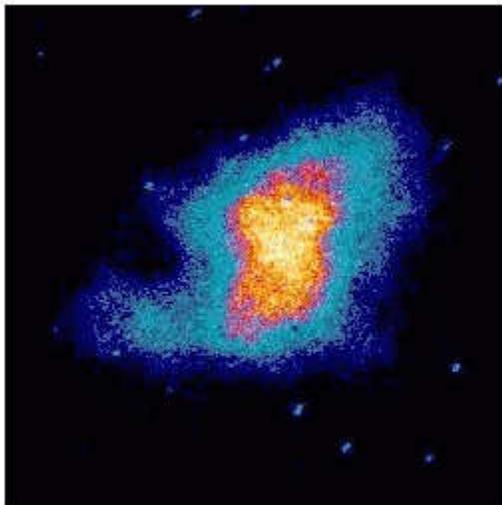
Radio wave (VLA)



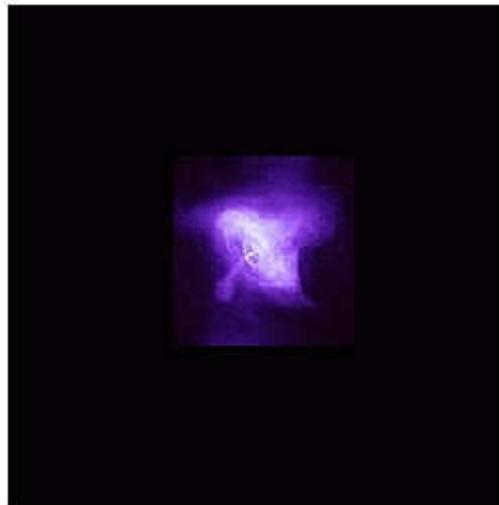
Infrared radiation (Spitzer)



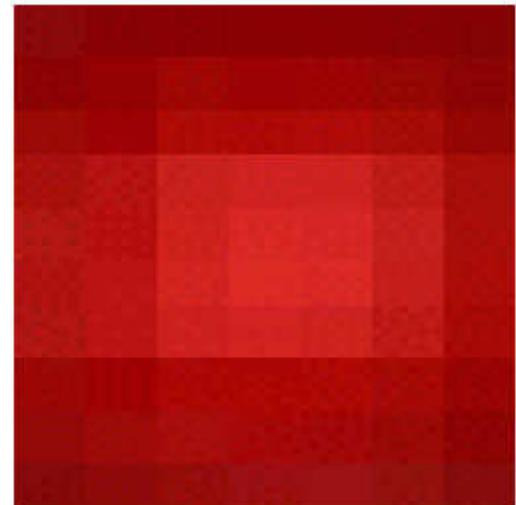
Visible light (Hubble)



Ultraviolet radiation (Astro-1)

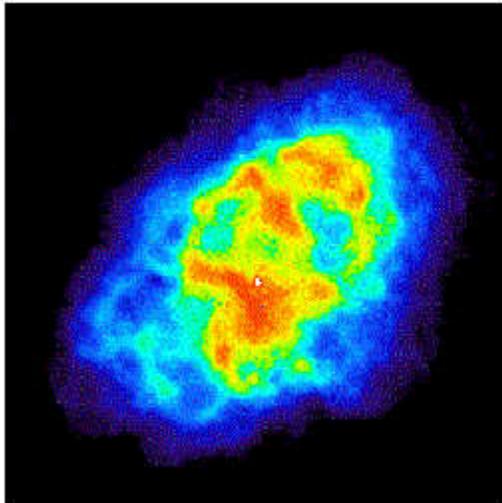


Low-energy X-ray (Chandra)



High-energy X-ray (Integral)

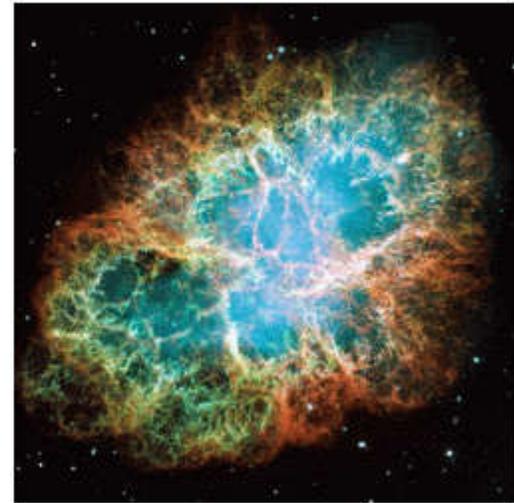
Crab Nebula: Remnant of an Exploded Star (Supernova)



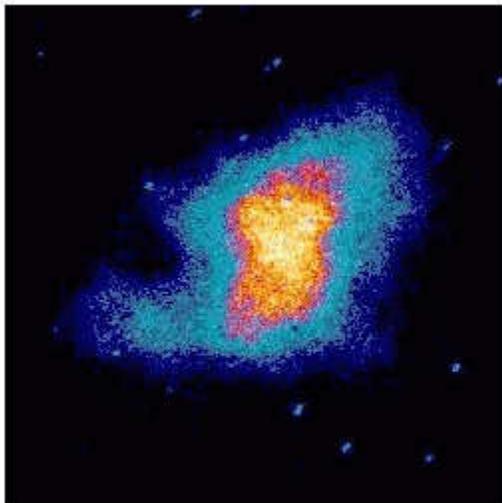
Radio wave (VLA)



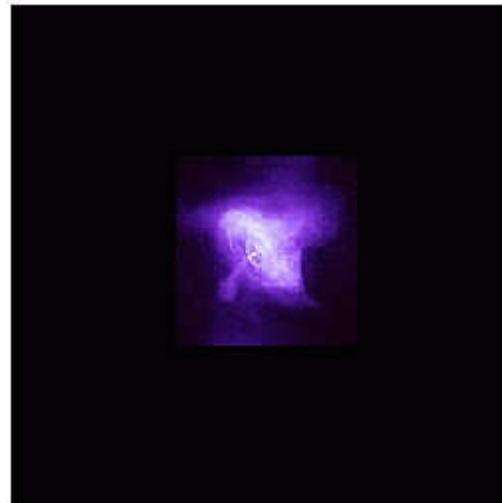
Infrared radiation (Spitzer)



