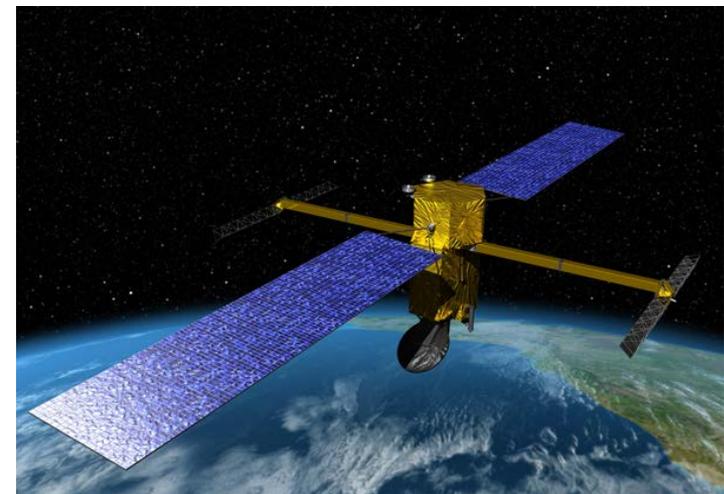


# SWOT Ground System

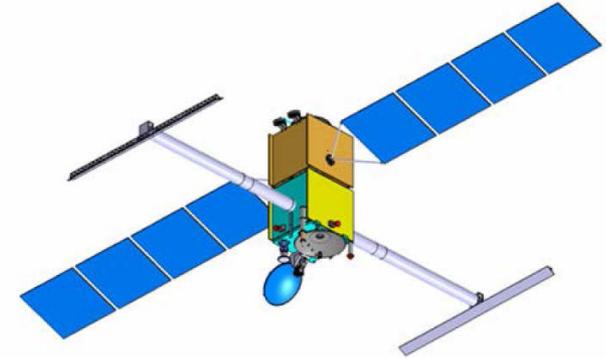
IGARSS – August 2011

Helene Vadon (CNES, Toulouse)

Phil Callahan (Jet Propulsion Lab, California Institute of  
Technology)



- Introduction
  - SWOT mission
  - Mission phases
  - SWOT ground segment overview
- KaRIN acquisition and products
  - Acquisition characteristics
  - SWOT Karin processing
  - Products
  - Sizing considerations
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- Mission

- Extraction of water surface heights over ocean and continental water bodies
- The principal instrument is **KaRIN**, a Ka-band interferometric SAR system.
- Other instruments are a nadir altimeter, radiometer, and precision tracking systems (classical Jason Altimeter suite)

