

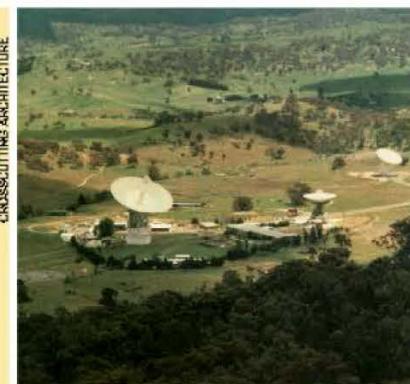
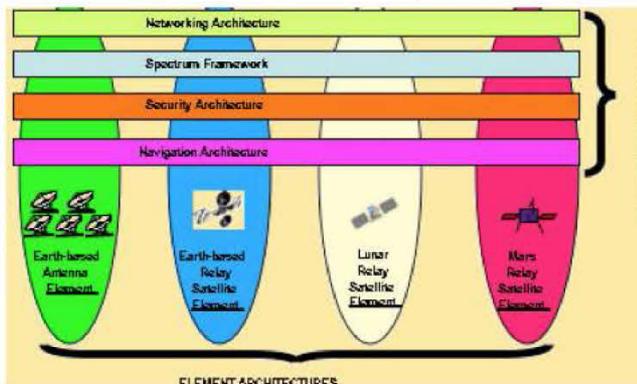
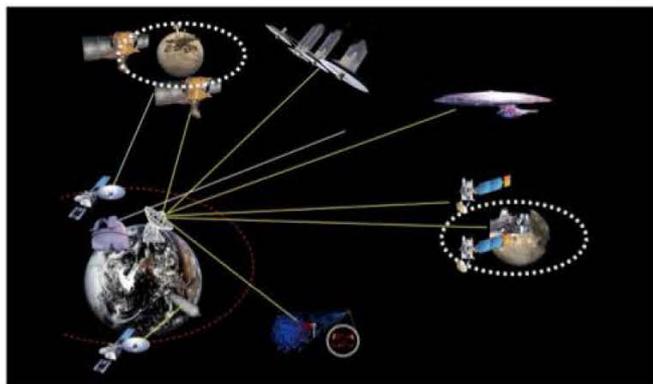
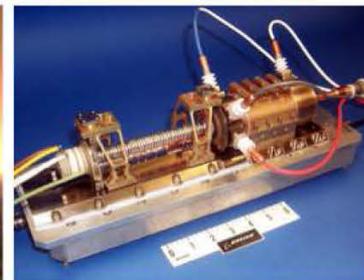
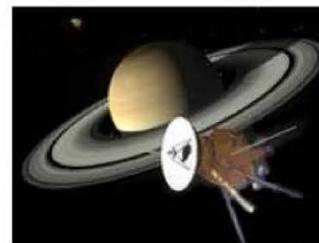


Jet Propulsion Laboratory  
California Institute of Technology

# Improving by a Factor of $10^{16}$

## A History of Pushing All the Boundaries in Deep Space Communications

Dr. Les Deutsch  
Interplanetary Network Chief Technologist  
Jet Propulsion Laboratory  
California Institute of Technology  
June 28, 2012



