

# Active Optics for UV/Vis/IR Space Telescopes

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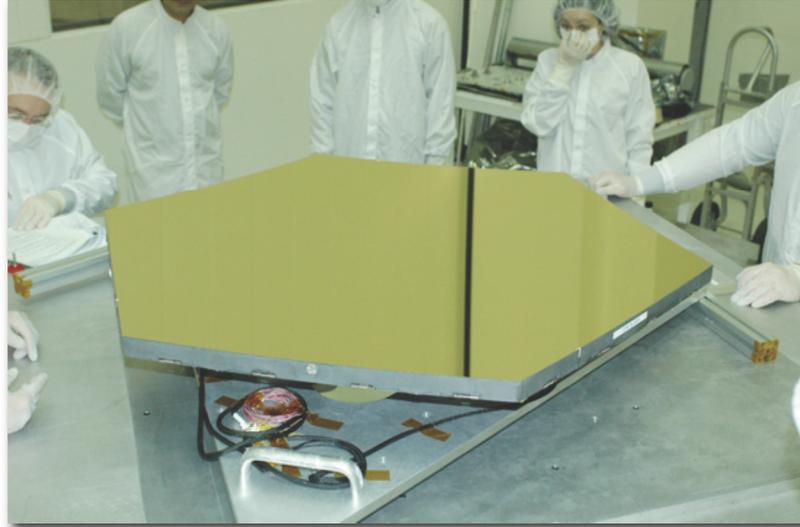
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- 1. Jet Propulsion Laboratory, California Institute of Technology**
- 2. Adaptive Optics Xinetics**
- 3. Northrup Grumman Aeronautical Systems**

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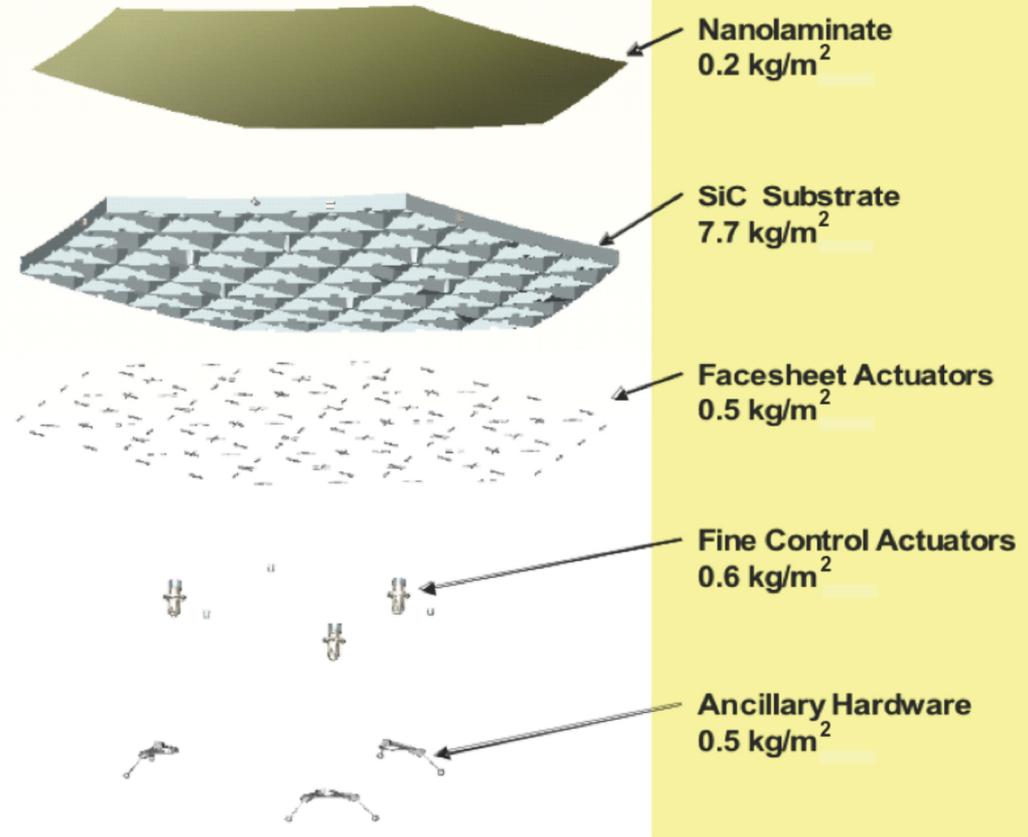
# Active Optics for Space Telescopes



- **Active Optics: Mirrors that can be reshaped after launch; and the Wavefront Sensing and Control system to command them**
  - *Reduce mission risk*
    - Correct any optical problem that might arise
    - Enable testing to spec during system assembly and integration
  - *Reduce mission cost*
    - Reduce mission mass
    - Relax fabrication and assembly tolerances
    - Speed up Assembly, Integration and Test phases

