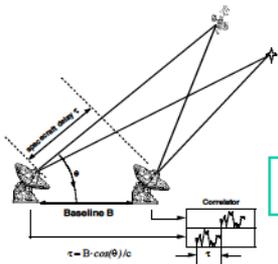


DSN Delta Differential One-way Range History and Current Operations

26 March 2013

Delta-DOR Engineering
Jet Propulsion Laboratory



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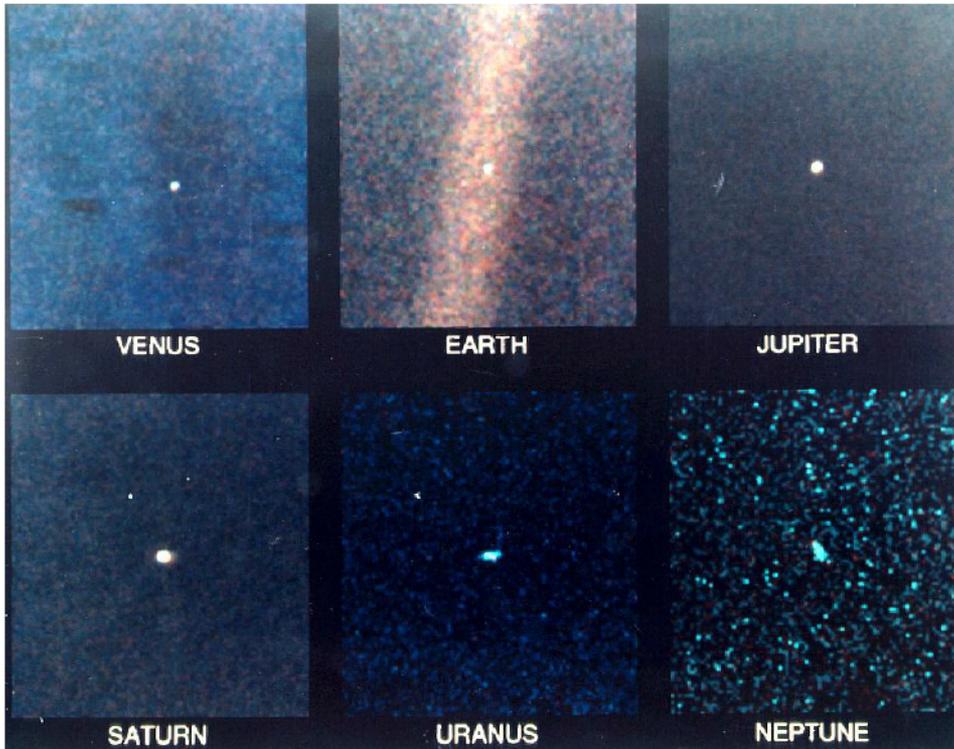


Part I. Δ DOR History

Angle and Distance: What Was Known in 1961 ?

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Planetary Images

- Planetary angular coordinates were known fairly well ($\sim 1 \mu\text{rad}$) from years of optical observations
- The distance scale for the solar system was poorly known – different estimates for the AU differed by 100,000 km
- Original goal for Mariner 2 was to determine AU based on miss distance of Venus flyby

Distance Coordinate Is Improved

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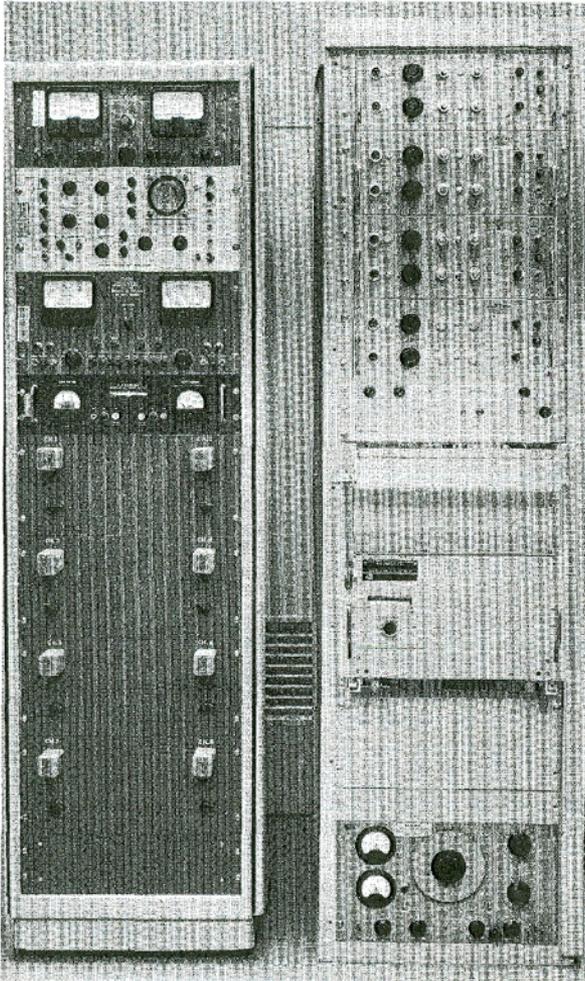


