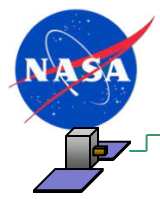


# Analysis of Phoenix Anomalies and IV&V Findings Applied to the GRAIL Mission

March 5, 2012

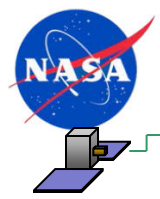
Steve Larson  
Jet Propulsion Laboratory  
California Institute of Technology



# Introduction



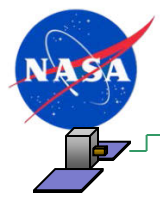
- Motivation
- GRAIL Project Overview
- Analysis of Phoenix IV&V and Post-Launch Anomalies
- Phoenix Lessons Applied to GRAIL
- Phoenix/GRAIL Post IV&V Comparison
- GRAIL Post-Launch Experience



# Motivation



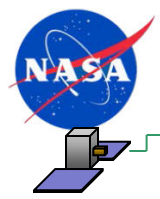
- GRAIL is a cost-capped mission
  - IV&V is recognized as a contributor to mission reliability, but funds were limited
  - Improving the efficiency of the IV&V effort by learning from earlier missions would increase the value added



# GRAIL Project Overview



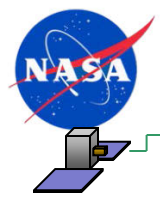
- GRAIL = Gravity Recovery And Interior Laboratory
- Launch September 2011
- Mission
  - Determine the structure of the lunar interior, from crust to core
  - Advance understanding of the thermal evolution of the Moon.
  - Extend knowledge gained from the Moon to the other terrestrial planets.
- Project Management: JPL
- PI: Dr. Maria Zuber, MIT
- Spacecraft: Lockheed Martin S&ES
- Instrument: JPL



# Analysis of Phoenix IV&V Results



- NASA IV&V provided a dump of all issues written for the Phoenix project
  - 893 issues categorized into 16 bins
    - Multiple categorizations allowed
  - Analyzed according to whether the project responded by changing the affected artifacts or using as-is.
  - Looked for other patterns that would suggest areas for improvement

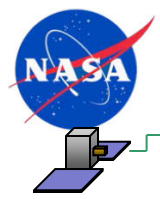


# Analysis of Phoenix IV&V Results

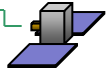


- Code-related issues were much more likely to be accepted “as-is” by the project
- Issues that could be addressed by updating documentation were more likely to be fixed
- Possible explanations:
  - Code issues developed late in the development process
  - Lower effort barrier to changing documents vs. code (e.g., don’t need to regression test documents)

Description	Fix	UAI	Total
Array Bounds violation	55%	45%	1%
Conflicting Code Statements	20%	80%	2%
Coding Error	17%	83%	8%
Dead Code	30%	70%	5%
Design/Code Discrepancy	57%	43%	3%
Loss of Precision	20%	80%	1%
Memory Leak	0%	100%	0%
NULL Pointer	0%	100%	1%
Type Mismatch	25%	75%	2%
Uninitialized Variable	41%	59%	4%
<b>Code-related Subtotal</b>	<b>29%</b>	<b>71%</b>	<b>27%</b>
Document Discrepancy	53%	47%	17%
Design/Requirements Discrepancy	71%	29%	10%
Missing Requirement	72%	28%	6%
Requirement Not Verified	93%	7%	13%
Requirements Quality	63%	37%	20%
Requirements Trace	59%	41%	8%
<b>Documentation-related Subtotal</b>	<b>68%</b>	<b>32%</b>	<b>73%</b>
<b>Combined Total</b>	<b>57%</b>	<b>43%</b>	<b>100%</b>



# Analysis of Phoenix IV&V Results



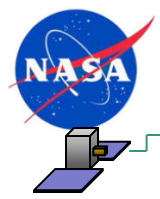
- Fix vs. Use As-Is decisions were correlated with issue severity
  - All Severity 2 issues were addressed by the project
    - IV&V agreed with the 2 UAI responses, and closed the issues
  - Severity 5 issues are the exception
    - Less numerous, may have been fixed as part of normal process or addressing higher severity issues
  - For the most part, IV&V agreed with UAI decisions
- Fix/Use as-is decisions were correlated with the analysis approach
  - Roughly 2/3 of issues found with automated tools were accepted as-is
  - The proportion was reversed in the case of manual analysis
  - Lockheed code relatively mature, often able to show that code would behave properly
    - Later in life cycle, more difficult to fix
  - Manual analysis more likely to be applied to tests, documentation
    - Earlier in life cycle and/or easier to fix

