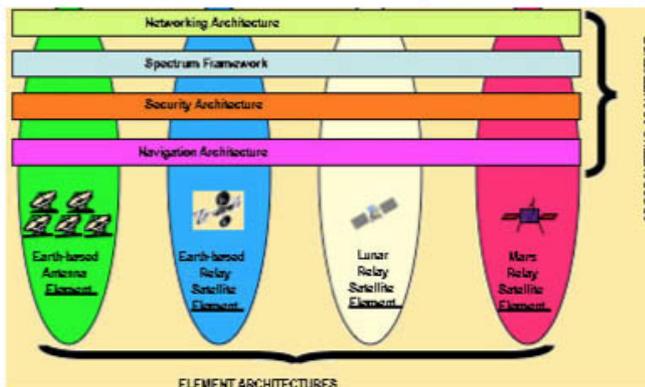
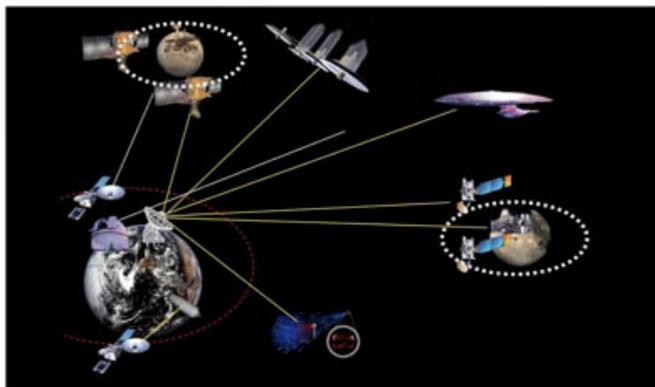
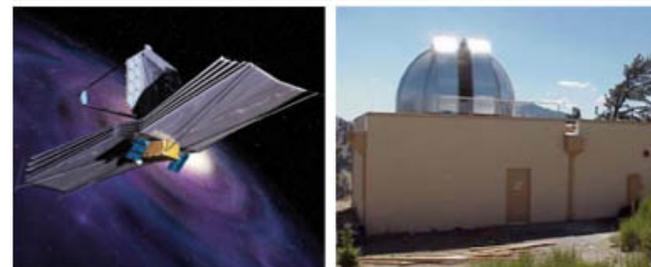
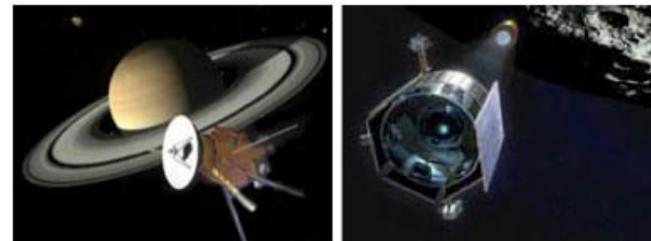


# Interesting Problems in Deep Space Communications

Dr. Les Deutsch  
Jet Propulsion Laboratory, California Institute of Technology





# Agenda

- What makes deep space unique?
- Case study 1: Galileo mission to Jupiter
- Case study 2: Cassini/Huygens mission to Saturn
- Future trends in deep space communications
- What worries me today?



# Some Current Deep Space Missions



Cassini: Saturn



SIRTF: Astronomy

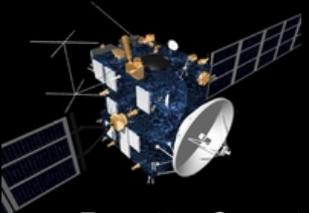


Mars Global  
Surveyor

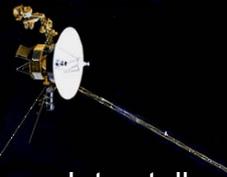
Mars Odyssey



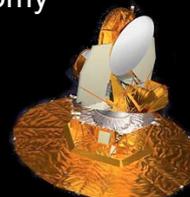
Mars Science  
Laboratory



Rosetta: Comet



Voyager: Interstellar



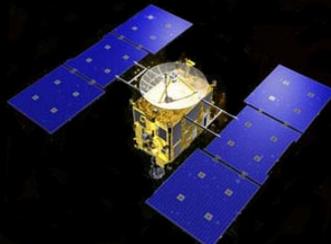
WMAP: Astronomy



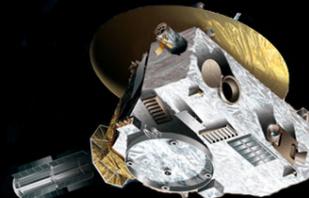
Mars Express



Mars  
Reconnaissance  
Orbiter



Hayabusa: Asteroid



New Horizons:  
Pluto



MESSENGER:  
Mercury

Kepler: Extrasolar  
Planets

GRAIL: Moon



Dawn: Asteroids





# Deep Space is Unique

- **Spacecraft mass and power are precious**
- **Spacecraft go huge distances from Earth**
- **Navigation is highly dependent on Earth**
- **Communications system is a mission science instrument**
- **Every mission is unique**



# Space Loss

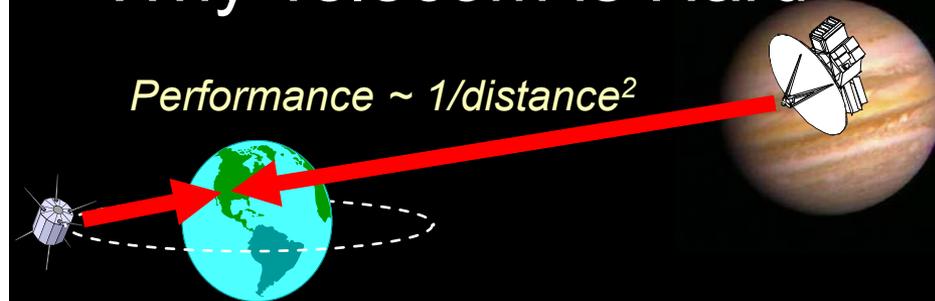
- All else being equal, communications performance is inversely proportional to distance squared

$$P_T/N_0 = \text{constant} / d^2$$

- Need to overcome this problem of physics to be successful in deep space

## Why Telecom is Hard

Performance  $\sim 1/\text{distance}^2$



Relative Difficulty		
Place	Distance	Difficulty
Geo	$4 \times 10^4$ km	Baseline
Moon	$4 \times 10^5$ km	100
Mars	$3 \times 10^8$ km	$5.6 \times 10^7$
Jupiter	$8 \times 10^8$ km	$4.0 \times 10^8$
Pluto	$5 \times 10^9$ km	$1.6 \times 10^{10}$



# NASA's Deep Space Network



- **Giant (34m and 70m) antennas in three locations around the world**
- **Provides communications with missions beyond GEO**





# Spacecraft Mass & Power

**JPL**



- **Deep space missions must leave Earth's gravity well – very difficult**
  - An Atlas V 551 can lift about 19,000 kg to LEO but only ~500 kg to deep space
- **Power generation is very difficult for a spacecraft far from the sun**
  - Solar flux goes down by a factor of four each time the distance from the Sun doubles, so a solar panel at Jupiter can only generate a billionth the power as at Earth
  - Nuclear-based generators are both expensive and politically sensitive



# Cannot Waste a dB

**JPL**

- **Deep space missions operate close to theoretical communications efficiency limit (within 1 dB, typically)**
- **Example: If a spacecraft, designed to work with a 70m antenna, lost a dB of performance, it would take an additional 32m antenna to make up the difference!**
  - **Cost for three 32m antennas = ~\$100M!**



# Autonomy

- **It can take minutes to hours for signals to travel between the spacecraft and Earth**
- **Decisions must often be made faster than this – requiring spacecraft autonomy**
- **Spacecraft are usually “sequenced”, meaning they are programmed to operate for long periods without commands from Earth**
- **Spacecraft manage the data they acquire, storing it until it can be sent back to Earth**
- **Emergencies require special “safing” algorithms**

# Case Study 1



## Galileo Mission to Jupiter

- **NASA flagship mission launched in October 1989**
  - Delayed several years by Challenger accident
  - Radioisotope generators already partially depleted
- **High gain antenna (HGA) failed to deploy**
- **Left the comm system with  $10^{-4}$  disadvantage**
  - Hemispherical antenna instead of HGA
  - S-band instead of X-band
- **Spacecraft was fully functional except for the HGA**
- **Lucky break**
  - Twice as much RAM onboard as in the original design

