



The Thermal Design Evolution of the Phoenix Robotic Arm

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Outline



- Phoenix Mission Description
- RA Mechanical Configuration
- RA Landed Thermal Environment
- RA Thermal Design Challenges
- Warm-up Heater Sizing
- Maximum Actuator Power Dissipation Sizing
- Failed-On Warm-Up Heater Accommodation
- Conclusions



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CJL/GTT-1



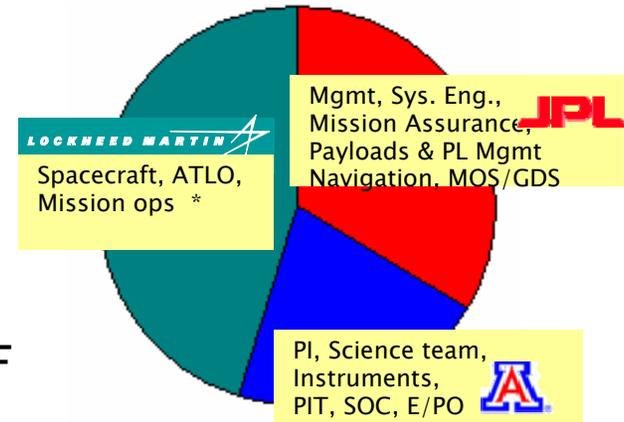
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Phoenix Overview

What is Phoenix?

- Phoenix will be the next NASA Mars landed mission
- Phoenix utilizes the terminated MSP'01 lander, improved through Return To Flight upgrades
- Phoenix will fly many of the lost MPL (Mars 98) payloads and some from MSP'01
- Phoenix utilizes a powered descent system unlike MPF and MER
 - More scalable
 - Provides soft landing capabilities
 - More precise placement on the surface
- Key Partners
 - The University of Arizona provides the PI, Peter Smith, and several instruments as well as the PIT and SOC
 - JPL provides Project Management, Systems Engineering, MOS/GDS, as well as the RA and MECA instruments
 - Lockheed Martin provides the Flight System and Operations support
 - Instruments are supported as well through contributions from all over the world



- CSA/MDR/Optech
 - Provides MET station and optical Lidar
- Max Planck Institute for Aeronomy (MPAe)
 - Provides RA camera ('01) and calibration
- University of Neuchatel/Swiss Federal Institute of Technology
 - Provides Atomic Force microscope for MECA ('01)
- University of Copenhagen
 - Provides magnets for MECA and SSI cal target



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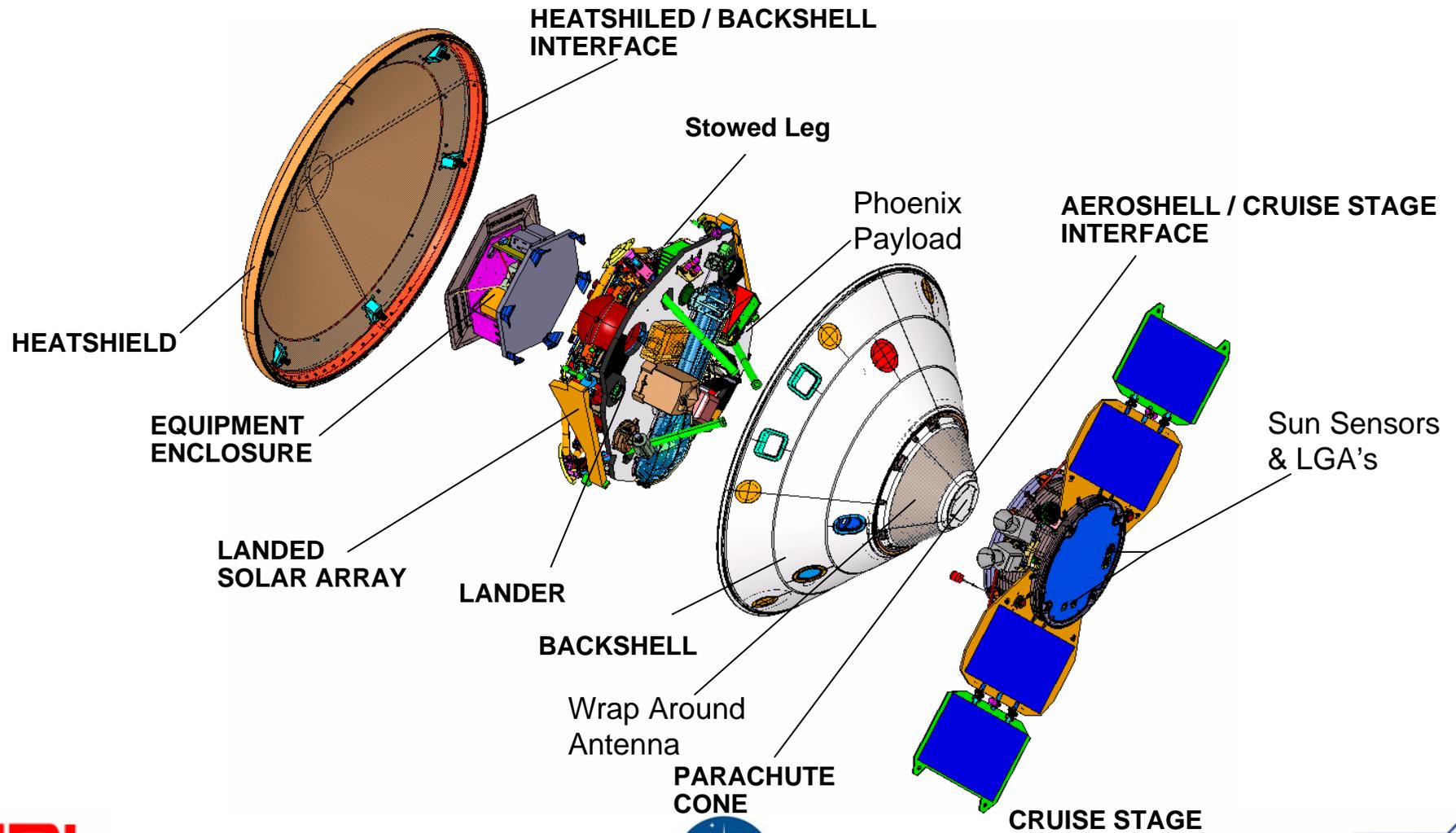
CJL/GTT-2



