



*K. Ayres 06 Structural ppt
12011*

JPL

Terrestrial Planet Finder Mission

TPF

A NASA
Origins
Mission

Configuration & Structural Design

Structural Details

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(with inputs from Tim Ho)



Presentation Overview



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- Materials Used
- Primary Mirror Design
- Secondary Tower Design



Materials Used



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ULE Glass (Ultra-Low Expansion Titanium Silicate Glass by Corning)

- Primary & Secondary Mirrors (good thermal stability)
- RT Properties: $E = 68 \text{ Gpa}$, $\text{Density} = 2200 \text{ kg/m}^3$, $\text{CTE} = 30\text{e-}9 / \text{C}$
CTE variability will be addressed in analysis section

K1100/954 Carbon Fiber Composite

- Primary & Secondary Mirror Thermal Enclosures (high conductivity)
- RT Properties: $E = 275 \text{ Gpa}$, $\text{Density} = 1886 \text{ kg/m}^3$, $\text{CTE} = -3.0\text{e-}7/\text{C}$

S-Glass Fiberglass Composite

- AMS/secondary tower bracket & SMA isolators, launch struts (low conductivity)
- RT Properties: $E = 47 \text{ Gpa}$, $\text{Density} = 2000 \text{ kg/m}^3$, $\text{CTE} = 8.0\text{e-}6/\text{C}$

M55J/954 GrEp

- AMS, secondary tower & bracket (good thermal stability & stiffness)
- RT Properties: $E = 110 \text{ Gpa}$, $\text{Density} = 1633 \text{ kg/m}^3$, $\text{CTE} = -1.8\text{e-}7/\text{C}$



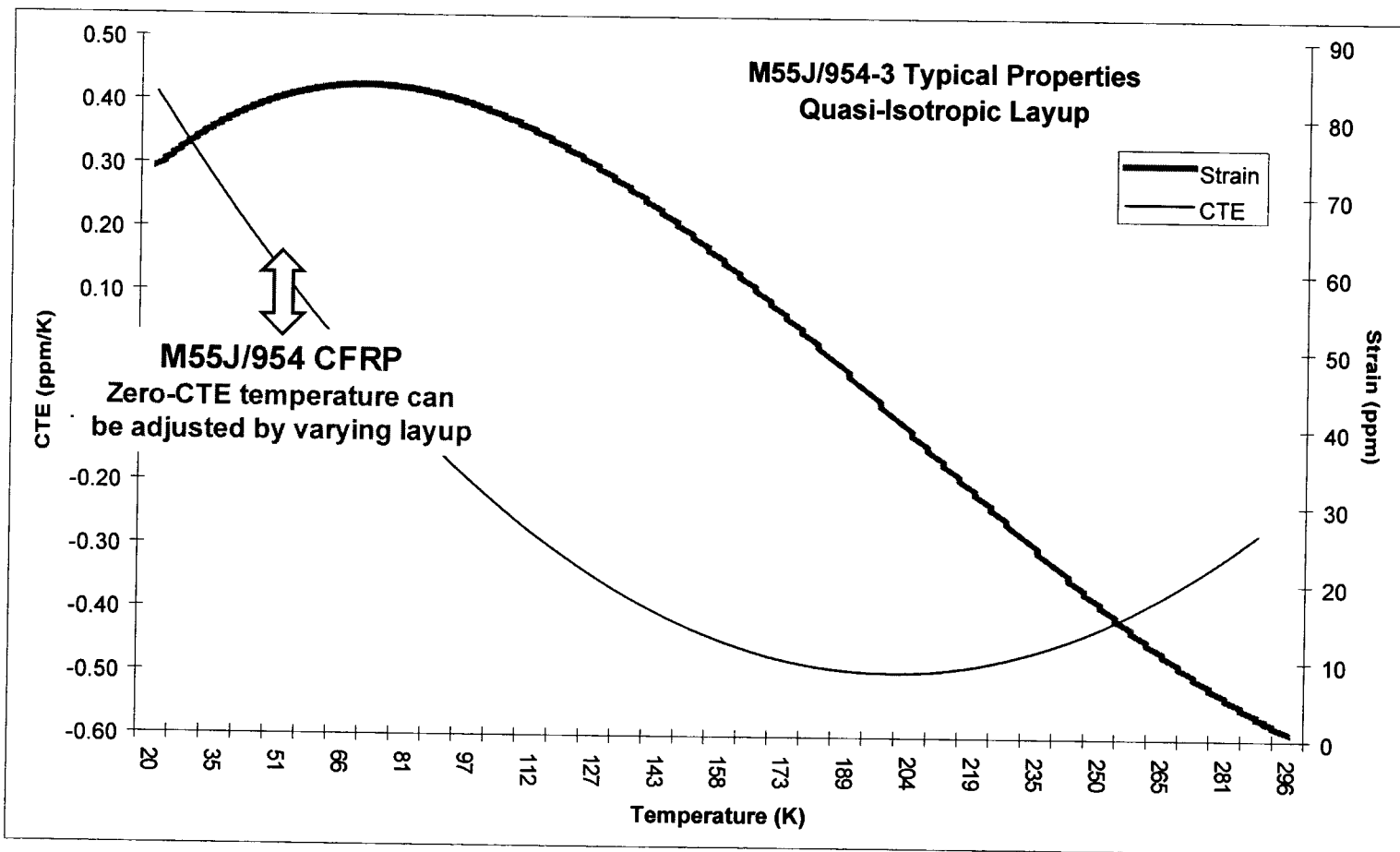
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Variation of CTE Properties for M55J/954-3 as a Function of Temperature



