

# HIGH-ACCURACY MULTIANGLE SPECTROPOLARIMETRIC IMAGING CONCEPT FOR AEROSOL REMOTE SENSING FROM SPACE

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## Science background

*A multiangle spectropolarimetric imager would provide many key aerosol measurements*

Aerosol direct  
radiative forcing

Separation of  
anthropogenic  
component

Air quality

Indirect  
radiative  
effects

- **Associating aerosol properties with radiation requires disentangling variables whose effects are intertwined**
  - optical depth, single scattering albedo, and phase function
  - constraints on vertical distribution
  - subpixel clouds and cirrus
- **Separating aerosol forcing into natural and anthropogenic forcing and assessing air quality requires identification of aerosol chemistry**
  - particle size, shape, refractive index
  - location of sources
- **Quantifying indirect forcing requires simultaneous cloud measurements**
  - cloud fraction and albedo
  - isolation of cloud-aerosol interactions from meteorological covariability

## Fusion of measurements

*A multiangle spectropolarimetric imager combines the salient attributes of multiple aerosol sensors into a single instrument*

	Spatial resolution	Along-track angle range	Spectral range	Polarization accuracy	Global coverage
<b>MISR</b>	275 m - 1.1 km	70° fore - 70° aft	446 - 866 nm	NA	9 days
<b>MODIS</b>	250 m - 1 km	NA	469 - 2130 nm	NA	2 days
<b>AATSR</b>	1 - 2 km	2 angles: 0°, 55° fore	550 - 1610 nm	NA	5 days
<b>POLDER</b>	6 - 7 km	50° fore - 50° aft	443 - 910 nm	~2%	2 days
<b>APS</b>	10 - 35 km	60° fore - 60° aft	412 - 2250 nm	0.2%	No
<b>TOMS</b>	40 km	NA	330 - 380 nm	NA	1 day
<b>Concept</b>	250 m - 2 km	70° fore - 70° aft	380 - 2130 nm	0.5%	3 - 4 days

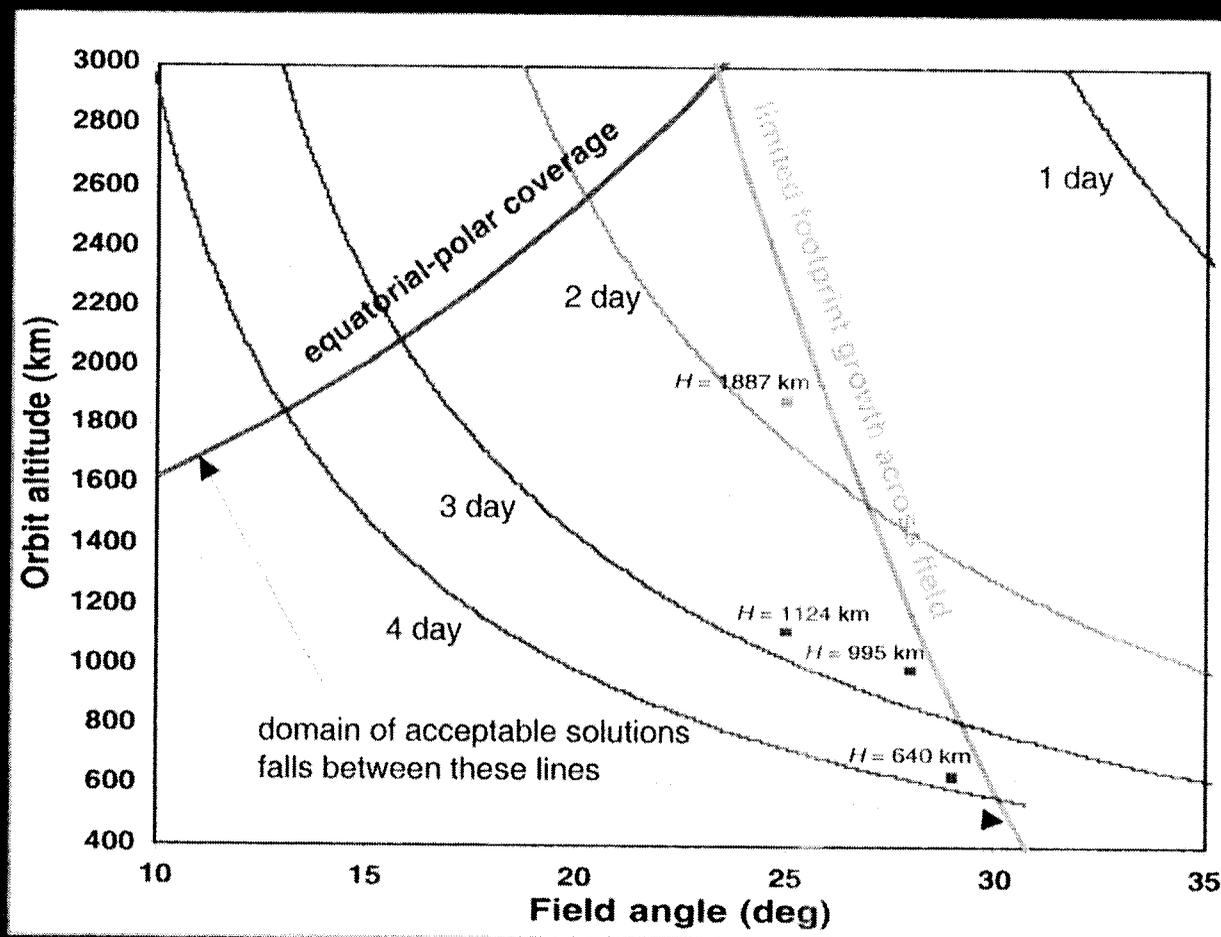
# Benefits of combination into a single instrument