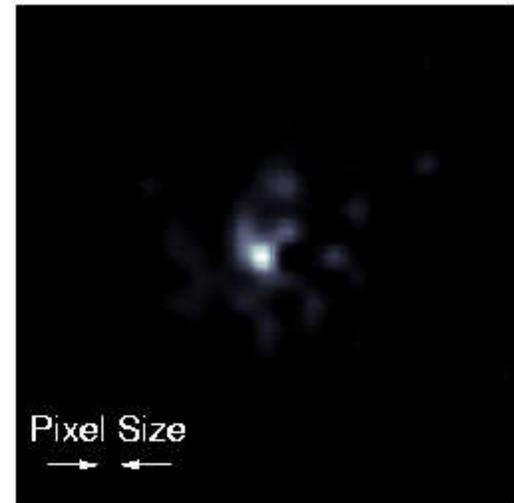
Visible light (Hubble)



Ultraviolet radiation (Astro-1)



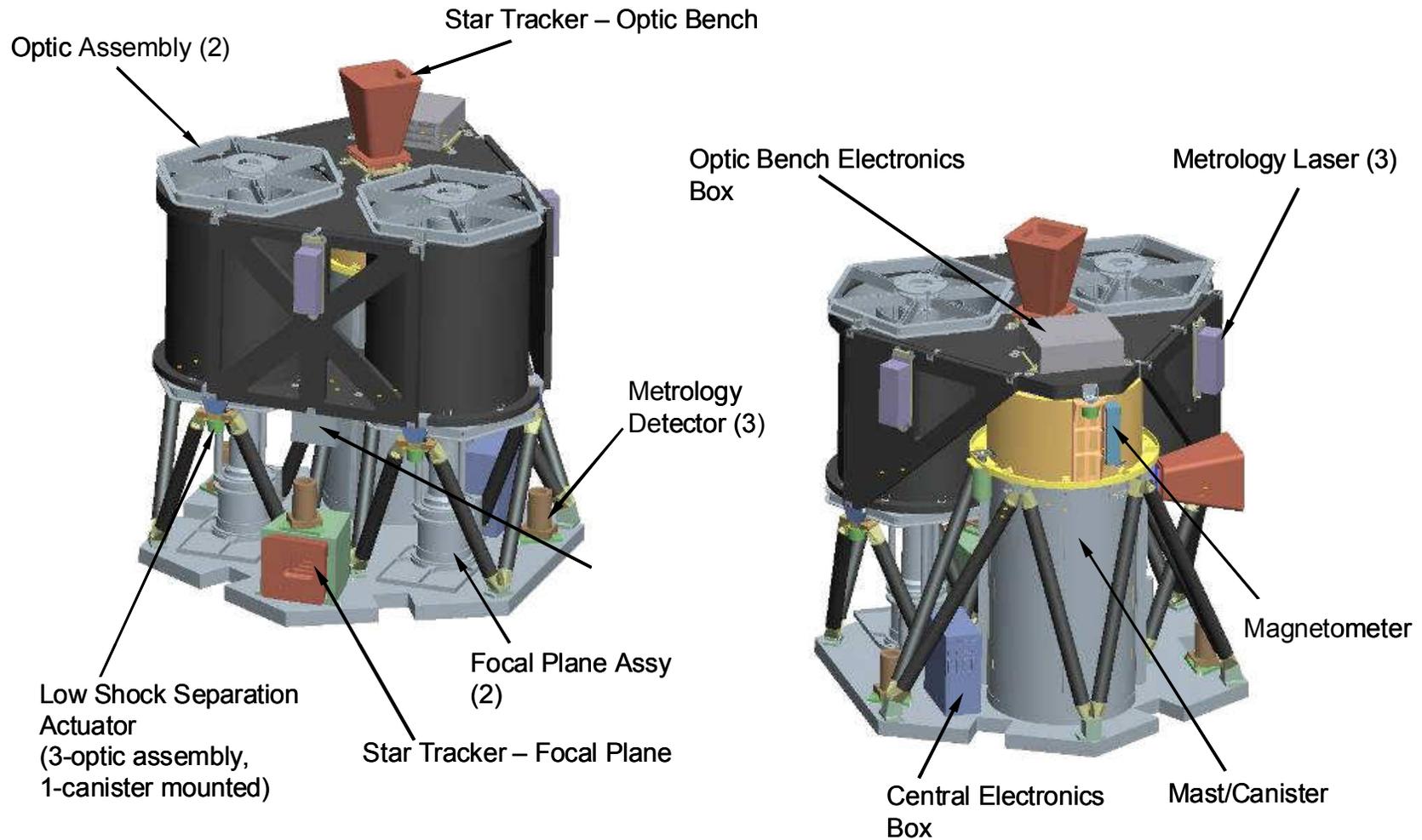
Low-energy X-ray (Chandra)



High-energy X-ray (HEFT)