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Spacecraft - Exploded View



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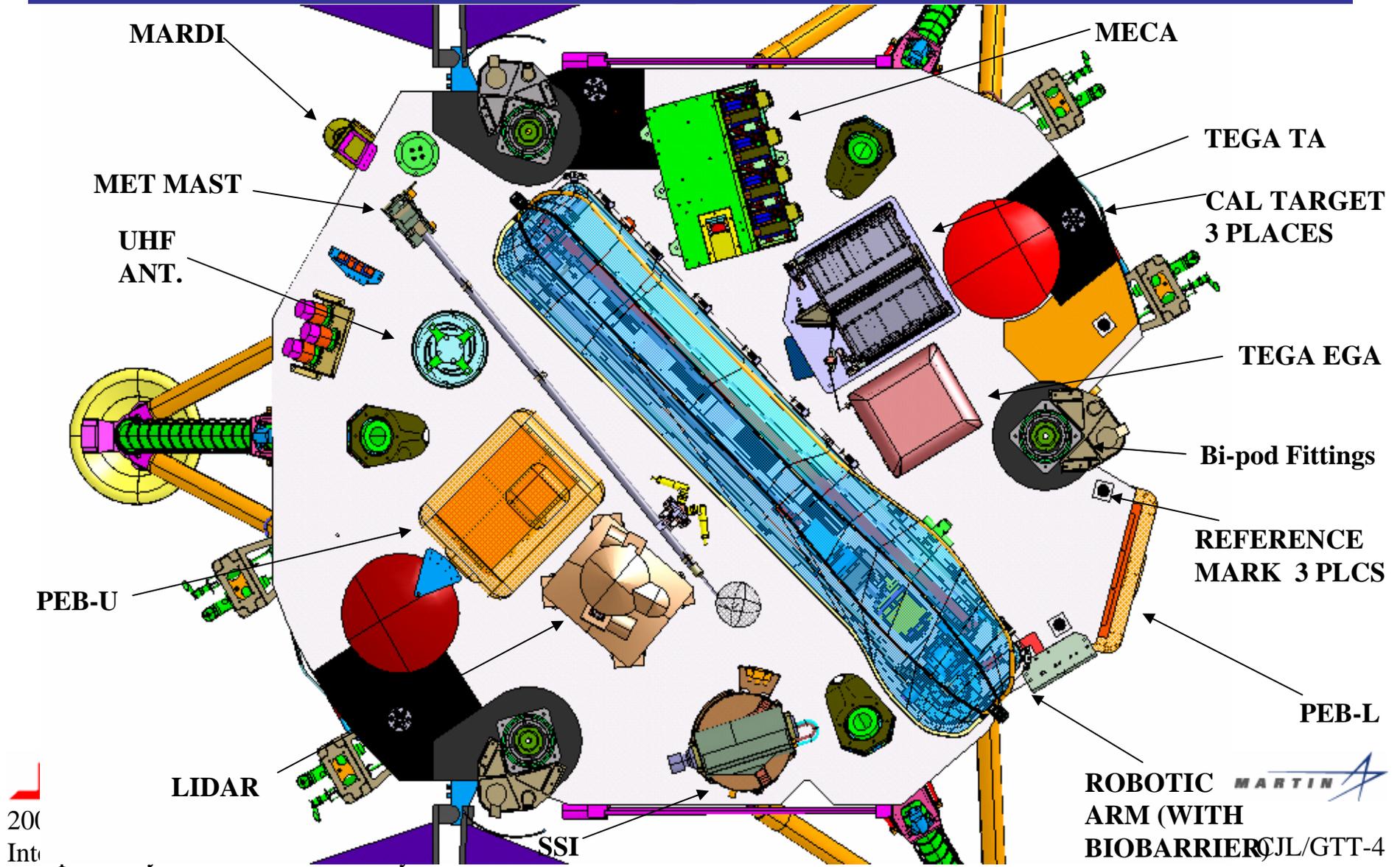
CJL/GTT-3