*Provides synergy, accuracy, simultaneity, imaging, and economy*

- **Synergy**
  - multiangle: particle size, shape, and retrievals over bright regions (deserts, cities)
  - multispectral: particle size (visible and SWIR), absorption and height (near-UV)
  - polarimetric: size-resolved refractive index
- **Accuracy**
  - $1 \text{ W m}^{-2}$  is estimated magnitude and uncertainty of aerosol climate forcing
  - corresponds to optical depth of  $\sim 0.04$ , which is current measurement uncertainty
  - combining multiple techniques expected to reduce errors by factor of 2-3 (NACIP)
- **Simultaneity**
  - measurements acquired from one platform
  - aerosol mesoscale variability requires simultaneity within  $\sim 20 \text{ km}$  and 3 hrs
- **Imaging**
  - resolution: can discern cloud and intercloud scales
  - coverage: frequent sampling of validation sites and usefulness for assimilation
- **Economy**
  - resource requirements for one instrument less than sum of individual components

# Orbit analysis

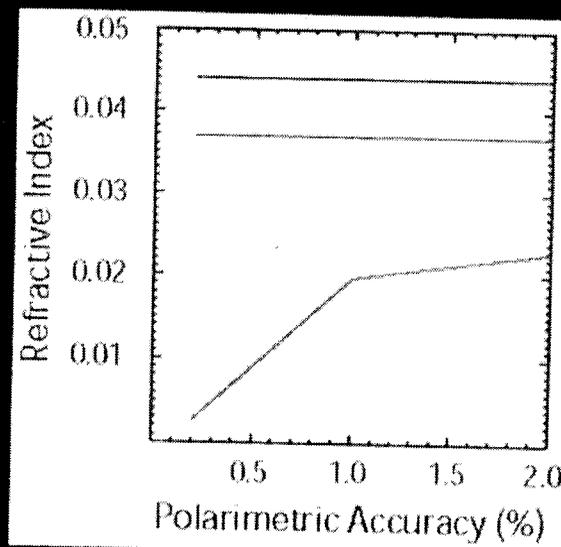
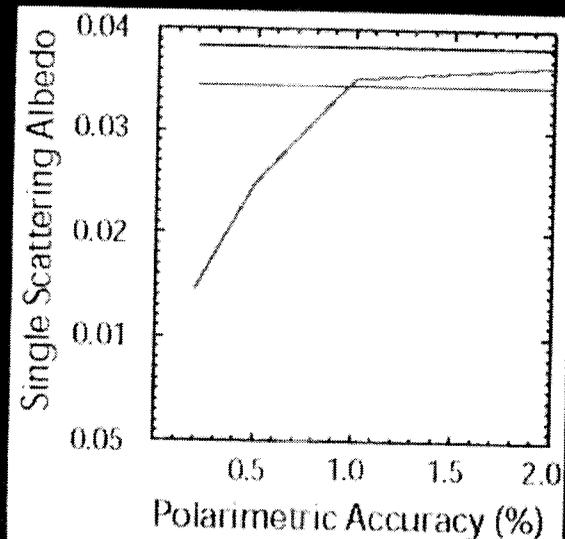
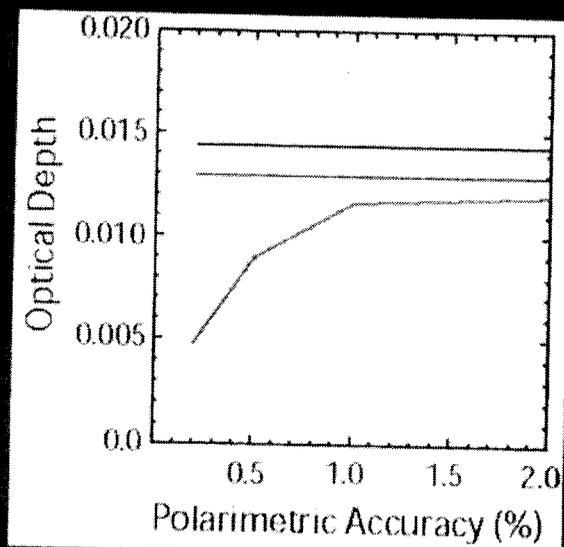
*Establishing orbit requirements and constraints is first step to defining instrument parameters, particularly size of the optics*