# Actuated Hybrid Mirrors (AHMs)

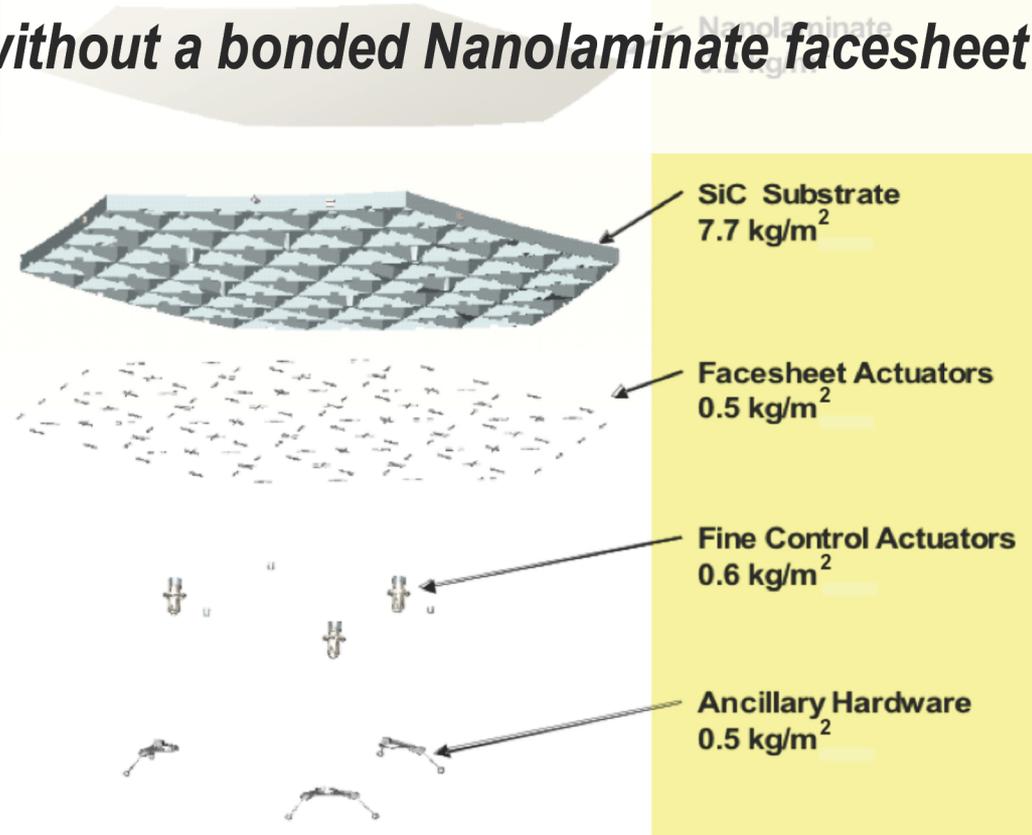
- **AHMs are large mirrors**
  - PMs or PM segments
- **Nanolaminate facesheet**
  - Multilayer metal foil, made by sputter deposition on a super-polished mandrel
- **SiC substrate**
  - Reaction-bonded Ceraform SiC is cast in a mold, fired, then bonded to facesheet
- **Electroceramic actuators**
  - Surface-parallel embedded actuators give large stroke and high accuracy
- **AHMs are *low mass and high strength***
  - Areal density < 20 kg/m<sup>2</sup> including electronics for meter-class AHMs
- **AHMs are *made by replication* for high optical quality and low cost**



# Polished SiC Mirrors

**Polished SiC mirrors are AHMs without a bonded Nanolaminate facesheet**

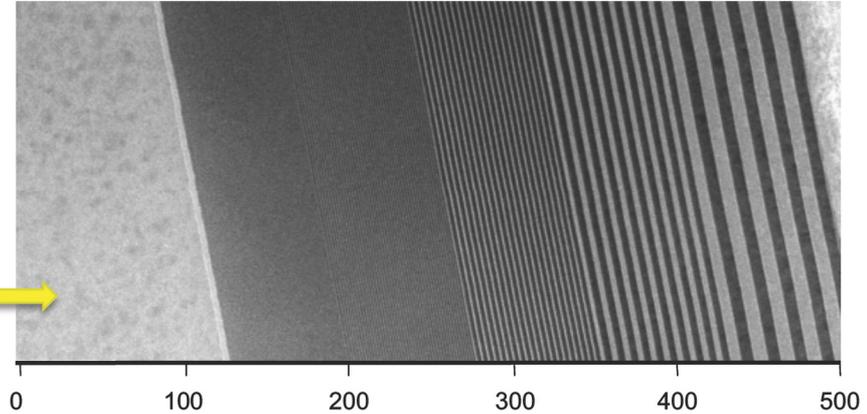
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- **Polished SiC mirrors are also low mass and high strength**
- **Polished SiC mirrors can be joined to create very large mirrors**
- **Polished SiC mirrors can be used at cold or even cryo temperatures**

# Nanolaminate Properties

- **Nanolaminates: multilayer solids with high interface concentration**
  - Have been made from 72 materials
  - Amorphous/crystalline layers for AHM
- **X-ray optic example has layer thicknesses from 0.4 nm to 32 nm**



- **AHM nanolaminate layers:**
  - A few Å of C for release layer
  - Au layer for outer surface
  - Conventional coating applied
  - 446 periods of:
    - 42 nm crystalline Zr layer
    - 3 nm amorphous Zr/Cu layer
- **Finished 1.52 m nanolaminate being removed from chamber**