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Science Deck Layout



200
Int

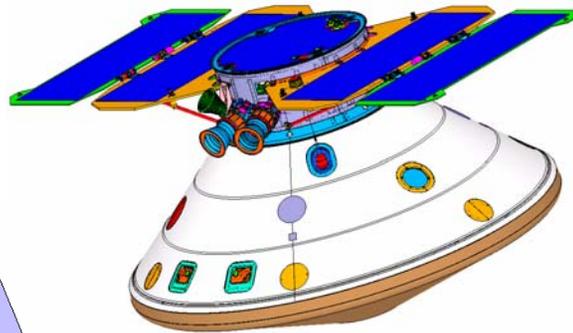
ROBOTIC ARM (WITH BIOBARRIER) CJL/GTT-4



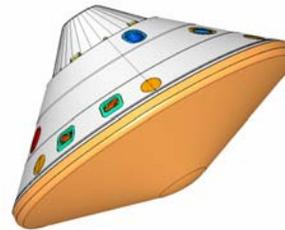
Phoenix



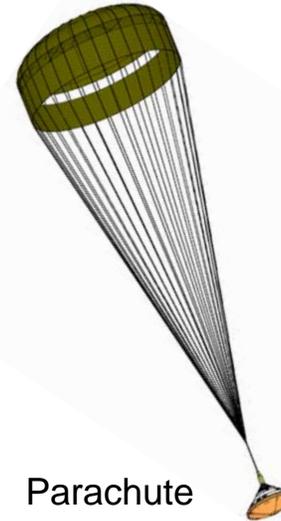
Mission Phase Overview



Cruise



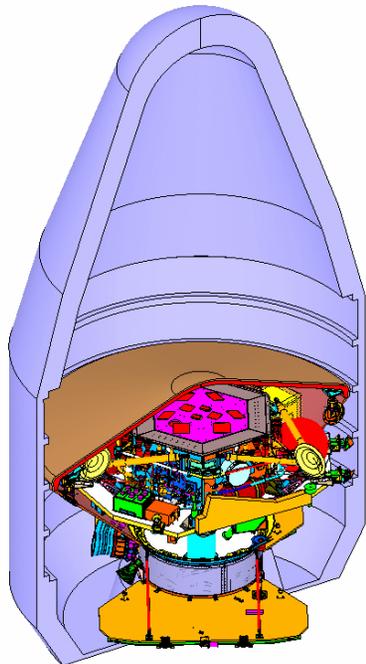
Hypersonic Entry With Guidance



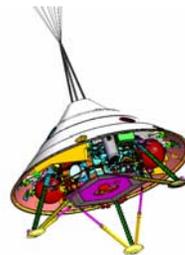
Parachute



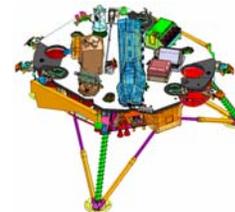
Heatshield Separation



Launch Configuration



Terminal Descent and Landing



Lander with Deployments



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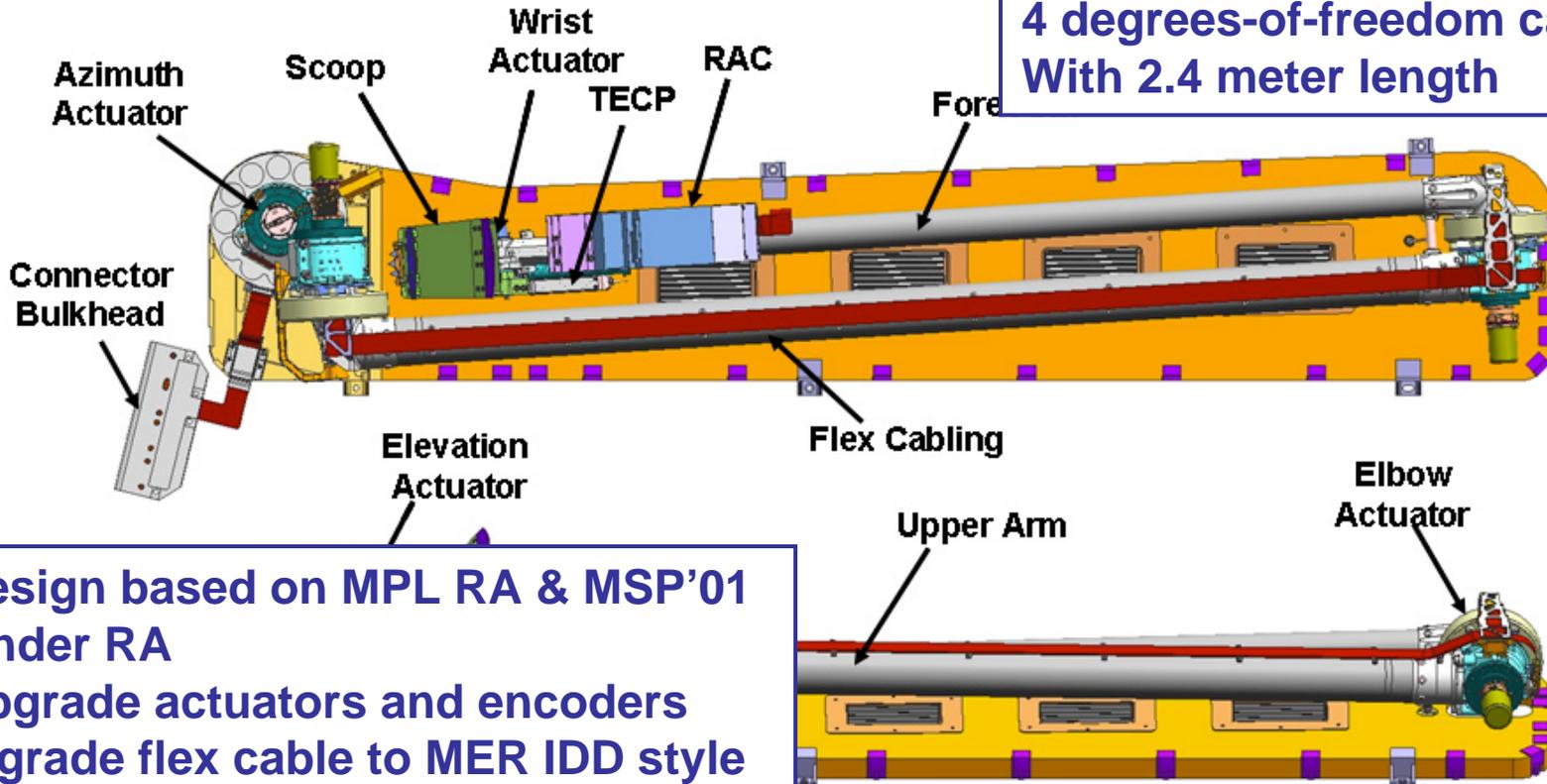