*** 15 min exposure ***

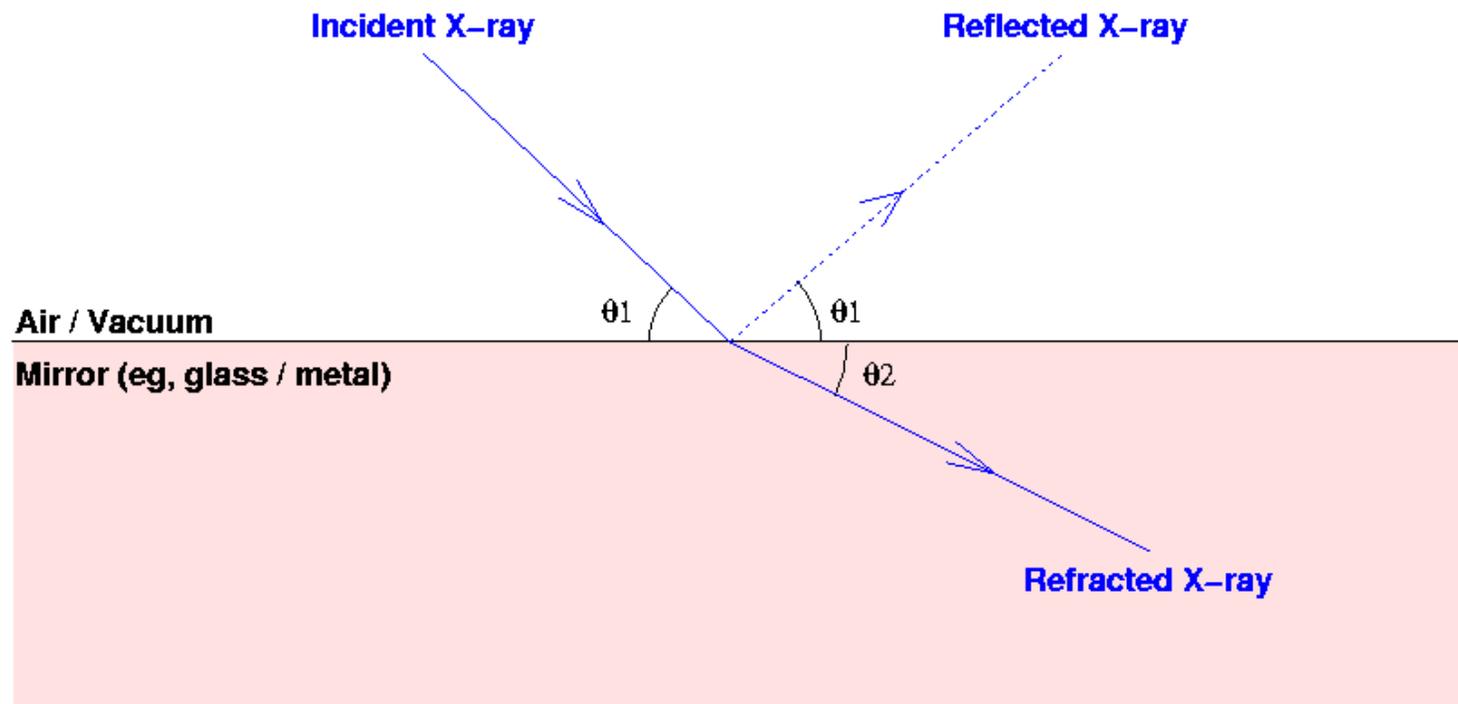
Instrument



Optics: X-ray Reflection

X-rays bend slightly toward the surface (opposite from visible light)

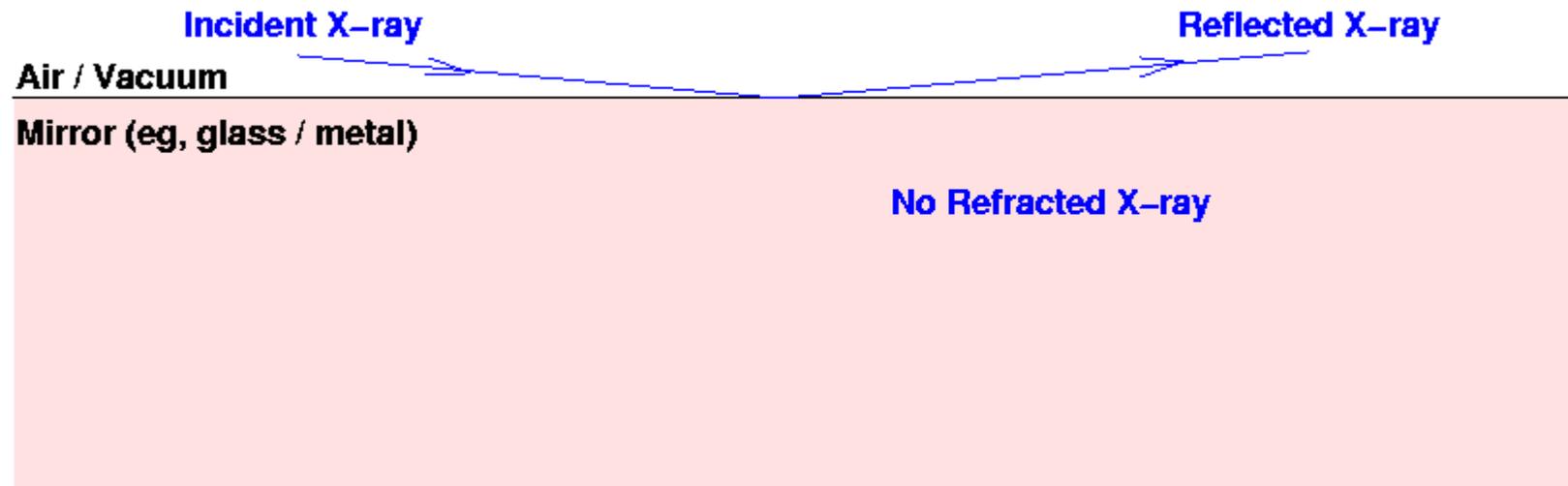
$$n = 1 - \delta - i\beta \quad \delta \sim 10^{-3}$$