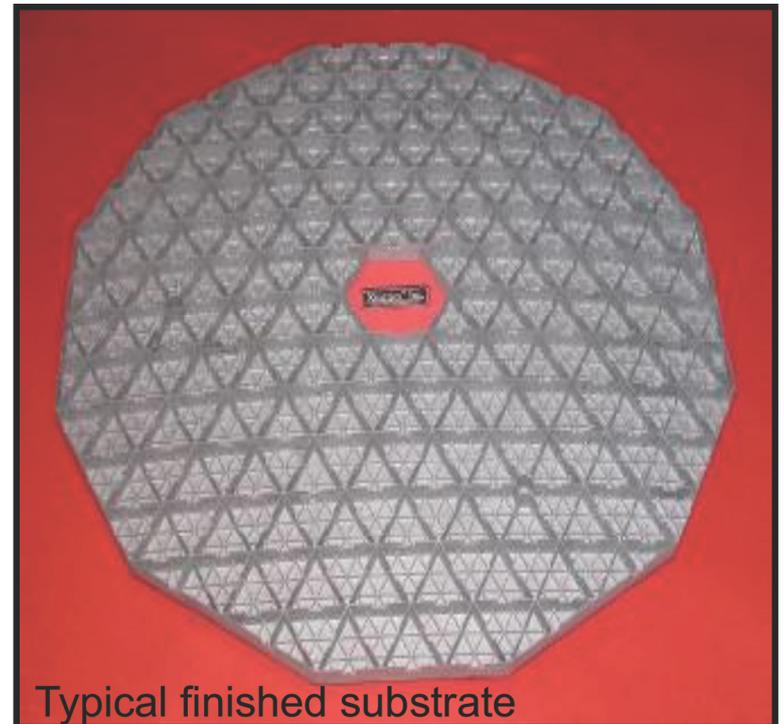
# Ceraform Silicon Carbide

- **Ceraform SiC:**

- Fugitive core foam mold created by CNC machining
- SiC nanopowder slip fills mold
- Part is freeze-dried
- Mold core is leached out
- First firing creates green state part
- Part is machined
- Second firing to full hardness

- **Final rough grind of SiC front surface matches the curvature of the mandrel/nanolaminate to  $\pm 5 \mu\text{m}$**

Property	Units	Aluminum	Beryllium	SiC	ULE	Desire
$\rho$ , Weight	g/cm <sup>3</sup>	2.71	1.85	2.95	2.21	Low
E, Stiffness	GPa	68.3	303	364	67.6	High
E/ $\rho$ , Specific Stiffness	KN-m/g	25	164	123	31	High
$\sigma/\rho$ , Stress Loading	N-m/g	46	11	24	3.2	High
$\alpha$ , Thermal Soaks	ppm/ $^{\circ}\text{C}$	22.7	11.4	3.38	$\pm 0.03$	Low
$\Delta\alpha$ Homogeneity	ppb/ $^{\circ}\text{C}$	100	100	30	10	Low
K/ $\alpha$ , Thermal Gradients	MW/m	6.9	19	51	44	High
K/rCp, Thermal Diffusivity	m <sup>2</sup> /s	6.55	6.07	8.7	0.08	High
K/ $\alpha$ E, Thermal Stress	MW-m/N	101	63	140	646	High



# Joined SiC Mirrors

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- **SiC substrates can be joined using brazing or bonding techniques, and then polished, to make very large, active mirrors**
  - 4 m or larger, using existing SiC fab infrastructure
  - Directly polished to  $<20\text{\AA}$  surface roughness
  - Superpolishable to  $<5\text{\AA}$  after Si cladding
  - With or without central hole
  - Can be used at cold temperatures

# AHM and SiC Mirror Technology Status

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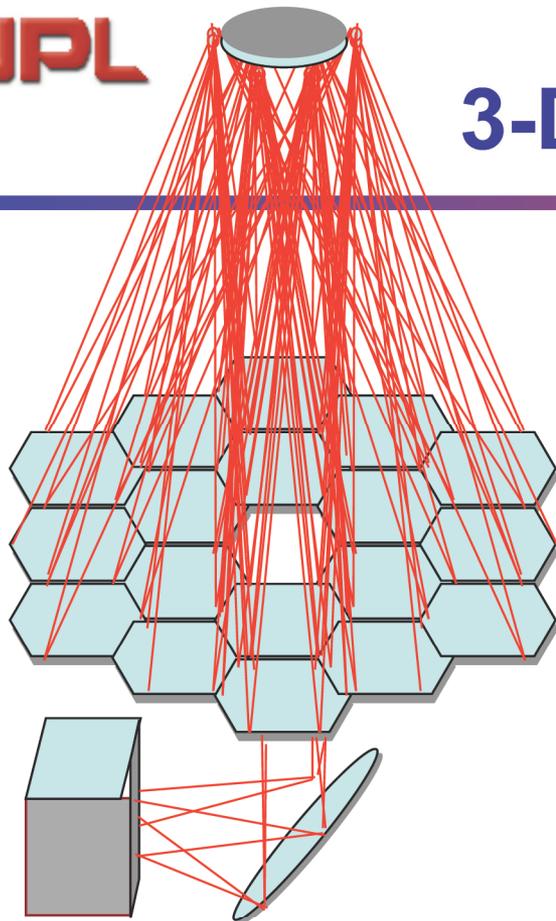
- **AHM mirror technologies are maturing rapidly**
  - To do: grow to larger sizes (1.8m, 2.5m, e.g.)
- **Very large polished SiC mirrors offer benefits in some cases**
  - Large monolithic primary mirrors (4m or larger)
  - Cold or cryogenic mirrors, active or not
- **Very large polished SiC mirrors require some further technology development**
  - To do: Lightweight mirror segment joining
  - To do: Low-stress Si cladding
  - To do: Superpolishing
  - To do: Cryogenic active mirrors, using actuators to correct cool-down stresses and avoid costly cryo-null figuring
- **Other active optics technologies needing development**
  - To do: “Self-sensing” for <10pm WFE stability
  - To do: Continuous, pm accuracy WFS for internal coronagraph or lensing applications

# Wavefront Sensing and Control

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- **Wavefront sensing and control methods are well established**
  - JWST
  - Active mirror testbeds
- **Proposed exoplanet-specific WFSC methods need further development**
  - Continuous pm-level WFS
  - Mirror self-sensing methods
- **(Details in backup charts)**