- Main scientific requirements

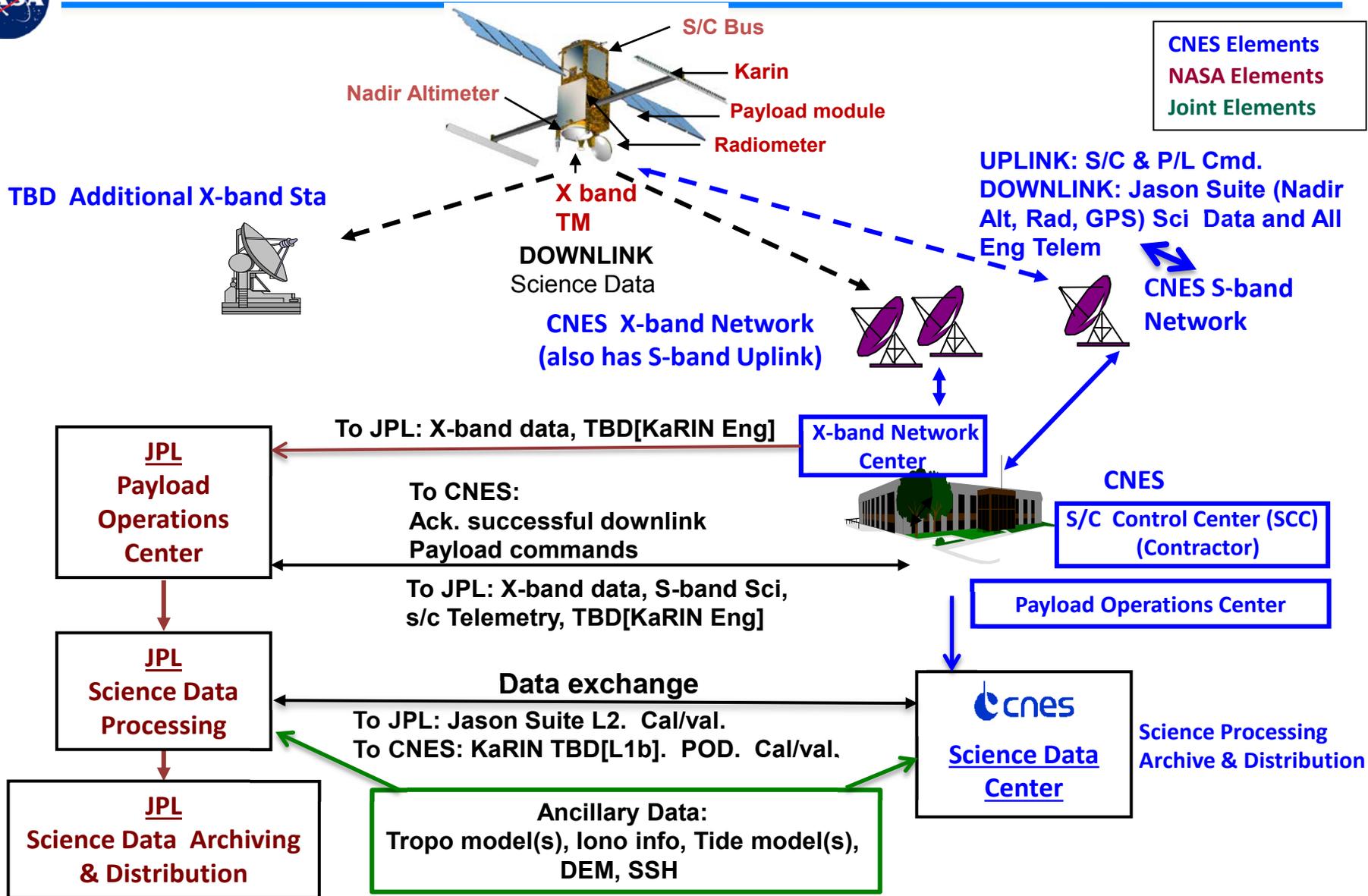
- **Oceanography**: Global coverage ( $< 78^\circ$ ) of oceans, sea surface height precision  $< 2$  cm at approximately 1 km resolution  $\Rightarrow$  LR mode
- **Hydrology**: Global inventory of rivers  $> 100$  m (50 m) and lakes  $> (250 \text{ m})^2$ , water surface height precision of the order of 10 cm (averaged over  $1 \text{ km}^2$ ), slope precision 1 cm/km (averaged over 10 km); average posting 50 m  $\Rightarrow$  HR mode

# Mission Phases

---

- Launch and Early Orbit Operations (~ 3 days)
  - Launch planned for late 2019
- Checkout / Commissioning (~ 30 - 45 days)
- Calibration / Validation (Cal/Val) (~ 90 days)
  - Fast repeat orbit at 3 (or 1) days
- Transition to Science Orbit, Operations (~ 15 days)
  - Non-sun-synchronous, 22 day exact repeat, 970 km altitude, 78 deg inclination
- Science Observations (= 3 years, starts ~4.5 months after launch)
- Disposal and Ground System Closeout (= 3 months)

# Architecture Diagram





# Ground System Elements (1 of 2)

---

- Tracking Stations
  - CNES S-band stations: Downlink for Nadir Alt, Radiometer, DORIS, GPS (“Jason Altimeter Suite”, JAS). Uplink commanding
  - X-band network (620 Mbps (nominal, min)): Downlink for high rate science (KaRIN), and associated Engineering data
- Front End Networks
  - Network from X-band Data Center(s) to JPL and CNES operation centers. Minimum data rate to return all downlink within approximately 20 mins is  $>\sim 300$  Mbps (similar to downlink)
  - Network from CNES to JPL for S-band science and engineering data
- Spacecraft Control Center (SCC, CNES/Contractor)
  - Real time monitoring of s/c and p/l via S-band housekeeping telemetry