# Polarization accuracy requirements

## *Polarimetric accuracy is design driver*

- “Satellite Instrument Calibration for Measuring Global Climate Change” (Ohring et al., 2003) establishes accuracy requirements
  - aerosol optical depth ( $\tau$ ) -- 0.01      aerosol single scatter albedo ( $\omega$ ) -- 0.03
- Limiting factors on accuracy:
  - instrument calibration, disentangling optical depth and particle microphysical properties
- Polarimetry can be done as a relative measurement. With accuracy of 0.5%:
  - potentially improves intensity retrievals of  $\tau$  and  $\omega$  by ~factor of 2
  - provides high sensitivity to refractive index

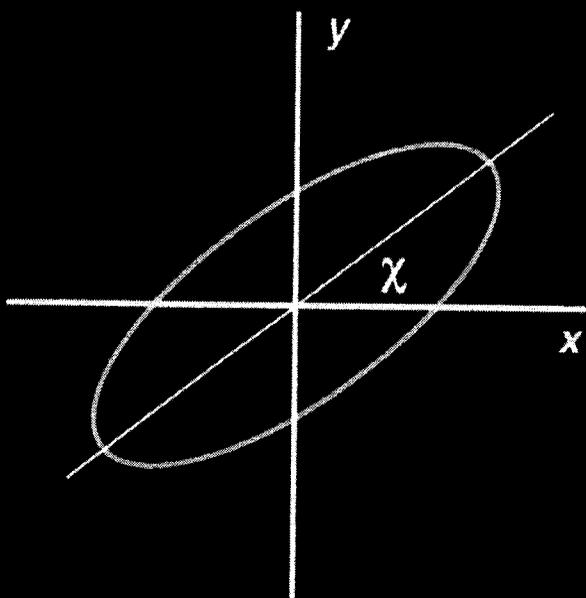


intensity only (9 angles), intensity only (17 angles), intensity + polarization (9 angles)

Sensitivities for model with  $\tau = 0.2$ ,  $\omega_0 = 0.9$ , fine mode  $0.2 \mu\text{m}$   $n_r = 1.45$ , coarse mode  $1 \mu\text{m}$   $n_r = 1.35$

## Required polarimetric measurements

*Measurement objectives are degree of linear polarization (DOLP) and orientation of polarization plane relative to the plane of scattering*



- Scattered light from the Earth is the sum of an unpolarized component and a polarized component, i.e.,  $I = I_{\text{unpol}} + I_{\text{pol}}$
- In general, characterizing the radiation field requires measuring 4 components of the Stokes vector:
  - I is intensity
  - Q is preference for horizontal over vertical polarization
  - U is preference for 45° over 135° polarization
  - V is preference for right-handed over left-handed circular polarization
- Circular polarization is negligible in practice ( $V/I < 0.1\%$ ) for aerosol scattering

$$DOLP = \frac{I_{\text{pol}}}{I} = [(Q/I)^2 + (U/I)^2]^{1/2}$$

$$\tan 2\chi = \frac{(U/I)}{(Q/I)}$$

## Measurement challenges

*Meeting 0.5% accuracy in DOLP is an exceptional challenge for an imaging instrument*

- Instrumental effects must be carefully controlled. High-performance optics are essential
  - all-reflective design to obtain UV-SWIR spectral coverage
  - high optical quality for imaging
  - low instrumentally-induced polarization
  - low depolarization (conversion of linearly polarized light to unpolarized or circular)
  - low diattenuation (difference in transmission for orthogonal polarizations)
- Intrinsic limitations of detector-to-detector gain differences must be overcome
- Image motion and spatial co-registration must be considered to avoid “false” polarization signals

