

Array Signal Processing in the NASA Deep Space Network

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Abstract— The NASA Deep Space Network (DSN) has recently been augmented with an array capability. This is a third-generation system employing the full spectrum combining technique. The first deployment at Canberra, Australia in 1996 was made specifically to support Galileo mission; the second deployment at Goldstone, California, U.S., followed four years later in 2000 and was of multi-mission capability. The third installation extended the Goldstone capability to the other two DSN sites, namely Canberra, Australia and Madrid, Spain.

The new array capability offers many benefits to deep space missions whose signal reception is often constrained by the vast interplanetary distance. With an increase in signal level obtained from combining the inputs of several smaller antennas, the new capability helps to extend the mission life time for those whose received reception from a single antenna falls below the tracking threshold. For other missions not constrained by this problem, arraying can help to increase telemetry data return relative to that received from a single antenna. Aside from benefiting telemetry data, for the first time in the DSN history, arraying via full spectrum combining technique also enhance navigation performance.

In this paper, we will describe the benefits of arraying and past as well as expected future use of this application. The signal processing aspects of array system are described. Field measurements via actual tracking spacecraft are also presented.

TABLE OF CONTENTS

1. INTRODUCTION
2. PAST AND FUTURE USE OF ARRAYING
3. MERITS OF FULL SPECTRUM COMBINING
4. EQUIPMENT DESCRIPTION
5. SIGNAL PROCESSING
6. RESULTS
7. CONCLUSIONS

1. INTRODUCTION

Arraying is a common technique used to improve reception of weak signals [1]. It is symbolically represented in Figure 1. These signals received simultaneously from different antennas are combined, creating the same effect as an enlarged aperture. This approach can be of benefit in deep space communications where the spacecraft transmitted signals traverse vast interplanetary distances.

This paper first provides an historical context for the development and application of arraying. It then follows with a system description and highlights of different aspects of signal processing. Also presented are the results obtained from field measurements on key performance parameters.

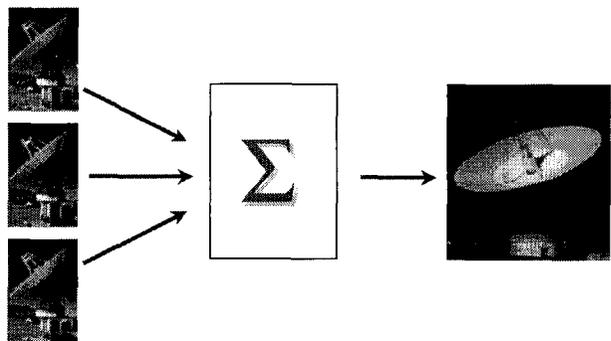


Figure 1 Benefit of arraying

2. PAST AND FUTURE USE OF ARRAYING

In the past, the Voyager mission relied on arraying to increase its data return during Uranus encounter in 1986 and Neptune encounter in 1989 [2, 3]. The Galileo mission was another recent example where arraying was used to significantly increase the science data return. The arraying employed up to five antennas, located at three different tracking facilities spread over the two continents of North America and Australia. It resulted in a factor of 3 improvement in data return. [4, 5, 6]

Future missions can also benefit from arraying. These include the class of missions where certain operational phases may require more performance than a single antenna can provide. For example, the Cassini mission requires only a single 34-meter antenna supported using ground-based antennas, but upon entering Saturn's orbit, in order to return 4G bits/day mapping data (averaging 4.6k bits/s), it requires the use of an array of seven 70-meter and 34-meter antennas [7]. Another class of potential missions is those that need to relay critical science data back to Earth in the shortest amount of time as possible. Stardust belongs to this group of customers. Upon entering the Wild 2 comet in 2004 and having analyzed the properties of cometary samples, the mission plan to reduce single-event risk by sending data back as fast as possible [8]. An array of two 34-meter antennas will enable the Stardust mission to cut the transmission time in half compared to a single antenna.