# Some Current Deep Space Missions

JPL



Cassini: Saturn



SIRTF: Astronomy

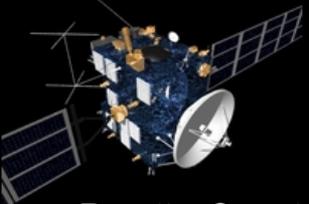


Mars Global  
Surveyor

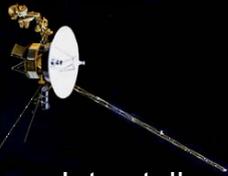
Mars Odyssey



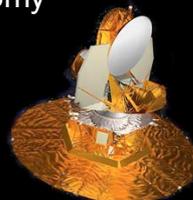
Mars Science  
Laboratory



Rosetta: Comet



Voyager: Interstellar



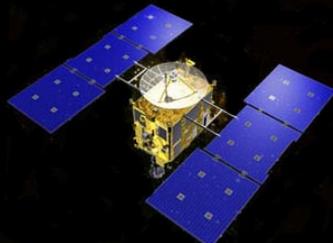
WMAP: Astronomy



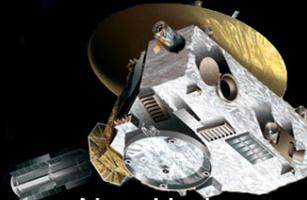