Figure 63. 8-channel receiver and recorder

- Richard Goldstein of JPL made first planetary radar measurements of Venus in March 1961
 - Sensitive open loop receiver and good signal processing algorithm were key
- Within a few months, AU was known to better than 1000 km, an improvement in knowledge of distance by x100
- Mariner 2 trajectory was redesigned and useful scientific data were acquired at Venus flyby in December 1962
- **How would angular tracking be improved and how long would it take ?**



Navigation and VLBI in the DSN Developed Separately

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Navigation Development (1962-1978)

- Doppler Measurement System
- Dynamic and Measurement Models
- Estimation and Filtering
- Higher Radio Frequency
- Range Data Type Introduced
- Media Models

VLBI Development (1971-1978)

- Celestial Reference Frame
- Precession / Nutation
- UT1 / Polar Motion
- Clock Synchronization
- Station Coordinates
- Media Models

Δ DOR Is Introduced (1979)

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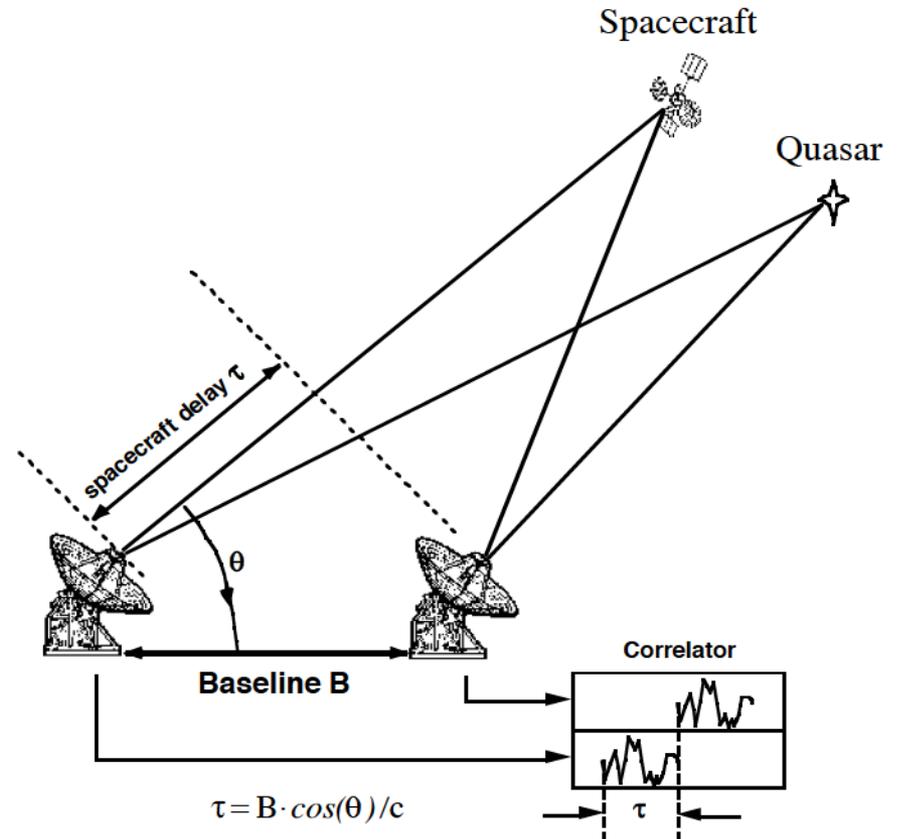
- Navigation system had one major deficiency: not sensitive to declination for declinations near zero
 - Voyager Saturn encounters (1980-1981) would be at zero declination
- VLBI tracking of spacecraft was studied to directly measure angles
 - Angular position could be determined more accurately
 - Spacecraft state estimates would be less sensitive to dynamic force mis-modeling
- David Curkendall of JPL realized that a measurement analogous to VLBI could be made using ranging signals
 - Hence the name Delta Differential One-way Range (Δ DOR)
- **But various difficulties would slow development**