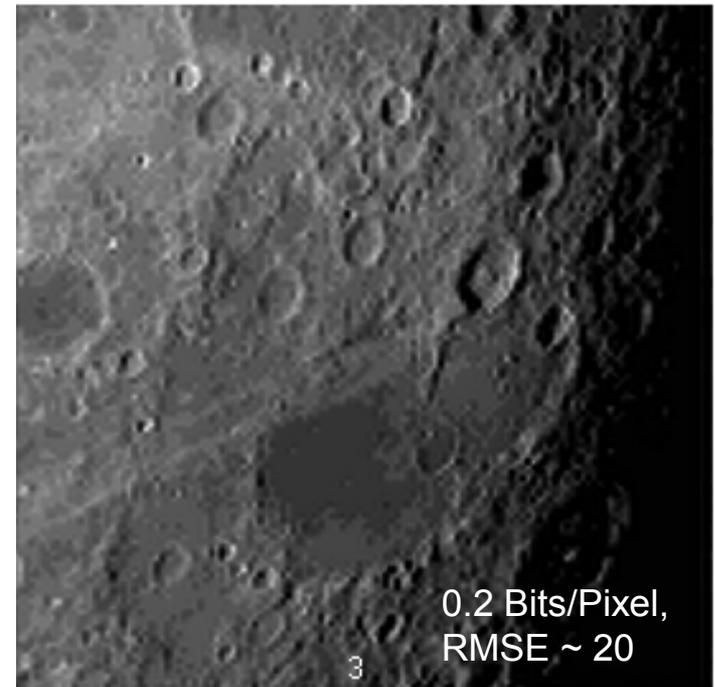




# Compression 4 GLL

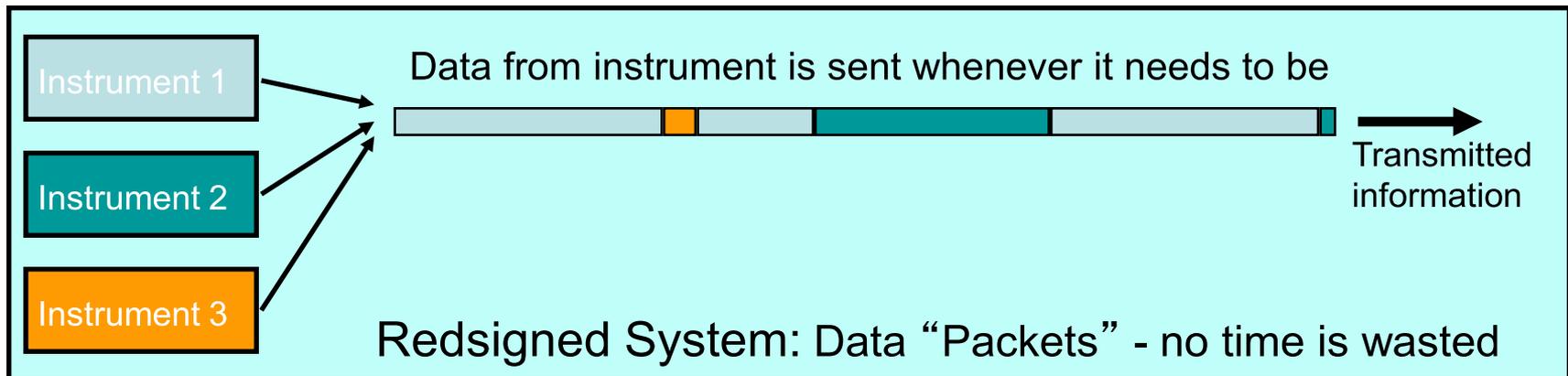
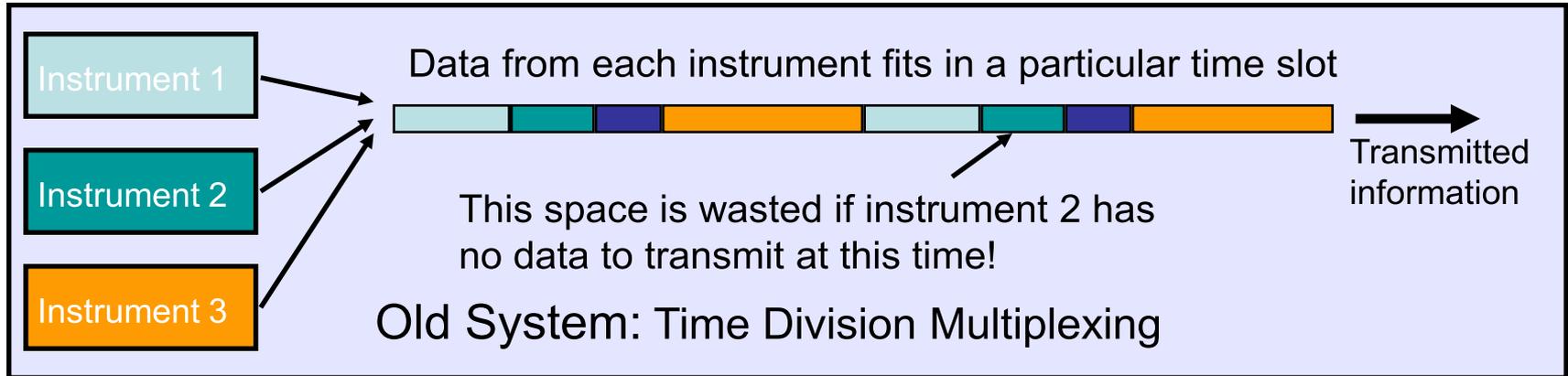
- **Before Galileo, deep space missions used “lossless” compression**
  - Low compression ratios - typically no better than 2:1
  - Well understood, deterministic errors, caused mainly by data overflow
- **We added Integer Cosine Transform algorithm – similar to JPEG**
  - Compression ratios could be set – typically between 5:1 and 20:1
  - New error containment strategies made this viable for deep space
  - Error artifacts are better understood due to research in support of Galileo
- **Galileo was forced to determine which information is most important from each instrument and subsystem – data editing**

**All subsequent deep space missions have used compression and editing**



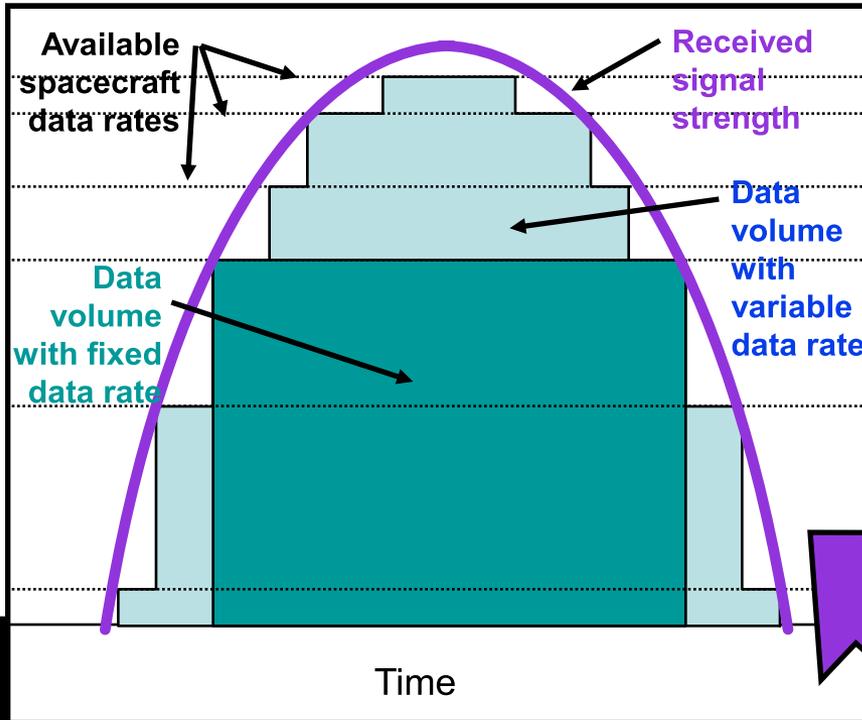


# Packetized Telemetry

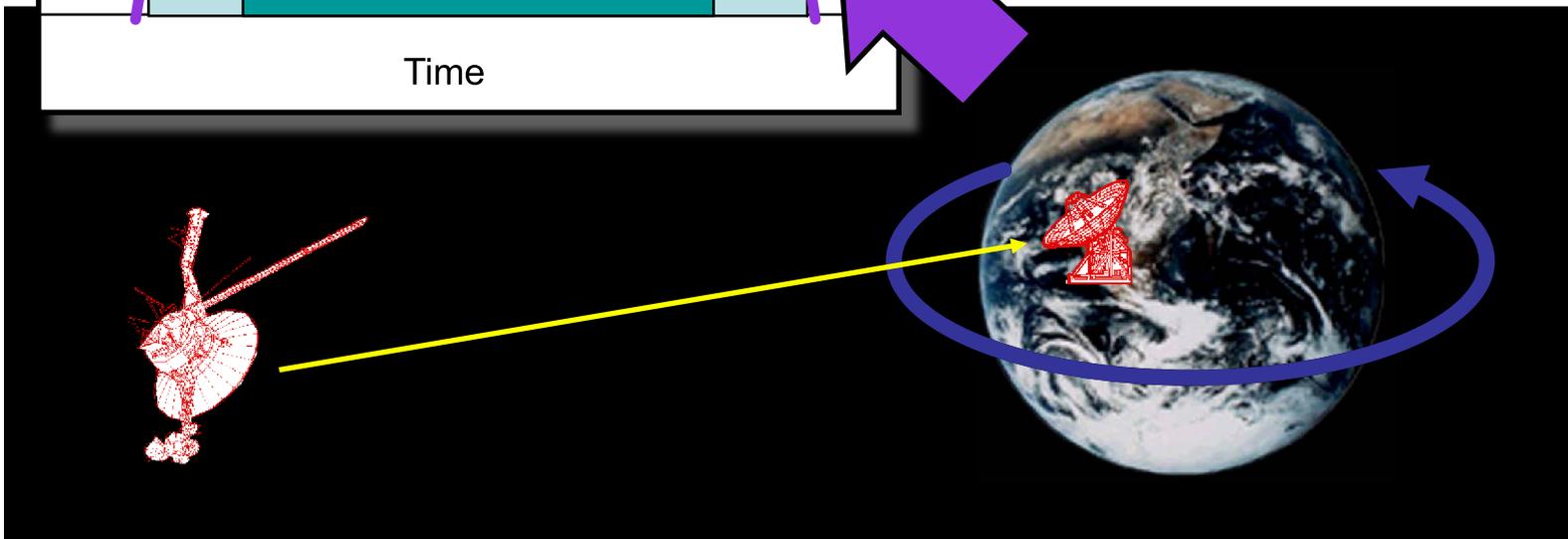


- **All subsequent deep space missions use packet telemetry to eliminate wasted space in the transmitted information**

