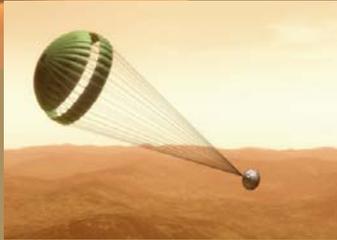
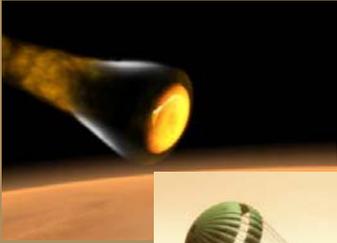


Mars Science Laboratory Entry, Descent and Landing Overview

*Jeffrey W. Umland
Jet Propulsion Laboratory*

*4th International Planetary Probe Workshop
Pasadena, CA
June 29, 2006*

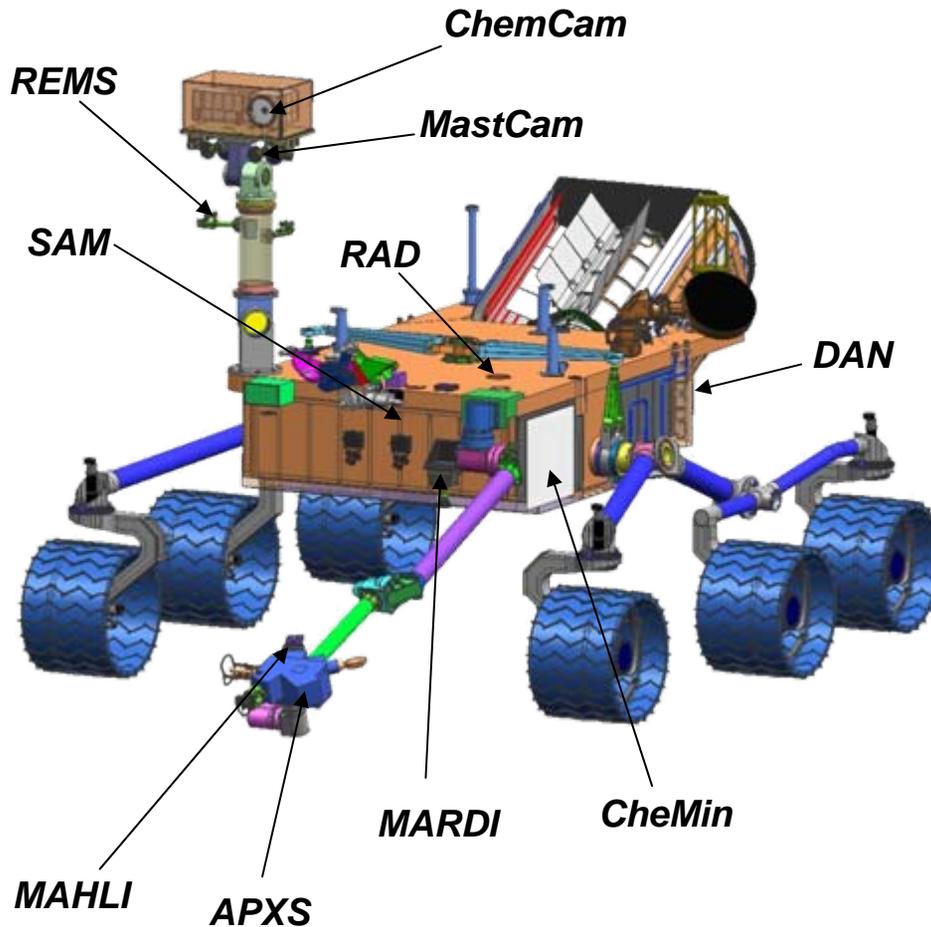




MSL Science Instruments



Mars Science Laboratory



Remote Sensing (Mast)

ChemCam – Laser Induced Breakdown Spectrometer

MastCam - Color Stereo Imager

Contact Instruments (Arm)

MAHLI - Microscopic Imager

APXS - Proton/X-ray Backscatter Spectrometer

Analytical Laboratory (Front Chassis)

SAM - Gas Chromatograph/Mass Spectrometer/
Tunable Laser Spectrometer
(Sample Composition / Organics Detection)

CheMin - X-ray Diffraction / Florescence
(Sample Mineralogy)

Environmental Characterization (Body-mount)