Delta Differential One-way Range

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- Δ DOR complements line-of-sight range and Doppler measurements
- Δ DOR uses interferometry to directly measure spacecraft angular position in the radio reference frame
- Development began in the late 1970's
- Accuracy has improved from 150 nrad in 1981 to 2 nrad today
- Many missions have used / will use Δ DOR
- This technique has become more relevant as recent missions have greater navigational challenges
- Observations from 2 baselines are needed to measure both components of angular position





ΔDOR Success Rates over Several Periods

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Spacecraft	VLBI System	Period	Number of Scheduled Observations	Success Rate
Voyager	Mk II + ODA	1981 – 1984	144	67%
VEGA	NCB	1985 – 1986	62	97%
Voyager	NCB	1986 – 1987	90	62%
Phobos	NCB	1988	21	81%
Magellan (cruise)	NCB	1989 – 1990	56	86%
Galileo	NCB	1990 – 1992	48	88%
Mars Observer	NCB	1992 – 1993	66	86%
Mars Odyssey	RSR	2001	48	98%
MER1, MER2	VSR	2003 – 2004	131	98%
MRO	VSR	2005 – 2006	65	98%
Phoenix	WVSR	2007 – 2008	105	100%
MSL	WVSR	2011-2012	81	96%



Δ DOR Error Budget Development

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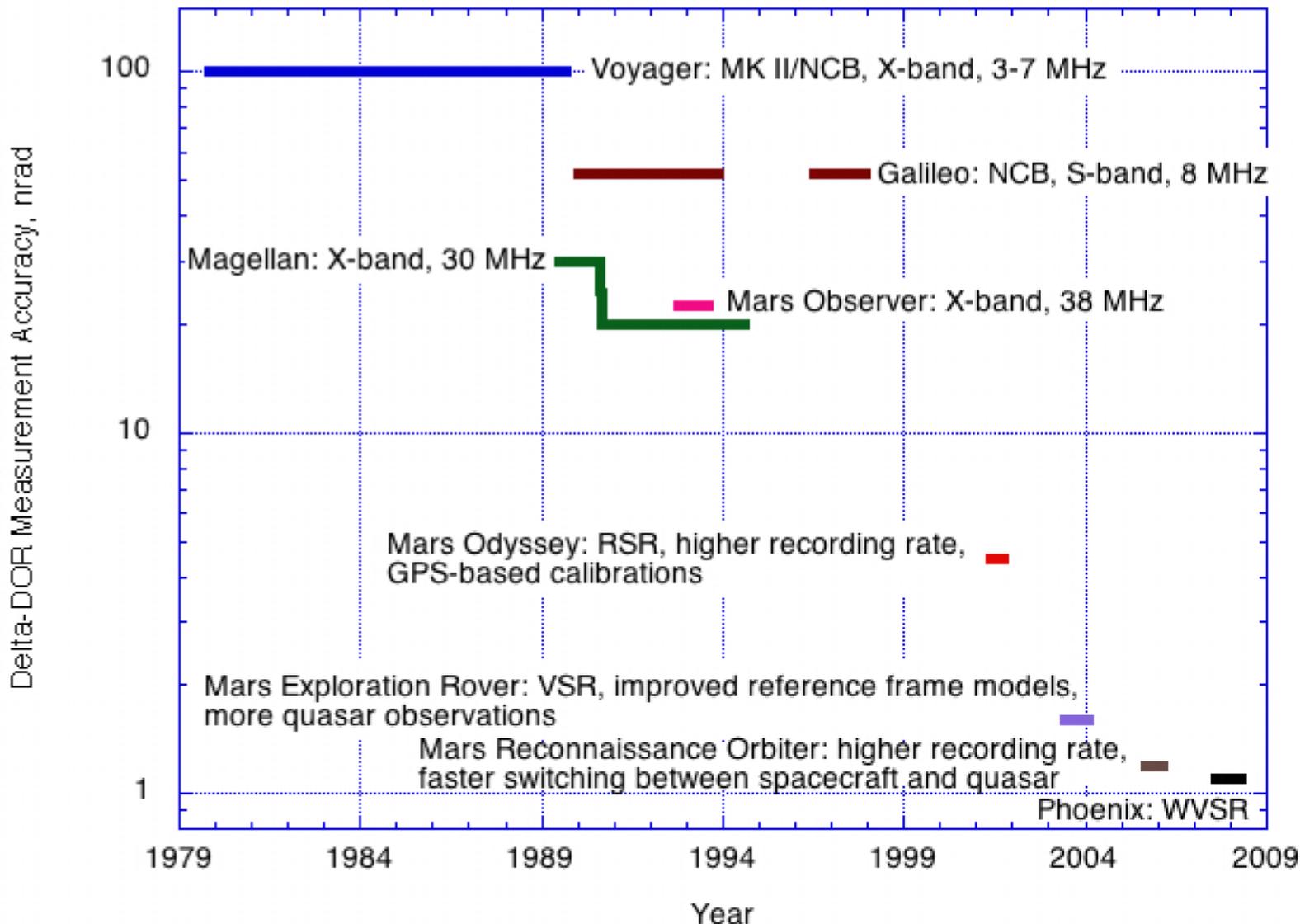
- Analyzing factors that contribute to measurement error has been an important aspect of technique
- Accuracy for early measurements was limited by thermal noise and instrumental phase
 - Due to restricted bandwidth of spacecraft signals
- These errors were reduced when ‘DOR Tones’ were added to spacecraft transponders and recording system was improved
- Focus then shifted to
 - Densify quasar catalog
 - Improve reference frame parameters
 - Improve calibrations for media and earth orientation
 - Improve observational techniques (i.e. fast switching between spacecraft and quasar)