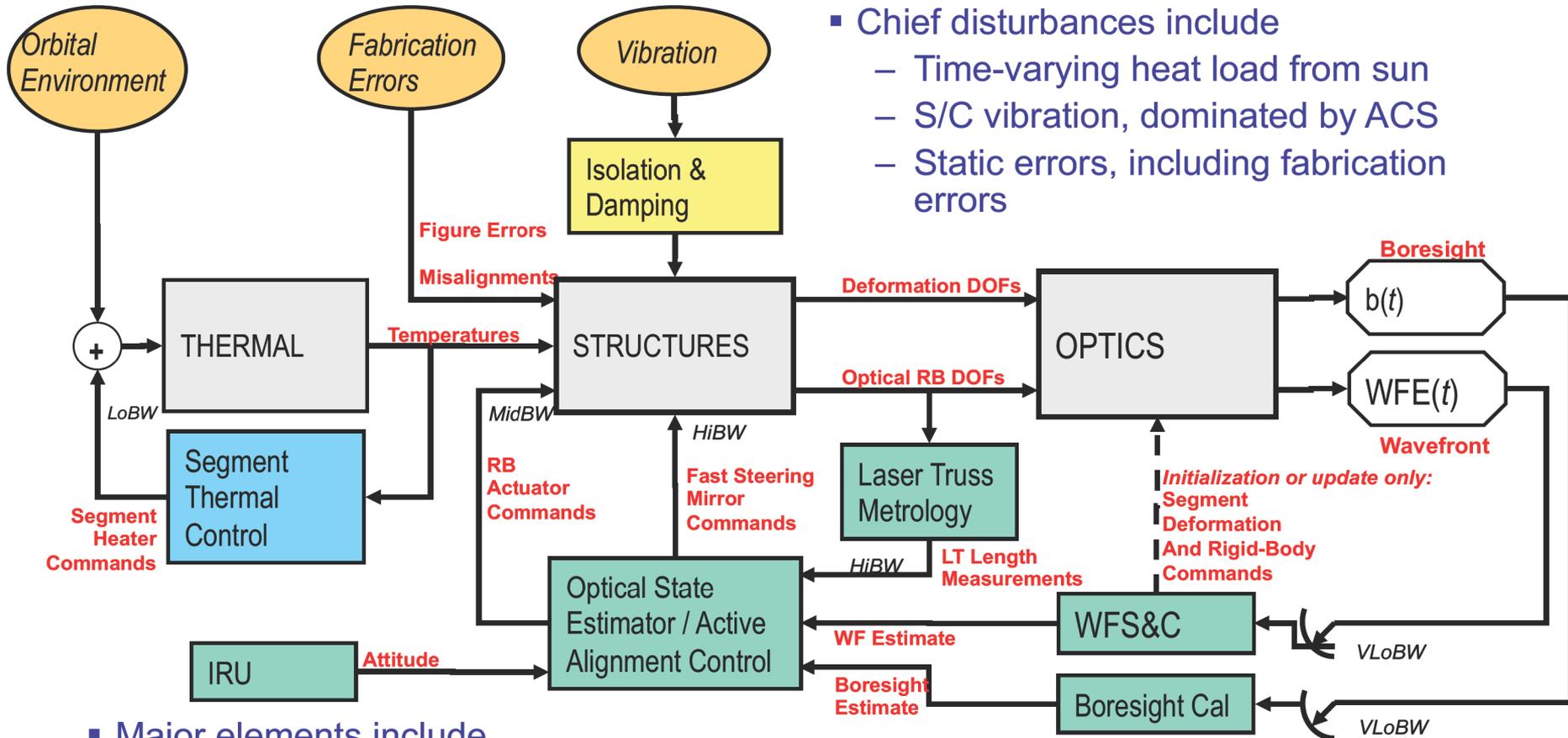
# 3-Dimensional Laser Truss



- **Uses Laser Distance Gauges (LDG)**
  - 6 LDGs per segment measure all relative RB DOFs in the entire OTA
    - All PM segments, the SM, FF, TM and OBA
  - The IRS is attached to the OBA, providing measurements of 6 more absolute DOFs wrt inertial space
- **Same measurement equation:**  $\delta = Cx$ 
  - Sensitivities computed from model kinematics

- **Measurement is invertible:**  $x = C^{-1}\delta$  is full rank
- ***Optical State Estimator* uses a Kalman Filter to estimate the RB state**
  - Balances measurement vs. prior knowledge for optimal estimate
  - Predicts WF and Boresight from state estimate
- **Feedback control using RB actuators and optimal control laws keeps performance in spec**
  - Integrated model will be used to evaluate performance