## Basic concept

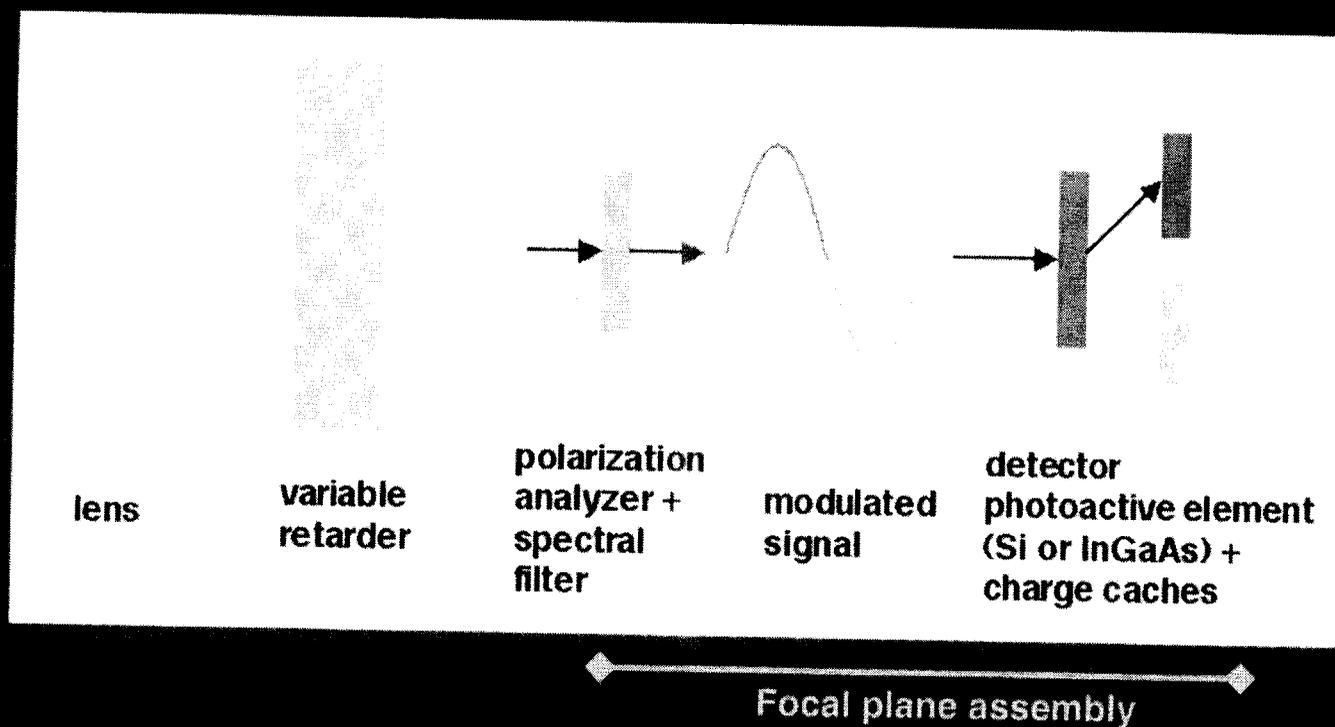
### *Rapidly interlaced time-modulated polarimetry*

- A rapidly varying retarder causes the plane of polarization to rotate at high speed
- In conjunction with a focal plane polarization analyzer, this causes the detected intensity to have a rapid sinusoidal modulation
  - the average signal records the intensity,  $I$
  - a single line array with a polarization analyzer can be used to retrieve  $Q$  and  $U$  from the modulated signal
  - another arrangement is to have two line arrays with analyzers at  $45^\circ$  to each other; one obtains  $Q$  from the modulated signal, the other obtains  $U$
- A charge-caching focal plane demodulates the time-varying signal

# Key elements of the approach

*Design has 4 key components*

- High-performance lens
- Rapid retardance modulator
- Polarization analyzer
- Synchronously demodulating focal plane



## Features of this approach

*Uses concepts demonstrated by the astronomy community*

- Has been used to obtain high precision (0.001%) polarimetric imaging for solar astronomy using rapid charge-shifting CCDs to demodulate the signals
- The rapid modulation causes measurements of I and Q (or I and U) to be interlaced at sub-pixel scale, thus insulating against false polarization (“self-registering”)
- The ratios  $Q/I$  and  $U/I$  are insensitive to detector gain or camera transmittance (“self-calibrating”)

# Candidate polarization retarder approaches

## (1) Mechanically rotating retarder

- consists of a half-wave retarder
- advantages: high polarization quality
- disadvantages: need for long-life, high-speed mechanical actuator (> 3000 RPM); magnetic levitation technology development could resolve this