Severity	Disposition	Count	Percent
2	F	24	92.3%
2	UAI	2	7.7%
3	F	329	66.1%
3	UAI	169	33.9%
4	F	80	33.2%
4	UAI	161	66.8%
5	F	26	65.0%
5	UAI	14	35.0%

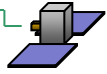
## Issue Severities

1	Partial or complete mission failure
2	Adverse effect with no workaround
3	Adverse effect with a workaround
4	An inconvenience
5	Anything else

ToolUsed	Fix	UAI
Flexe-Lint, V8.00Q	26	54
Klocwork inSpect	19	49
Manual Analysis	415	239
Understand for C/C++	0	4



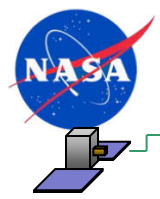
# Analysis of Phoenix IV&V Results



- Payloads accounted for roughly  $\frac{3}{4}$  of all issues, and had a somewhat higher proportion of higher severity issues
- Assuming that issue frequency is a predictor of in-flight performance, one might predict that payloads would account for the majority of post-launch FSW anomalies

CSCI	Severity				Totals	
	2	3	4	5	Count	%
GN&C	1	25	5	2	33	15.4%
I/O	0	28	5	0	33	15.4%
OS	0	16	9	1	26	12.1%
Spacecraft	5	38	50	3	96	44.9%
Telecom	0	14	12	0	26	12.1%
<b>Spacecraft Subtotal</b>	<b>8</b>	<b>124</b>	<b>85</b>	<b>11</b>	<b>214</b>	
<b>Spacecraft % Distribution</b>	<b>4%</b>	<b>58%</b>	<b>40%</b>	<b>5%</b>	<b>27%</b>	
Mardi	0	4	4	0	8	1.4%
MECA	4	130	44	4	182	30.8%
MET	0	39	9	3	51	8.6%
RA	5	48	32	7	92	15.6%
SSI/RAC	6	75	27	15	123	20.8%
TEGA	5	81	44	5	135	22.8%
<b>Payloads Subtotal</b>	<b>20</b>	<b>377</b>	<b>160</b>	<b>34</b>	<b>591</b>	
<b>Payloads % Distribution</b>	<b>3%</b>	<b>64%</b>	<b>27%</b>	<b>6%</b>	<b>73%</b>	

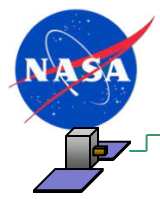




# Phoenix Anomaly Analysis



- 369 unique post-launch ISA (Incident/Surprise/Anomaly) reports were analyzed
  - Binned into 8 categories of contributing factors
    - Factors identify where in the development & test process a defect was likely to have been introduced, or could have been corrected but was not
    - Multiple factors allowed
  - Binned according to whether issue was discovered on flight vehicle or on the ground
  - 31 ISAs determined to be in flight software (next slide)
    - 7 in spacecraft, 24 in payloads
      - » Mirrored IV&V ration of spacecraft/payload issues
    - Most ISAs concerned ground software and hardware, and were not included in the study

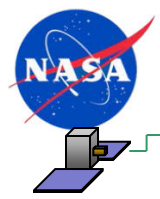


# Contributing Factor Distribution



	Contributing Factors								Total # Incidents	Occurred in Flight?
	Complexity	Heritage Process	Missing Requirement	Design	Inadequate Testing	Implementation	System Engineering	Insufficient Information		
<b>Spacecraft-</b>	0%	29%	43%	57%	43%	29%	57%	14%	23%	86%
<b>Payload-related</b>	13%	4%	38%	17%	46%	58%	29%	13%	77%	50%
<b>Combined</b>	10%	10%	39%	26%	45%	52%	35%	13%	100%	58%