In the DSN, the 70-meter antenna, being the largest tracking station, is often relied on for tracking spacecraft in the outermost region of deep space. This antenna, however, was built in the late 1960's. As the antenna structure ages and requires more maintenance service, there is a growing concern regarding its availability. An array of several smaller 34-meter antennas serve as a backup for the 70-meter during extended maintenance downtime.¹

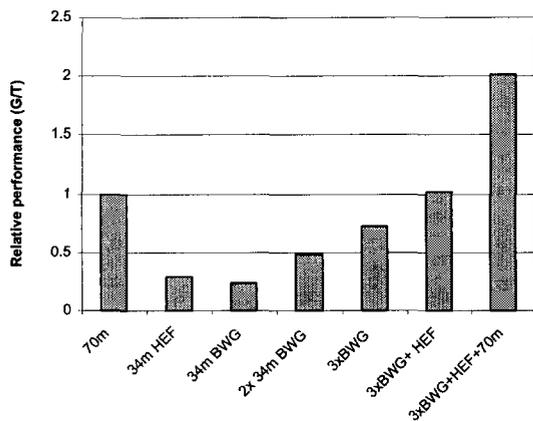


Figure 2. Relative performance of arraying.

Another way of looking at the issue is that arraying increases the flexibility of DSN scheduling and allows for better utilization of available resource. In the absence of an arraying capability, a shortfall in 34-meter link performance

¹It should be noted that the 70-meter substitution, or other array usage discussed in this paper, refers only to downlink processing. The capability, as built, does not support an arrayed uplink.

will require the use of the 70-meter. As a result, there exists a potential for over-subscription of the 70-meter antenna service. With arraying, however, the DSN has the option to schedule additional 34-meter antennas incrementally to meet mission requirements. Figure 2 provides a comparison on the relative performance, in terms of gain-to-noise-temperature ratio (G/T), of different array configurations². Note that an array of four 34-meter antennas will give the same performance as the 70-meter.

3. MERITS OF FULLY SPECIFIED ARRAYING

The main objective of arraying is to coherently combine signals from different antennas; however, because the antennas are geographically separated, the signal received at each site has a different delay and Doppler which are dependent on the antenna's position and motion relative to the spacecraft. The differential delay and Doppler must be removed so that the data streams can be coherently combined. In the arraying techniques described below, these quantities are modeled, and then the residuals between them and the actual values are obtained by cross-correlating the data streams from pairs of antennas.

These signals in deep space communication typically consist of a sinusoidal carrier, a sub-carrier that is phase modulated onto the carrier, and telemetry symbols that are, in turn, phase modulated onto the sub-carrier. The process of cross-correlation between the sub-carriers of different levels: symbol, carrier, or across the whole spectrum [9,10,11,12]. Figure 3 illustrates the three different schemes. The array implementation described in this paper employs the spectrum method. In the following paragraphs, we highlight the relative merits between this technique and other commonly used ones.

² Among the six antennas capable of receiving X-band at Goldstone, the largest being the single 70-meter antenna and the rest 34-meter's. The 34-meter group is further categorized into high efficiency (HEF) and beam waveguide (BWG) due to differences in their design and performance.

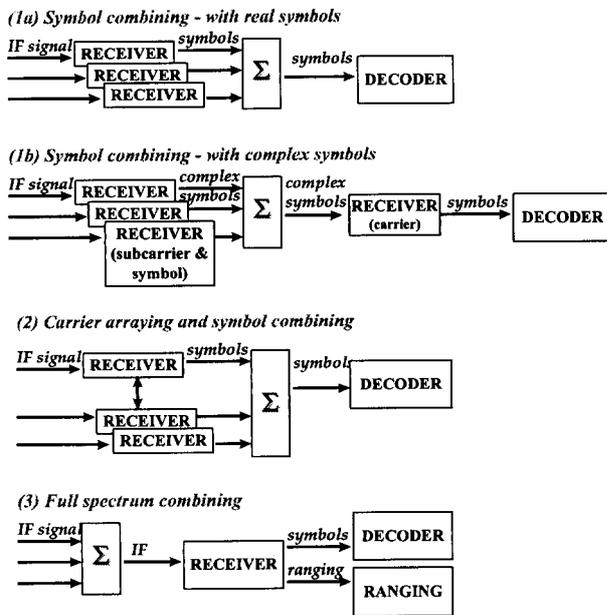


Figure 3: Different arraying methods

Symbol Combining

In symbol combining, demodulated symbols from different antennas are delay compensated, cross-correlated, and finally combined. Combining can be done with real or complex symbols. The resulting signal has a higher energy per bit to noise spectral density ratio (E_b/N_0), thus realizing a lower bit error rate performance. Thus, the combined signal allows for proper decoding and information formation whereas individual symbols would not. This technique successfully reduces the E_b/N_0 requirement.