Mars Express



Mars  
Reconnaissance  
Orbiter



Hayabusa: Asteroid

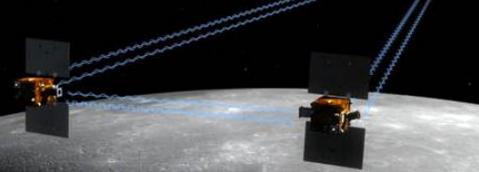


New Horizons:  
Pluto



MESSENGER:  
Mercury

GRAIL: Moon

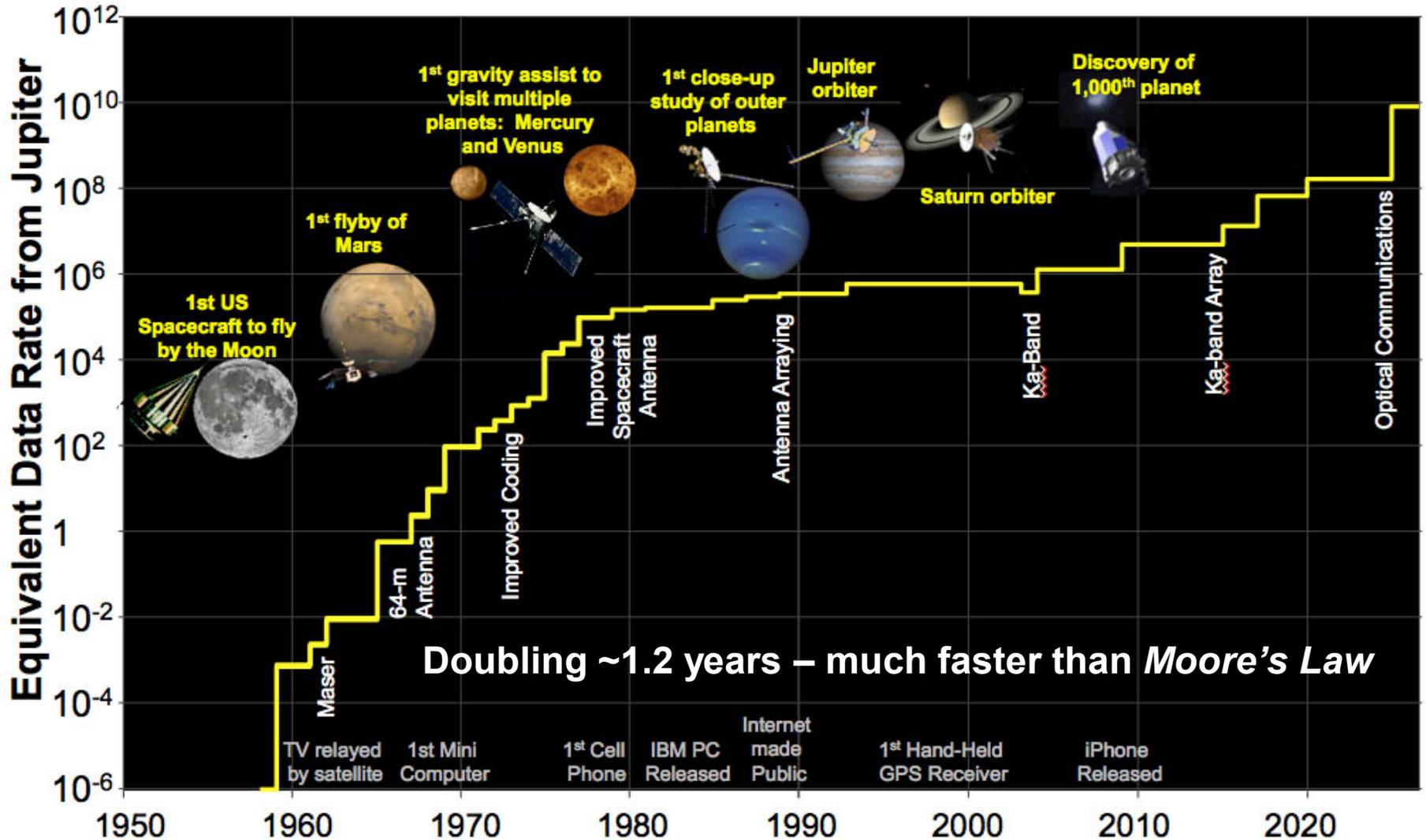


Dawn: Asteroids

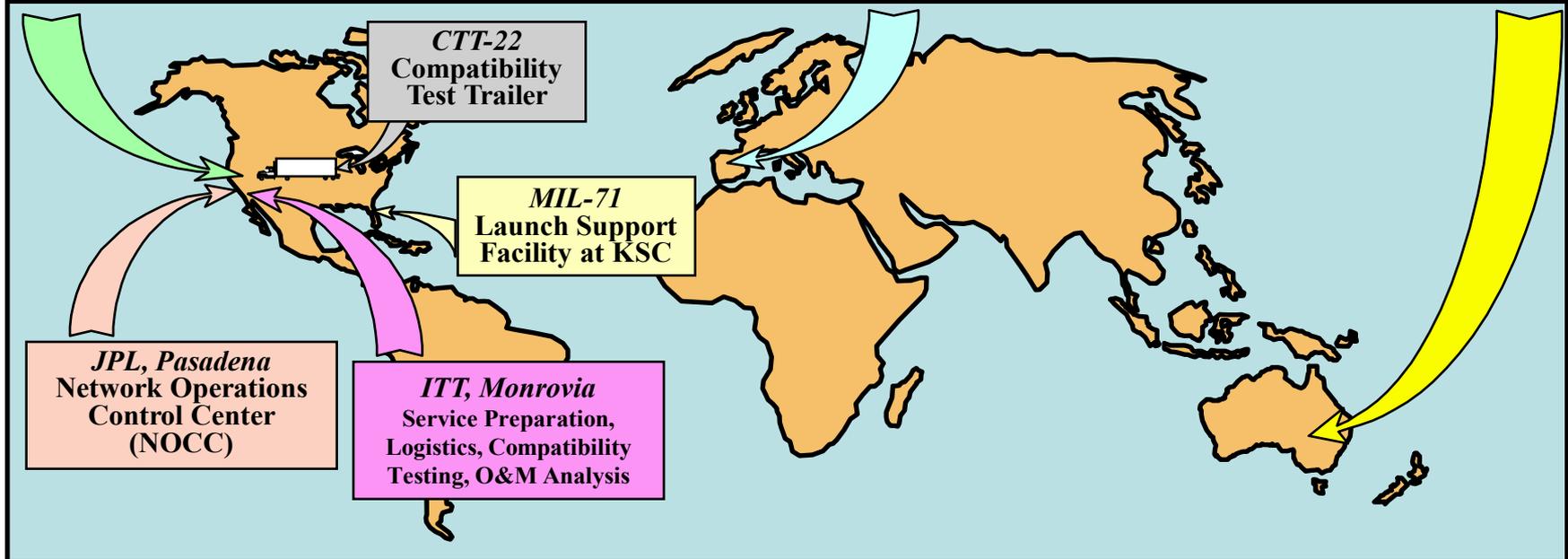
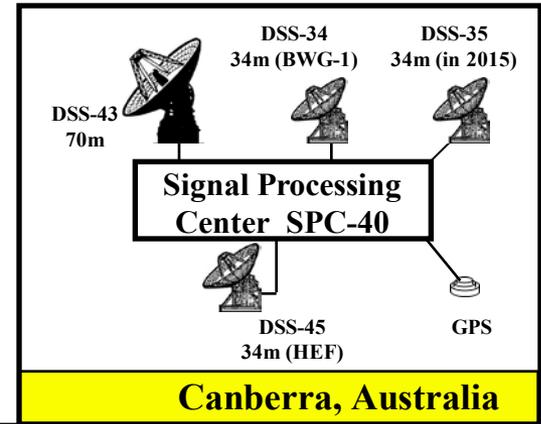
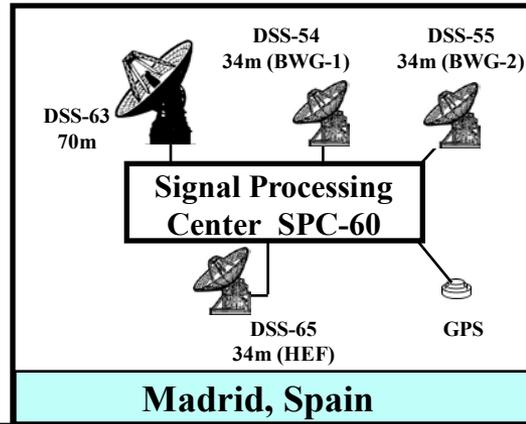
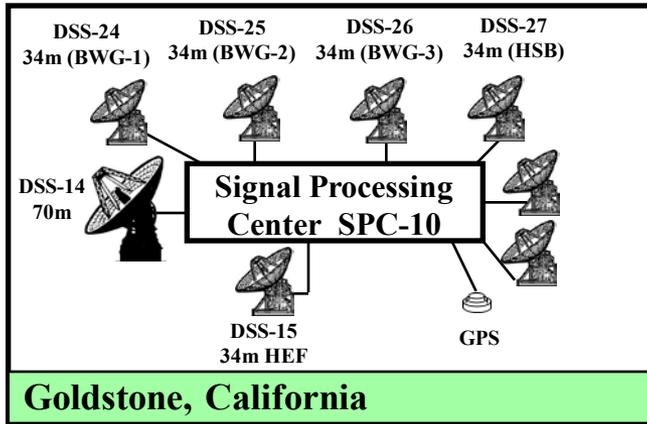
Kepler: Extrasolar  
Planets