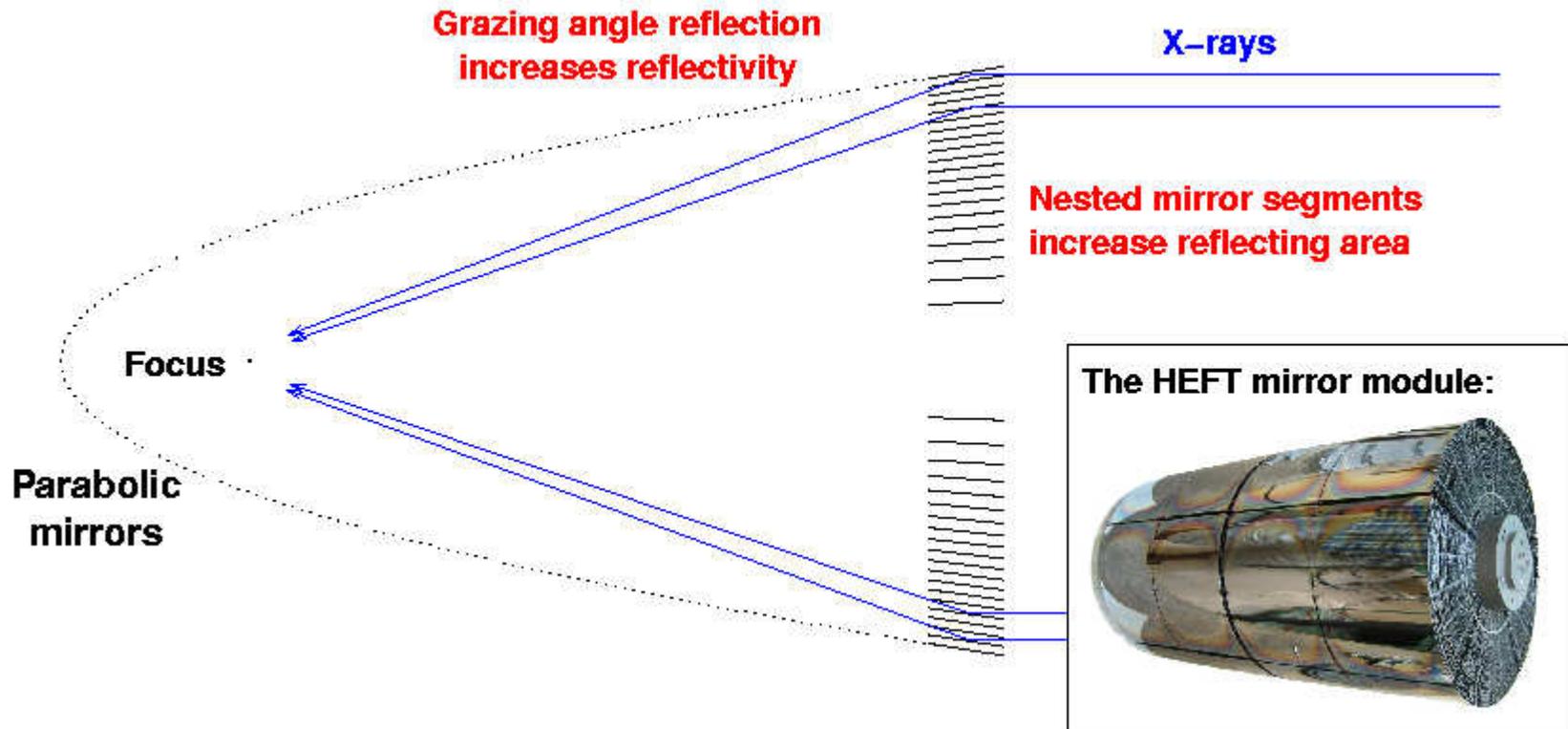
Optics: X-ray Total External Reflection

Critical condition for total external reflection

Critical angles are very small (< 1 degree)