# Variable data rates

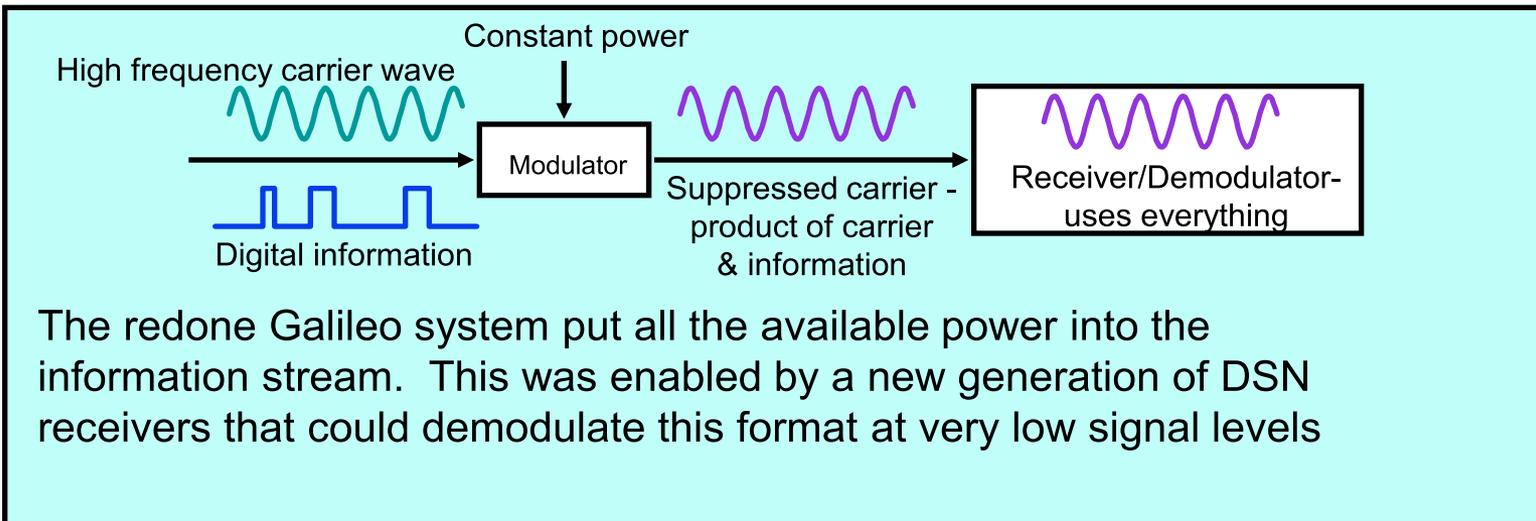
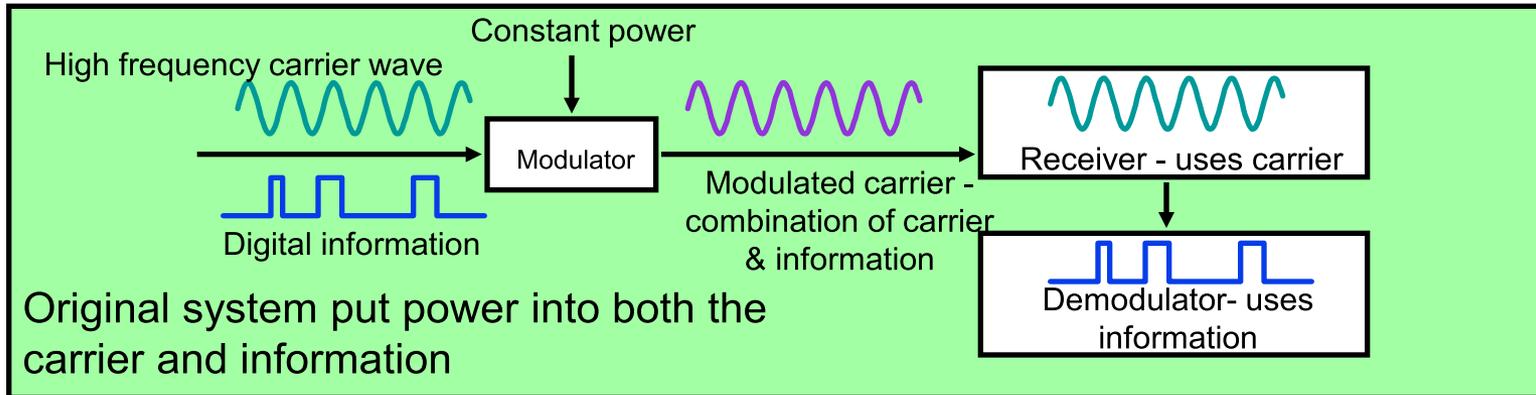


- Galileo was the first deep space mission to use variable data rates during DSN passes
- Used prediction of signal levels
- Led to investigation of Internet-like protocols – DTN





# Modulation

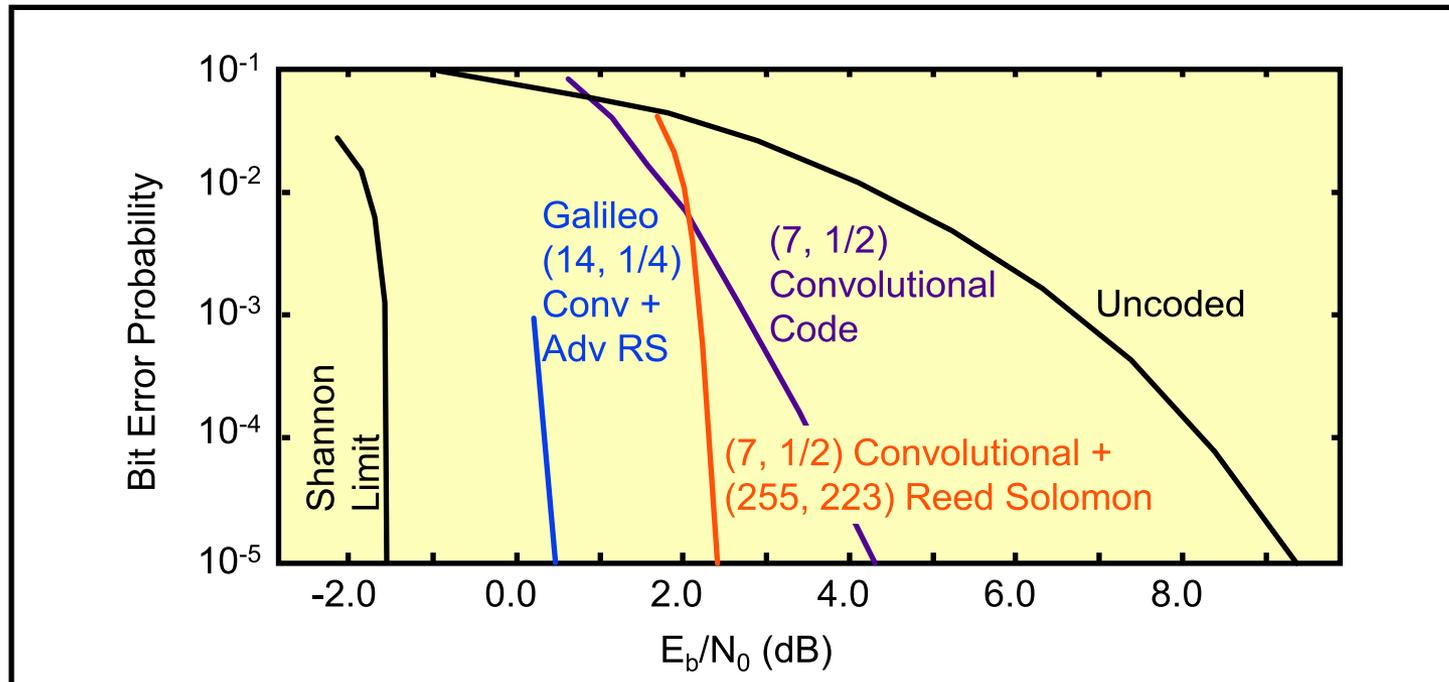


- **Suppressed carrier modulation is now available to all future deep space missions**