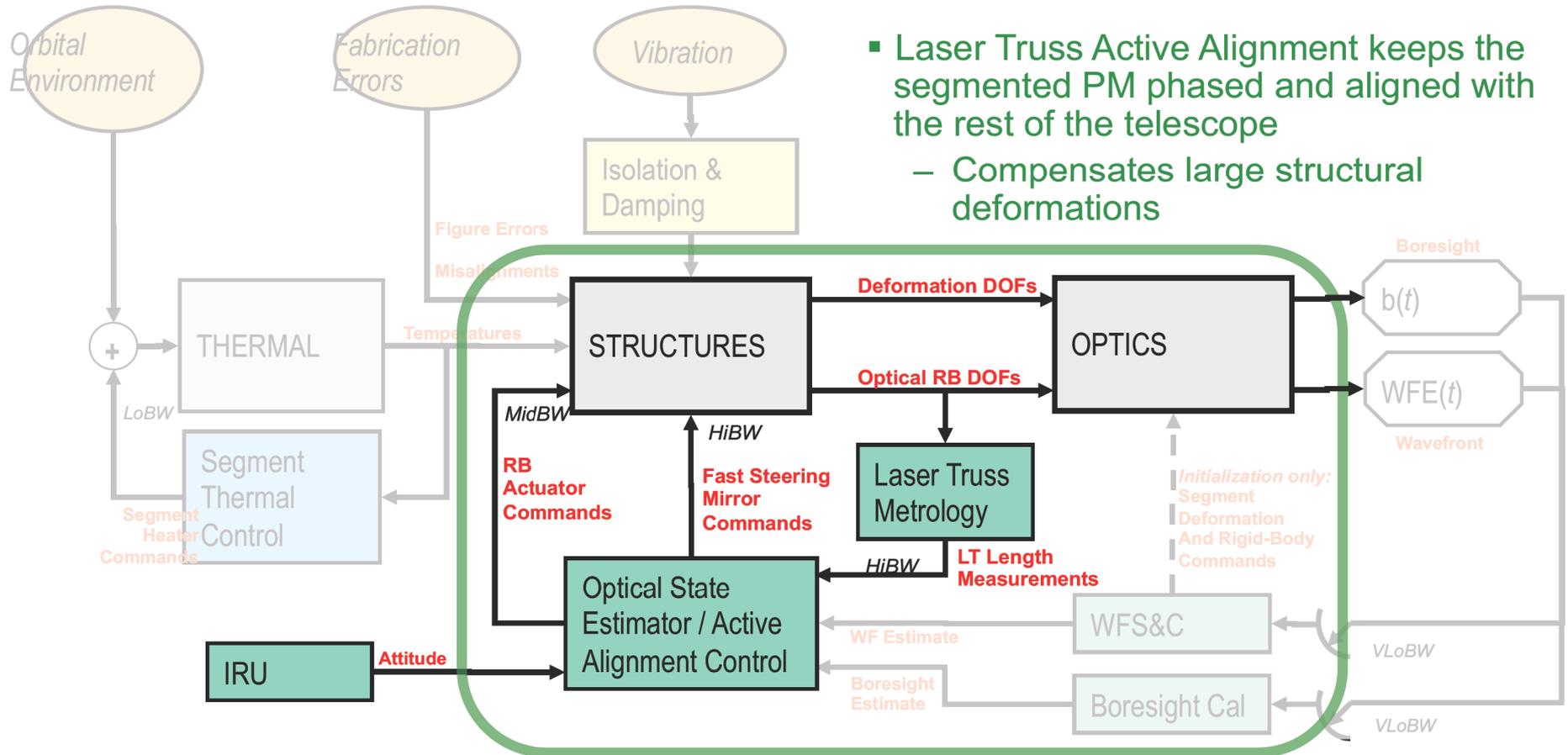
# Block Diagram



- Chief disturbances include
  - Time-varying heat load from sun
  - S/C vibration, dominated by ACS
  - Static errors, including fabrication errors

- Major elements include
  - Wavefront Sensing and Control
  - Laser Truss Active Alignment: active WF compensation and LOS pointing control
  - Segment Thermal Control to stabilize optical figure
  - Isolation and Damping to attenuate vibration disturbances

# Laser Truss Keeps All Optics Aligned



- Laser Truss Active Alignment keeps the segmented PM phased and aligned with the rest of the telescope
  - Compensates large structural deformations

- Laser Truss measurements at high BW are processed in a Kalman Filter to estimate the perturbation state of all the optics
- Estimated state is fed back to control WFE at low BW and boresight at high BW

# Laser Truss Pros and Cons

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## • Pros

- High accuracy –  $< 1$  nm per LDG when  $\Delta$  angle is small
- Observes all important RB states – including Primary and Secondary Mirrors, and Optical Bench
- Low drift – with 1 laser feeding all LDGs, require WFS update once per day
- Light weight beam launchers
- No on-segment power dissipation
- Does not require segments to be close together
- Does not require any particular gap geometry
- Works with missing segments (no degradation for the segments that remain)
- Useful for I&T
- Degrades gracefully if individual LDGs go out

## • Cons

- Requires 12 fibers into each segment for 6 DOF

# Conclusion

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- **AHMs provide high-quality, low-mass large optics**
  - Polished SiC mirrors promise the same advantages for very large mirrors, or cold mirrors
- **Active optics compensate typical space telescope errors to reduce mission risk and cost**
  - 10x to 300x for low-order errors, depending on actuator count
- **Active optics relax fabrication and assembly tolerances system-wide, lowering cost**
- **Active optics permit testing to spec performance on the ground, at multiple stages of assembly, without complex GSE**

# BACKUP

# Actuators

*Sintered body*

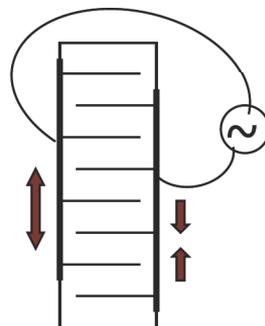


Active PMN Layer  
Thickness : 100 –152  $\mu\text{m}$

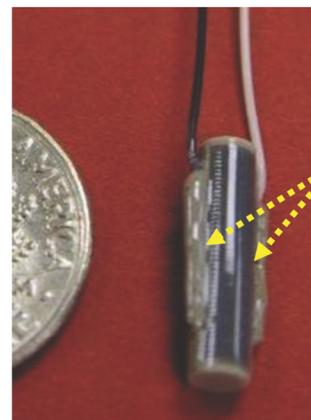
Pt Electrode Layer  
Thickness : 2-4  $\mu\text{m}$