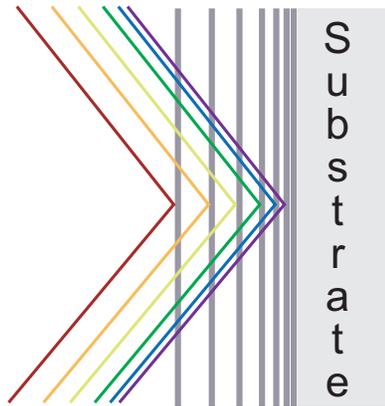


Optics: Nested Mirror Segments

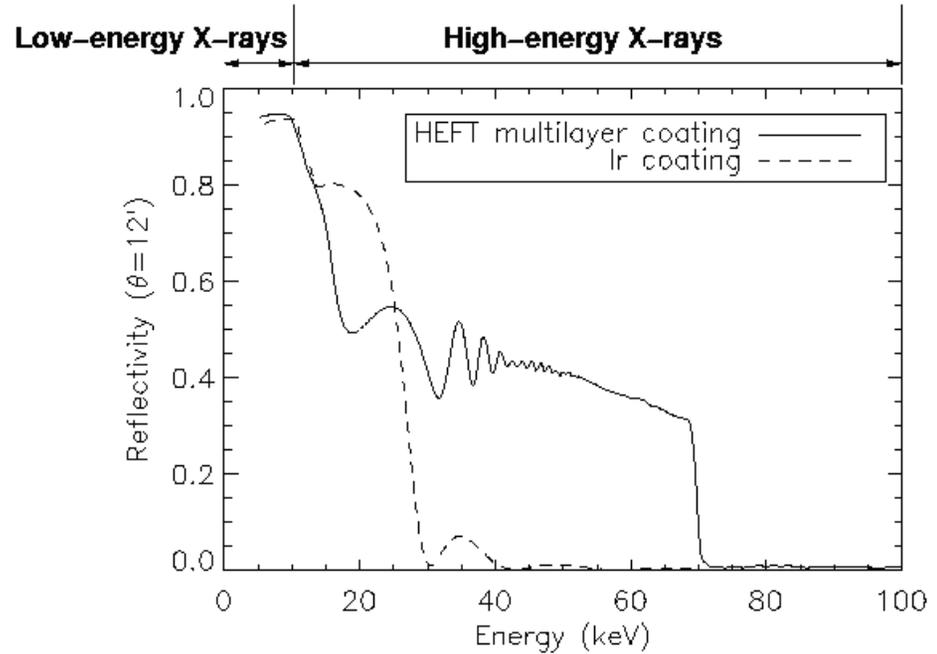


Need two conic sections to eliminate aberrations

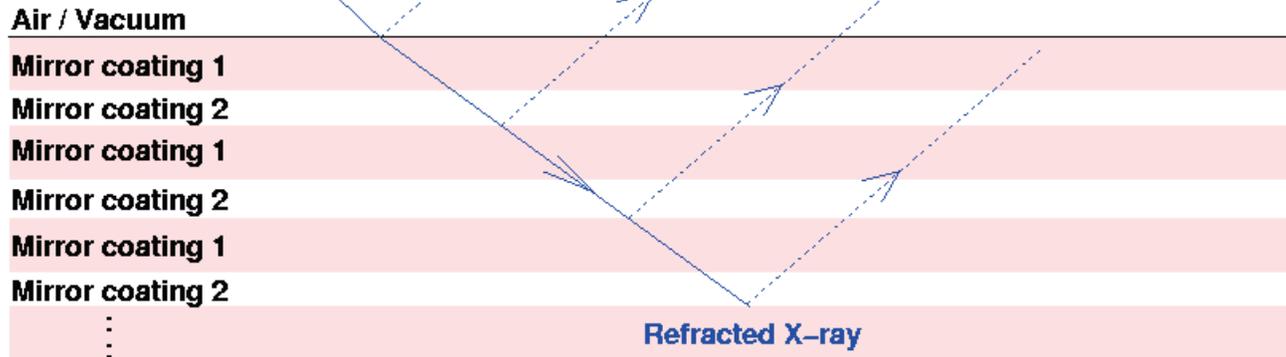
Optics: Multilayer Coating



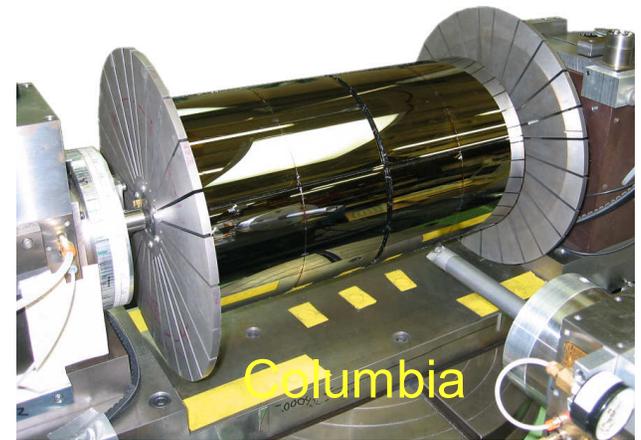
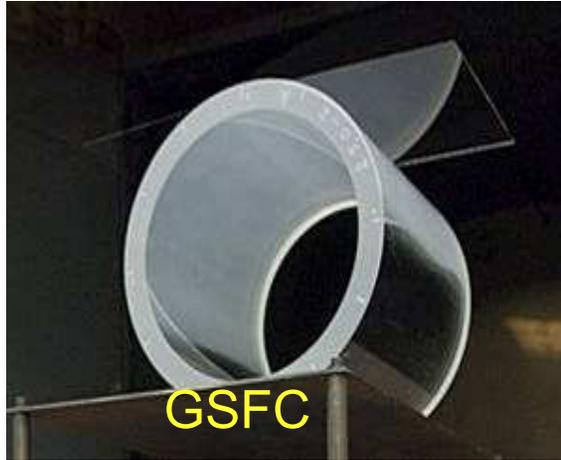
Incident X-ray



Constructive interference



Optics Construction



Combine small angles with multiayer coatings
many nested shells (72 HEFT, 130 NuSTAR)
coatings with up to 700 layer pairs
thinnest pairs: 25 angstroms!



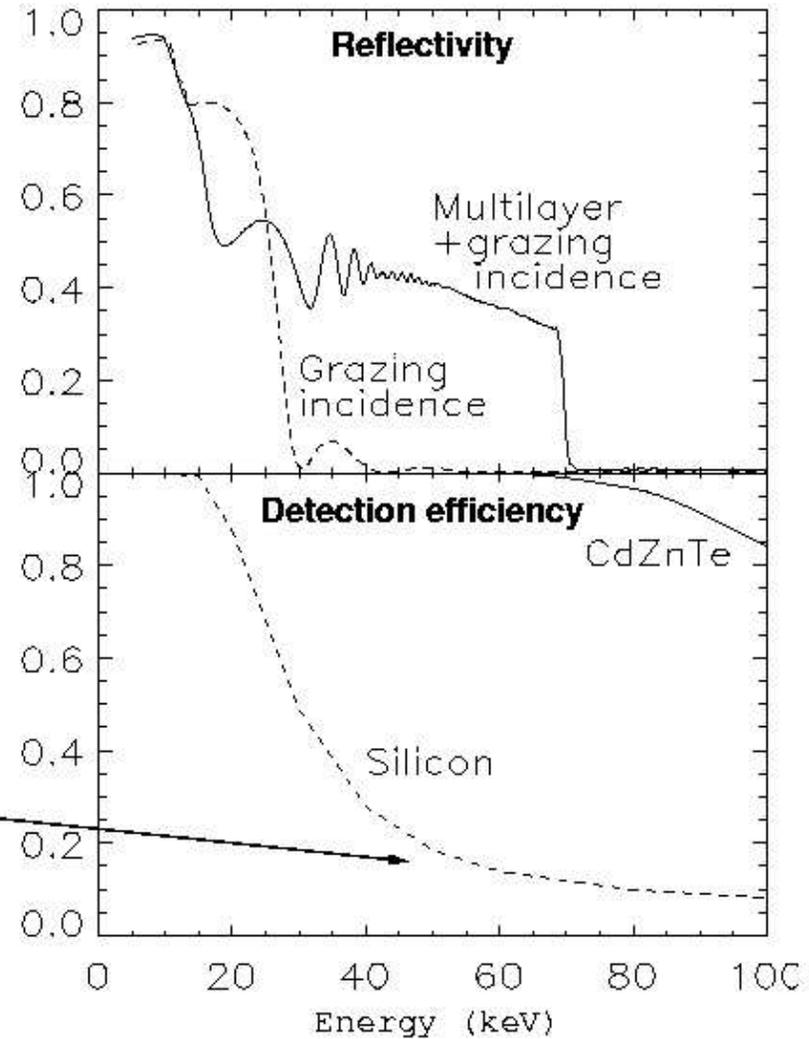
Detector: CdZnTe



Doctors' approach: photographic film
Efficiency: 2%

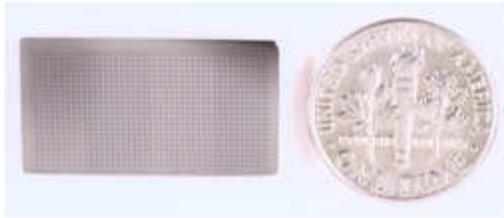


Low-energy X-ray astronomy: silicon CCDs
Efficiency for high-energy X-rays: 10%

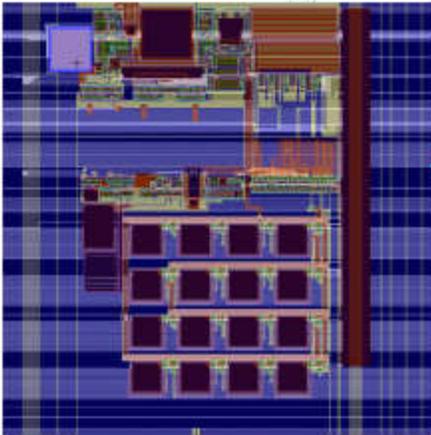


Conventional detectors are too inefficient.