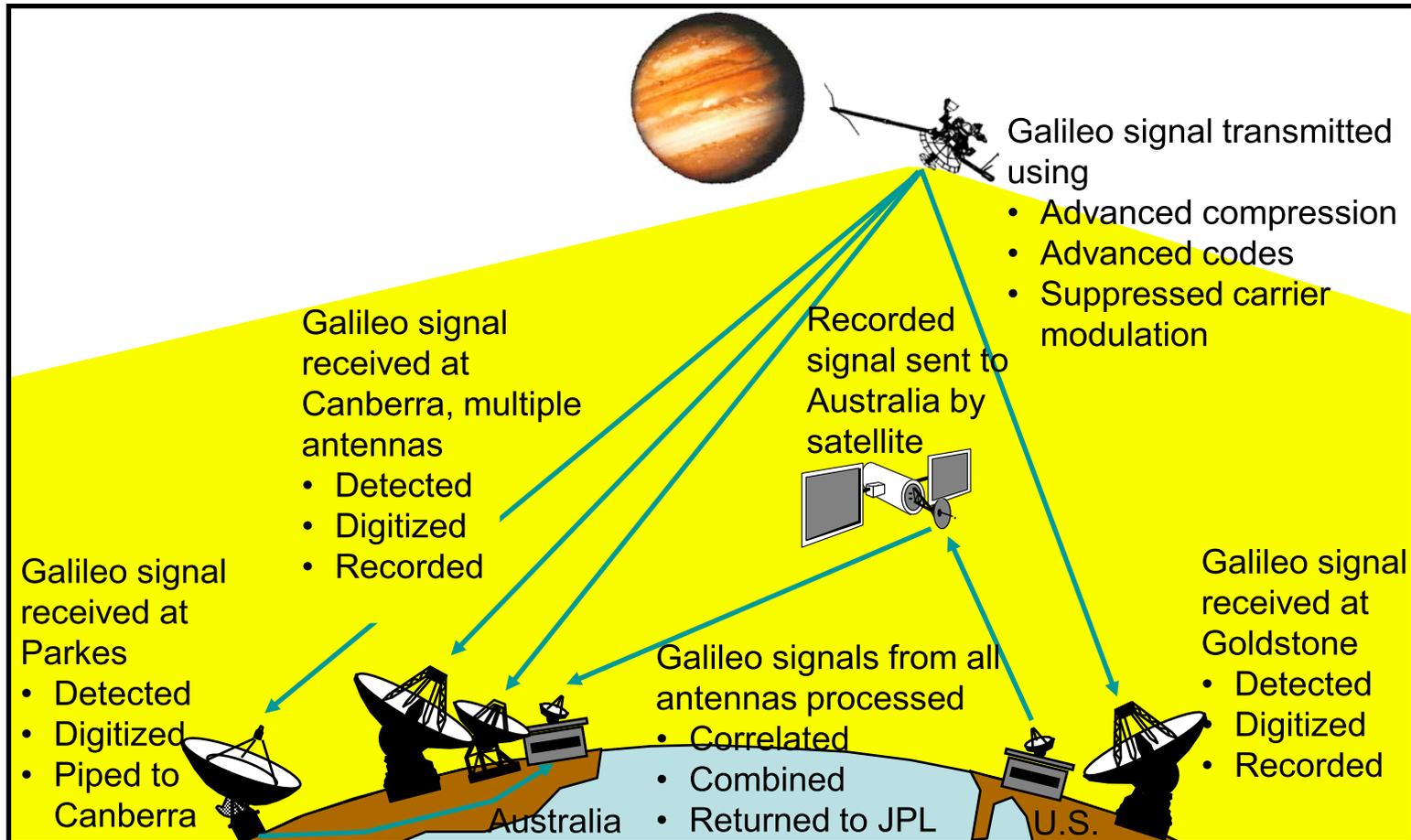


# Error-correcting codes

- Error correcting codes had been used for many years in deep space
- Galileo's redesigned codes were the best yet flown
- Among the new coding techniques on Galileo were variable redundancy, and redecoding – applicable to future codes as well
- Subsequent missions leverage these techniques and continue to evolve better codes

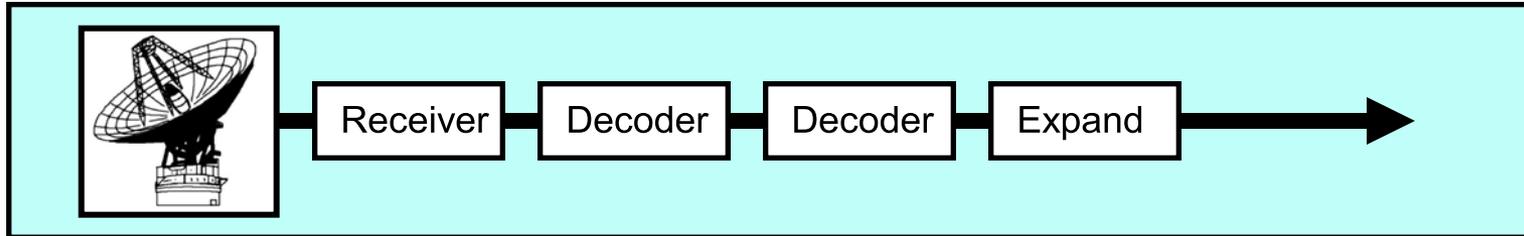


# JPL Antenna+Antenna+Antenna... Arraying

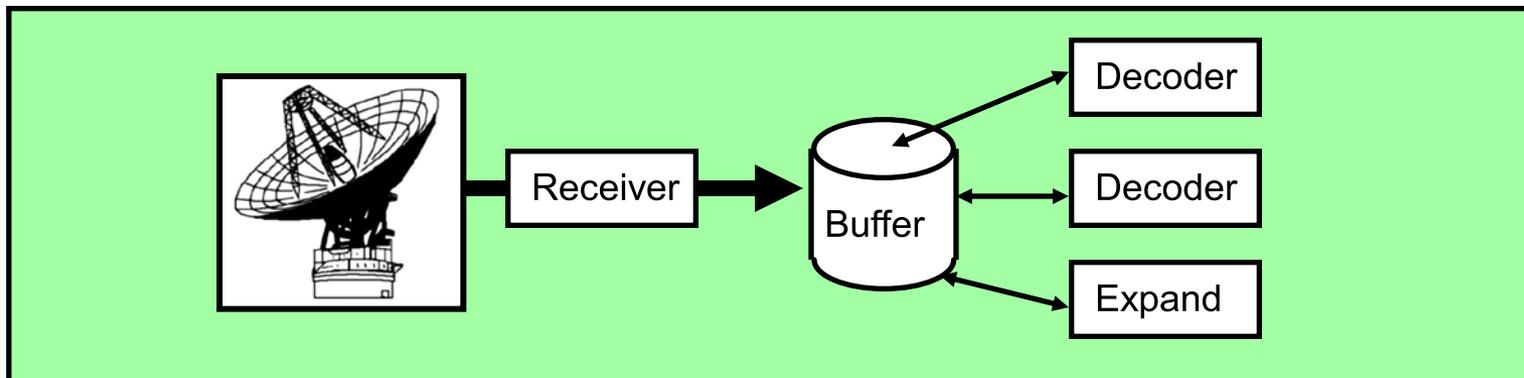


- **The DSN now routinely supports arrays – but only within the DSN**
- **Arrays are be used for special mission events and to synthesize 70m antenna performance from smaller antennas – serving as backups for the 70m antennas**

# Asynchronous processing



- **For Galileo, all DSN processing after bit detection was performed on demand**
- **This system is more modular, efficient, and cheaper to maintain than the old serial system**
- **Although this does not increase link efficiency, the DSN is evolving to this architecture – a lower-cost legacy for all future missions**





# Summary of Galileo repairs



**Optimized Signal Detector**

**Antenna arraying**

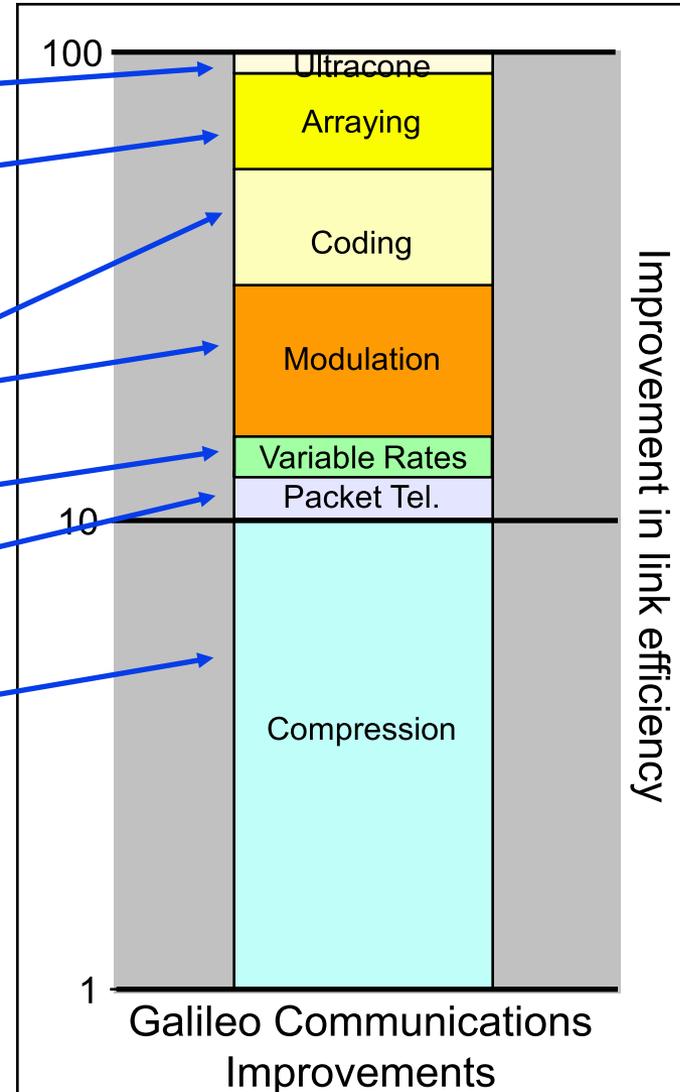
**Advanced error-correcting codes**

**Efficient modulation**

**Variable data rates**

**Packet telemetry**

**Data compression**





# Galileo – Conclusion



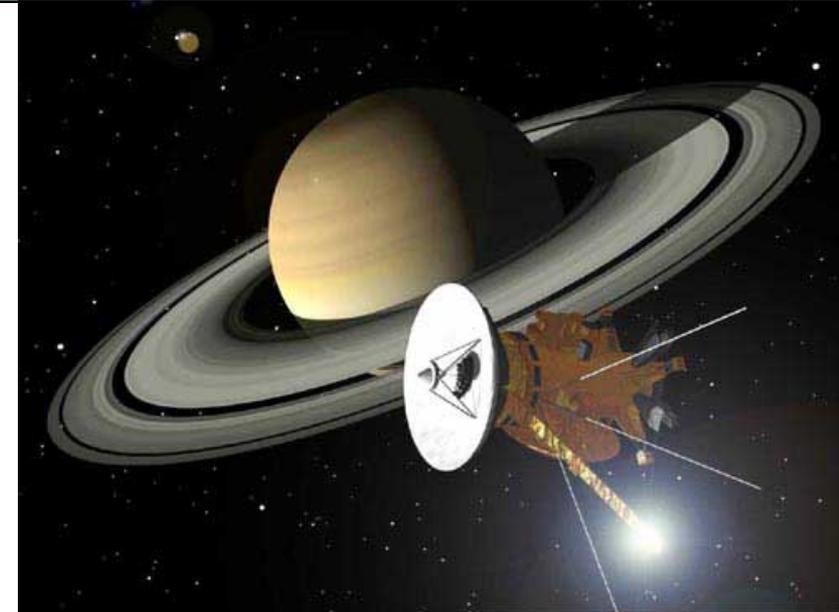
- **Galileo was able to increase its effective data rate from less than 10 bps to around 1 kbps**
- **The mission was a complete success, achieving more than 75% of its original science goals during the 2-year prime mission**
- **Galileo went into extended mission and was eventually crashed (intentionally) into Jupiter to avoid possible contamination of the moons**
- **All of the communications improvements that were made to Galileo became standard on subsequent missions**
- **Additional “lesson learned”: It is good to have reprogrammability on a deep space mission**