# of active layers:  
100 - 200

*Electrical Connection  
(conceptual)*



*XiRE 0313 Photo,  
XiRE 0416 similar*



Conductive polymer

Top surface:  
Conformal coating

- **NGX actuators use PMN-PT electrostrictive ceramics**

- Multiple layers of ceramic and conductive electrode are co-fired to form a solid body
- Conductive polymers for external electrode and wire bonding (no soldering)
- Conformal insulating polymer coating

- **High stroke, low voltage**

- $\pm 2.5 \mu\text{m}$  stroke at 20C
- 0-100V operating range

- **Used for astronomical Deformable Mirrors**

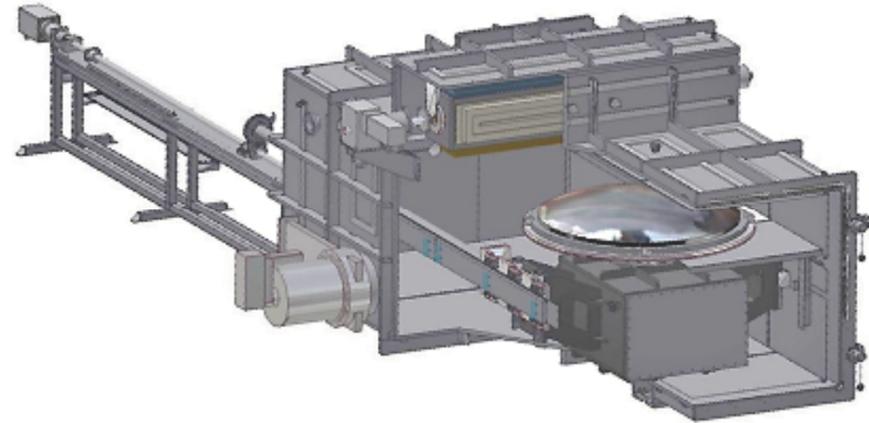
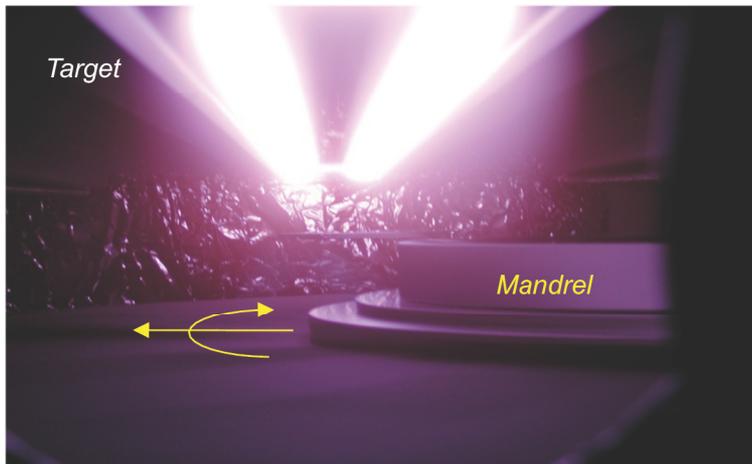
- High reliability

*Actuator with Mounting Tabs*



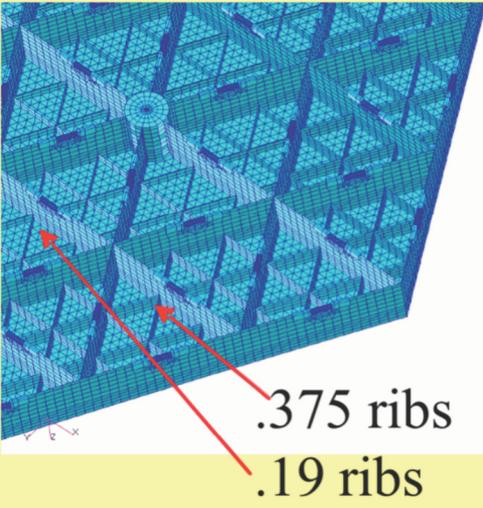
# Nanolaminate Facesheet

- **AHM nanolaminates are made at LLNL, in the Very Large Optic Coater (VLOC)**
- **Mandrel is a nanoclean, superpolished glass tool with figure opposite to final AHM**
- **Mandrel is translated and rotated under “targets:” the deposition sources**



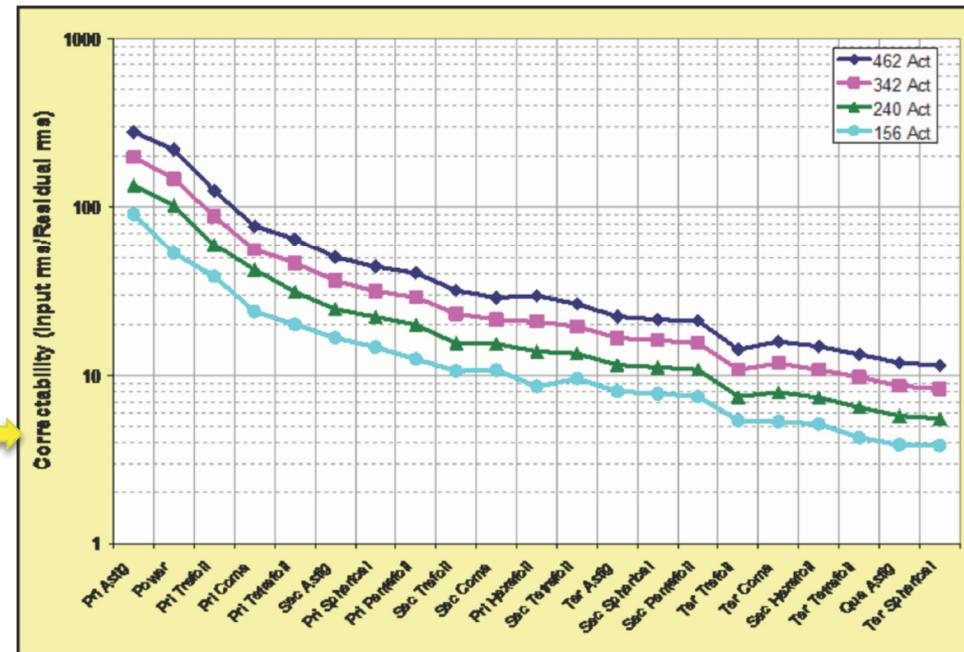
- **Magnetrons create Ar+ plasma to drive atoms off the targets and onto the mandrel**
- **Switching between multiple targets creates multilayers**
- **Nanolaminate uniformity and strength assured by: ultra-stable processes**
- **Nanolaminate surface smoothness replicates mandrel**