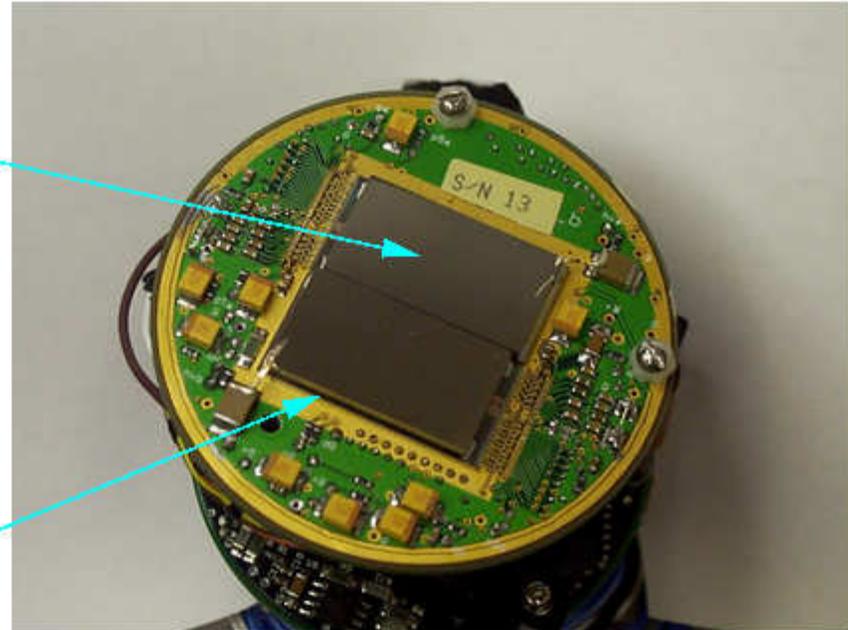
Focal Plane Detector



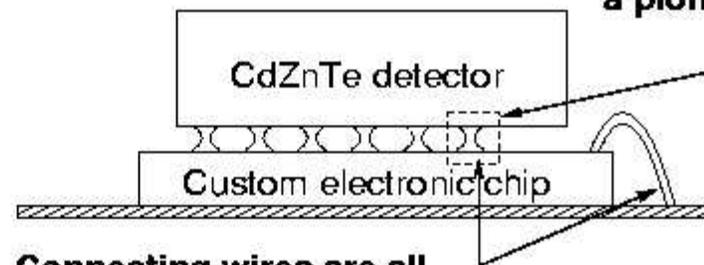
Cadmium zinc telluride (CdZnTe):
High atomic numbers ($Z=48,30,52$)
=> High electron binding energy
=> Over 99% efficiency in trapping high-energy X-rays.



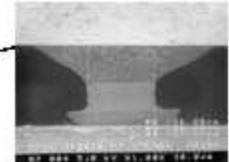
Underneath: Custom electronics for X-ray detection and amplification; built as integrated circuit (ASIC); compact, low-power, low-noise.



Side view schematics:



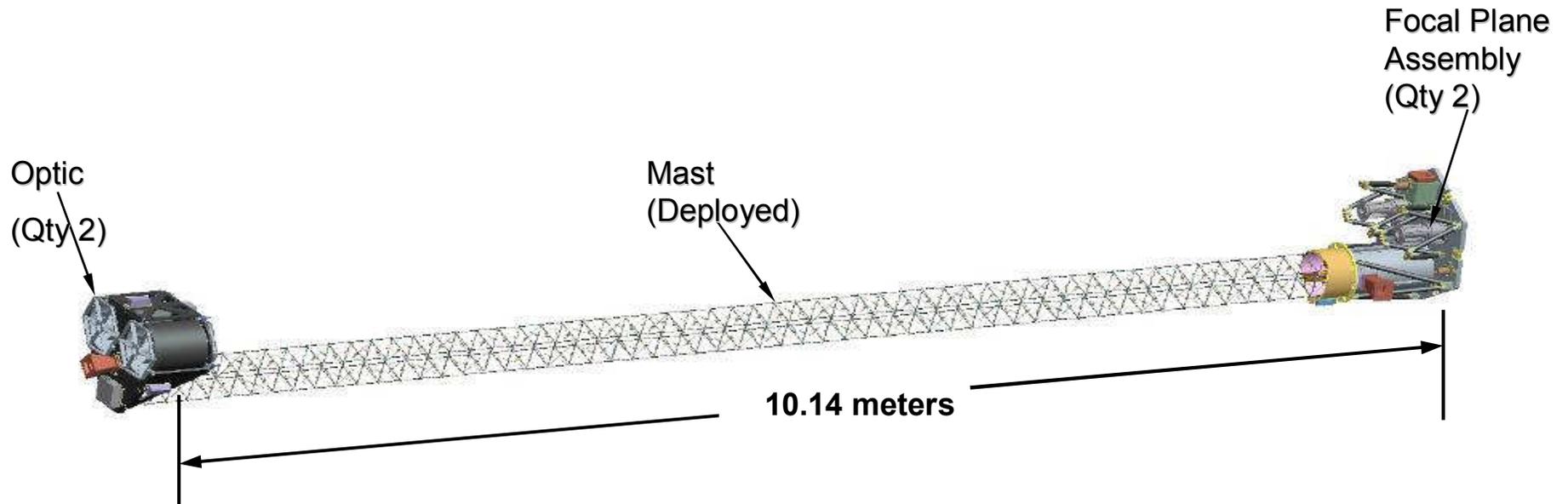
Flip-chip bonding: a pioneer in 1990s



Connecting wires are all narrower than human hair

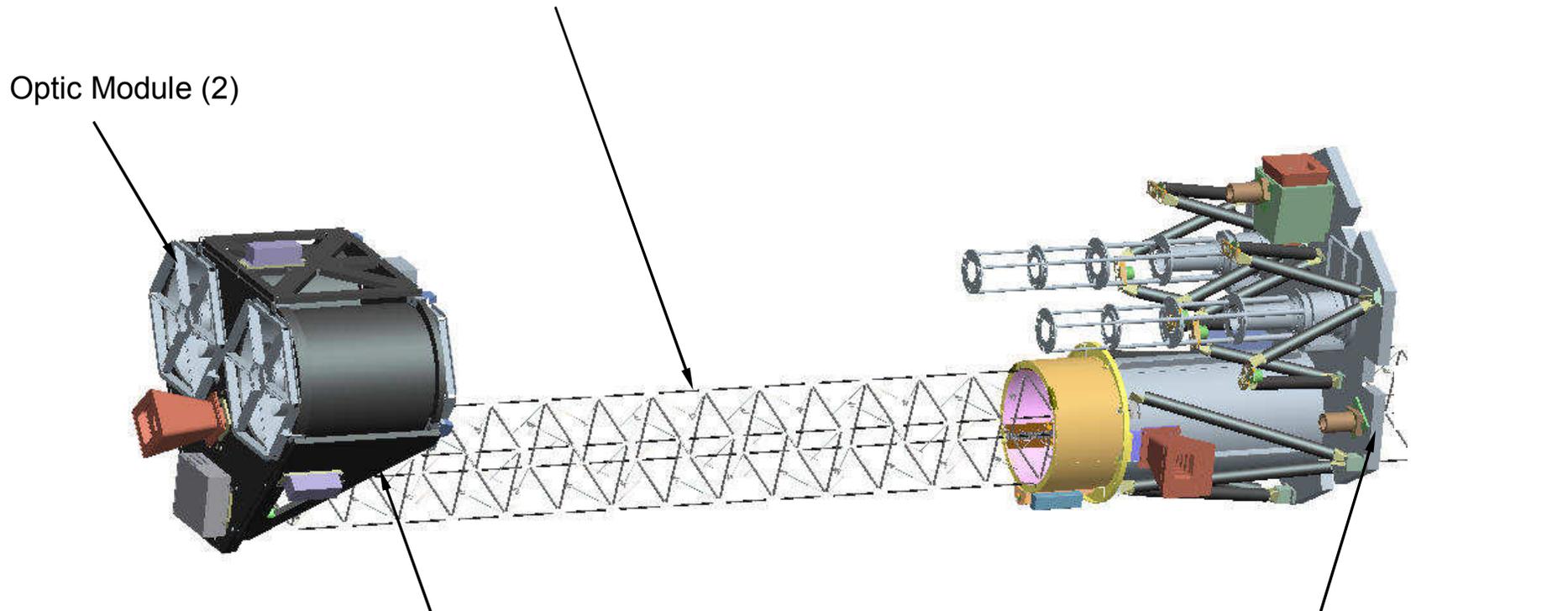
Mast: Focal Length

- The thermally stable deployable mast provides a 10.14 meter focal length between the tip-mounted optic assemblies and their corresponding base-mounted focal plane assemblies



Mast: Deployment

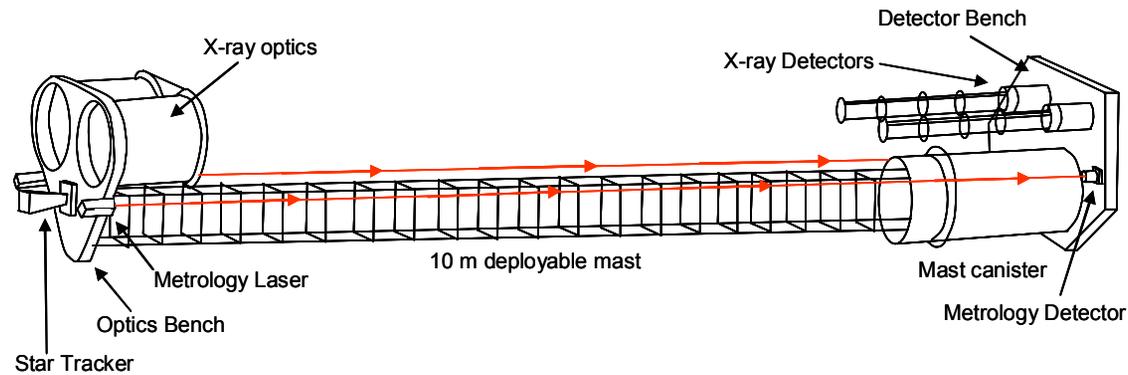
Mast and Adjustment Mechanism - The mast and adjustment mechanism are responsible for providing a stable, 10.14 meter separation between the focal plane assemblies and the tip mounted optic modules



Optical Bench – The optical bench is responsible for structurally supporting the two optics modules, three metrology lasers, an optic bench electronics box, a star tracker, and a rear-facing camera.

Focal Plane Bench Assembly – The focal plane bench assembly is responsible for structurally supporting the mast/canister, two focal plane assemblies, three metrology detectors, the central electronics box, two star trackers, and a magnetometer

Metrology



NuSTAR metrology System measures orientation that individual X-ray photons arrives from

All data processing is post processing

All measurements are relative to initial in-flight calibration

Metrology System consists of

- Optical Bench Mounted Star Tracker**
- 2 laser beams and corresponding position sensitive detectors**

Metrology system measures:

- Inertial orientation of optical bench**
- Lateral displacement of focal bench relative to optical bench**

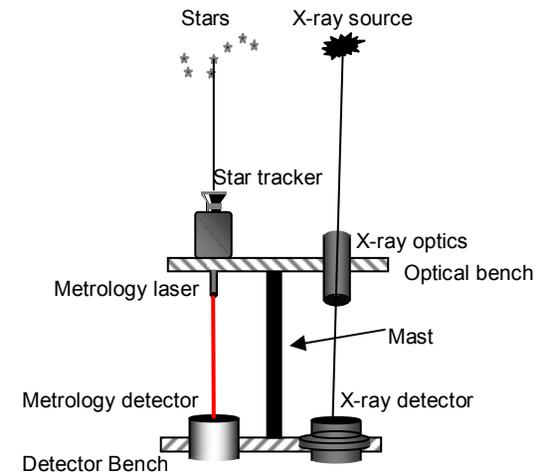
Metrology system does NOT measure:

- Distance between optical bench and focal plane bench**
- Inertial orientation of focal bench**

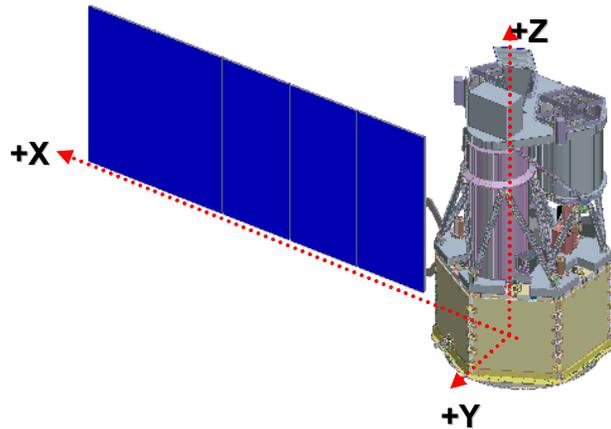
Update rate: ~100 Hz

Accuracy: 150 microns

NuSTAR simplified instrument model



Spacecraft



**Orbital Sciences Corporation
heritage LEOStar spacecraft**

Mechanical

- Aluminum Facesheet Over Aluminum Honeycomb Bus
- Carbon Fiber Over Aluminum Honeycomb Arrays

Attitude Control

- 3-axis Stabilized Zero Momentum System
- Star Tracker Provides Attitude Data
- Gyro Provides Rate Data
- Sun Sensors Provide Attitude Data in Safe Hold
- TAM & Torquers for Momentum Control

Command & Data Handling

- Standard MIL-STD-1553 and RS-422 Interfaces
- Standard CEU + Mission Unique Card (MUC)
Accommodates Payload Needs

Electrical Power

- Single Wing Deployed Single Axis GaAs Solar Arrays
- 16 A-Hr Battery
- Heritage Power Regulation Electronics
- Power Converter Box

Software and Fault Protection

- Execution of Stored and Real-Time Command Sequences
- Safe Hold Mode Protects Spacecraft During Faults

Thermal Control

- MLI & Silver Teflon for Passive Thermal Control
- Thermistors and Heaters for Component Thermal Control

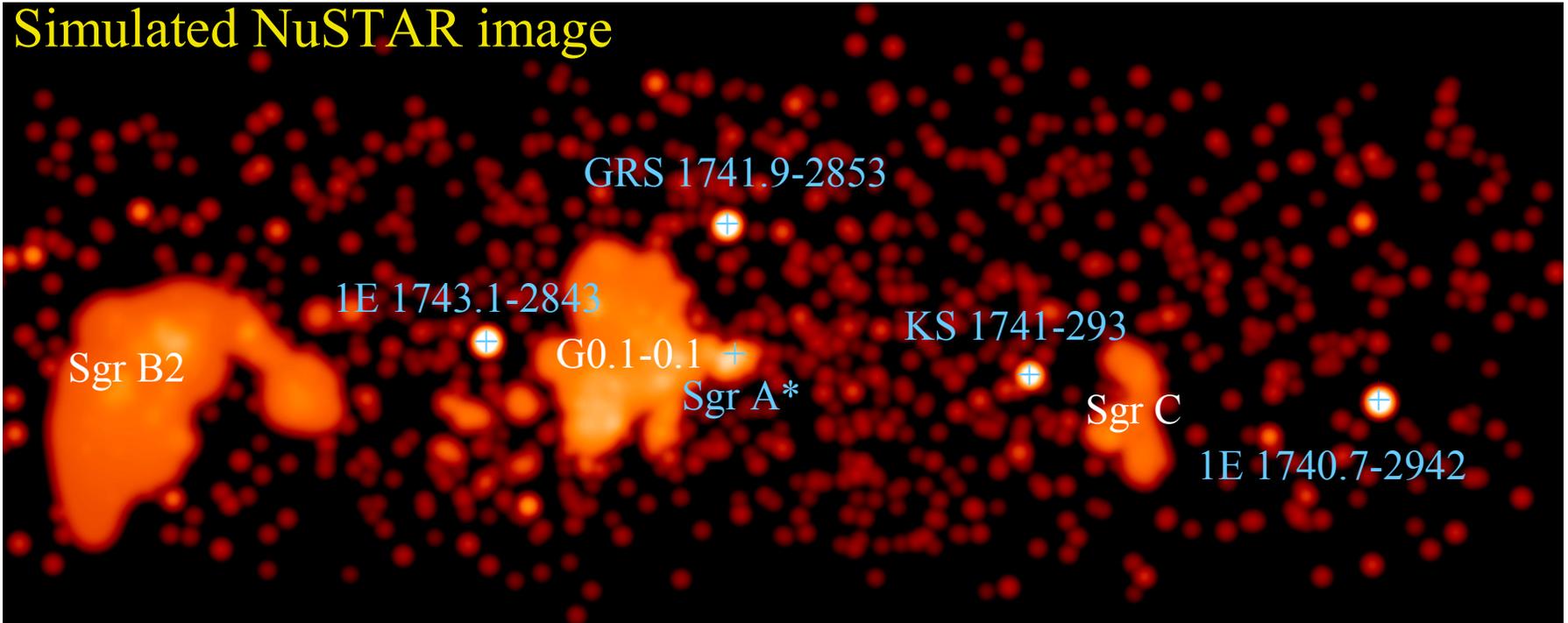
RF Communication

- 32 kbps S-band Uplink, 2 Mbps S-band Downlink
- 2 Omni Antennas Provide Near Spherical Coverage

Mission Schedule

- System Requirements Review
 - **July 9 , 2008**
- Project Preliminary Design Review
 - **June 9, 2009**
- Project Critical Design Review
 - **February 2, 2010**
- Instrument Delivery to Observatory I&T
 - **November 12, 2010**
- Observatory I&T
 - **November 13, 2010 - June 24, 2011**
- Launch
 - **August 15, 2011**
- Science Operations
 - **September 15, 2011 – October 1, 2013**

The Future: *NuSTAR*



- 40" angular resolution
- Sensitivity $\sim 10^{-15}$ erg cm $^{-2}$ s $^{-1}$ (20-40 keV; 10 6 s).

Michael Muno (Caltech) for the *NuSTAR* team