# Deep Space Telemetry

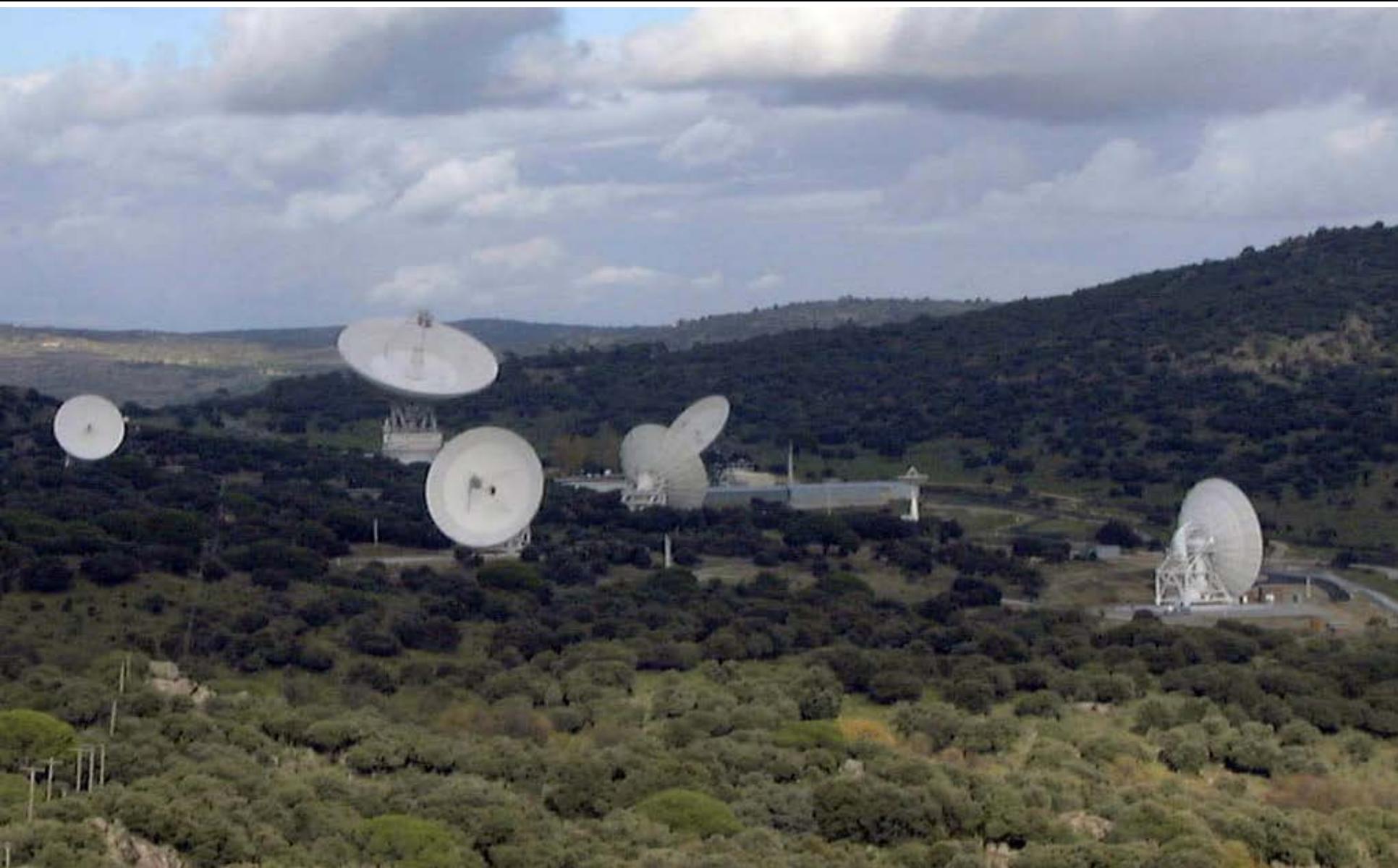


# DSN Facilities



**JPL**

# DSN Antennas in Madrid, Spain

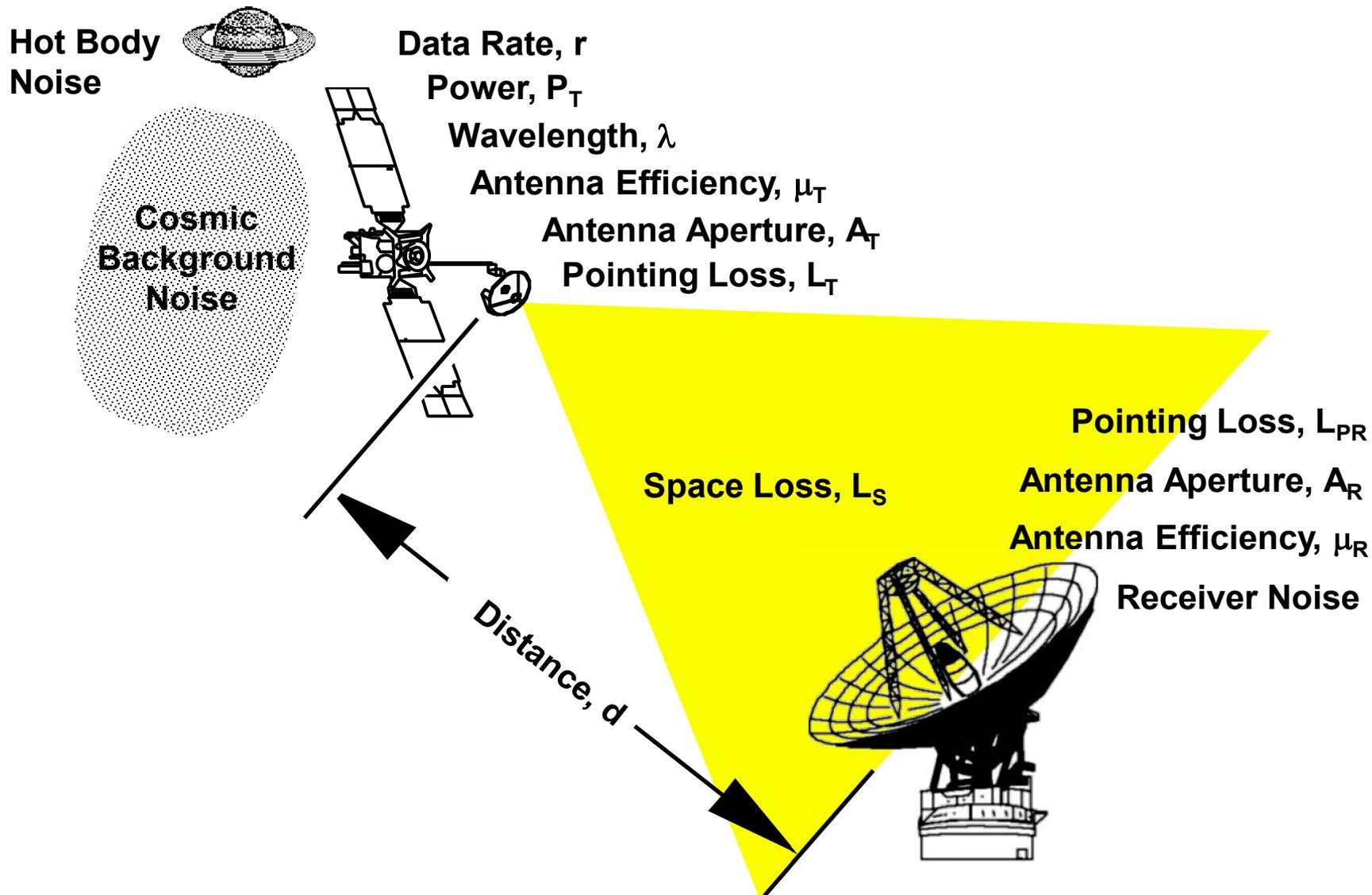




# This is a Small Part of a Big Story

- **The history of communications with deep space is full of interesting problems that have been solved**
- **This talk concentrates on only one of these: receiving digital information from spacecraft far from Earth**
- **Other interesting subjects include**
  - **Getting information from Earth to deep space**
  - **Navigating spacecraft across the solar system using radio and/or optical signals**
  - **Using the communications link as a science instrument to probe planets, moons, small bodies, and the bounds of theoretical physics**

# Deep Space Link Parameters



# Deep Space Link Equations

## Space Loss

$$L_s = \left( \frac{\lambda}{4\pi d} \right)^2$$

## Antenna Gain

$$G_i = \frac{4\pi\mu_i A_i}{\lambda^2}$$

## Received Power Per Bit

$$P_R = P_T G_T L_{PT} L_S L_{PR} G_R / r$$

## Noise Spectral Density

$$N_0 = kT$$

## Noise Sources

$$\begin{aligned} T &= T_{\text{cosmic background}} \\ &+ T_{\text{hot bodies}} \\ &+ T_{\text{RFI}} \\ &+ T_{\text{atmosphere}} \\ &+ T_{\text{receiver}} \end{aligned}$$

Overall performance is a function of Signal-to-Noise ratio

$$\frac{P_R}{N_0} = \frac{P_T G_T L_{PT} L_S L_{PR} G_R / r}{kT}$$

**The goal is to maximize data rate ( $r$ ), while maintaining reasonable (affordable) values of all the other parameters!**

# Space Loss

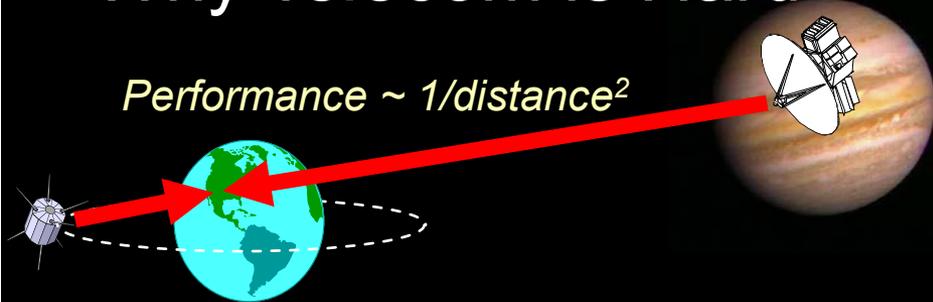
- All else being equal, communications performance is inversely proportional to distance squared

$$P_R/N_0 = \text{constant} / d^2$$

- Need to overcome this problem of physics to be successful in deep space

## Why Telecom is Hard

Performance  $\sim 1/\text{distance}^2$



Relative Difficulty		
Place	Distance	Difficulty
Geo	$4 \times 10^4$ km	Baseline
Moon	$4 \times 10^5$ km	100
Mars	$3 \times 10^8$ km	$5.6 \times 10^7$
Jupiter	$8 \times 10^8$ km	$4.0 \times 10^8$
Pluto	$5 \times 10^9$ km	$1.6 \times 10^{10}$