Δ DOR Accuracy Improvements

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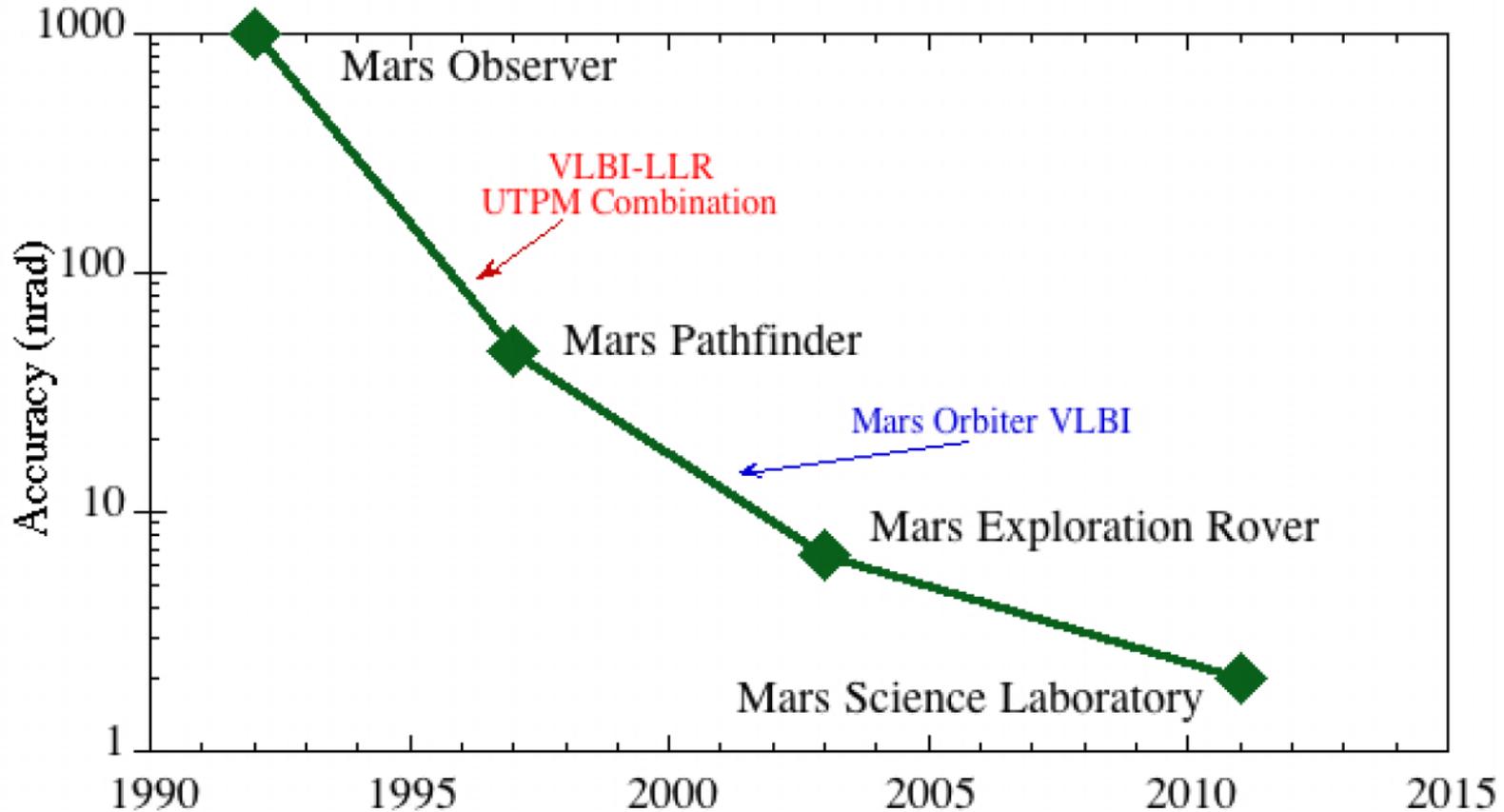