The symbol combining technique offers advantages for doing signal processing at relatively low symbol rates, typically in the range of a few to a few hundred kHz. Requirements for accurate alignment are therefore less constrained. Transferring of symbols over a long distance is also easier, thus, allowing antennas at large distances to participate in the array.

A drawback, however, is that it requires the signal level at individual antennas to be well above the threshold of the receiver. Otherwise, valid symbols cannot be derived.

Carrier arraying and symbol combining

In carrier arraying, information derived from each carrier signal is detected at the antenna and then used to achieve acquisition at the supporting antenna. On each receiver, lock-up symbols are not combined.

The advantage of carrier arraying is that it pushes down the signal threshold required for normal operation. Only the reference antenna in the array must acquire the signal on its own.

Full spectrum combining

In full spectrum combining, the entire signal spectrum of interest containing the carrier, sub-carrier and symbols is combined at once. During reception, the incoming signals having this spectrum are also combined. The result is an improvement in radiometric observability as well. Carrier demodulation together with sub-carrier and symbol synchronization takes place only after signals are combined. The main advantage of full spectrum combining is a lowering of the acquisition threshold required in the receiver, decoder and ranging correlator.

The challenge of full spectrum combining is in the correlation process. The error in the estimation of relative delay between pairs of antennas becomes more pronounced in carrier processing now applied to IF frequency, which is higher than symbol frequency.

Full spectrum arraying was first employed in the Galileo mission. The Galileo sub-optimal equipment, however, is tailored to low data rates (below 1 k sym/s). The new capability described in this paper tends to be supported at a rate of 6 Msym/s. Unfortunately, because of the high transmission bandwidth between antennas, the data rate is limited to those antennas within a tracking complex, i.e., no inter-complex arraying across two continents is supported, as in the case of Galileo arraying.

In summary, symbol combining is achievable as long as individual antennas in the array can acquire and demodulate the signal. Its benefit, thus, applies mostly to the decoding process. Carrier arraying helps to overcome the shortfalls in the receiver carrier-tracking loop at the supporting antenna. Full spectrum arraying further reduces the required threshold, enabling proper demodulation of the combined signal, but such processing cannot be done on the individual signals.

4. EQUIPMENT DESCRIPTION

Signal processing for arraying is done by two main assemblies: the Full Spectrum Receiver (FSR) and the Full Spectrum Combiner (FSC), see Figure 4. The FSR inputs are individual 300 MHz IF signals derived from the RF that have been received by the antenna, amplified by the microwave equipment, and downconverted by an RF-to-IF downconverter. The FSR outputs are digital samples at 16 MHz, 8-bit resolution. Once digitally combined by the FSC, the signal is converted back to analog form. Except for averaging higher SNR, the signal is virtually identical to the signal arrives at the reference antenna. Downstream

processing (such as demodulation, decoding, and range detection) cannot be accomplished without the help of digital output to the digital signal processing and engineering data products.

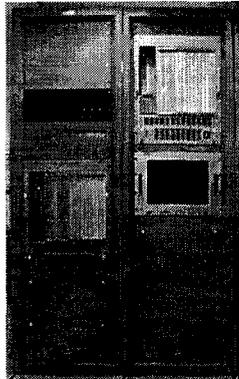


Figure 4 Array signal processing equipment

Major components of the FSR are illustrated in Figure 5. The analog/digital converter and the digital-to-analog converter capture the variations of the 300 MHz IF analog signal in a 16 MHz, 8-bit/s, in-phase and quadrature-phase digital data stream. The delay line and phase rotator boards correct signal delay and phase distortion from the predicted together with feedback from the FSC derived residuals. These signals monitor board samples the digital data streams and transform them to measurements of carrier and telemetry signal-to-noise ratios (SNRs). These values are provided to operators for monitoring. They are also relayed to the FSC for proper setting of the combining coefficients. Measurements of the carrier SNR is obtainable directly from the standard Fast Fourier Transform. Measurements of the telemetry SNR, however, require a measurement involving the correlation of the upper and lower harmonics of the subcarrier. The Real-time digital processor handles high-level monitoring and controls in the FSR.

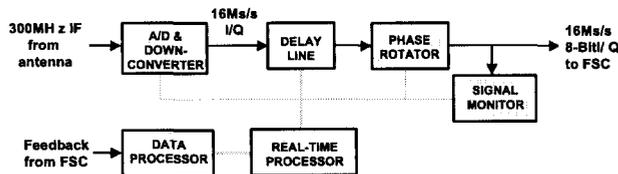


Figure 5 Processing in the full spectrum receiver.