# Substrate Design Considerations

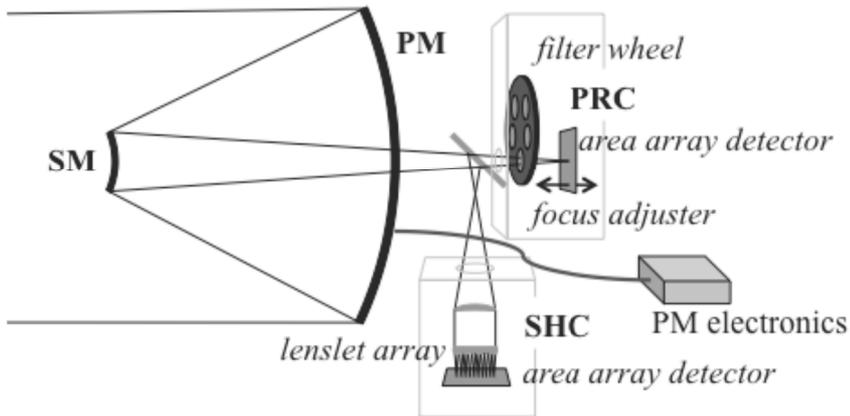


- **Substrate design must meet multiple objectives**
  - Optical performance is improved with more cells/actuators
    - At the expense of mass and complexity
    - Improved with multiple levels of ribs with differing heights
  - Stiffness: first mode  $\gg 100$  Hz
  - Mass: areal density typically 7-10 kg/m<sup>2</sup> for meter-class AHMs
  - CTE balanced by selection of actuator interface tabs

- **FEA models are built for candidate designs**
- **Structural/optical analyses are used to trade design objectives and constraints**
  - Correctability
  - Mass
  - Stiffness
  - Actuator tab material



# Wavefront Sensing and Control



- **WFS&C Elements**
  - Imaging Camera (“PRC”) with area array detector and narrow-band filter
  - Focus adjust mechanism
  - PM actuator electronics
  - Shack-Hartmann Camera (“SHC”), with area array detector, pupil imaging lens and lenslet array

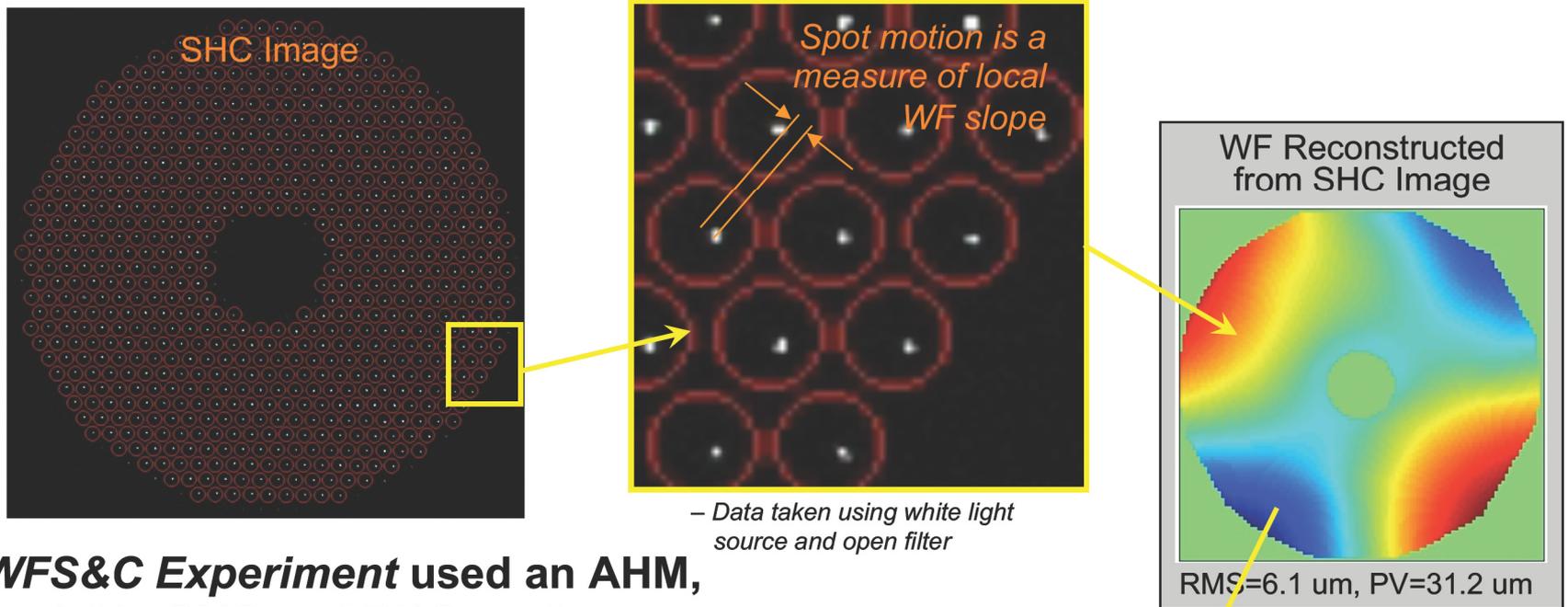
- **WFS&C Operations are performed while observing a star**