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RA Mechanical Configuration

4 degrees-of-freedom capability
With 2.4 meter length



- Design based on MPL RA & MSP'01 Lander RA
- Upgrade actuators and encoders
- Upgrade flex cable to MER IDD style
- Change ripper tines to tungsten carbide

Hardware omitted for clarity

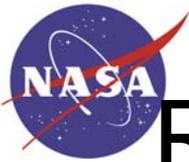


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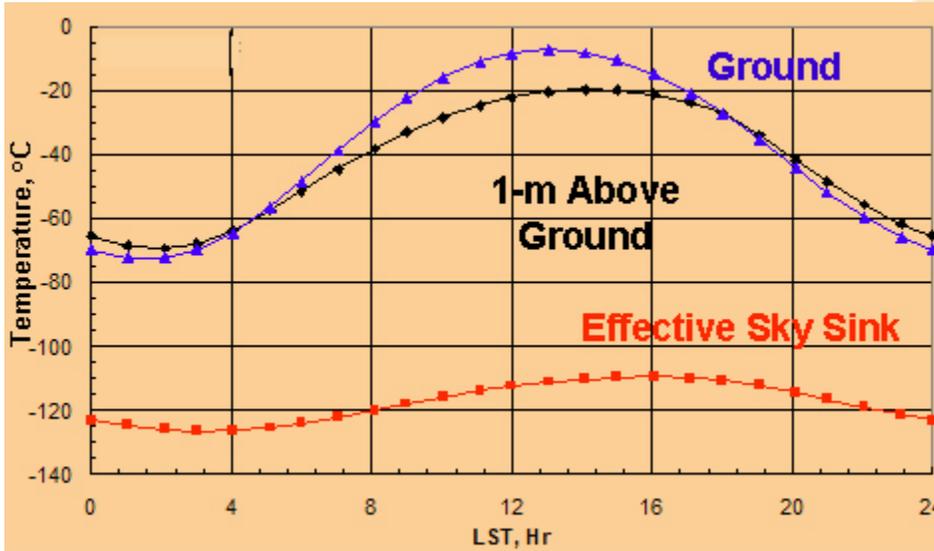
CJL/GTT-6



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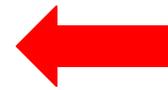


RA Landed Thermal Environment



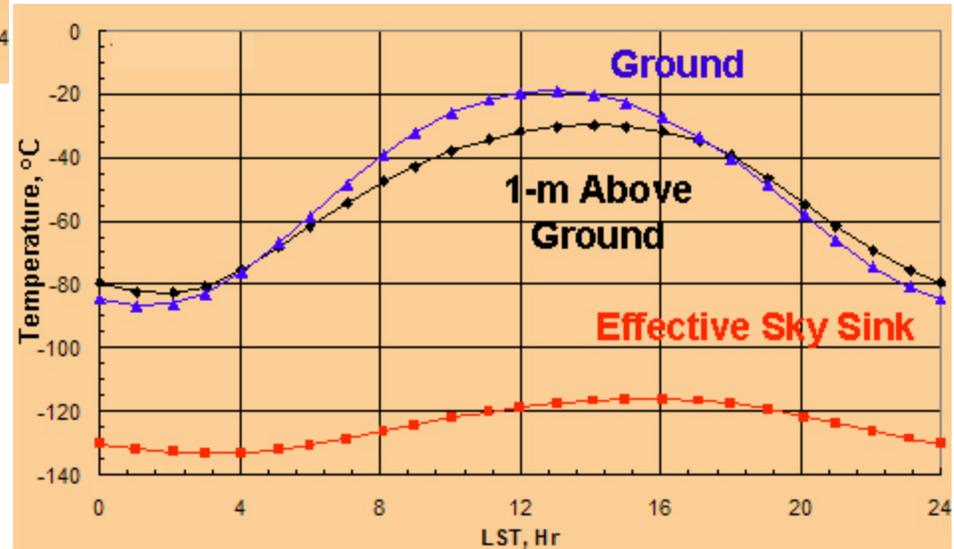
Hot Case:

- $L_s = 110.7^\circ$
- Opacity = 0.2
- Latitude = $70^\circ N$



Cold Case:

- $L_s = 73.3^\circ$
- Opacity = 0.2
- Latitude = $70^\circ N$



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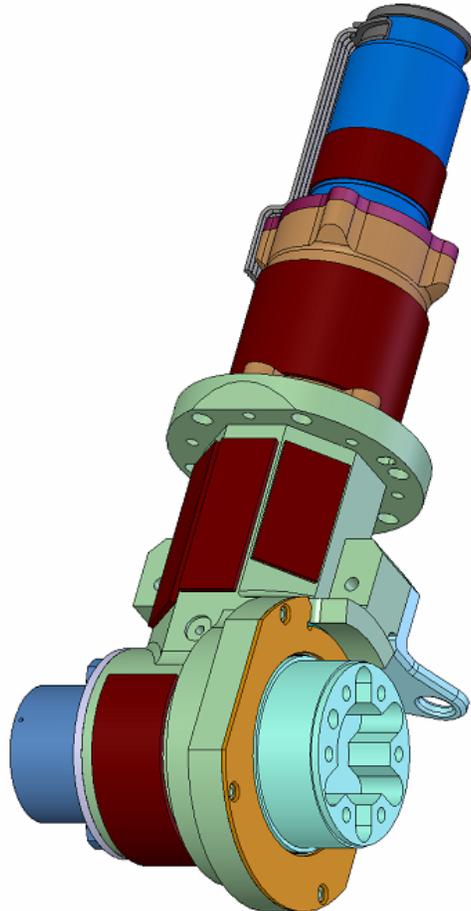
CJL/GTT-7



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RA Thermal Design Approach



- Film heaters located on motor, gearbox, and output to warm-up actuator for early morning operation
- No insulation to prevent entanglement
- “As-manufactured” surface finishes

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LOCKHEED MARTIN 

CJL/GTT-8



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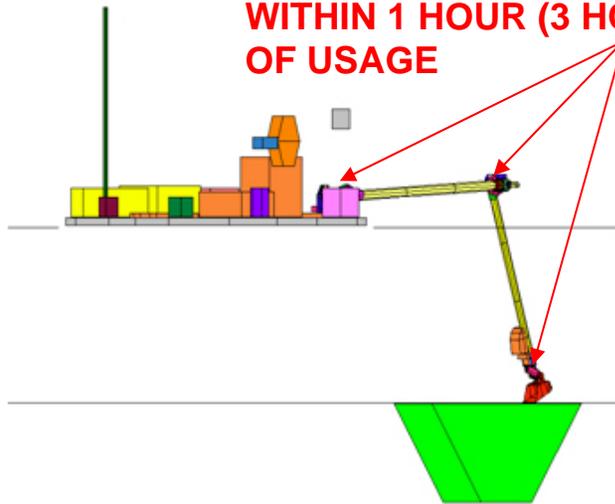
RA Thermal Design Challenge (Driving Requirements) 1/2



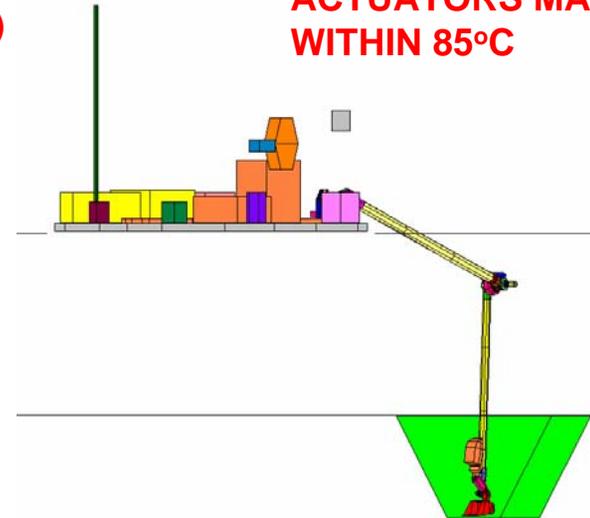
Daytime: Enable early morning operation

Daytime: Enable 3.5 hours of operation

**WARM UP ACTUATORS TO -55°C
WITHIN 1 HOUR (3 HOURS FOR WRIST)
OF USAGE**



**ACTUATORS MAINTAINED
WITHIN 85°C**



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LOCKHEED MARTIN

CJL/GTT-9



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RA Thermal Design Challenge (Driving Requirements) 2/2