Figure 6 presents major components in the FSC. The cross correlations of upper and lower sideband of different antennas are used to derive differential phase and delay values for feedback to the FSRs. At the same time, the Weight and Sum combine the weighted FSRs input, to produce optimal output. The D/A and Upconverter transforms the digital baseband stream to an analog 300 MHz IF. The Signal Monitor as well as the Realtime and

Data processors carry out functions similar to those in the FSR.

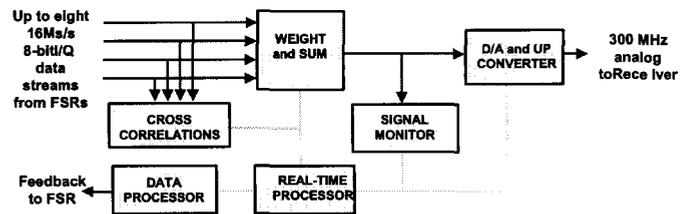


Figure 6 Processing in the full spectrum combiner

5. SIGNAL PROCESSING

This section will highlight the aspects of the signal processing system. The focus is on correlation, delay compensation and combining.

Correlation

The success of combining depends on good correlation results. Correlation is an essential process without which proper combining cannot be done.

Figure 7 shows the detailed processing of correlation. With the aid of Doppler predicts, the upper (UPR) and lower (LWR) sidebands of the signal received at each antenna are captured. The upper sideband from one antenna is correlated with the same component of the array reference, from which the phase difference at the upper sideband is measured. The same process is simultaneously performed on the lower sideband component. An average of the two phase measurements then yields the phase offset, while the ratio of the difference to twice the sideband frequency provides the time delay.

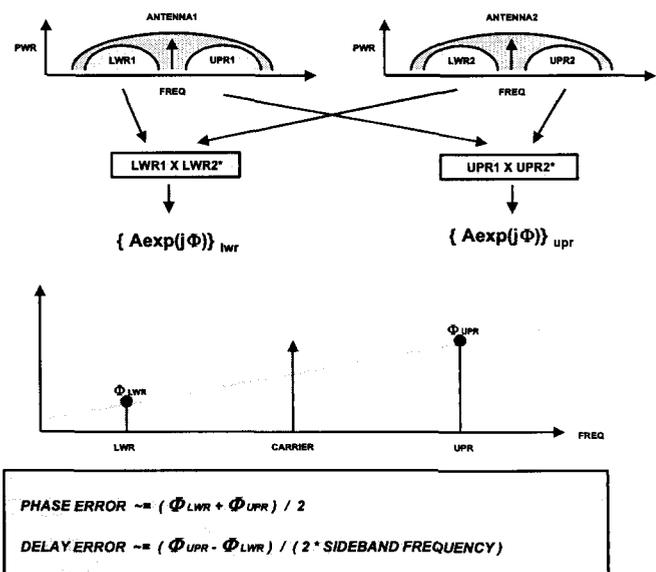


Figure 7 Correlation processing

There are different ways so far implementing the correlation process. The full spectrum arraying equipment supports two approaches, both successfully tested. The simpler scheme ("simple") involves choosing the antenna signal of largest SNR to be the reference, against which all other antenna signals are correlated. This scheme works well when the elements of the array have significantly higher SNR than the others, as in the case of an arraying of 70-meter and 34-meter antennas. The second method, referred to as "simple", treats the array reference as a sum of all antennas except the one under consideration. In other words, one antenna will be cross-correlated against the sum of all others. Simulation results indicate that the rotating sum method performs better than the fixed reference, and that the final solution emerges within a few iterations, see Figure 8 [13]. Note that the simulation includes a third approach using eigenvalue method [14]; however, this algorithm is not implemented in the FSC.