# Ground System Elements (2 of 2)

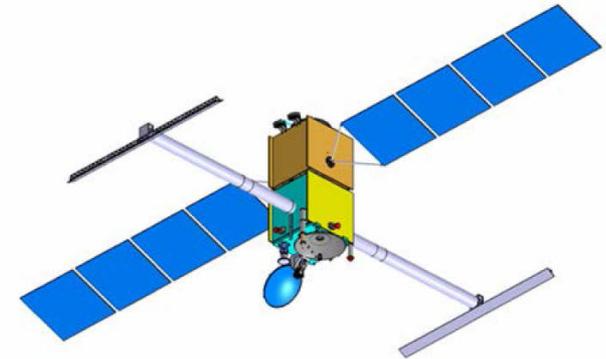
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- Payload Operations Centers (POC)
  - JPL: Near Real Time health monitoring of KaRIN, GPS, Radiometer. Determine X-band data quality, need for replay.
  - CNES: Near Real Time health monitoring of Jason Altimeter Suite
- Science Processing Centers
  - JPL: Processing of KaRIN, POD
  - CNES: Jason Suite, POD
- Data Archiving and Distribution Centers
  - JPL: PODAAC
  - CNES: AVISO
- Project Data Distribution Networks
  - JPL-CNES link
  - Science Processing Centers to Archiving/Distribution Centers