## Case Study 2



# JPL Cassini/Huygens Mission to Saturn

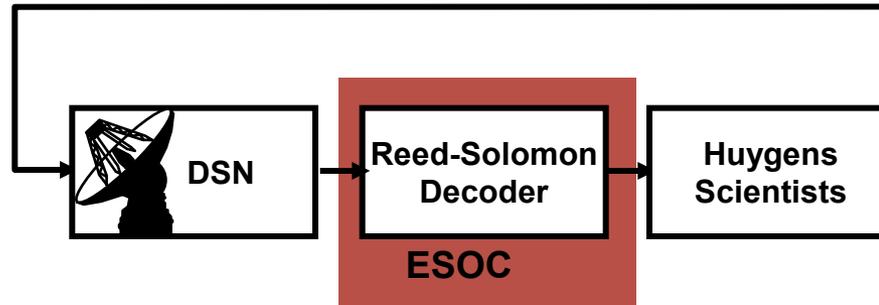
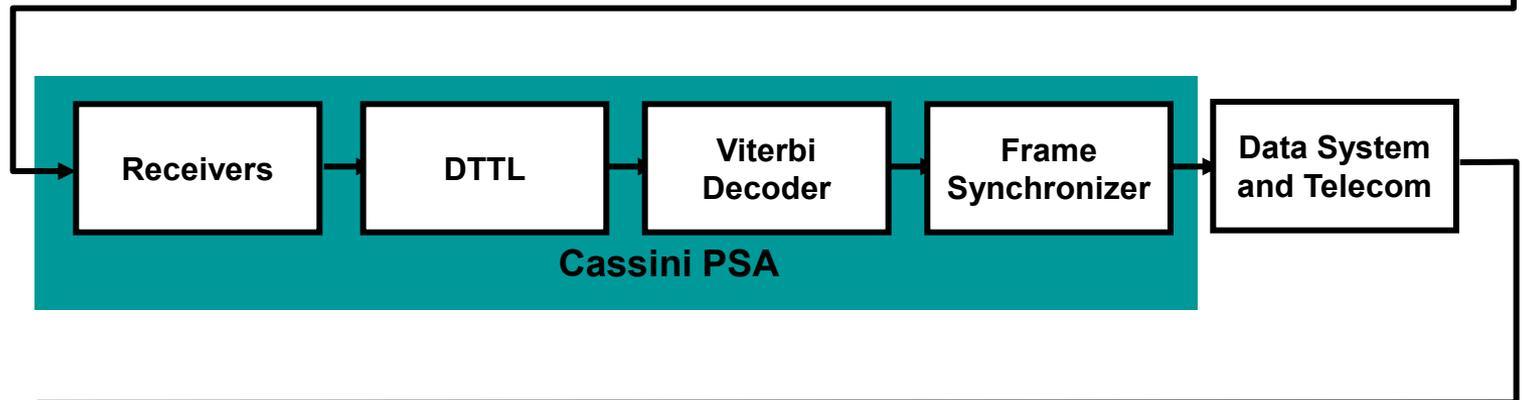
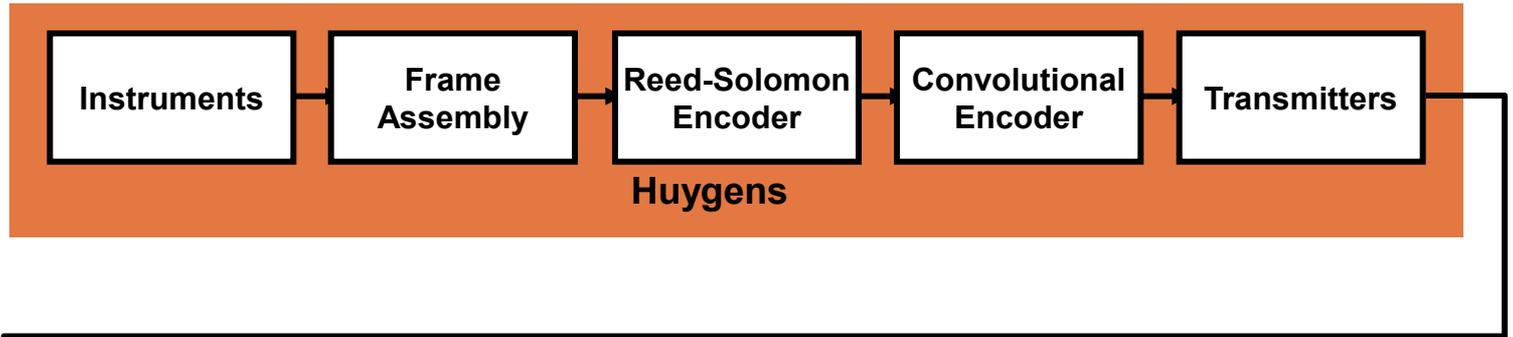
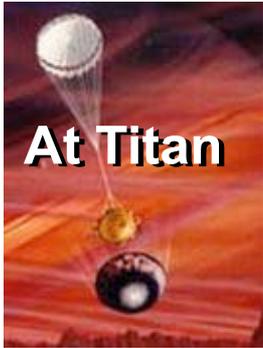
- **Launched in July 2004**
- **Cassini was the NASA flagship mission to Saturn, Huygens was an ESA-built daughter probe for exploring Titan**
  - Huygens would descend into the atmosphere to Titan and use Cassini to relay signals to Earth
- **Everything looked great during Cassini/Huygens cruise to Saturn**
- **A very clever test of Cassini relay radio, using the DSN to mimic Huygens, showed that Huygens was going to fail**





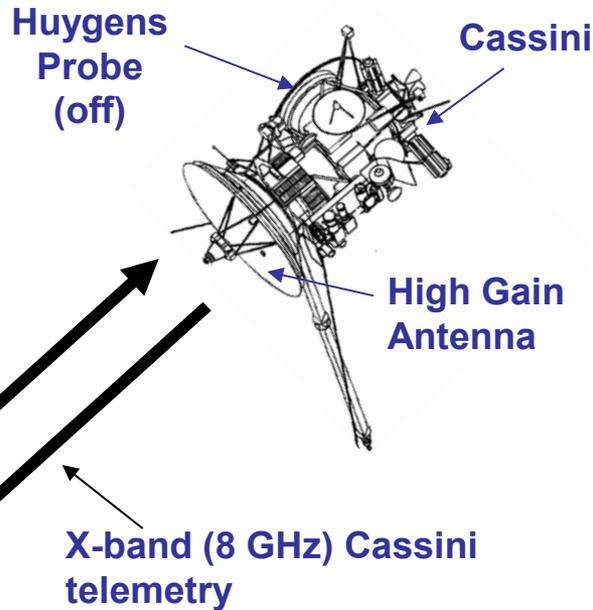
# Huygens Telecommunications System

JPL



# Discovering the Problem

S-band (2 GHz) radio  
signal transmission  
from Goldstone  
simulating Huygens  
Probe transmission  
(one-way light time  
was ~ 40 min)