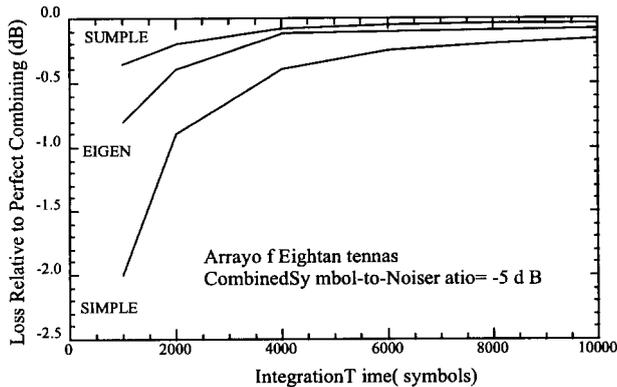


Figure 8: Different methods of correlation

Consideration needs to be given to the setting of integration time for the correlation process. From thermal noise considerations, a long integration period is preferred since it would yield an estimate of small error. Obviously, the lower the signal level, the more integration time required. The problem is that the signals received at different antennas travel through the Earth's troposphere and therefore are subjected to different delay. These tropospheric delays vary on a relatively short timescale, resulting in less correlation for long integrations. An illustration is provided in Figure 9 for a fixed combined symbol SNR at -5 dB/Hz, with equal aperture antennas separated by a baseline of 1 - 10 km [15, 16]. At X-band, the tropospheric limit for a 20-degree correlation error is about 20 seconds. The operating region is the triangular area in the lower right corner of Figure 9. It is upper-bounded by the tropospheric noise and lower bounded by thermal noise. Note also that the graph is expressed in terms of symbol rate, rather than in received signal to noise ratio. Given a fixed symbol SNR, these two quantities are equivalent.

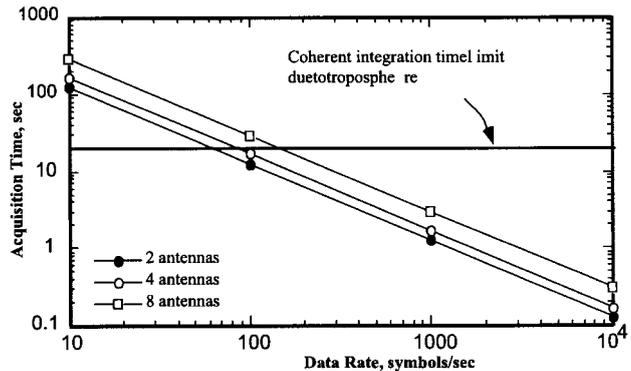


Figure 9: Limit of correlation integration time

Care must also be given toward the use of correlation measurements. Invalid cross correlation arises when one or both signals encounter problems. For example, one antenna could be mis-pointed, or the spacecraft could go behind a planet. As in any control feedback system, care must be given to the design so that bad estimation of the error signal does not drive the system away from a stable condition. This prevention is achieved with a filter on the FSC correlation estimates.

Delay compensation

The delay compensation process is done in two steps. First, each FSR is provided with two sets of delay predictions, one for the reference antenna, the other for the antenna under processing. Using a prediction model, the FSR removes a majority of differential delays of the signals and aligns with the reference. These predictions are computed based on the spacecraft trajectory and the location of tracking antennas. The second step involves minor corrections based on residual error being measured from the correlation of the FSC and the data of the FSR.

Over the track, the relative position of different antennas in the array changes with respect to the spacecraft. The delay of the non-reference signal varies relative to that of the reference. The relative delay is corrected by adjusting the physical delay line in the non-reference FSR. Since such adjustment is only possible with positive values, a delay bias is introduced to all antennas. This bias is typically set at a value at least equal to the maximum delay among array antennas. It is later corrected in the following telemetry and radiometric processing, with proper adjustment of the Earth-received telemetry, Doppler, and range data.

Another consideration is the determination of delay. In order to arrive at the correct determination of relative delay between two antennas, both sidebands are required. Information is needed. The reason is due to the 2π ambiguity in the phase difference from upper and lower sidebands. The sideband measurement alone can only point

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Combining

Combining is done relatively straightforward. The 16-MHz samples from different FSRs are weighted according to their relative SNRs. The system allows disabling of certain inputs where signal is not detected, so that the non-contributing elements would not affect the gain performance.

6. RESULTS

Results of field demonstrations at Goldstone with missions currently in flight are discussed below. Emphasis will be placed on the array performance in radiometric data.

Telemetry array gain

Figure 10 shows the measurement of individual antenna SNR (Pd/No) at each of the two Goldstone 34-m antennas at the combined signal using one of the Mars98 Climate Orbiter racks. The profiles vary as a function of time because of the changing elevation. An average array gain of 2.9 +/- 0.2 dB was observed, compared to a 3.0 dB theoretical improvement. The 0.1 dB difference is attributed to error in the correlation between the noise as well as signal processing loss in the hardware. Laboratory measurement with calibrated test signals showed that SNR degradation, caused by hardware, is within 0.2 dB.