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Time History of Project Requirements for Mars Ephemeris Accuracy

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Doppler, Range, ΔDOR, VLBI, Earth Orientation, Media Calibrations, Reference Frames

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DSN Radiometric Tracking System Performance



ΔDOR: 2 nrad

Celestial
reference
frame: 1 nrad

Doppler: 0.05 mm/sec

Range: 2 m

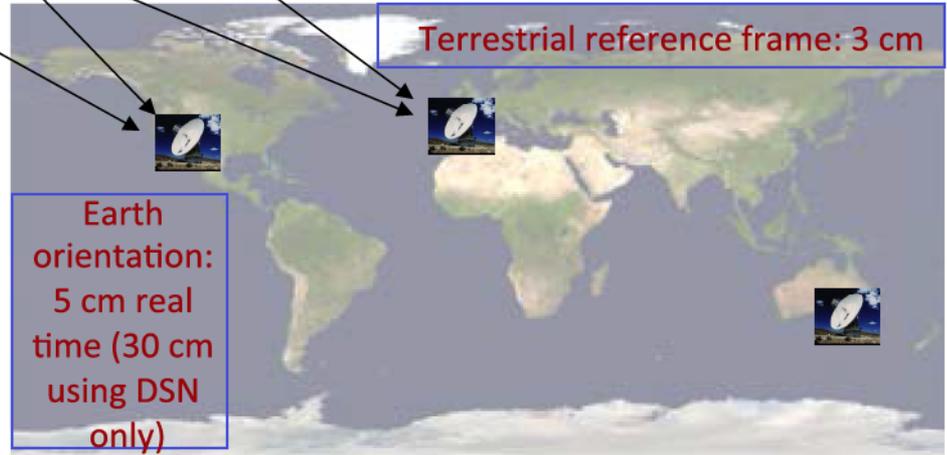


Ionosphere: 3 cm @ X-band (5 TEC)
Troposphere: 1 cm zenith



Terrestrial reference frame: 3 cm

Earth
orientation:
5 cm real
time (30 cm
using DSN
only)