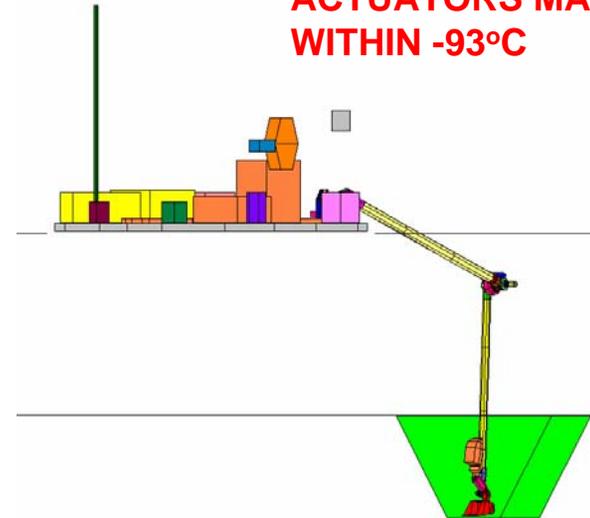
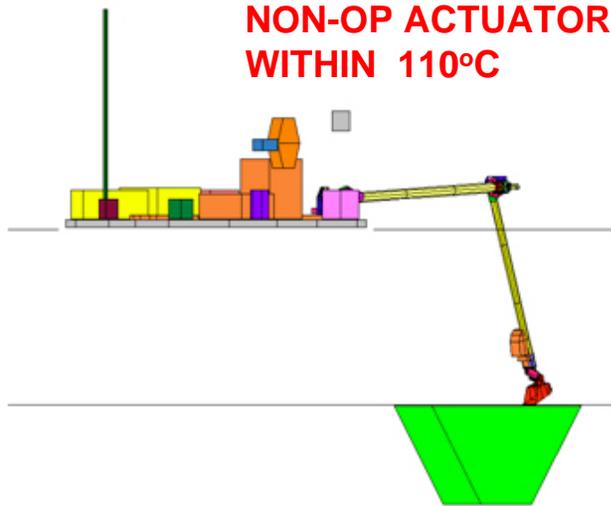


Daytime – Fault Condition: Tolerate a stuck-on actuator heater

Nighttime: Tolerate overnight environment

NON-OP ACTUATORS MAINTAINED WITHIN 110°C

ACTUATORS MAINTAINED WITHIN -93°C



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Warm-up Heater Sizing

Heater Location	Azimuth	Elevation	Elbow	Wrist
Motor	0.5	0.3		
Gearbox	0.0	0.3		
House-1	0.8	1.3		9.1
House-2	0.8		1.0	
House-3				6.3
Total	2.0	1.8	1.0	15.3

Total Warm-up Heater Power is 20.1 W

Heater power sized to warm actuators to -55°C in 1 hour (3 hours for the wrist) prior to early morning actuator use



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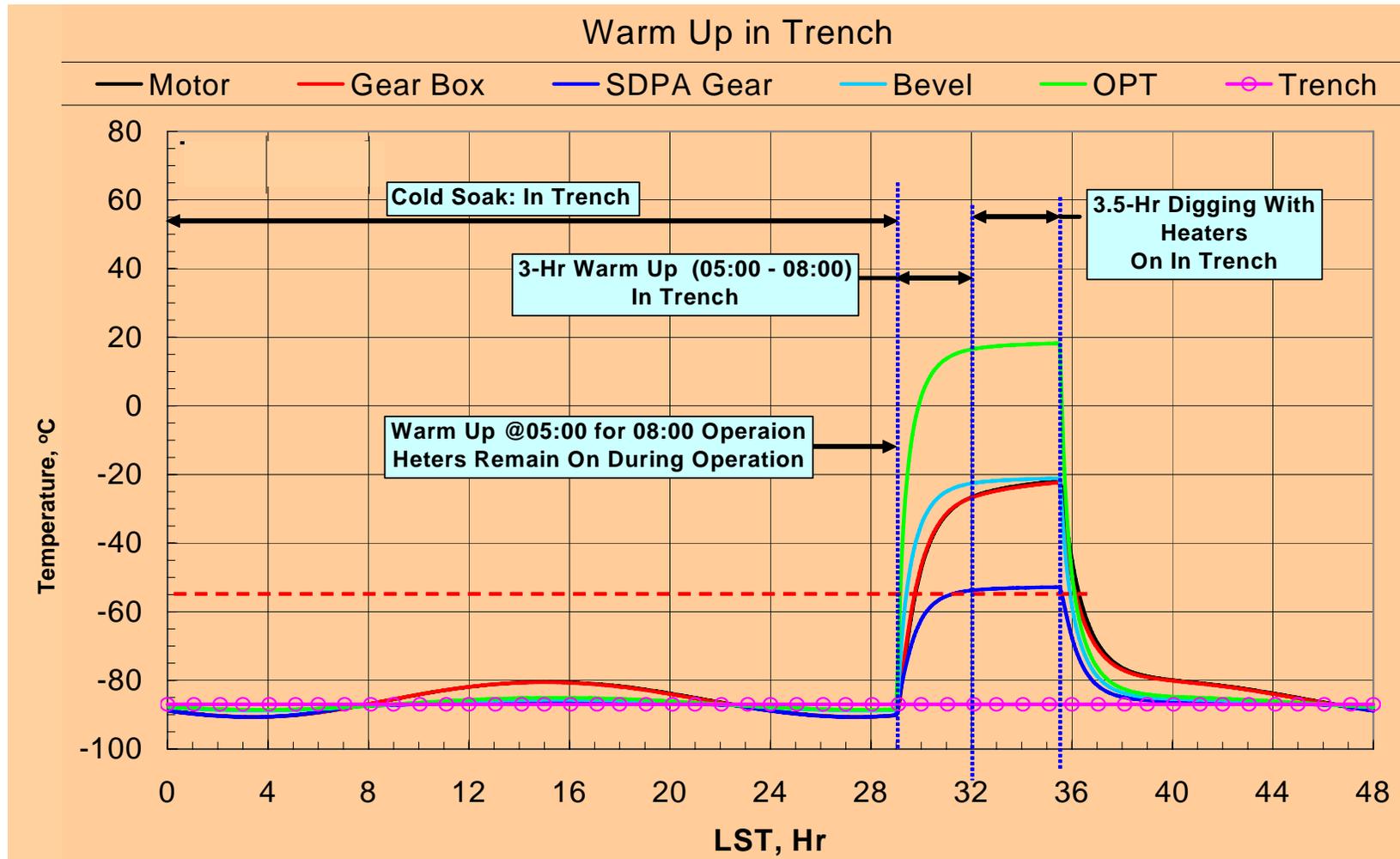