# Big Antennas

**JPL**

$$P_R/N_0 = \text{constant} * A_T * A_R$$

- **Big antennas are good for deep space**
- **Spacecraft antennas have grown as the size of spacecraft fairings have grown**
- **The big payoff is in ground antennas**
  - **A single large investment serves all space missions**
- **This is why the DSN has the largest steerable communications antennas in the world**

# History of DSN Antennas



1958, 26m Station



1966, 64m Station



1979, 34m Station

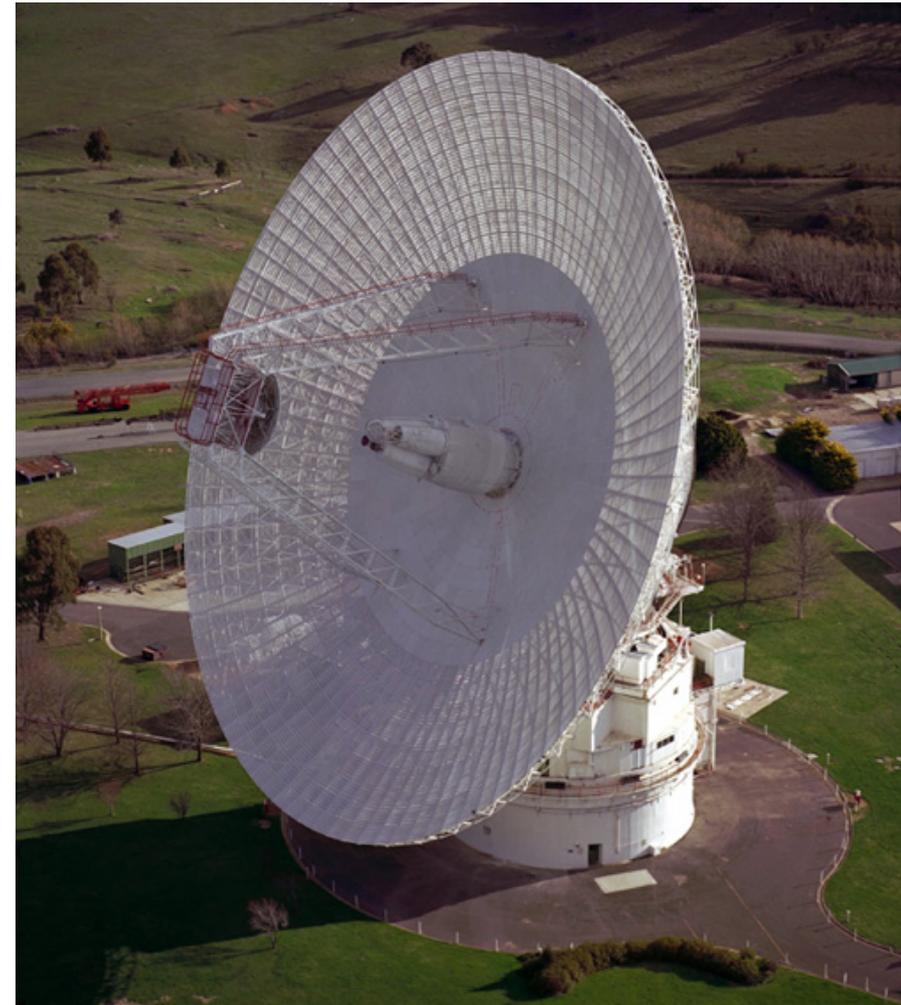


1988, 70m Station  
(converted from prior 64 antennas)

# The DSN's Huge Antennas

**JPL**

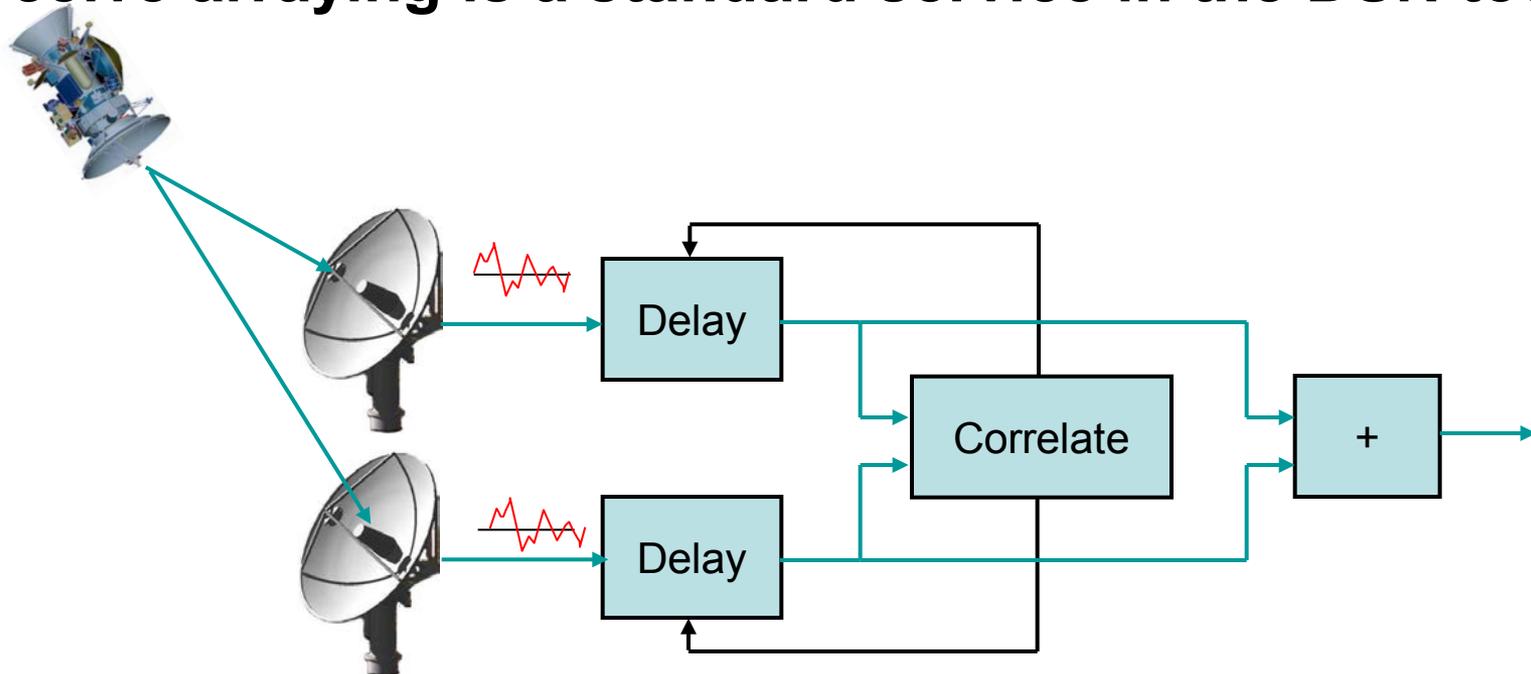
- This is what people recognize most about the DSN
- DSN's 70m antennas are the largest steerable communication antennas in the world
- Each also has a 20 KW transmitter