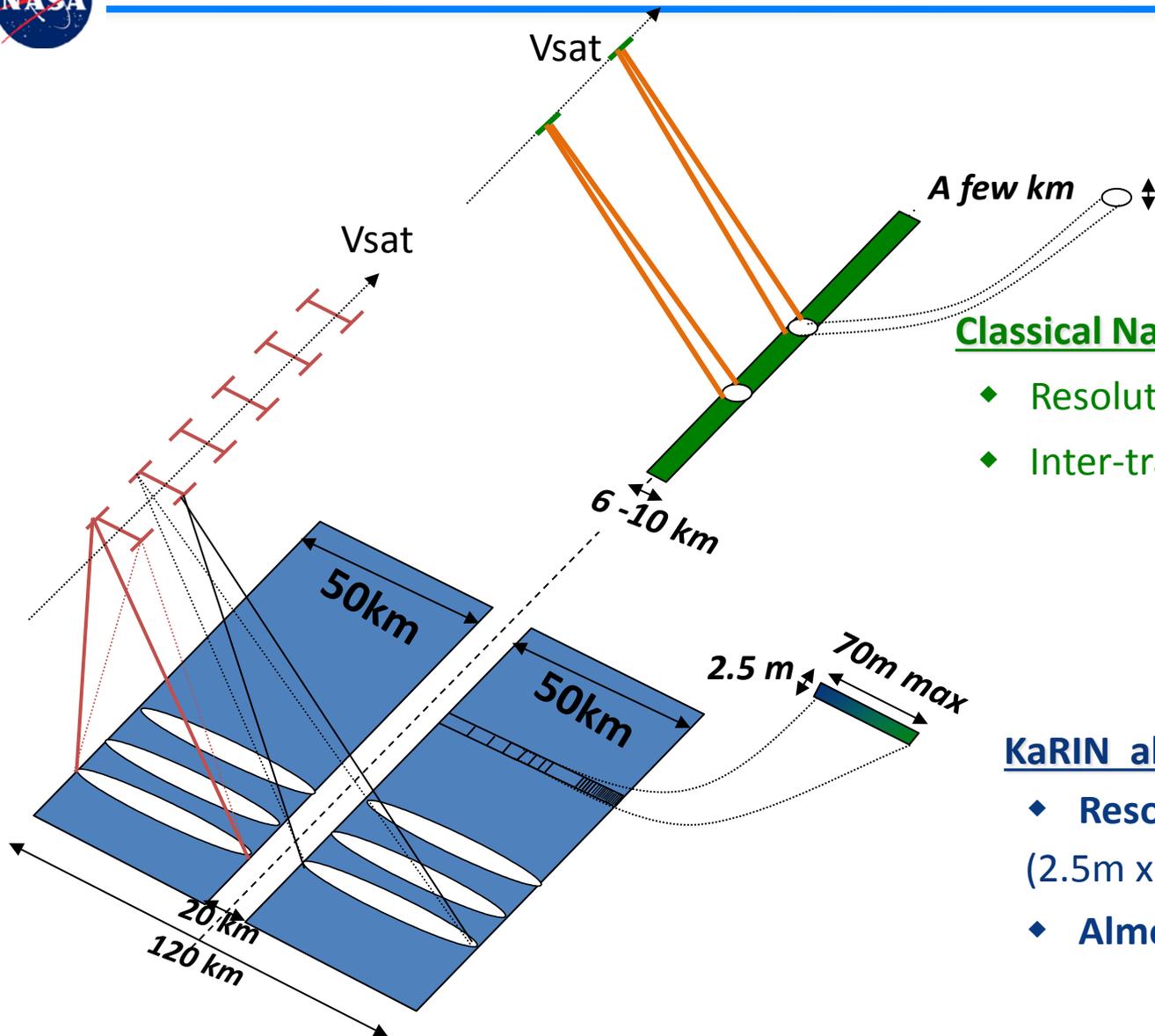


# Outline

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# KaRIN Data Acquisition



## Classical Nadir Altimetry

- ◆ Resolution = Swath ( ~ 10km)
- ◆ Inter-track at equator ~ 100km

## KaRIN altimetry

- ◆ Resolution << Swath :  
(2.5m x 10m to 70m) << ~100km
- ◆ Almost full coverage

# KaRIN Processing Steps

---

- SAR processing (range and azimuth compression)
- Interferometric processing (co-registration, computation of interferometric phase and coherence)
- Application of “acquisition geometry” in a wide sense (geolocation, precise orbit determination, correction of roll, baseline variations, tropospheric delay, ...)
- Estimation of geophysical parameters (water surface detection (HR only), computation of water surface heights, slopes etc.)
- Establish/refine floodplain DEM (yearly or whole mission)
  
- High-level and multi-temporal processing is a science teams activity



- Level 0 ( ~ 1 TB/day)
  - Ocean KaRIN: raw data (L1 (partial interferograms) processing onboard)
  - Hydro KaRIN: raw data
  - Nadir altimeter and Radiometer data: instruments raw data
  - GPSP and DORIS data: POD raw data
  
- Level 1: Nadir Alt, Radiometer, KaRIN ( ~ 12 TB/day)
  - L1a: Engineering unit conversion, calibration data separated for additional processing
  - L1b: Final instrument level data with calibrations appended
    - Ocean: 9-11 onboard interferograms (1km x 1 km), correlations, amplitudes with instrument calibrations and phase variations corrected combined
    - Hydro (HR): KaRIN phase flattened Interferograms
    - Altimeter, radiometer compliant with Jason-2 Science Data record processing (SGDR)



- L2A

- Ocean KaRIN : As L1B + Geophysical corrections. Nadir data merged for cross calibration.
- Ocean Crossover Attitude Correction: Estimate attitude errors from ocean crossovers for correcting both ocean and land data.
- Intermediate Hydro: L1B + water mask detection; rain, snow, ice/frozen detections

- L2B (standard distributed products)

Remark: Every field will be associated with an uncertainty estimate

- Ocean:
  - Sea surface (SS) heights in latitude/longitude grid
  - SS slope at each point of the grid
  - Sigma 0
  - Wind speed
  - Ocean Sea Wave Height on same grid (goal)

- L2B (standard distributed products) (continued)
  - Remark: Every field will be associated with an uncertainty estimate
  - Hydro:
    - Geo-localized water mask (lakes > 250 m<sup>2</sup>, rivers >100 m)
    - Estimated water elevations (same sampling as mask) -> Triangular interpolated network (+ uncertainties)
    - Topographic map of the flood plain surrounding the water surface and channel cross sections (requirement to be produced by the end of the mission)
  - Ocean Nadir: Similar to current Jason-2 SGDR
- NRT (goal)
  - Ocean: as Level 2 products but with lower accuracy (degraded orbit and calibration using past cross-overs)
  - Hydro: Level 1 products (SLC + interferogram), no calibration data, use of ancillary DEM for phase flattening
  - Nadir products: OGDR/ IGDR products (see Jason-2 OSS)



- POD – determined from GPS and DORIS
- Ancillary Data – updated daily
  - Atmospheric models: Dry Tropo (surface pressure), Wet Tropo (water vapor)
  - Input to ionospheric model
- Geophysical Models – updated infrequently
  - Mean Sea Surface, Geoid
  - DEM, Water mask (a priori, then self derived)
  - Elastic Ocean Tides, Earth Tides, “Pole Tide”