MARDI - Descent Imager

REMS - Meteorological monitoring

RAD - Surface Radiation Flux Monitor
(future human health & safety)

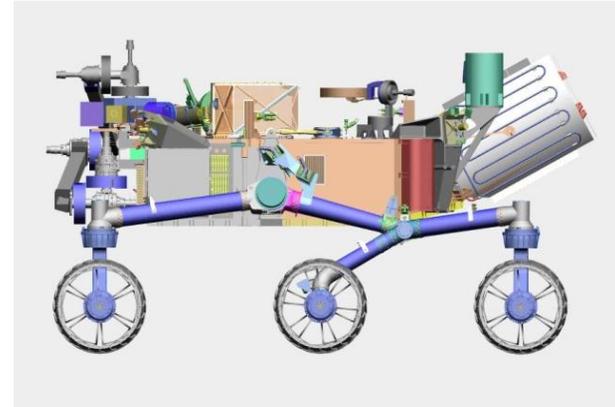
DAN - Neutron Backscatter subsurface hydrogen
(water/ice) detection



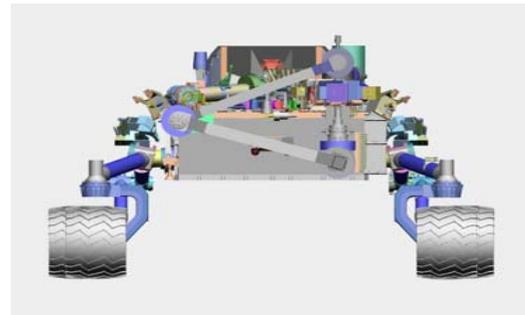
MSL Size Comparison



 2006 MINI Cooper S



JPL 2009 MSL Rover



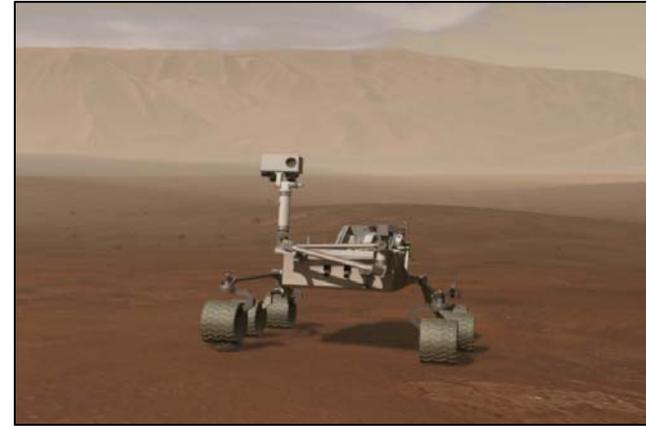


MSL Requirements



Key Driving EDL Requirements:

- Deliver 800 kg rover
- Landed altitude < [0 to 1] km (MOLA)
- Landing with a maximum error of 10 km from target



	<i>Viking</i>	<i>MPF</i>	<i>MPL</i>	<i>MER</i>	<i>Phoenix</i>	<i>MSL</i>
<i>Landed Mass (kg)</i>	603	360	290	540	364	1541*
<i>Delivered Mass (kg)</i>	603	360	290	173	364	800*
<i>Entry Ballistic Coefficient (kg/m²)</i>	63	62	71	88	71	121
<i>Landing Altitude (km, MOLA)</i>	-3.5	-1.5	+2.4	-1.3	-3.5	+[0 to 1]
<i>Landing Accuracy (km)</i>	420 x 200	100 x 50	75x 38	80 x 20	75 x 20	<10 x 10