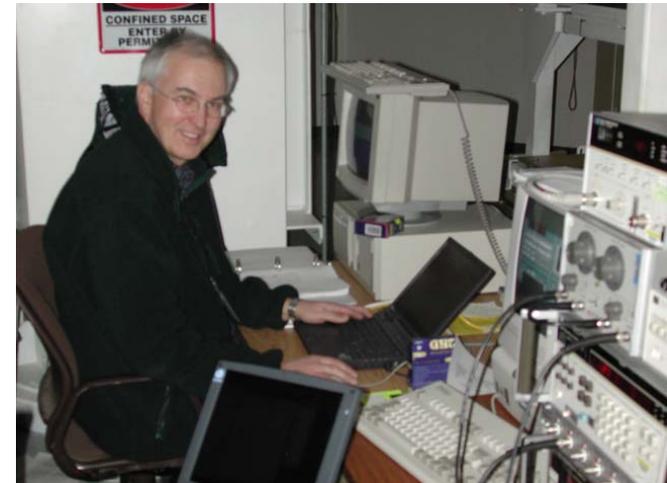


DSS 24  
Deep Space  
Network Antenna

JPL

ESOC

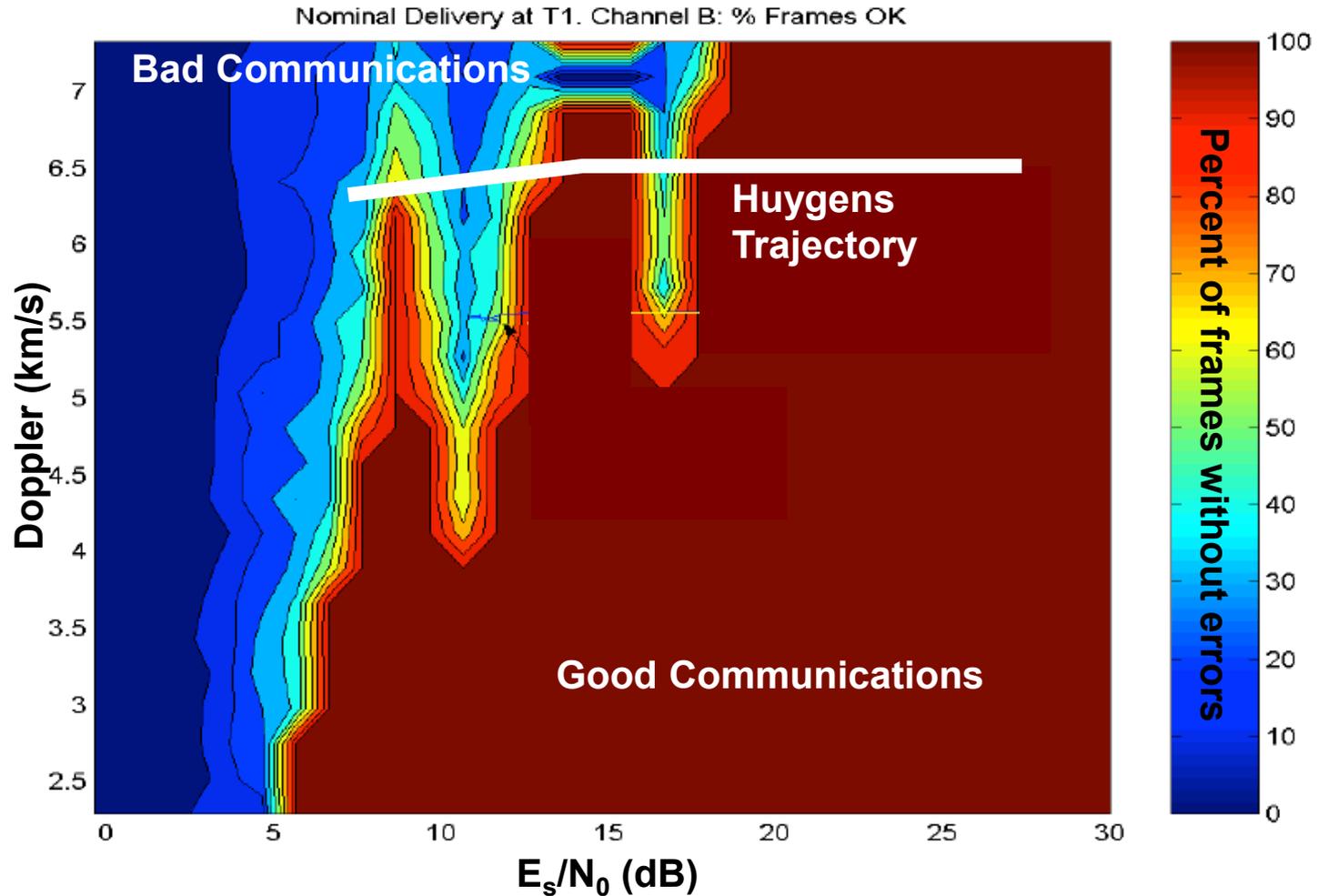
Data Evaluation  
Data arrive at  
ESOC  
80 min after  
transmission  
From Goldstone!



Boris Smets and Test Equipment



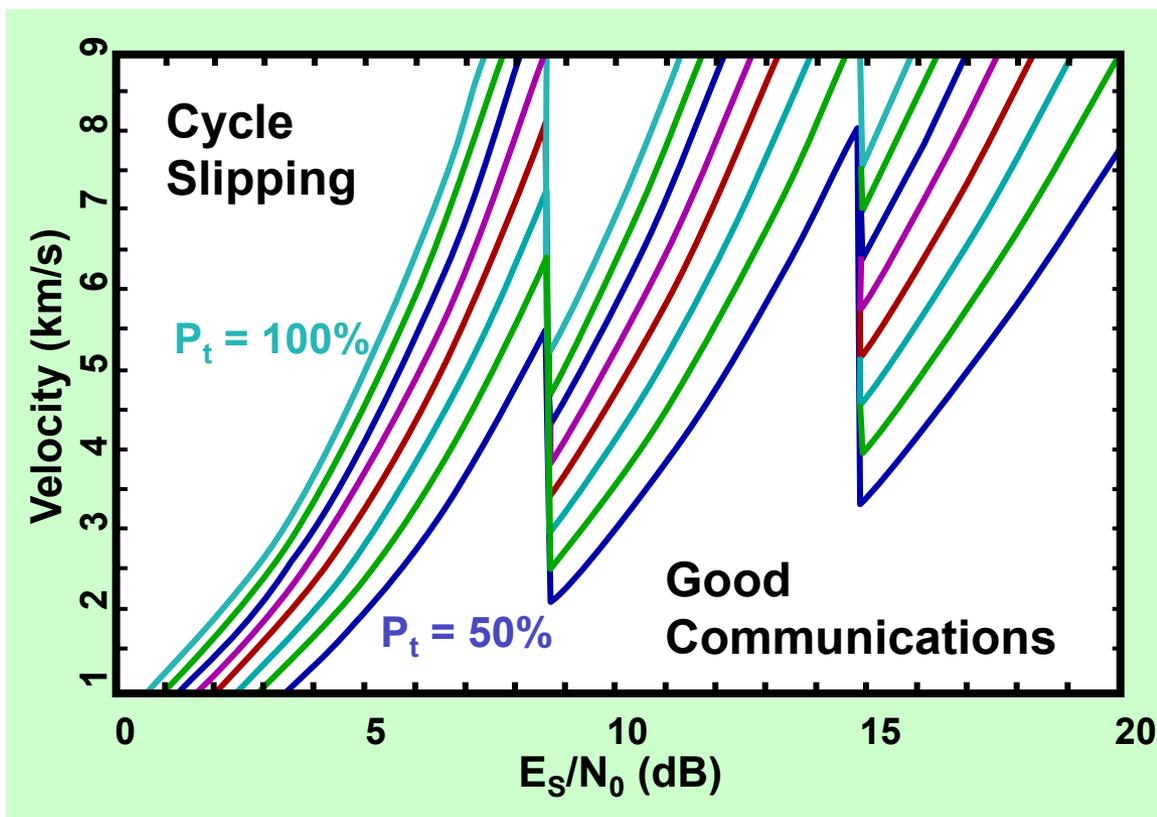
# Test Results





# JPL Modeling the Failure Mechanism

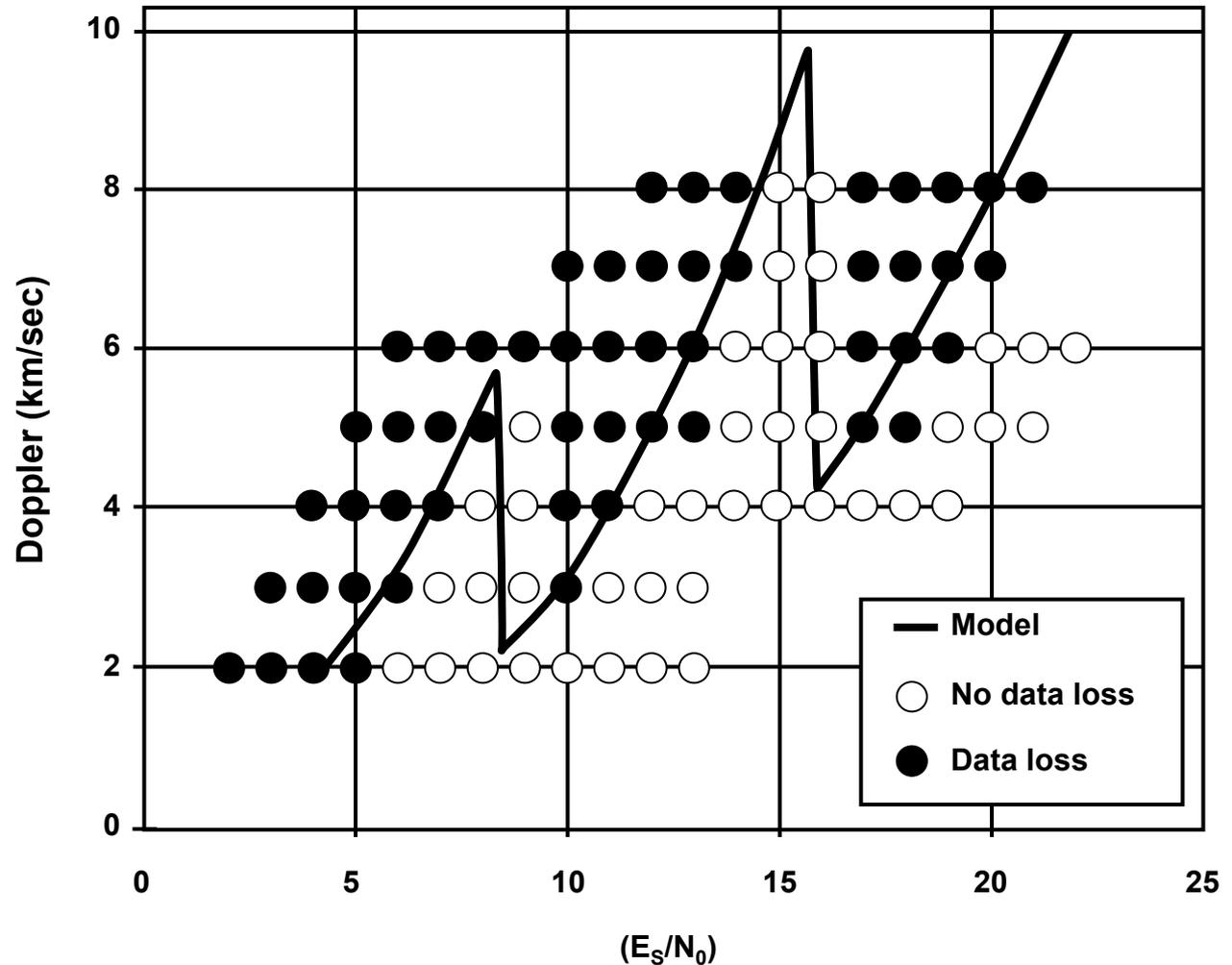
- A complete model of the relay system was developed including
  - Symbol loop dynamics as a function of bit transition probability
  - Coding and synchronization
- Symbol cycle slips were generated with a period that is a function of  $E_b/N_0$ ,  $D_f$ , and  $P_t$
- Fingers turned out to be caused by AGC function
- Analysis showed the anomaly was caused by symbol loop bandwidths
- Unfortunately, these parameters were hard-coded and could not be changed in flight
- The Galileo “lesson learned” was not learned well enough