# Even More: Antenna Arraying

JPL

- By carefully aligning and adding the signals from multiple antennas, a performance approaching that of the sum of the apertures is achieved
- Used to help “save” the Galileo mission to Jupiter when its deployable antenna failed to open
- Receive arraying is a standard service in the DSN today

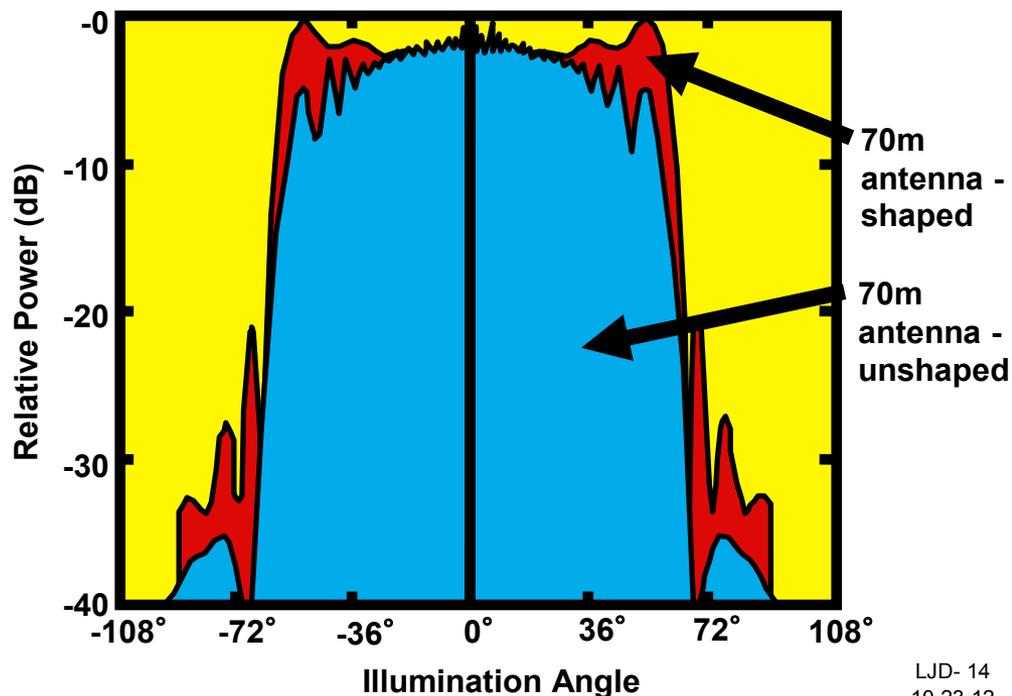


# Antenna Efficiency

JPL

$$P_R/N_0 = \text{constant} * \mu_T * \mu_R$$

- Spacecraft and DSN antennas are at least 70%
- DSN maintains this efficiency even as the huge antennas are rotated and elevated, by:
  - Adjusting the subreflector
  - Using master equatorials
  - Using the received signal to adjust pointing
- DSN antennas have “shaped reflectors”





# Aperture: So Far

**JPL**

- **Increases in spacecraft and ground apertures, improvements in antenna efficiency, and the use of arraying have so far led to an total improvement of**

**44.3 dB**

**or**

**a factor of more than 27,000**

# Higher Frequency is Good

**JPL**

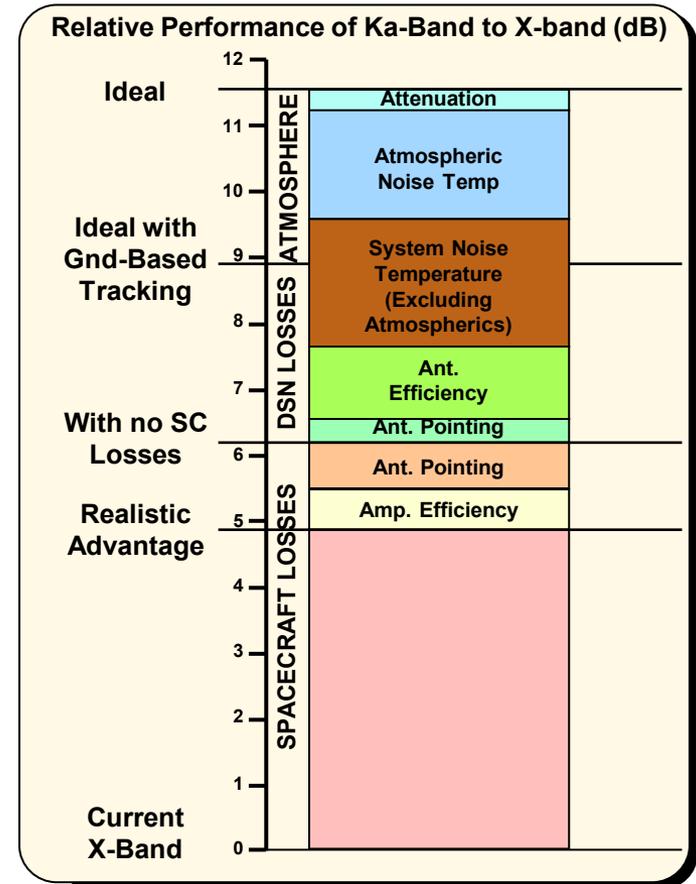
$$P_R/N_0 = \text{constant}/\lambda^2$$

- The first deep space missions transmitted at 960 MHz
- 2.2 GHz (S-band) became standard in 1969
- 8.4 GHz (X-band) became prevalent in the early 1970s
- 32 GHz (Ka-band) is now becoming the standard
- Optical communications is currently in demonstration phase and will become operational in the next decade