Part II. Δ DOR Current Operations

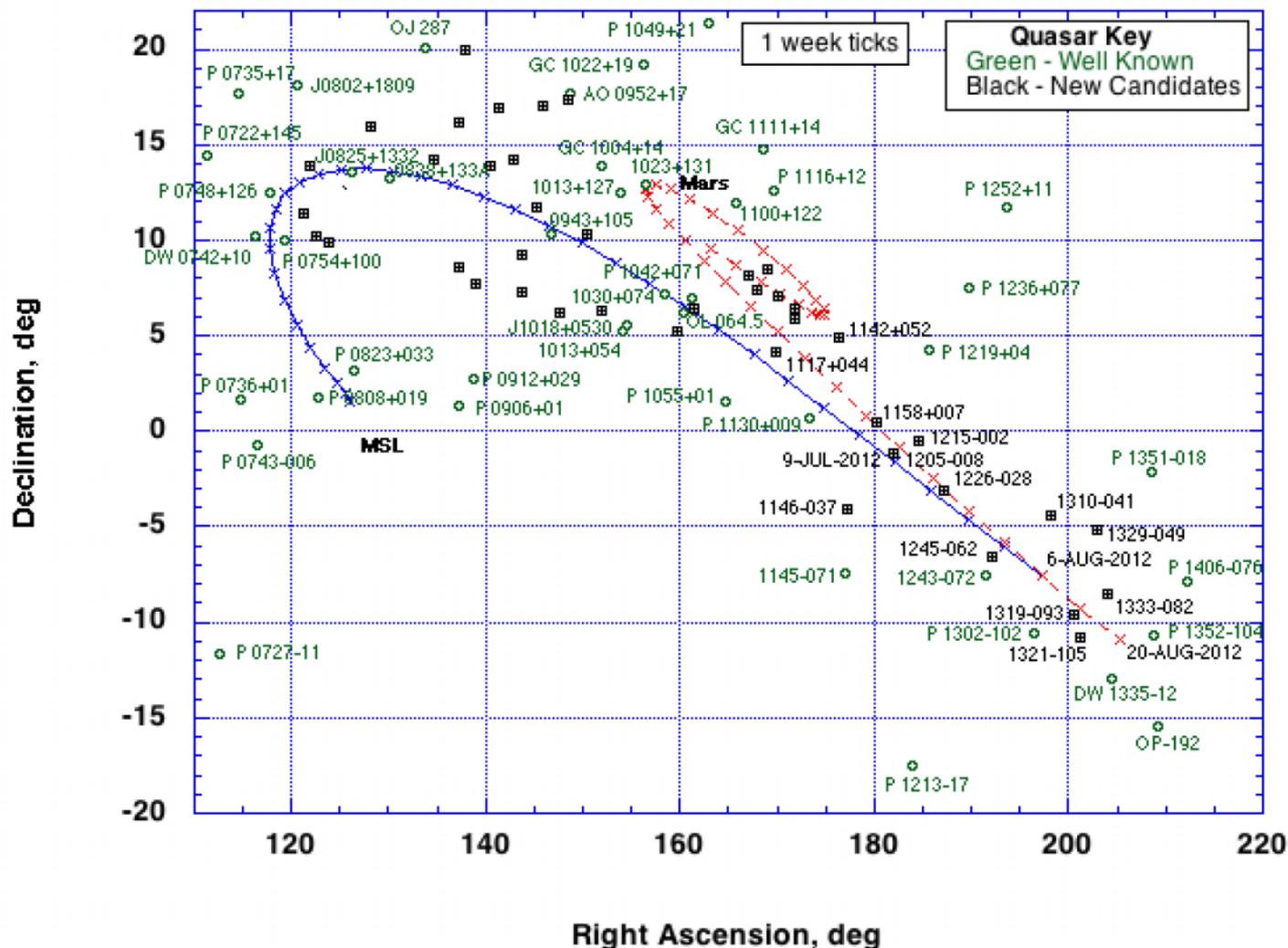
- **Mission Planning:** A strategy is developed for number of measurements to be requested, to meet navigation requirements
- **Scheduling:** Two DSN antennas are allocated during view period overlap for each measurement
- **Sequencing:** A detailed observation sequence is developed for each measurement; spacecraft DOR tones commanded on
- **Data Acquisition:** Spacecraft and quasar radio signals are recorded open loop using Wideband VLBI Science Receiver
- **Correlation Processing:** Data are transferred to a common site and correlated to develop time delay observables (typically 48 GBytes reduces to 7 time delay observables)
- **Navigation Processing:** Measurement models are developed, data are combined, and spacecraft state is estimated