# Verifying the Model

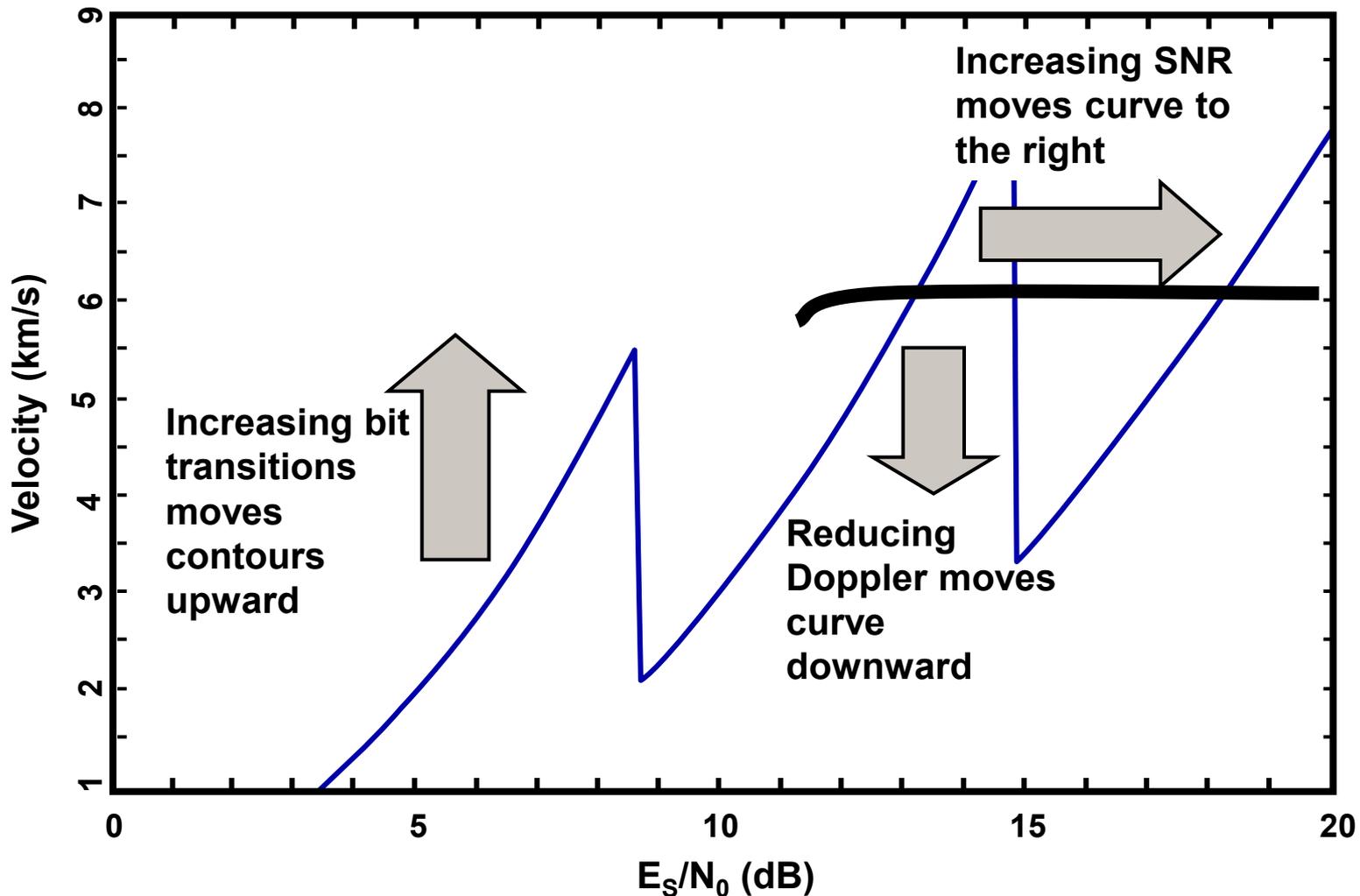
- Additional DSN tests were run to verify and calibrate the model
- Results were excellent





# Ways to Improve Data Return

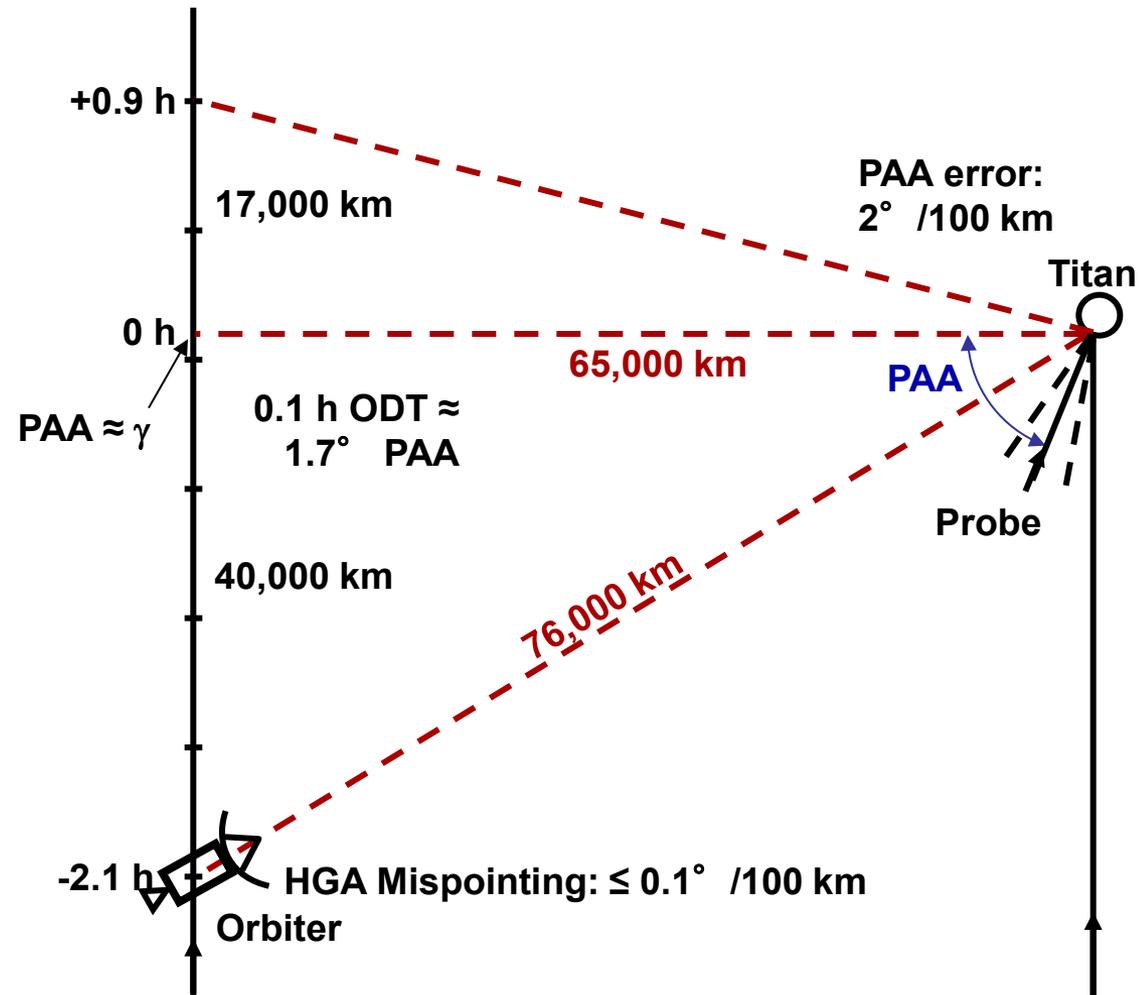
Varying the critical parameters “moves” the probe curve with respect to the contours - resulting in more good data returned to Earth





# JPL A New Retrograde Flyby Saves the Day

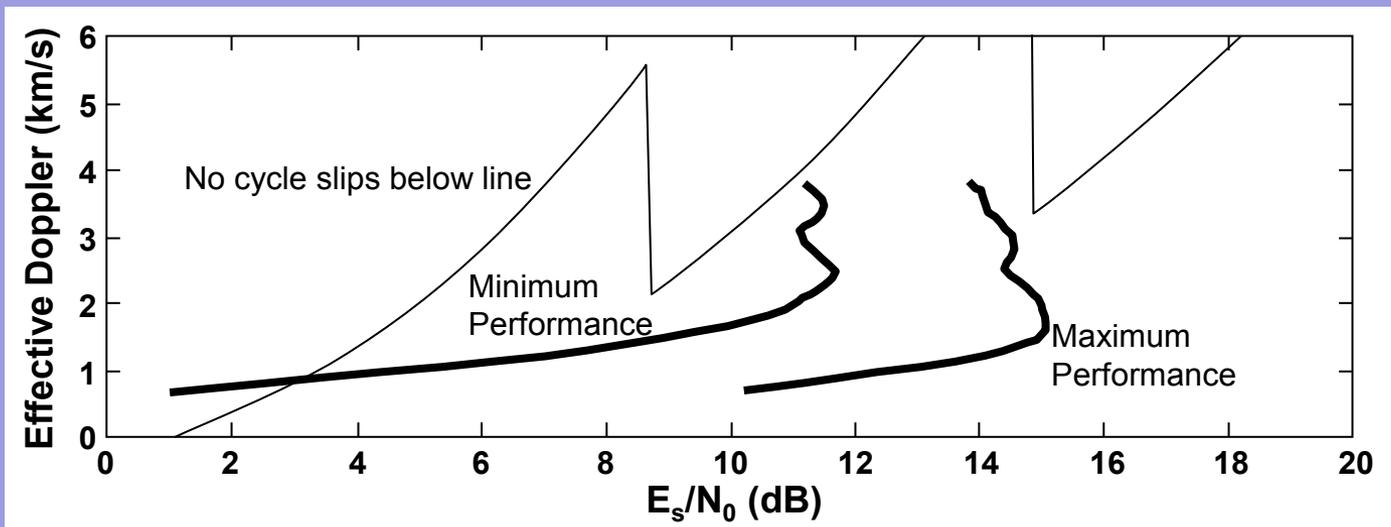
- The trick now was to find a high altitude flyby that uses minimal fuel ( $\sim 150$  m/s for simple altitude increase)
- JPL navigators came up with the idea of flying by the opposite side (retrograde) of Titan
- This uses Titan's gravity to help more with Cassini maneuvers
- New trajectory minimized additional propellant needs ( $\sim 100$  m/s)
- There is actually a class of these trajectories with one ultimately chosen by the Project