# Sizing Considerations

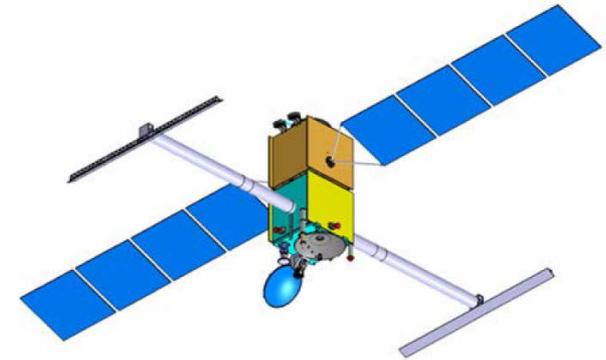
- Global coverage, continuous acquisition
- Acquisition hypothesis
  - Sea 70% and Land 30% =>

In the average orbit, KaRIN will acquire 67 GB of data to be transmitted to ground=> **~1 TB /day** (~ 99% HR data)
- Archive Plan
  - Nominal mission duration: 3 years
  - Long term archive of (main contributors in terms of volume) Raw HR + SLC (L1A) + Interferograms + masks (L2A) + geo-localized hydro products (L2B) => **~10 000TBytes**
- Comparison to other missions
  - CNES: more than **5** times Pleiades satellite archive (dual civilian/defense optical imagery, to be launched soon)
  - NASA: similar to proposed DESDynI (Earth deformation and sea ice from L-band Radar repeat pass interferometer)



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# General Levels of Commonality

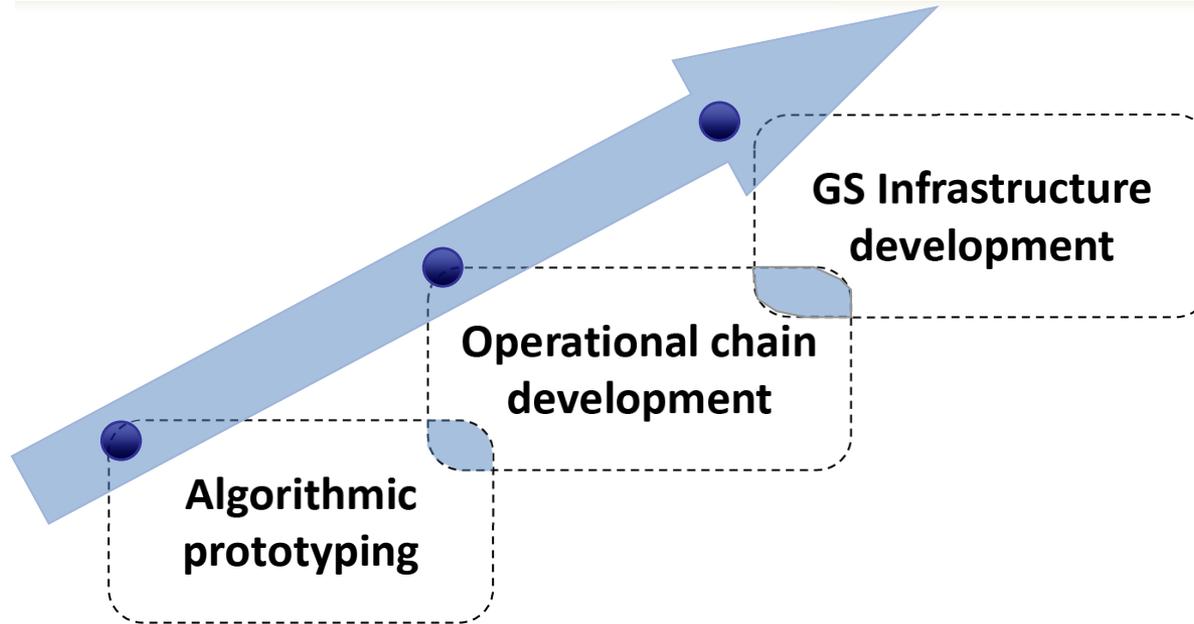
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- Commonality between two processing systems may be achieved at different levels
  - Commonality of the **full production system** (clone, e.g. Jason)
    - Prevents existing systems reuse. Not necessary for SWOT development
  - Commonality of the **algorithms, software, and products**
    - Products are guaranteed 100% similar. May be the right level.
  - Commonality of **specification and reference data**
    - Independent developments. Risky because reference data does not cover all cases so products will not be 100% similar.