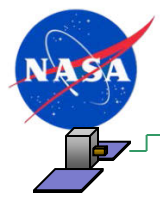
- Most ISAs had more than 1 contributing factor
  - For example, the heritage process introduced a defect in spacecraft battery control that testing should have caught, had fidelity to flight conditions been adequate
- Systems Engineering and S/W Design were leading causes of spacecraft issues
  - Inverse of what might be predicted from IV&V distribution (58% for Implementation vs. 37% for Requirements & Design)
- Code & Test were leading causes of payload issues
  - Also inverted from IV&V issue distribution(68% for Requirements & Design vs. 16% for Implementation)
- Effects of complexity is an industry-wide concern, but did not appear to be a dominant factor
  - Complexity-related issues in the payloads did not fit the classic “Normal Accidents” model



# Predictive Skill



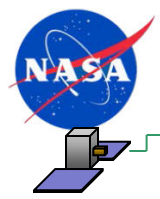
- IV&V broadly predicted (based on the data) that payloads would be the primary source of in-flight anomalies
  - This was borne out in flight
- Inverse relationship between distribution of IV&V issues and contributing factors for flight anomalies suggests additional analysis needs to be done to understand this relationship.



# Phoenix Lessons Applied to GRAIL



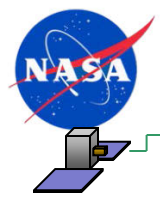
- “Newness” is a risk
  - Payload issues dominated, and were all either first-of-a-kind or modifications to previous products
  - PBRA process used on GRAIL emphasized “newness” as a risk
- Product Line FSW should be approached differently than first-of-a-kind/low heritage software
  - Discussion of the effects of heritage allowed us to close issues more easily
- Unnecessary issues can be avoided by waiting for products to mature
  - Structured discussion of potential issues generated by review of early versions of FSW and requirements allowed us to resolve a large number of issues without excess formality
- Problems that escape both the developer and IV&V tend to be “difficult”
  - Hardware interfaces of greatest concern, received thorough IV&V review
  - However, proprietary nature of source data limited the analysis
- Augmenting the normal IV&V process with less formal analysis improves understanding and effectiveness
  - GRAIL able to vet a much larger number of potential issues and focus on important findings



# IV&V Results—GRAIL/Phoenix



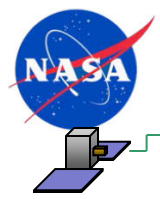
- Upon disposition of the last GRAIL issue, Phoenix (spacecraft only) and GRAIL issue distribution was analyzed
  - Overall roughly 10% drop in number of issues
    - Change could simply be due to different IV&V team or bundling strategies
  - Overall increase in Fix rate
    - Previous IV&V work on code base likely reduced number of false positives in code
    - Better communication eliminated more false positives in all categories
  - Big drops in Dead Code, Design/Code Discrepancy, Document Discrepancy categories
    - Better communication helped eliminate false positives and issues due to examining immature products
    - Dead code reduction may be due to prior IV&V work, but not analyzed for cause
  - A new category (Code/Requirements Discrepancy) was introduced for the GRAIL analysis
    - Better alignment with the way IV&V does their work
    - On Phoenix, these would have shown up in either Design/Code Discrepancy or Coding Error
  - Increase in Requirements Quality, Requirements Trace, Missing Requirements categories
    - IV&V changed approach, started with modeling and top-down requirements assessment (many more documents examined)
    - Different IV&V personnel may have also contributed



# GRAIL Post-launch Experience



- No flight software anomalies in either spacecraft or instrument since launch
  - Instrument has been off most of the time
- One FSW patch for issue discovered prelaunch
  - Conformance with BAE RAD750 errata on register usage
  - No issues in 1000+ boot cycles prelaunch
  - Fixed out of an abundance of caution
- Two minor bugs found, but not fixed
  - Pyro firing
  - ACS parameter usage
- All three would have been very difficult for IV&V to catch
  - RAD750 errata nonconformance would have required expert level knowledge of the board
  - Pyro firing code error conformed to C language standard, subtle error in use of enums
  - ACS parameter usage would have required expert level knowledge of ACS algorithms and access to proprietary information



# Conclusions



- Analysis of patterns in IV&V findings and their correlation with post-launch anomalies allowed GRAIL to make more efficient use of IV&V services
  - Fewer issues
  - Higher fix rate
  - Better communication
  - Increased volume of potential issues vetted, at lower cost
- Hard to make predictions of post-launch performance based on IV&V findings
  - Phoenix made sound fix/use as-is decisions
    - Things that were fixed eliminated some problems, but hard to quantify
  - Broad predictive success in one area, but inverse relationship in others