* Includes required JPL practices 30% margin on allocation

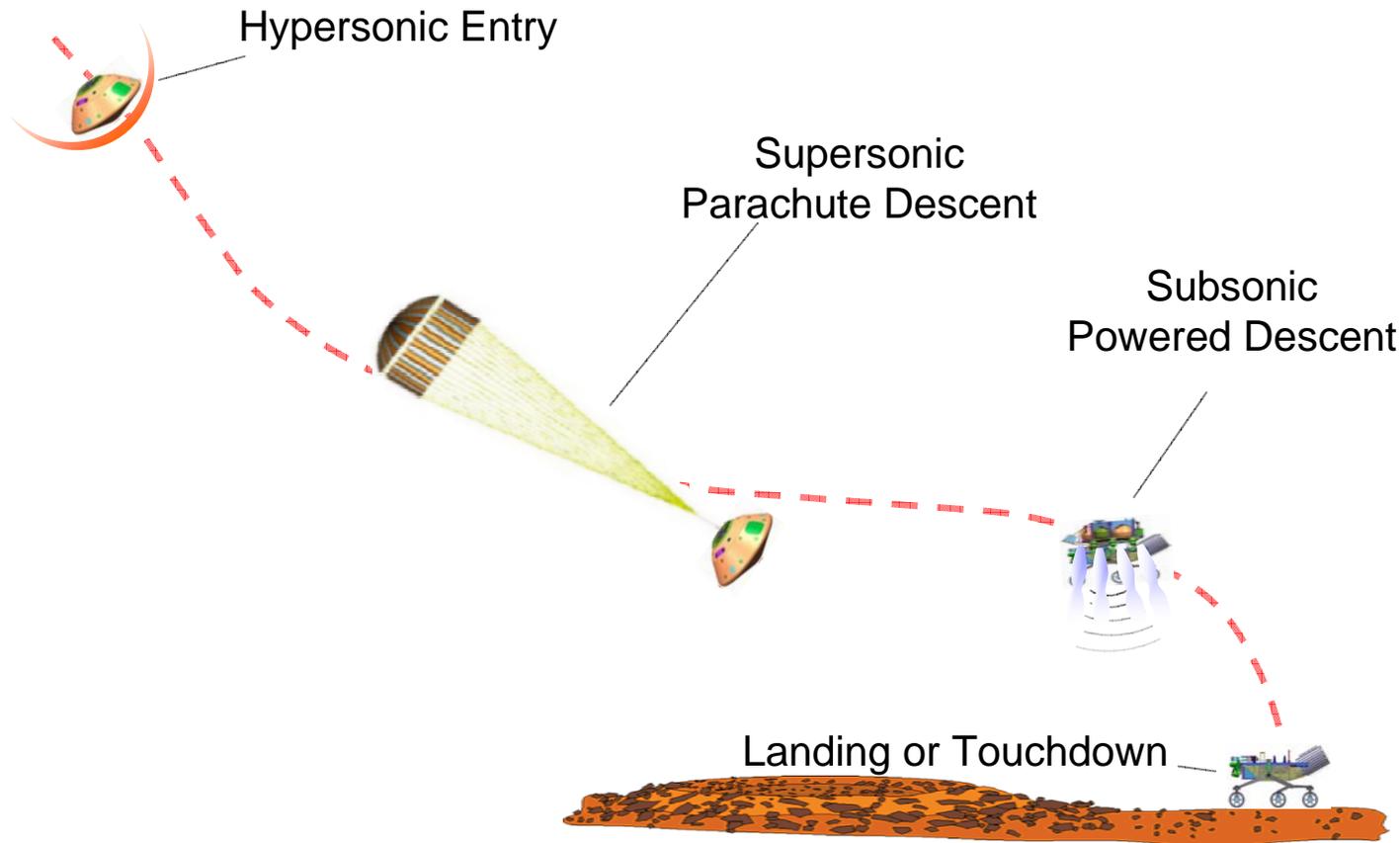


Martian Entry, Descent and Landing



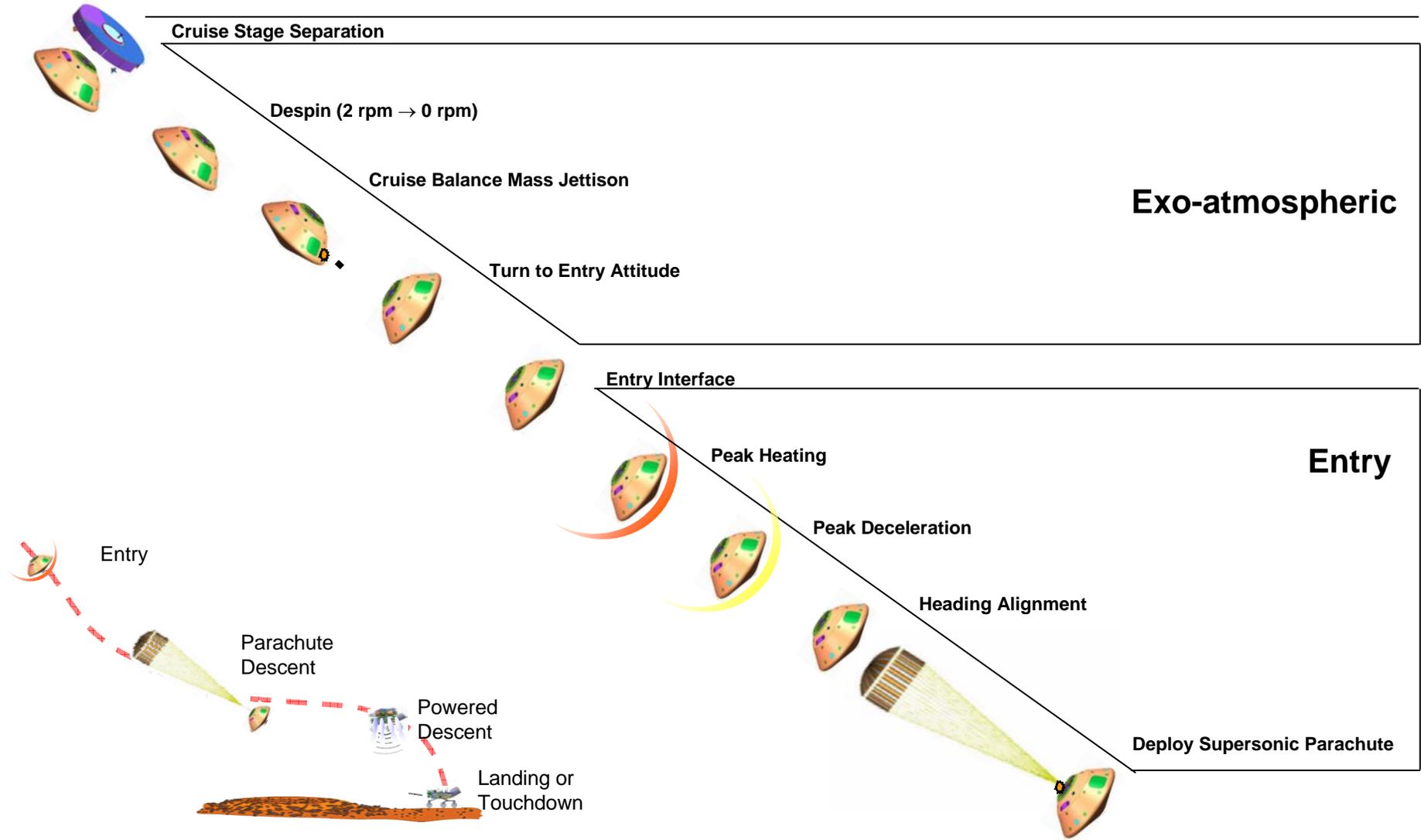
Generic Martian Entry, Descent and Landing System:

- Four Major Elements present historically for all large scale payloads
- MSL must incorporate improved performance for each EDL element