## (2) Photoelastic modulator (PEM)

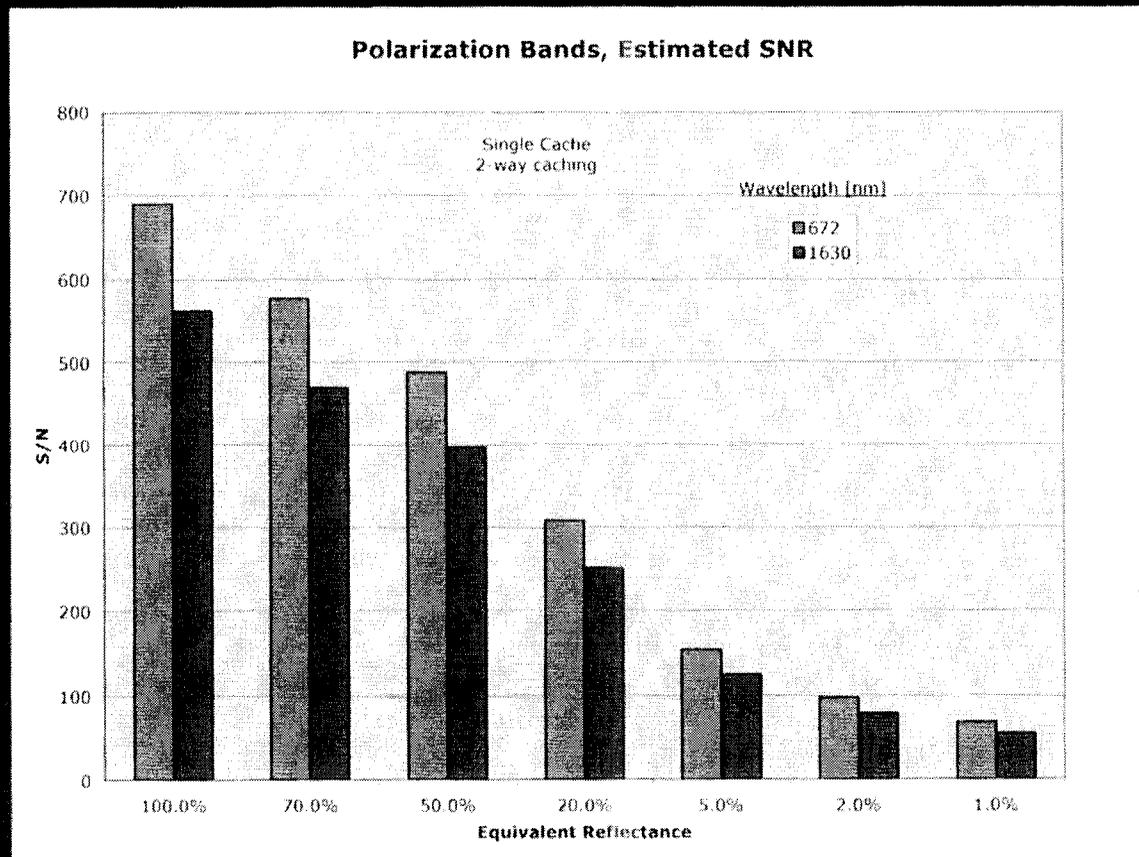
- consists of a glass bar bonded to a piezoelectric transducer (PZT) that induces modulated stress birefringence (25-50 kHz)
- advantages: low power, no moving parts, high polarization quality
- disadvantages: devices susceptible to mechanical shocks

## (3) Liquid crystals

- consists of birefringent fluids with elongated molecules that tip perpendicular to the substrate when an electrical field is applied, reducing the effective birefringence
- advantages: low power, no moving parts, high polarization quality
- disadvantages: slow switching speeds, limited transmittance range

# Radiometric performance

*Required signal-to-noise ratios for polarization channels at very low signal levels can be met by averaging pixels to 1-2 km resolution*



**Signal-to-noise ratio for single pixel (250 m in nadir)  
- can be improved by pixel averaging**

# Instrument system parameters

*The integrated instrument concept has modest resource requirements*

## Estimated power

51-96 W (depending on retardance modulator option)

## Estimated mass

< 50 kg

## Estimated volume

0.5 m x 0.5 m x 0.3 m

## Estimated orbit-averaged data rate (no compression)

10-20 Mbps (with pixel summing and editing)

These numbers are based on an assumed 640-km altitude orbit, which would provide 4-day global coverage