- Science processing algorithms will be developed primarily by supplier of each instrument
- Algorithm flow will be iterated within entire development team
- Data product definitions will be iterated with Science Team
- Algorithm testbed will be set up and prototype processing developed
  - Testbed will read and write specified products
  - Test data will be (or emulate) specified products
- After initial development cycle, algorithm specifications will be written for review by System Engineering team and selected subset of Science Team



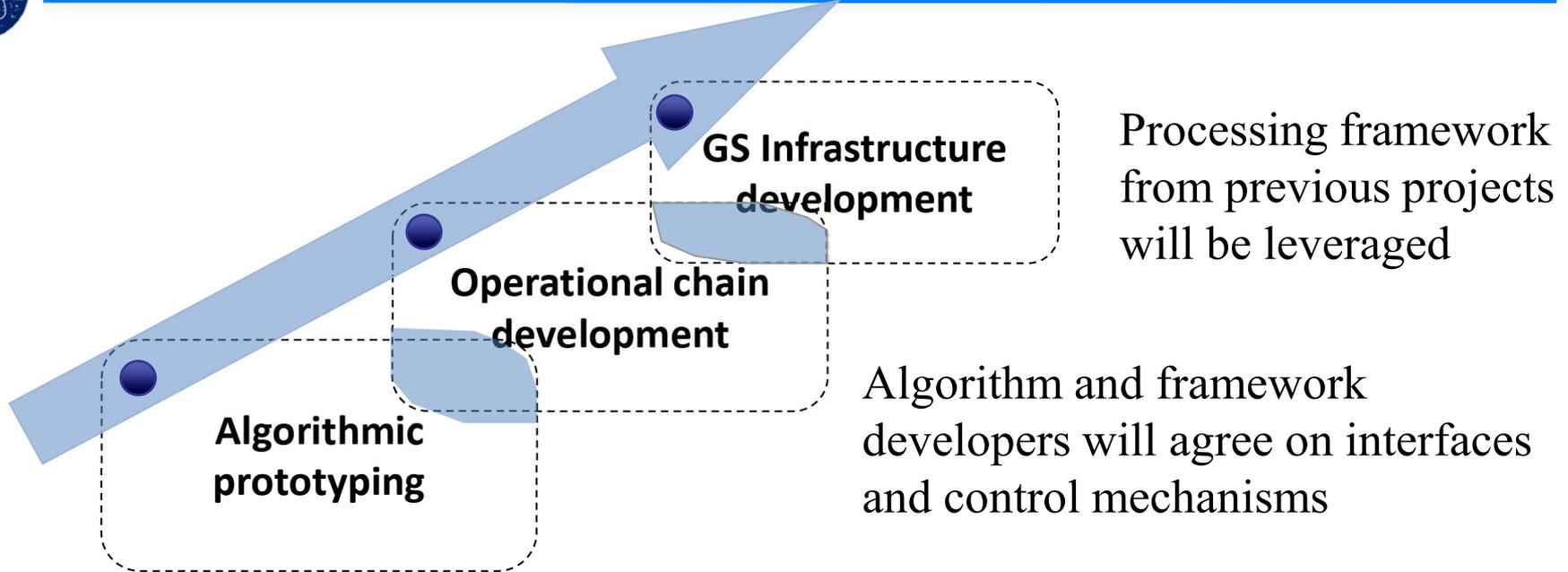
# CNES Usual Development Approach



- Algorithms prototyping activities and operational chain development are relatively independent
  - ◆ The scope of prototype development may be larger than scope of operational system development
  - ◆ But not true the other way: prototypes are needed to build the reference data for the validation of the operational chain



# JPL Usual Development Approach



- Testbed prototype will incorporate as much of the processing framework as possible
- **Notes:**
- NRT processing will be based on standard algorithms but may not use the processing framework
- Existing POD processing will be implemented in project environment