Spacecraft and Quasars in Plane-of-Sky

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MSL-2011



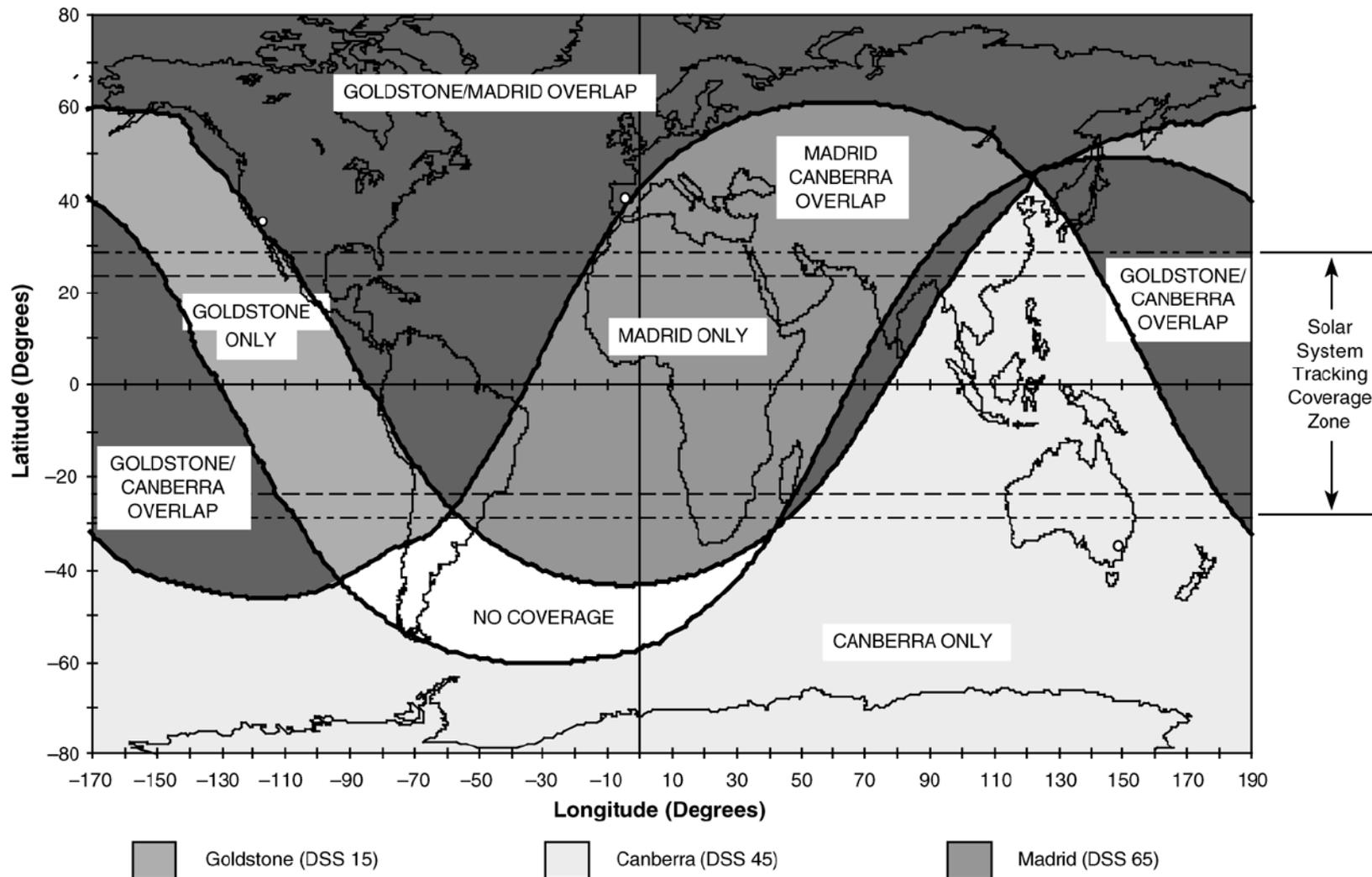
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View Period Overlaps

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- A nominal Sequence of Events is developed for all scheduled DDOR tracking passes
- For each scheduled DDOR pass, the DSN will develop a detailed observation sequence, based on spacecraft trajectory, quasar catalog, and stations to be used
- The DSN develops support products to control antenna pointing and frequency channel selection during data acquisition
- Recorded data are transferred to JPL and correlated to generate time delay observables for Navigation

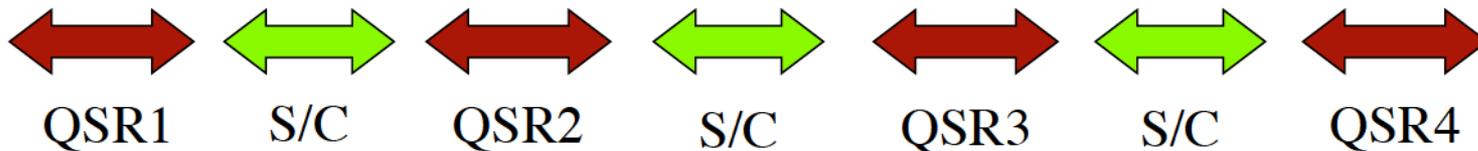


The Δ DOR Measurement

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- High rate recordings are made of frequency channels containing spacecraft signals and quasar signals
 - Spacecraft ‘DOR Tones’ provide wide spanned bandwidth
- Antennas slew from quasar to spacecraft to quasar (QSQ Sequence)
- Several QSQ sequences are recorded in 1 hour
- A time delay observable is provided for each observation of a radio source