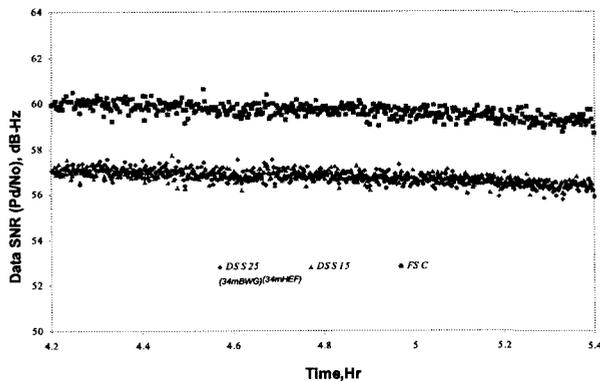


Figure 10 Two-antenna array with Mars Climate Orbiter.

Figure 11 presents results from an array of maximum configuration. It employs all operational antennas available for X-band deep space support at Goldstone. The track was conducted with the Saturn-bound Cassini spacecraft. Relative to the performance of the 70-m antenna, the array

yielded a gain of 1.8 +/- 0.6 dB. Theoretical improvement would have been 2.0 dB.

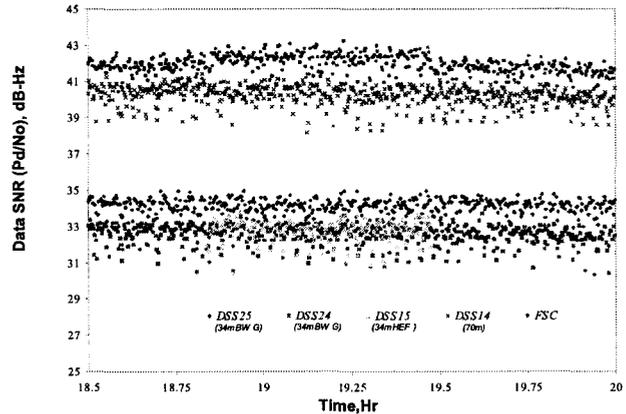


Figure 11 Four-antenna array with Cassini.

Radiometric array gain

Ranging measurement was also obtained on a different rack with Cassini with an array of two 34-meter antennas. The realized gain for ranging is less than that of telemetry. A 1.6 +/- 0.3 dB gain was measured, relative to 2.4 dB predicted which was confirmed by a measured 2.3 dB on telemetry. The less favorable performance is caused by the fact that ranging signal lies much further away from the carrier, compared to the telemetry subcarrier. In the presence of noise and ever-changing Doppler frequency, the error in the phase and delay estimation of the 22 kHz sideband gets magnified when extrapolated to the 1 MHz ranging signal.

7. CONCLUSIONS

In summary, this paper discusses the recently deployed array capability within the NASA Deep Space Network. It provides a brief history of arraying and how this particular method of full spectrum arraying is beneficial to missions, compared to other array techniques. A general description of equipment and special considerations on signal processing are presented. Performance improvement observed with spacecraft in flight is consistent with expectation.

This new product enables the NASA Deep Space Network to provide better support to missions. It is now possible to provide extra performance in radiometric data as well as telemetry. Also, with advances in digital signal processing, the delay within the new system is no longer subjected to the phase drift of analog electronic components. This in turn eliminates the need for pre-track calibration that was often required with past equipment, resulting in an easy-to-operate and more schedule-efficient system.