# Recommended Solution

- The team developed several point designs that showed possible solutions to the anomaly
  - Each of these would return close to 100% of the data with margin
- The trick here was that communication performance had to be bounded both from above and below
  - Too much margin is bad!
- Our experience with Galileo (operating 0.5 dB from theoretical) helped





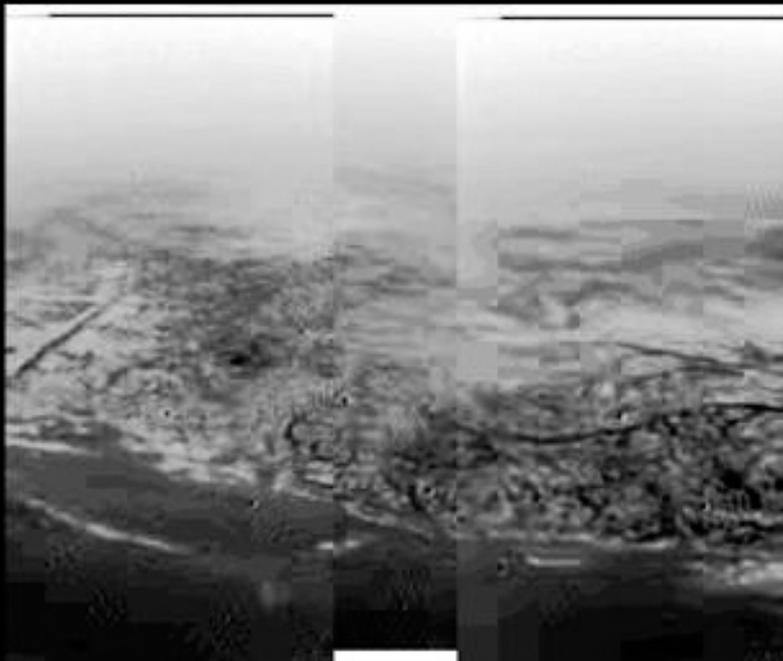
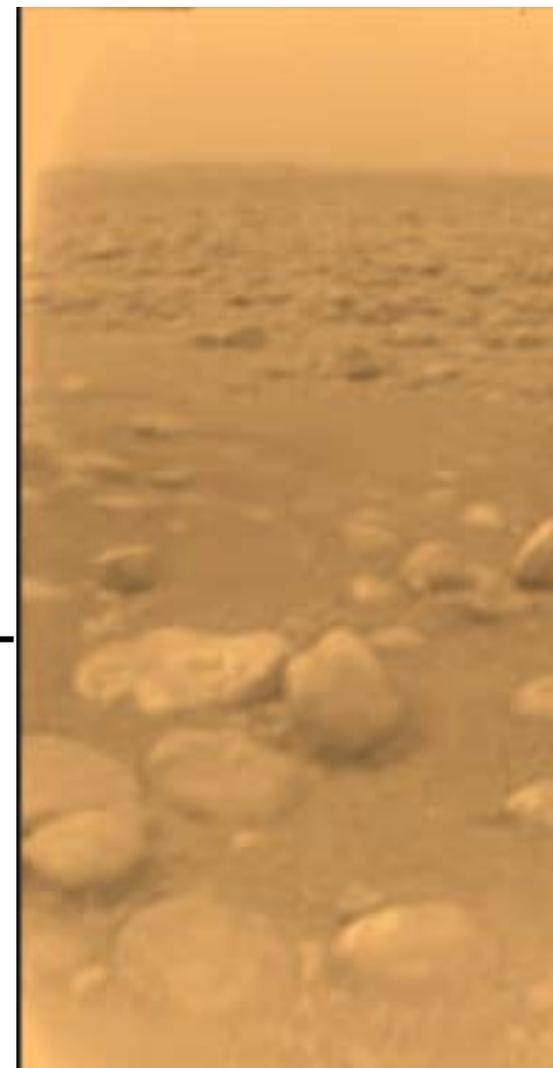
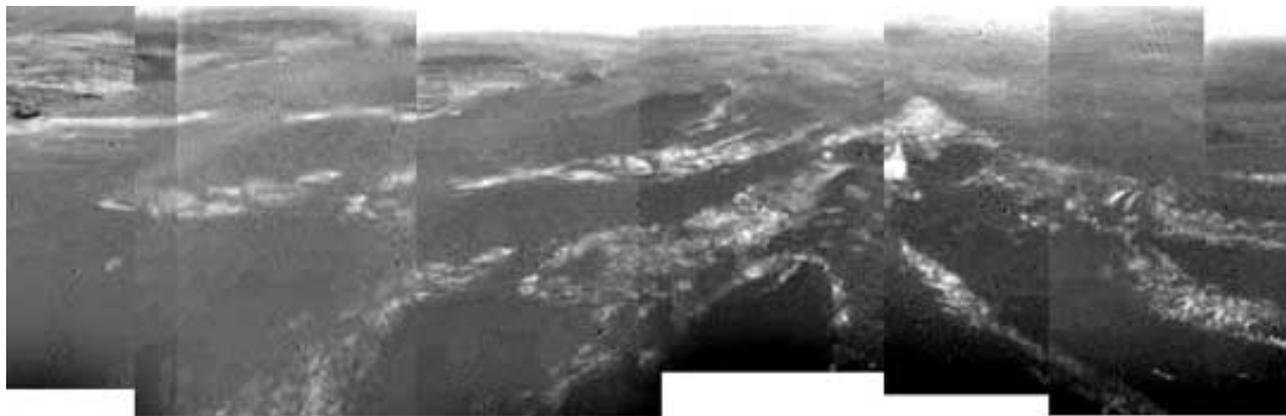
# Probe Data Playback

- **Cassini turned its HGA to receive probe data during actual descent**
  - No visibility during actual probe mission was possible
- **After turning to Earth, Cassini commenced playing the probe data to the DSN**
  - Eight full copies were sent, over all DSN complexes
  - Provided redundancy for the data playback
  - Provided resiliency in case of any two 70m failures
- **One of the two communications channels on Huygens failed (no data)**
  - Likely caused by operator error in commanding
  - This was a known risk (this is why there were two channels!)
  - All housekeeping data was redundant on the two channels
  - Most scientists planned their data campaign to satisfy their main goals with either single channel



# In the End: Success!

# JPL

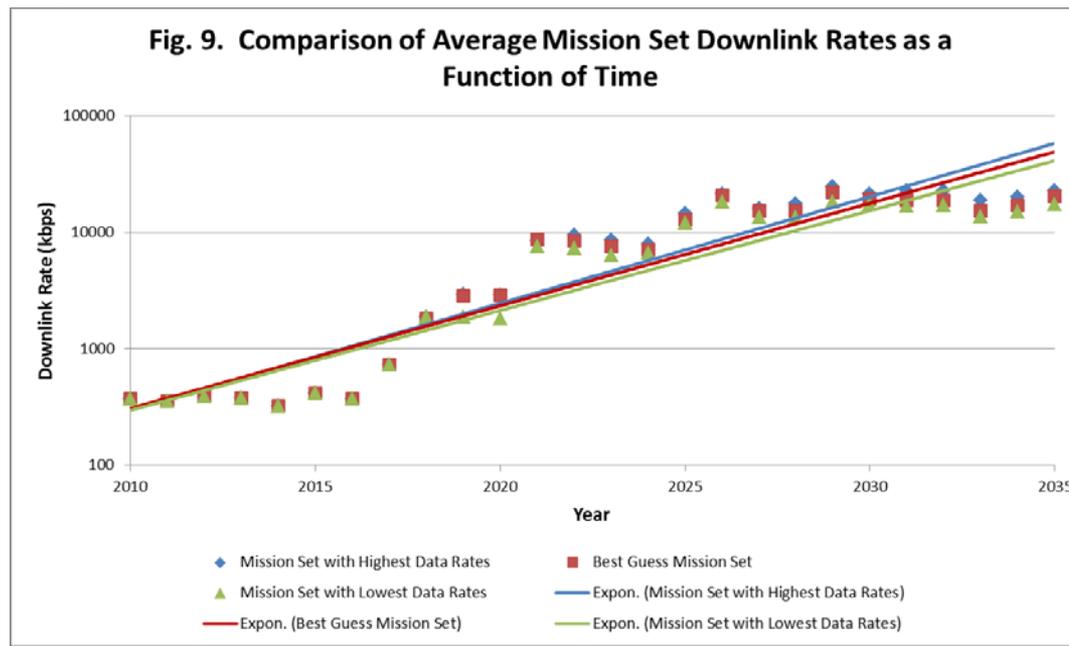


15cm



# Future Trends

- **Deep space missions are limited by the communications link**
  - Mars Reconnaissance Orbiter has only mapped 1% of Mars at high resolution
- **Things will only get worse as missions carry more data-hungry instruments**
  - Radars, multispectral and hyperspectral imagers
- **When humans go into deep space, they will need much higher bandwidth comm**





# Emerging Technologies

- **Among the technologies that will likely come into play:**
  - **Software defined radios**
  - **High frequency radio communications (e.g. 32 GHz and above)**
  - **Optical communications**
  - **More advanced data compression**
  - **More onboard intelligence and autonomy**
  - **Space internetworking**
  - **Quantum communication and sensing**



# What Keeps Me Awake Today?



- **Deciding which emerging technologies to pursue**
  - Finite funding and finite researchers mean we have to place our bets carefully
- **Ensuring we have future researchers**
  - We see fewer students pursuing communications today
- **Making sure missions use the new technology**
  - Missions are, by nature, risk averse
  - We need new processes that make it easier to infuse these technologies



# Conclusion



- **There are real problems to be solved in deep space communications**
- **Most often, smart solutions lead to new kinds of space missions**
- **Sometimes, smart solutions save spacecraft that are in jeopardy**
- **The space business needs a next generation of smart problem solvers**