- There are two measurement opportunities per day using DSN baselines
 - 2 baselines needed to observe 2 components of angular position
 - Madrid/Goldstone and Goldstone/Canberra baselines are used
- Navigation plans normally call for several measurements on each baseline, for each segment of cruise
 - Additional measurements for maneuvers, flybys, encounters
- Cooperative agreements with other agencies can provide additional baselines and improved geometry for Δ DOR measurements

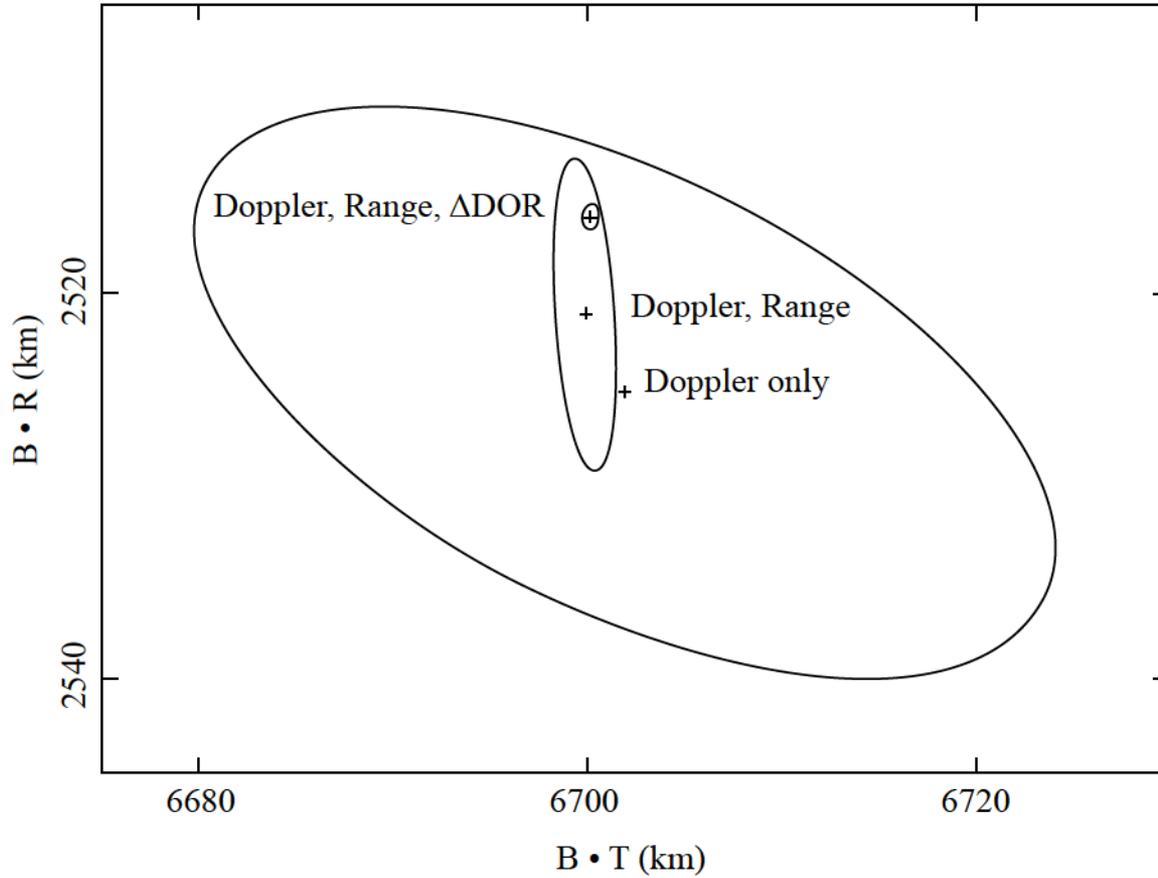


Δ DOR Benefits: Target Plane Error Ellipses for Several Data Combinations

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Mars B-plane (Mars Equatorial of Date)



Orthogonal Baselines from Combined NASA, ESA, JAXA Antennas

DSN

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