# Commonality Approach for SWOT

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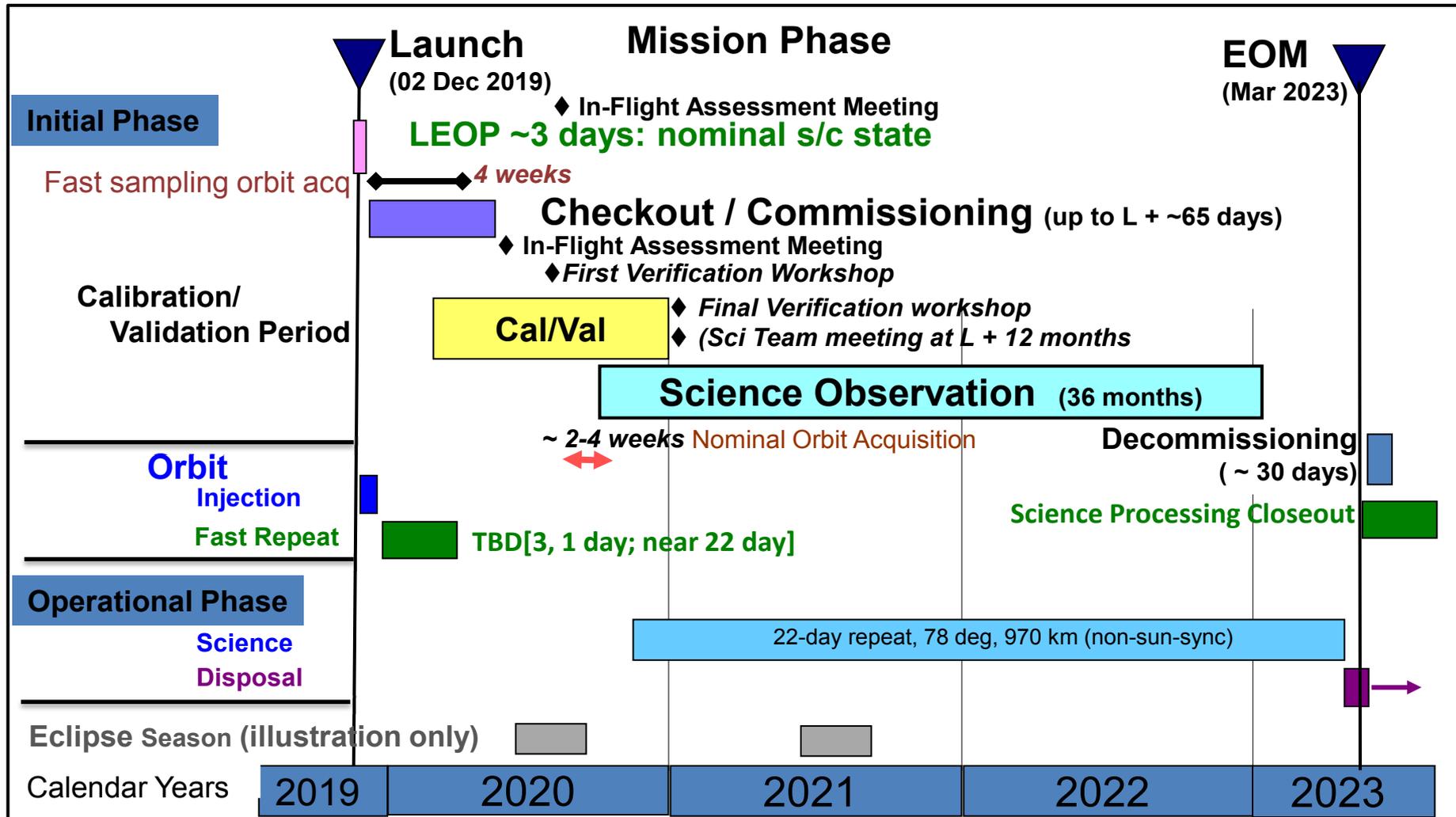
- Complementary prototyping activities will be conducted by JPL and CNES
- The order and nature of the processing steps of the operational LR and HR processing chains will be discussed and agreed on by JPL/NASA and CNES
- The detailed work share is not yet defined, but the operational algorithms and related binary code will be common

# BACKUP



# SWOT 40 month Mission Timeline

Backup or here ?





# Science Operations (1 of 1)

- Science data collection phase starts ~ 4.5 months after launch. Collect data for 3 full years in observational orbit
- Full validation period: 6 months of data in science orbit + 2 months processing, internal review
  - Science Team meeting to assess initial data products ~ 12 months after launch
  - Processing update based on Science Team assessment implemented ~ 2 months after meeting
- Ready to reprocess all data with updated system ~ 16 months after launch.
- Science Requirements on Reprocessing (assume “caught up” at 24 months) puts requirements on total Processing Center throughput (Suggest project aggregate > 5x). Budget/scenario:
  - Cal/val data (~3 months) – as science phase starts
  - Initial science phase data (~12 months +3 months cal/val) – within TBD [ $< 3$ ] months. Result: at ~20 months after launch version 2 of all usable data available. Support Science Team meeting at ~24 months to validate data
  - End of nominal mission: Reprocess all data a second time with SDS update done from ~ 30 – 36 months after launch; takes 6 – 8 months (runs until 3 months after end of mission [assumes no extended mission])