# Higher frequencies = some losses!



- The higher the frequency
  - the better you have to point the antennas
  - the more loss there will be in Earth's atmosphere
  - the less efficient the electronics on the spacecraft
- This focuses technology development





# Higher Frequencies: So Far

**JPL**

- **As of today, the improvements to the system from using higher frequencies have amounted to**

**20.6 dB, a factor of ~115**

- **When we add optical communications, the total will increase to**

**37.6 dB, a factor of ~5,800**

# Lowering the System Noise

$$P_R/N_0 = \text{constant}/T$$

- Some elements to  $T$  cannot be controlled
- We concentrate on the contributions of spacecraft and DSN electronics to  $T$
- We carefully avoid RFI
  - Deep space research has its own spectrum assignments from the ITU
- DSN detectors use the best low noise amplifiers we can build or buy
  - Hydrogen masers or HEMTs
  - Physical temperature is  $\sim 12$  K



Ka-band (32 GHz) low noise amplifier

# System Noise: So Far

**JPL**

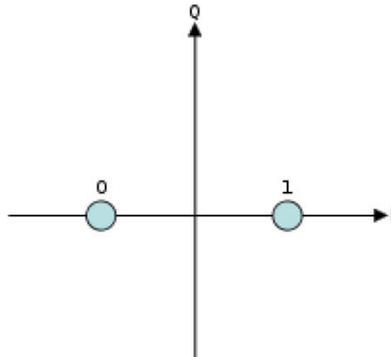
- **This one is harder to track**
- **Many other major system improvements have come with some associated lowering of T**
  - e.g. larger antennas mean narrower beams, so less background noise enters the system
- **As far as improvements that were directed specifically at T, so far we have had an improvement of**

**17.5 dB, or a factor of ~57**

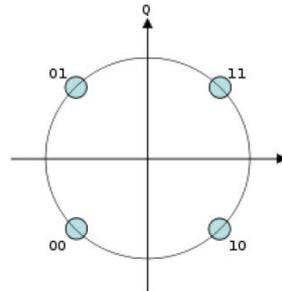
# Modulation – Optimizing P



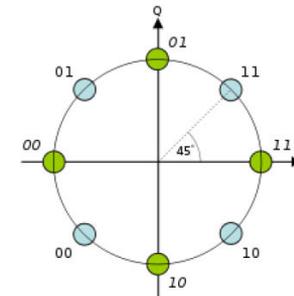
- The way in which data is modulated onto a carrier plays a big part in communications performance
- Consider these standard signaling sets



Binary Phase Shift Keying (BPSK)



Quadrature Phase Shift Keying (QPSK)



Eight Phase Shift Keying (8PSK)

- BPSK has the best performance because the distance between adjacent signals is greatest for the same power

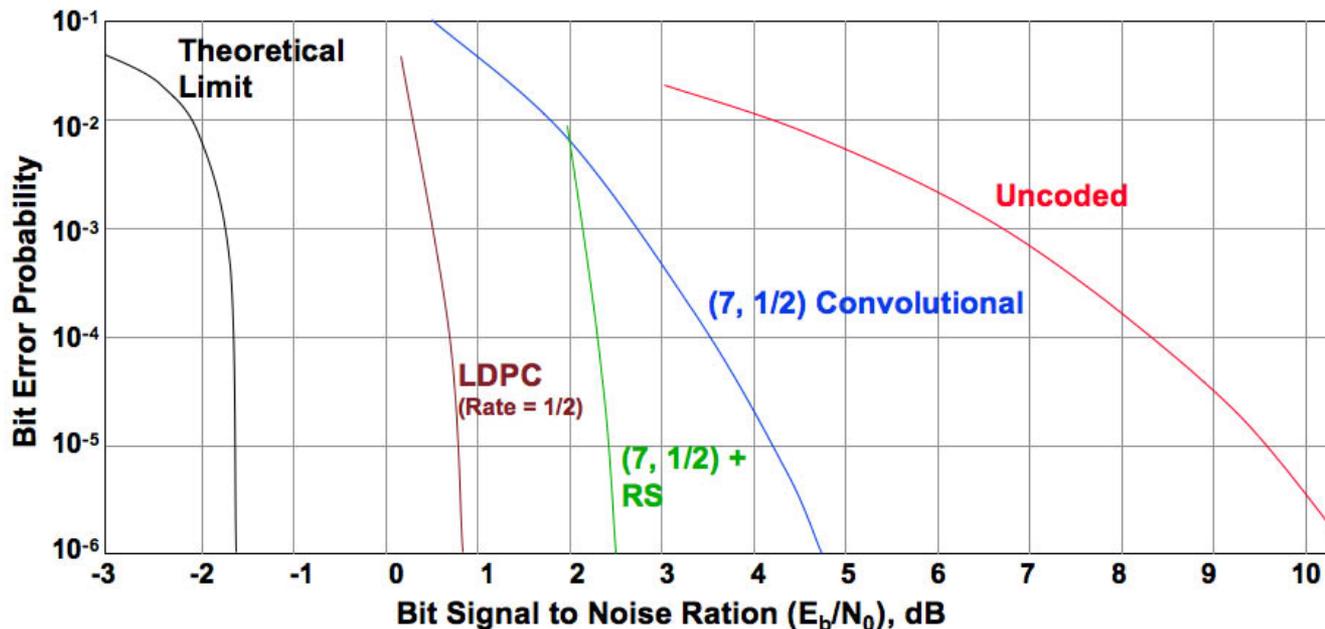
# Deep Space vs. Earth Orbiters

- **Because power is at such a premium for deep space missions, BPSK is typically the preferred modulation scheme**
- **For spacecraft closer to home – Earth orbiters – the situation is different**
  - **They have “power to burn” because they are so close to home**
  - **Because of this, they can fly massive, data-hungry instruments requiring Gbps of return link**
  - **This leads to their use of higher-order modulation types like QPSK and 8PSK – or even higher in some cases!**

# Error-Correcting Codes



- Controlling redundancy in the data stream can result in the ability to correct errors in reception
- Shannon theory showed that it is “easy” to come out ahead – which is a very non-intuitive result: it is better, and even easy, in most cases to add bits to the data stream without adding information!