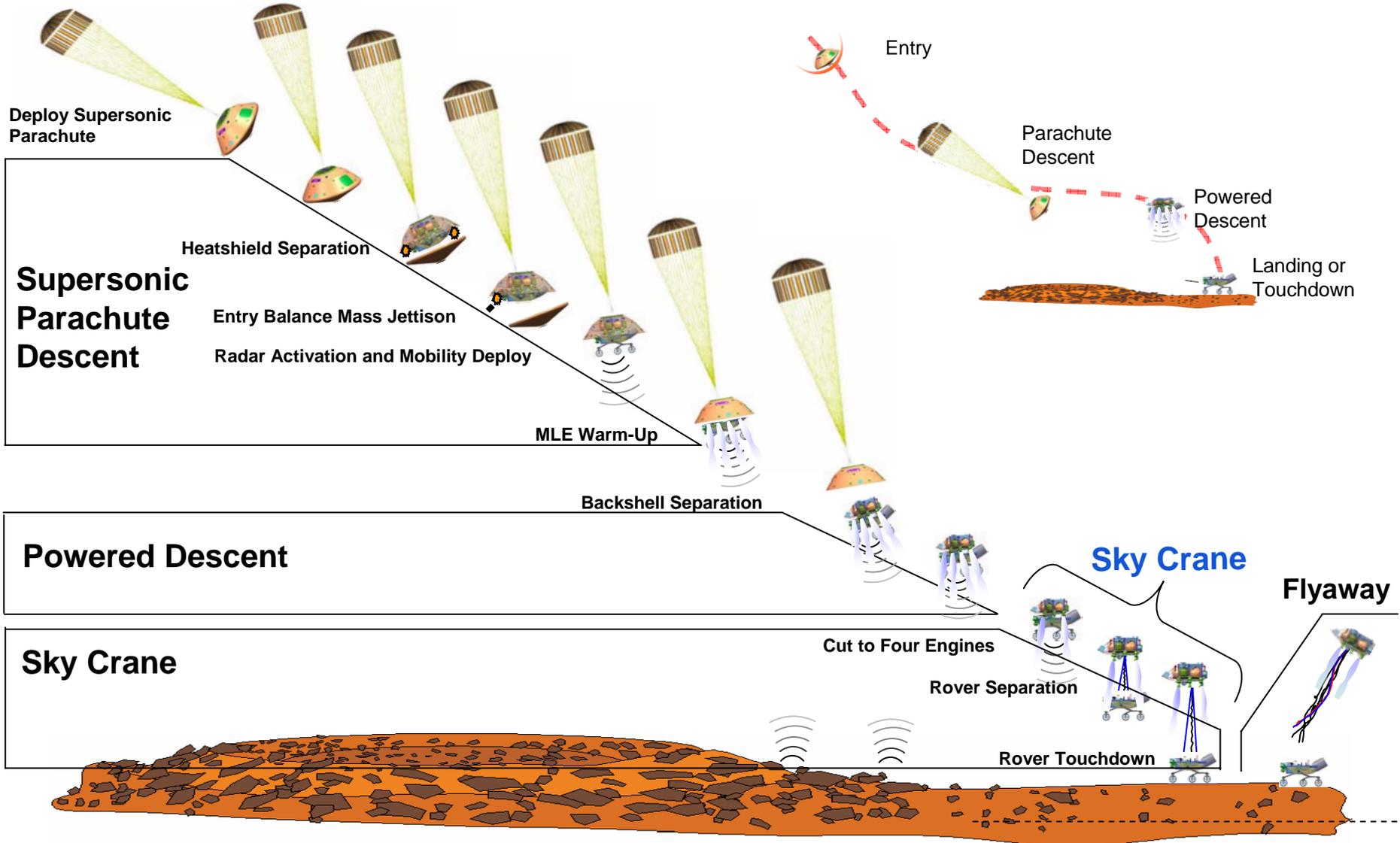


Event Timeline 1/2





Event Timeline 2/2





Sky Crane Touchdown



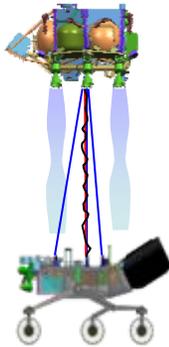
Mars Science Laboratory

Descent Stage commanded to follow Reference Trajectory: $V_{Vertical} = 0.75 \text{ m/sec}$ & $V_{Horizontal} = 0.0 \text{ m/sec}$

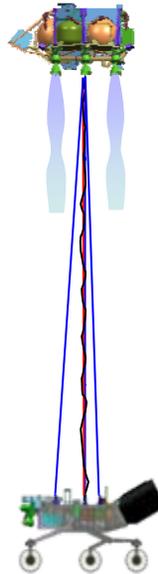
One-Body Phase
Duration = ~2 sec



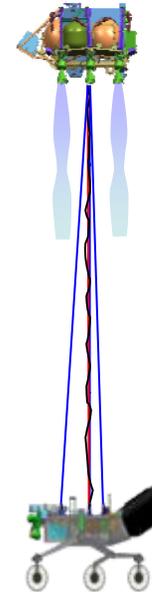
Deployment Phase
Duration = ~6 sec



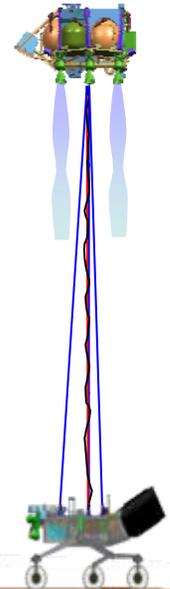
Post-Deploy Settling Phase
Duration = ~2 sec



Ready for Touchdown Phase
Duration = 0-8 sec



Touchdown Phase
Duration < 2 sec



Skycrane Event Timeline