Primary Mirror Design



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Design Drivers: Stability & Weight

Mirror Aberration during Observation:

- Stability in **Thermal** Environment

20 deg Dither maneuver needs ~2hrs of observation w/o DM adjust

Minimize thermal gradients (& delta gradients)

Silicon Carbide (to be investigated)

Minimize sensitivity to gradients (& delta gradients)

ULE Glass selection (Baseline design)

- Stability in **Dynamic** Environment

Reaction Wheel Jitter

Micro-dynamic noise (sunshield, joints, etc.)

High frequency desired (want > 35 Hz for launch)



Primary Mirror Design



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Description of Current Primary Mirror Point-Design:

Size: 6m by 3.5 m ellipse, with thickness= 0.25m (9.8in)

Material: ULE Glass

Light-weighted (~7% mass of solid):

Honeycomb Sandwich: 6 cm core hexagonal cell size (flat-flat)
(100 cells across 6m dia)

1.4 mm cell wall Thickness

6 mm front & 3 mm back face-sheet thickness

Mass Properties: 754 kg, 45 kg/m² (16.6 m²)

(Hubble 2.4m PM was 180 kg/m²)

Mounted Frequencies:

32 Hz 1st mode, 35 Hz 2nd mode (fixed at base of 3 bipods)

43 Hz 1st mode, 46 Hz 2nd mode (pinned at mount pads)

Segmented Core:

Total of 21 hex shaped core segments

Each segment would be ~1.2m (flat-flat)

Core comprised of 2 boules stacked & fused

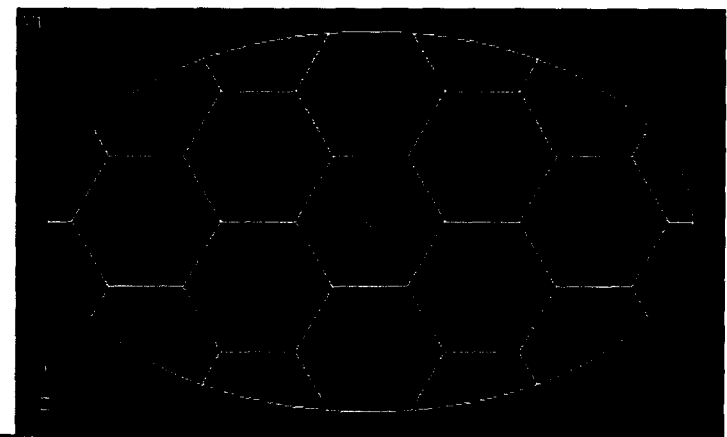
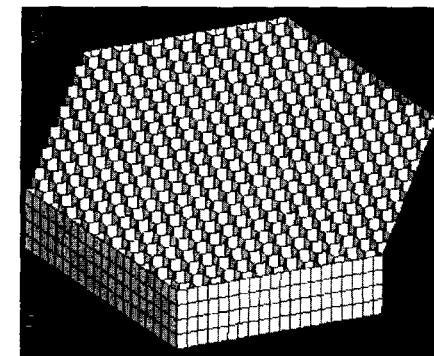
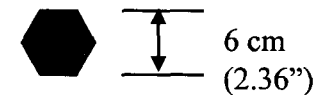
Edge-wall around periphery of each segment