# Being stingy with Spacecraft bits



- **Data compression**
  - “Lossless” image compression has been used since Voyager
  - Today, more advanced algorithms can reduce transmitted data by more than a factor of 10 without detectable losses in fidelity
  - Larger compression ratios are used for less sensitive data
- **Onboard processing**
  - Advances in spacecraft computers have allowed preprocessing of science data onboard, resulting in fewer bits transmitted to Earth
- **Autonomous operations**
  - If Earth-based teams can be removed from decision loops, the associated data need not be moved to the Earth at all
  - Some of or modern spacecraft make major decisions autonomously: navigation, mobility, targeted science observations



# Modulation, Coding, Compression... so far



- **These improvements typically are inexpensive and quick – my favorite kind!**
- **I cannot even monitor all of them – since improvements in onboard processing and autonomy are bookkept within each flight project and not even considered part of the communications system**
- **So far, improvements in this area – in modulation and coding alone – have amounted to a total of**

**23.8dB or a factor of ~240**



# A Rack-Up



- Here is a summary of the various improvements and showing their contribution to the  $10^{16}$  result.

Area	Improvement to date (dB)	Improvement by 2025 (dB)
Aperture	35.8	44.3
Frequency	20.6	37.6
Power	38.4	38.4
Noise	17.5	17.5
Modulation, Coding, Compression	16.7	23.8
<b>Total</b>	<b>129.2</b>	<b>161.6</b>



# Downlink Data Rate Possibilities

	Data Rate Today		Data Rate ~2020		Data Rate ~2030	
Spacecraft Capabilities	3m Antenna X-Band 100 W Xmitter		3m Antenna Ka-Band 180 W Xmitter		5m Antenna Ka-band 200 W Xmitter	
DSN Antennas	1 x 34m	3 x 34m	1 x 34m	Equiv to 3 x 34m	1 x 34m	Equiv to 7 x 34m
Mars (0.6 AU)	20 Mbps	60 Mbps	400 Mbps	*1.2 Gbps	*1.3 Gbps	*9.3 Gbps
Mars (2.6 AU)	1 Mbps	3 Mbps	21 Mbps	64 Mbps	71 Mbps	*500 Mbps
Jupiter	250 Kbps	750 Kbps	5 Mbps	15 Mbps	16 Mbps	115 Mbps
Saturn	71 Kbps	213 Kbps	1.4 Mbps	4 Mbps	4.7 Mbps	33 Mbps
Neptune	8 Kbps	24 Kbps	160 Kbps	470 Kbps	520 Kbps	3.7 Mbps

\* Reference spacecraft is MRO-class (power and antenna), Rate 1/6 Turbo Coding, 3 dB margin, 90% weather, and 20° DSN antenna elevation

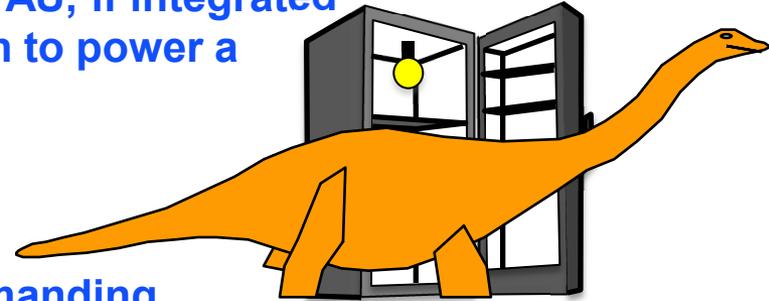
\*\* Performance will likely be 2 to three times lower due to need for bandwidth-efficient modulation to remain in allocated spectrum

# Some Amazing DSN Facts

## Received Signal Sensitivity:

The received energy from Voyager at 100 AU, if integrated for 10 trillion years, would be just enough to power a refrigerator light bulb for one second!

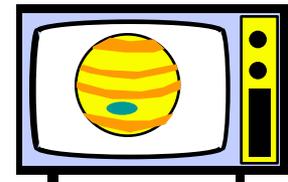
$$\text{Received power} = 6.3 \times 10^{-19} \text{ W}$$



## Command Power:

The DSN puts out enough power in commanding Voyager that it could easily provide high quality commercial TV at Jupiter!

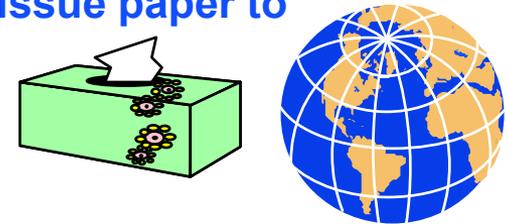
$$\text{Transmitted power} = 400 \text{ kW}$$



## Dynamic Range of the DSN:

The ratio of the received signal power to the DSN transmitting power is like comparing the thickness of a sheet of tissue paper to the entire Earth!

$$\text{Ratio} = 10^{27}$$



## Reference Clock Stabilities:

The clocks used in the DSN are so stable that they would drift only about 5 minutes if operated over the age of the universe!

$$1 \text{ part in } 10^{15}$$



# Future Challenges for the DSN

- **Space mission communication needs follow a “Moore’s Law” requiring ~factor of 10 improvement per decade**
- **Human spaceflight will venture beyond low Earth orbit into deep space**
  - **Data rates will have to be much larger to support both the needs of the astronauts and the desires of the public**
- **Deep space optical communication will come into its own in the next couple of decades**
- **The DSN will evolve to meet these challenges and continue to enable space missions for at least the next 50 years**