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Warm-up Heater Sizing Wrist Actuator



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Maximum Actuator Power Dissipation



Actuator	Max Allowable Motor Power Dissipation, W
Azimuth	4.4
Elevation	5.2
Elbow	3.6
Wrist	5.3

Actuator power sized to maintain temperature below 85°C during 3.5 hours of operation



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CJL/GTT-13



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Failed-on Warm-up Heaters

Wrist Actuator Component	Maximum Temperature, °C
Motor	69
Gearbox	70
SDPA	109
Bevel	119
OPT	113

Maximum allowable non-operating limit is 110°C in an event of a heater failure

- Overheating of actuator components is not readily detected by sensing motor case temperature
- Power cycling will be necessary during hot conditions



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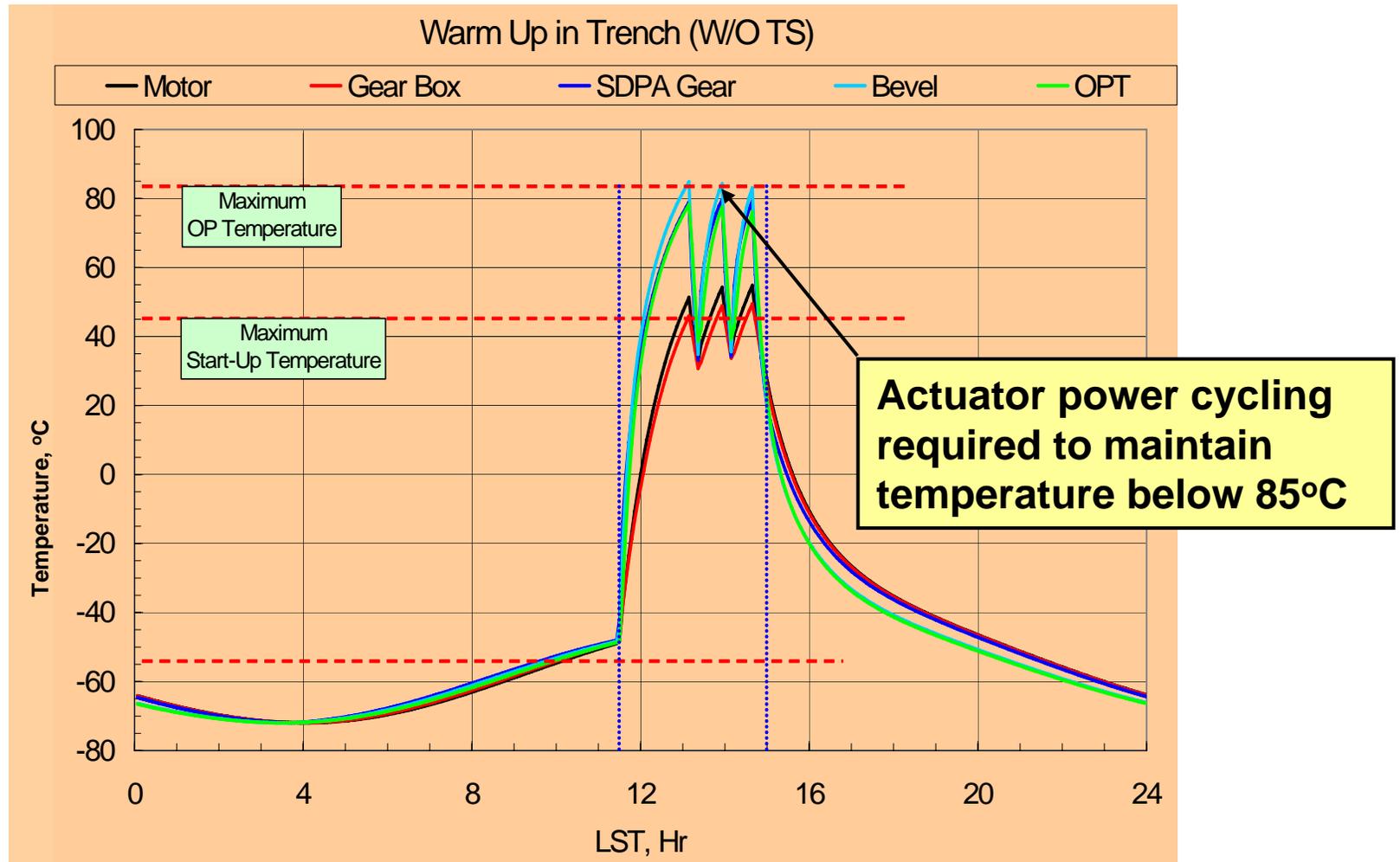
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FAILED-ON WARM-UP HEATERS