Assembly:

Core segments would be fused to front &
back face-sheets

Core segments are adjacent but not attached
to each other (except through face-sheet)

Edge-wall encompasses entire outer edge of mirror





Secondary Tower Design



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Design Drivers: Stability & Packaging

Minimize thermal distortion – low CTE

Some distortion compensated with
secondary mirror hexapod

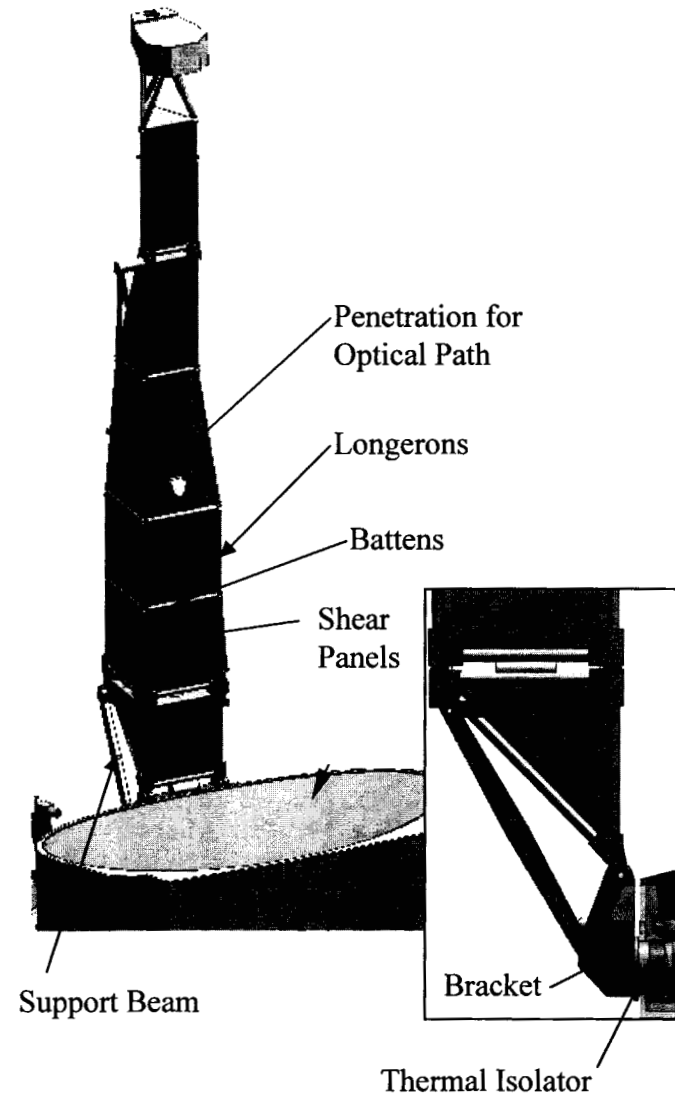
High Frequency desired for vib isolation

Looked at various designs:
truss, circular & triangular x-sections

Rectangular x-section near base was
best for given packaging constraints

Tower freq is ~20 Hz fixed at bracket

Thermal Isolator between bracket &
AMS reduces 1st mode to ~13 Hz



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