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REFERENCES

- [1] Yuen, J.H.: *Deep Space Telecommunications Systems Engineering*, Plenum Press, 1983.
- [2] Brown, D. W., H. W. Cooper, J. W. Armstrong, and S. S. Kent, "Parkes-CDSCTEL telemetry Array: Equipment Design," *JPL Telecommunications Data Acquisition Progress Report 42-85*, January-March 1986, Jet Propulsion Laboratory, Pasadena, CA, pp. 85-110, May 15, 1986.
- [3] Brown, D. W., W. D. B. Rundage, J. S. Ulvestad, S. S. Kent, and K. P. Bartos, "Interagency Telemetry Arraying for Voyager-Neptune Encounter," *JPL Telecommunications Data Acquisition Progress Report 42-102*, April-June 1990, Jet Propulsion Laboratory, Pasadena, CA, pp. 91-118, August 15, 1990.
- [4] Layland, J. W., F. D. McLaughlin, P. E. Beyer, D. J. Mudgway, D. W. Brown, R. W. Burton, R. J. Wallace, J. M. Ludwinski, B. D. Madsen, J. C. McKinney, N. A. Renzetti, and J. S. Ulvestad, "Galileo Array Study Team Report," *JPL Telecommunications Data Acquisition Progress Report 42-103*, July-September 1990, Jet Propulsion Laboratory, Pasadena, CA, pp. 161-169, November 15, 1990.
- [5] Statman, J. I., "Optimizing the Galileo Space Communication Link," *JPL Telecommunications Data Acquisition Progress Report 42-116*, October-December 1993, Jet Propulsion Laboratory, Pasadena, CA, pp. 114-120, February 15, 1994.
- [6] Pham, T. T., S. Shambayati, D. E. Haridi, and S. G. Finley, "Tracking the Galileo Spacecraft with the DS-CC Galileo Telemetry Prototype," *JPL Telecommunications Data Acquisition Progress Report 42-119*, July-September 1994, Jet Propulsion Laboratory, Pasadena, CA, pp. 21-235, November 15, 1994.
- [7] Deep Space Network, Near Earth and Deep Space Mission Support Requirements, JPL D-0787 (internal document, Jet Propulsion Laboratory, Pasadena, CA), October 1996.
- [8] Deep Space Network, Near Earth and Deep Space Mission Support Requirements, JPL D-0787 (internal document, Jet Propulsion Laboratory, Pasadena, CA), October 1996.
- [9] Mileant, A., and S. Hinedi, "Overview of Arraying Techniques in the Deep Space Network," *JPL Telecommunications Data Acquisition Progress Report 42-104*, October-December 1990, Jet Propulsion Laboratory, Pasadena, CA, pp. 109-139, February 15, 1991.
- [10] Divsalar, D., "Symbol Stream Combining Versus Baseband Combining for Telemetry Arraying," *JPL Telecommunications Data Acquisition Progress Report 42-74*, April-June 1983, pp. 13-28, Jet Propulsion Laboratory, Pasadena, CA, August 15, 1983.
- [11] Brockman, M. H., "Enhanced Radio Frequency Carrier Margin Improvement for an Array of Receiving Systems with Unequal Prediction Signal-to-Noise Ratios," *JPL Telecommunications Data Acquisition Progress Report 42-76*, October-December 1983, Jet Propulsion Laboratory, Pasadena, CA, pp. 170-188, February 15, 1984.
- [12] Rogstad, D. H., "Superpressed Carrier Full-Spectrum Combining," *JPL Telecommunications Data Acquisition Progress Report 42-107*, July-September 1991, Jet Propulsion Laboratory, Pasadena, CA, pp. 12-20, November 15, 1991.
- [13] Fort, D., *Array Preliminary Design Review* (internal document, Jet Propulsion Laboratory, Pasadena, CA), January 1998.
- [14] Cheung, K.-M., "Eigen Theory for Optimal Signal Combining: A Unified Approach," *JPL Telecommunications Data Acquisition Progress Report 42-126*, April-June 1996, Jet Propulsion Laboratory, Pasadena, CA, pp. 1-9, August 15, 1996.
- [15] Kahn, R., *Array Preliminary Design Review* (internal document, Jet Propulsion Laboratory, Pasadena, CA), January 1998.
- [16] Dewey, R. J., "The Effects of Correlated Noise in Intra-Complex DSNA Arrays for S-Band Galileo Telemetry Reception," *JPL Telecommunications Data Acquisition Progress Report 42-111*, July-September 1992, Jet Propulsion Laboratory, Pasadena, CA, pp. 129-152, November 15, 1992.

BIOGRAPHY



Timothy Pham is a member of the technical staff at the Jet Propulsion Laboratory. He led the development of several systems, ranging from demonstration of carrier arraying, radio science, Galileo telemetry support and more recently, 34-meter antenna arraying. He received a BSEE from the California Institute of Technology and a MSEE from the University of Southern California.



Andre Jongeling is a member of the technical staff at the Jet Propulsion Laboratory. He participated in the development of the arraying system for the Galileo mission, and has led the software team in the development of the 34m arraying system. He received a BSEE from the California State Polytechnic University in Los Angeles.