Wrist Actuator: Hot Case in Trench

WITHOUT Cut-off Thermostat





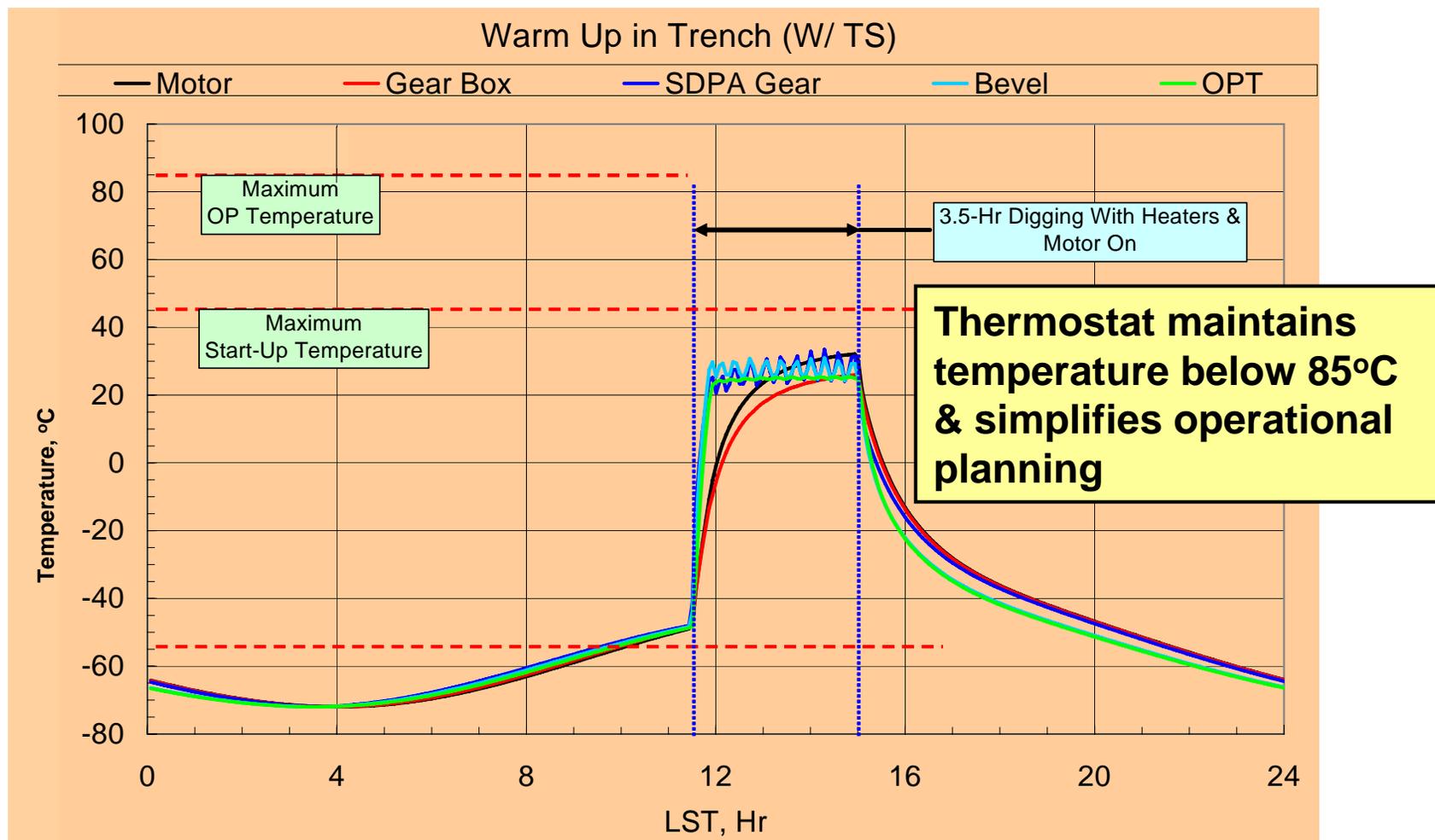
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FAILED-ON WARM-UP HEATERS

Wrist Actuator: Hot Case in Trench

WITH Cut-off Thermostat





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Conclusions

- An adequate warm-up heater approach was developed using analysis & previous Mars surface experience
- In the worst-cold environment, all the actuators were able to warm to their minimum operating temperature within the required time
- In the worst-hot environment, the maximum continuous actuator power dissipation during normal operations was determined
- A mechanical thermostat provides best means to cope with a failed-on wrist heater & to ease RA operations planning

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LOCKHEED MARTIN 

CJL/GTT-17