- *Initialization* WFS&C uses SHC for large WF capture range ( $> 30 \lambda$ ), and PRC for high resolution and high accuracy
  - Use of PRC Imaging Camera measures WF in the main science camera – no non-common path
  - Run once at the beginning of the mission
- *Maintenance* WFS&C uses PRC only, with minimal/no impact on science ops
  - Keeps WFE within spec
  - Run periodically throughout the mission (1/day to 1/week rate)

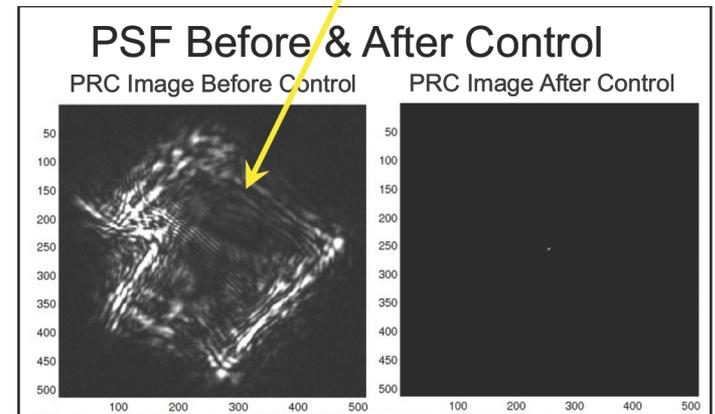
- **Image-based WF sensing using the PRC**

- Modified Gerchberg-Saxton (MGS) phase retrieval software proven through operations on many platforms

# SHC Large Capture Example

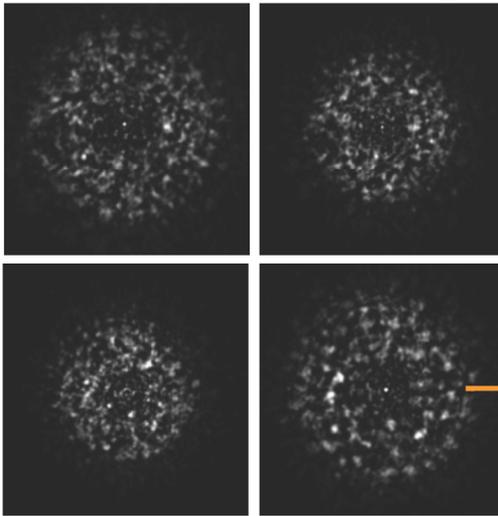


- **WFS&C Experiment** used an AHM, portable SHC and PRC, and autocollimating flat
- **SHC results show large capture range WF control**
  - Initial SHC WF error was 31 um (P-V), 6 um (RMS), double-pass
- **After SHC control, WF error was 80 nm RMS in the SHC, 116 nm in the PRC**



# PRC Fine Control Example

Defocussed PRC Images

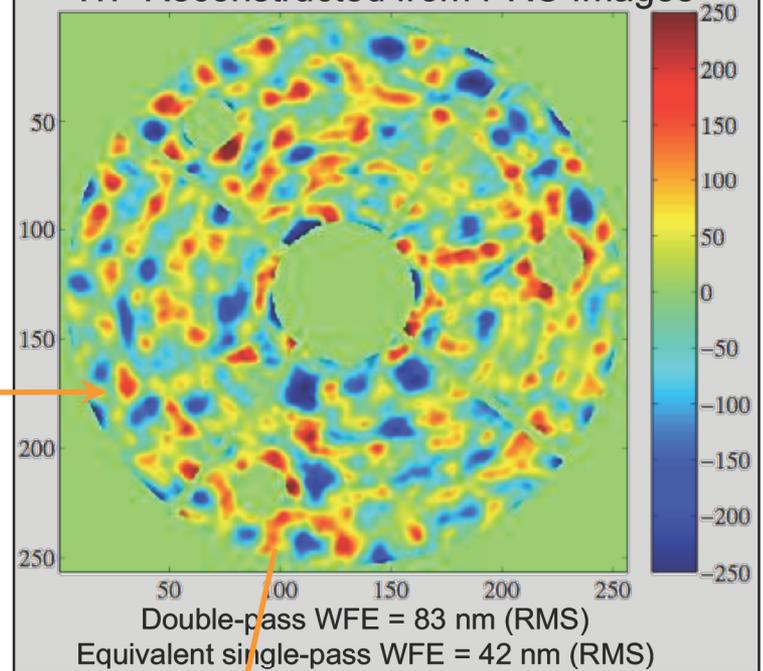


- Data taken using white light source
- Narrow-band Filter

Control used 18% of capacity, 9 V on average

- **WFS&C Experiment continues using the PRC imaging camera for *image-based WF sensing***
- **One or more iterations of control to achieve diffraction-limited WFE**
- **Performance is confirmed by the high-quality “single-pixel” in-focus PSF**

WF Reconstructed from PRC Images



Typical Double-Pass In-Focus PSF

